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INSTITUTE OF SCIENCE AND TECHNOLOGY
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**SCHOOL OF MECHANICAL ENGINEERING
DEPARTMENT OF AERONAUTICAL ENGINEERING**

UNIT – I - INTRODUCTION – SAE1605

CHAPTER 1

The Nature of Civil Aviation and Airports

Introduction

Since its beginning in the early twentieth century, civil aviation has become one of the most fascinating, important, and complex industries in the world. The civil aviation system, particularly its airports, has come to be the backbone of world transport and a necessity to twenty-first-century trade and commerce.

In 2008, the commercial service segment of civil aviation, consisting of more than 900 airlines and 22,000 aircraft, carried more than 2 billion passengers and 85 million tons of cargo on more than 74 million flights to more than 1700 airports in more than 180 countries worldwide. Millions more private, corporate, and charter “general aviation” operations were conducted at thousands of commercial and general aviation airports throughout the world. In many parts of the world, commercial service and general aviation serve as the primary, if not the only method of transportation between communities.

The magnitude of the impact of the commercial air transportation industry on the world economy is tremendous, contributing more than \$2.6 trillion in economic activity, equivalent to 8 percent of the world gross domestic product, and supporting 29 million jobs. In the United States alone civil aviation is responsible for \$900 billion in economic activity and 11 million jobs. General aviation serves an equally important role in the world’s economy, providing charter, cargo, corporate, medical, and private transport, as well as such services as aerial photography, firefighting, surveillance, and recreation. In the United States alone, there are more than 225,000 registered general aviation aircraft and more than 600,000 registered pilots.

The presence of civil aviation has affected our economic way of life, it has made changes in our social and cultural viewpoints, and has had a hand in shaping the course of political history.

The sociological changes brought about by air transportation are perhaps as important as those it has brought about in the economy. People have been brought closer together and so have reached a better understanding of interregional problems. Industry has found new ways to do business. The opportunity for more frequent exchanges of information has been facilitated, and air transport is enabling more people to enjoy the cultures and traditions of distant lands.

In recent years, profound changes in technology and policy have had significant impacts on civil aviation and its supporting airport infrastructure. The industry continues to grow in numbers of aircraft, passengers and cargo carried, and markets served, from nonstop service on superjumbo aircraft between cities half-way across the planet, to privately operated “very light jets” between any of thousands of small airports domestically. Growth encouraged from technological advancements countered with increased constraints on the civil aviation system due to increased capacity limitations, security regulations, and financial constraints have resulted in ever increasing challenges to airport planning and design.

Civil aviation is typically considered in three sectors, commercial service aviation (more commonly known as air carriers or airlines), air cargo, and general aviation. Although the lines between these traditional sectors are becoming increasingly blurred, the regulations and characteristics regarding their individual operations are often mutually exclusive, and as such, those involved in airport planning and design should have an understanding of each sector.

Commercial Service Aviation

Commercial service aviation, supported by the world’s airlines, is by far the most well known, most utilized, and most highly regulated segment of civil aviation. It is the segment of the industry responsible for providing public air transportation between the world’s cities.

In the United States, domestic commercial air service accommodated nearly 650 million enplaning passengers in 2008, flying approximately 570 billion passenger-miles, reflecting a slight decline following the most recent surge in the growth of air transportation since the mid-twentieth century, and forecasted to carry more than 1 billion passengers by 2020, as illustrated in Fig. 1-1.

Intercity travel, of course, is not solely available through commercial service aviation. Intercity travel may be accommodated using either private modes of transportation, most commonly via private automobile travel, or through other modes of public transportation, such as bus, rail, or ship. Private automobile travel, accounts for

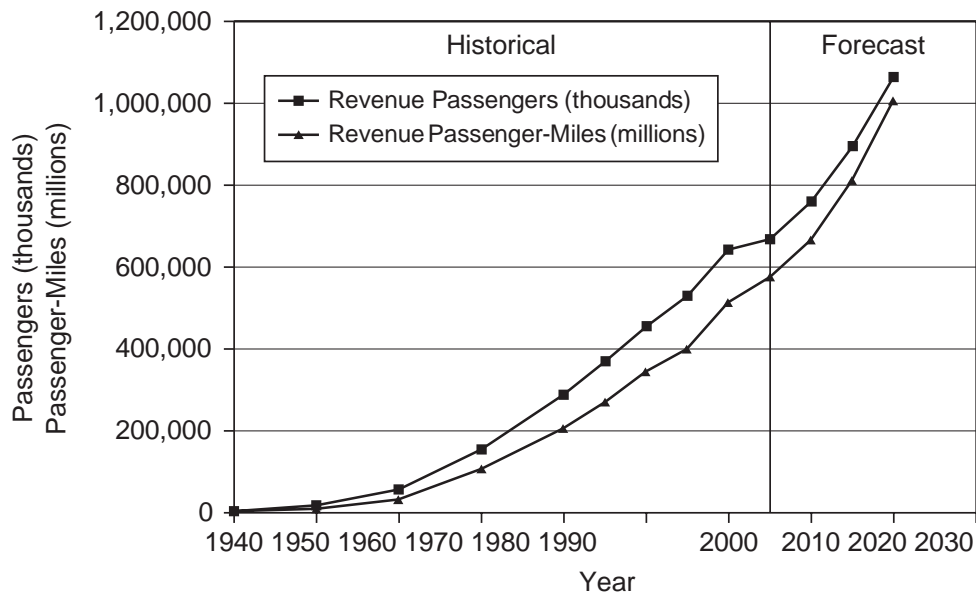


FIGURE 1-1 Total scheduled U.S. domestic passenger traffic: 1940 to 2020 (U.S. Bureau of Transportation Statistics).

nearly 90 percent of the total intercity (defined as trips greater than 50 mi in distance) travel in the United States, and public transportation or common-carrier travel (bus, rail, and air) accounts for the remaining portion.

Since the later half of the twentieth century, there has been a steady increase in overall travel, by private automobile and public transportation. Air transportation has had the greatest increase in overall passengers served. In the United States, this period has also witnessed a dramatic reduction in rail travel except for the rail markets in the Northeast United States. These relationships are shown in Table 1-1.

As illustrated in Table 1-2, air transportation in the United States accounts for the vast majority of domestic travel for trips exceeding 750 mi, and approximately one third of trips 500 to 750 mi in length. In all, air transportation accounts for approximately 70 percent of the United States' public intercity transportation. With the exception of travel to Canada and Mexico, air transportation serves nearly 100 percent of travel between the United States and international destinations.

In many parts of the world the use of private automobile is much less significant, and the use of rail transportation is much more prevalent. However, growth in commercial aviation in markets such as Europe and India are forecast to take greater numbers of passengers off the rails and onto airlines.

Much of the historical growth in air carrier transportation has been largely credited to the 1978 Federal Airline Deregulation Act, which allowed air carriers to freely enter and compete in domestic markets in

Year	Highway	Annual Growth, %	Civil Aviation	Annual Growth, %	Rail	Annual Growth, %
1960	1,272,078		33,399		21,261	
1965	1,555,237	22	57,626	73	17,388	-18
1970	2,042,002	31	117,542	104	10,771	-38
1975	2,404,954	18	147,400	25	8,444	-22
1980	2,653,510	10	219,068	49	11,019	30
1985	3,012,953	14	290,136	32	11,359	3
1990	3,561,209	18	358,873	24	13,139	16
1995	3,868,070	9	414,688	16	13,789	5
2000	4,390,076	13	531,329	28	14,900	8
2005	4,884,557	11	603,689	14	15,381	3

Source: U.S. Bureau of Transportation Statistics.

TABLE 1-1 U.S. Passenger Travel by Mode

the United States, and “open skies” agreements throughout the 1990s between nations to allow for more service between international destinations. The most recent growth in air transportation is attributable to changing airline business models, such as the emergence of the “low-cost carrier” (LCC), as well as increasing numbers of international open skies agreements that have proliferated since 2000.

Mode	One-Way Distance				
	50–499 Miles	500–749 Miles	750–999 Miles	1000–1499 Miles	1500+ Miles
Personal vehicle	95.4	61.8	42.3	31.5	14.8
Air	1.6	33.7	55.2	65.6	82.1
Bus	2.1	3.3	1.5	1.5	1.4
Train	0.8	1.0	0.9	0.7	0.8
Other	0.2	0.1	0.1	0.7	1.0
Total	89.8	3.1	2.0	2.3	2.8

TABLE 1-2 Percent of Trips by Mode for One-Way Travel Distance

Passenger Air Carriers

Commercial air carriers are defined in the United States as those that operate under Title 14 Part 121 of the U.S. Code of Federal Regulations to provide scheduled air transportation to the public. In the United States, these airlines are categorized by their annual revenues. Major airlines are those that generate at least \$1 billion in annual revenues. National carriers generate between \$100 million and \$1 billion in annual revenues, and regional carriers generate between \$25 million and \$100 million in annual revenues.

International air carriers receive operating certificates as prescribed by standards set by the International Civil Aviation Organization (ICAO) and defined by the country in which the airline is based. Historically, international air carriers were owned and operated by their nations, hence the term “flag” carriers. In recent years, most of the traditional international carriers have been transferred to private ownership. In addition, there has been an emergence of new international air carriers, most following the LCC model of serving point-to-point markets for fares that are on the whole far lower than their historical airline counterparts. The emergence of the LCC models in Europe and more recently in the Far East and India are resulting in a tremendous growth in aviation activity in these regions.

Air carriers using aircraft with less than 75 seats providing scheduled or unscheduled air charter services operating under Title 14 Part 121 of the U.S. Code of Federal Regulations are known as regional air carriers. Those carriers operating aircraft with less than 30 seats and those that operate under Title 14 Part 135 of the U.S. Code of Federal Regulations are known as commuter air carriers. If service frequency between city pairs is provided less than 5 times weekly, these carriers are known as air taxi operators.

In 2008, there were over 700 air taxis, commuter and small regional air carriers operating more than 2750 aircraft, over 50 percent of which were regional jet aircraft. Six hundred and forty two airports in the United States received service by small regional and commuter airlines. Regional and commuter air service is the sole provider of public air transportation to 492 airports in the United States (source: RAA). Throughout the 1980s and 1990s, commuter airline growth was encouraged by their increasing roles as code-share partners with major air carriers. In 2006, over 95 percent of all passengers traveling on commuter and regional air carriers purchased their tickets through these code-share partnerships. Table 1-3 illustrates the growth of the commuter and regional carriers since 1970.

International Air Transportation

Although international air transport was inaugurated in the mid-1930s, rapid growth did not begin until 1950. Since that time the average annual growth rate in the number of worldwide passengers

Year	Enplaned Passengers (thousands)	Passenger Miles (millions)	Average Trip Length (miles)	Number of Aircraft	Average Seats per Aircraft
1970	4,270	399	98	741	11
1975	7,243	689	110	948	13
1980	14,810	1,920	129	1,339	14
1985	26,000	4,410	173	1,745	19
1990	42,099	7,610	183	1,917	22
1995	55,800	11,461	213	2,109	30
2000	82,800	23,638	285	2,275	39
2003	112,120	43,100	384	2,189	45
2007	161,390	73,690	457	2,579	51

TABLE 1-3 Commuter and Regional Airline Statistics 1970 to 2007

was nearly 14 percent in the 1960s, slightly less than 7 percent in the 1970s, and slightly less than 5 percent in the 1980s. As illustrated in Fig. 1-2, worldwide growth in air transportation has increased by more than 60 percent between 1990 and 2005.

Air Cargo

Originating as the transport of mail by air in the early part of the twentieth century, air cargo has come to be defined as a \$40 billion

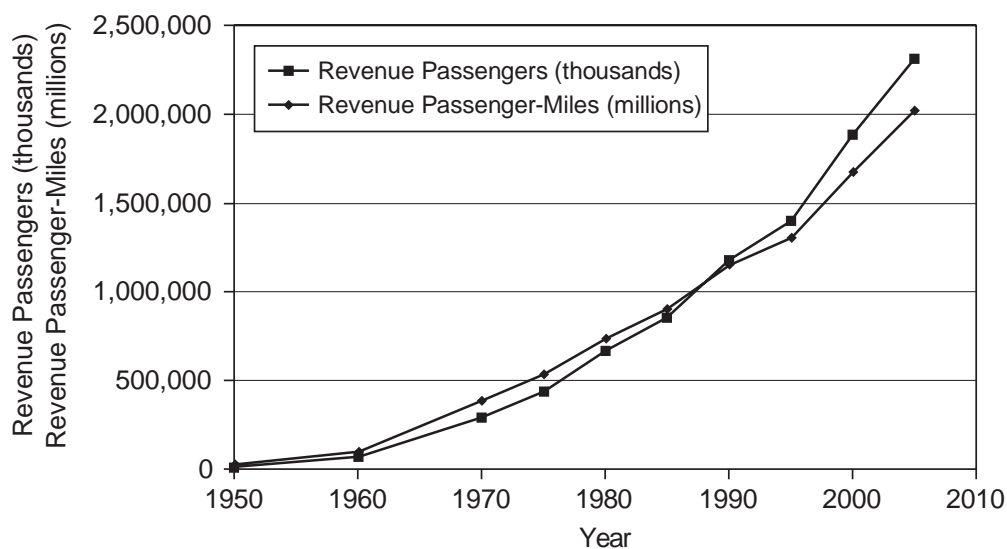


FIGURE 1-2 Worldwide growth in civil air transport passenger traffic.

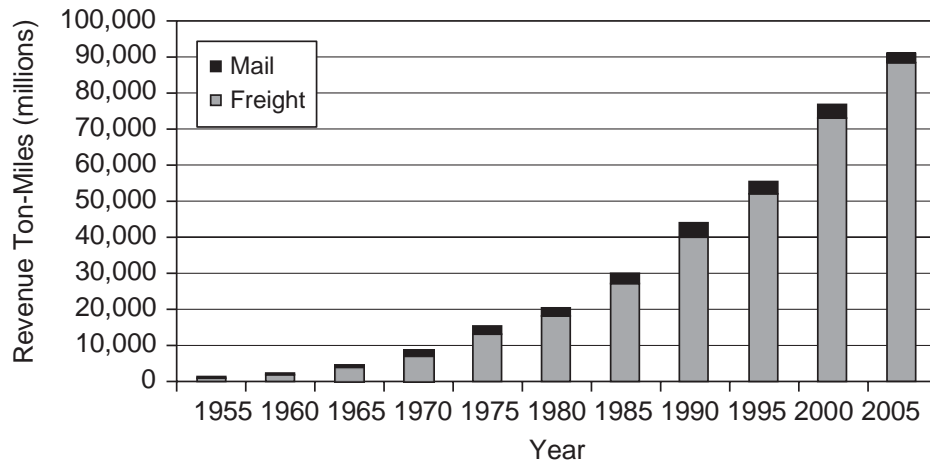


FIGURE 1-3 Worldwide air cargo.

industry focused on the air transport of mail, bulk freight, high-value goods, and all other revenue generating payload other than passengers and their luggage. As illustrated in Fig. 1-3, the transport of air cargo has increased tremendously since the mid-twentieth century, with its greatest rate of growth occurring since the late 1980s.

The top 50 carriers of air cargo in the global air cargo industry carried nearly one-hundred billion freight-ton-miles of cargo in 2008. Approximately 15 percent of the air cargo transported globally is performed by industry leaders and exclusive cargo carriers FedEx and UPS. The majority of air cargo is transported by air carriers, using aircraft designed exclusively for air cargo carriage, as well as on commercial passenger aircraft. Cargo carried on commercial passenger aircraft is often referred to as “belly cargo” as the cargo is stowed in the belly of the passenger aircraft. Cargo carried on aircraft designed exclusively for the carriage of cargo is often referred to as “palette” or “containerized” cargo, describing the containers within which cargo is stowed and the palettes used to load and unload cargo. Cargo operations using each type aircraft pose unique challenges for airport planning and design.

The geographic distribution of world air transport is also of interest. For statistical purposes ICAO has divided the world into six regions: Asia and Pacific, Europe, North America, Latin American, Caribbean, and the Middle East.

While slightly more than 60 percent of all traffic is generated in North America and Europe, the relative growth rates of traffic in the Asian and Pacific region, as well as in the Middle East, is expected to dominate worldwide air transportation growth, reflecting the growth of importance of this area in the political, social, and economic sectors.

Year	Asia and Pacific	Europe	North America	Latin America and Caribbean	Africa	Middle East
1972	8.4	35.9	47.6	4.4	2	1.7
1976	12.4	36.5	41	4.8	2.5	2.8
1980	15.5	35	38.6	5.5	2.6	2.8
1984	17.6	33	38	4.9	2.8	3.8
1988	19.8	31	39.3	4.6	2.2	3
1990	19.8	31.9	38.5	4.7	2.2	2.9
1992	21.2	27.3	41.8	4.7	2.2	2.8
2002	26.7	26.2	36.8	4.5	2.2	3.6
2005	33.2	25.4	30.3	4.3	2.1	4.7

TABLE 1-4 Percentage of Worldwide Distribution of Air Cargo Traffic

The air cargo market is forecast to triple between 2008 and 2030, led by the growth of air freight demand to China, as illustrated by the forecast percentage distribution of worldwide air cargo activity in Table 1-4. It is forecast that increasing percentages of air cargo would be shipped on dedicated cargo aircraft, requiring the need for expanded exclusive air cargo facilities at airports throughout the world.

General Aviation

General aviation is the term used to designate all flying done other than by the commercial air service carriers. General aviation operations range from local recreational flying to global business transport, performed on aircraft not operating under the federal aviation regulations for commercial air carriers.

While, by definition, general aviation operations carry no “commercial” passengers, it is estimated that more than 166 million people traveled by general aviation on nearly 20 million flights in 2008. During 2007, general aviation accounted for nearly 75 percent of all aircraft operations in the United States (source: FAA TAF). General aviation supports more than 1.3 million jobs and contributes more than \$103 billion annually to the United States economy.

As of 2008, there were approximately 225,000 general aviation aircraft registered in the United States and an estimated 340,000 aircraft worldwide (source: GAMA). These aircraft range in type and size from small single-engine propeller aircraft to large jet aircraft, to

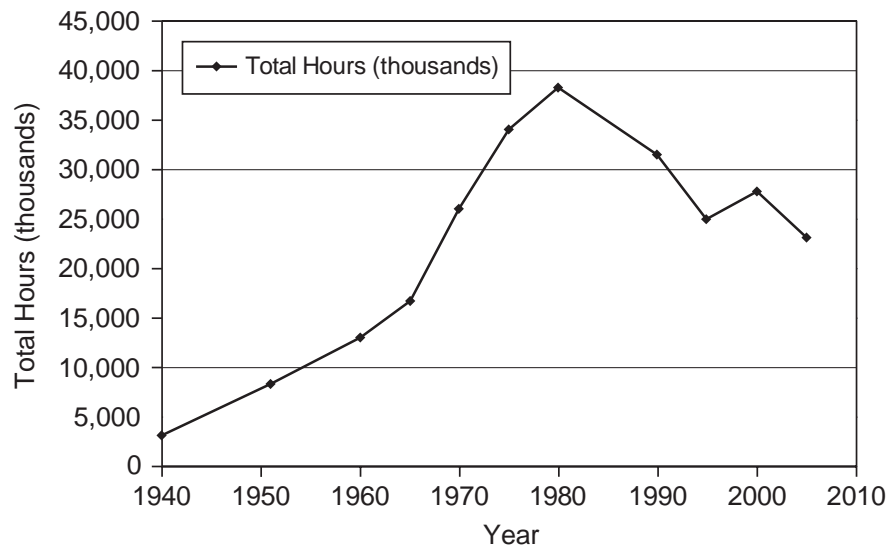


FIGURE 1-4 Total flight hours in general aviation aircraft in the United States.

“ultralight” aircraft, to helicopters. General aviation aircraft are served by nearly 20,000 landing facilities in the United States alone.

General aviation activity has experienced a decline in activity between 1980 and 2008, as illustrated in Fig. 1-4. Despite this recent historical decline, general aviation activity is forecast to increase with the proliferation of new aircraft technology which is expected to reduce the cost of general aviation operations. This forecast growth in general aviation, combined with new technologies, will pose interesting challenges for airport planning and design.

Civil Aviation Airports

Airports serving civil aviation range from private nonpaved strips that serve less than one privately operated aircraft per day to major international airports covering tens of thousands of acres, serving hundreds of thousands of flights and hundreds of millions of passengers annually. In the United States there are approximately 20,000 recognized civil airports, most of which are privately owned and closed to general public use. Of the approximately 5200 airports open to the public, approximately 700 are certified to accommodate commercial air service, with the remaining serving general aviation exclusively.

Airports currently serving at least 2500 enplaned passengers using commercial air service are known as *commercial service airports*. *Primary airports* are designated as those commercial airports serving at least 10,000 annual enplaned passengers. Airports serving less than 2500 annual enplaned passengers are considered *general aviation airports*. General aviation airports designed to accommodate smaller

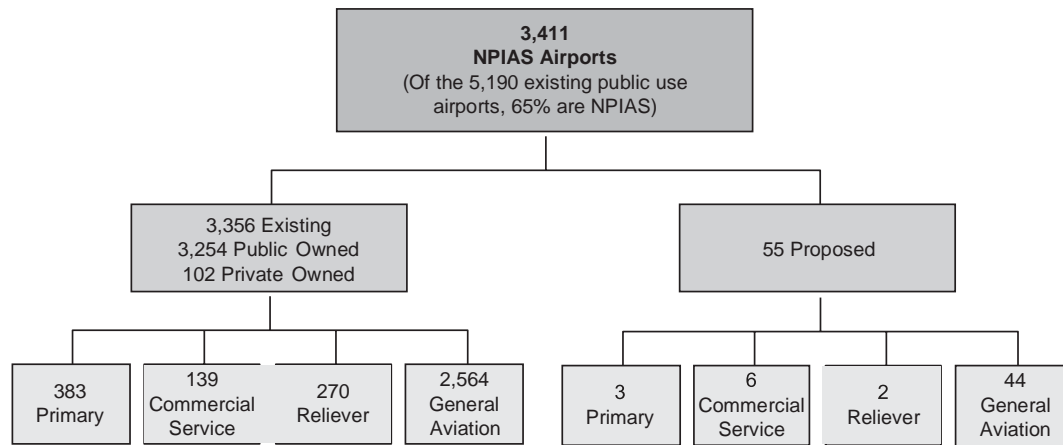


FIGURE 1-5 NPIAS categories.

single and twin-engine aircraft are considered basic utility airports. Those general aviation airports that accommodate larger aircraft are considered general utility airports.

The United States' Federal Aviation Administration (FAA), the governmental body with administrative oversight to the nation's civil aviation system, categorizes airports through its National Plan of Integrated Airport Systems (NPIAS). As illustrated in Fig. 1-5 the NPIAS recognizes approximately 3400 airports considered by the FAA to be essential to civil aviation and classifies these airports by the levels of commercial service activity within their respective stand- and metropolitan statistical areas (SMSAs).

Primary airports are further classified into what are known as "hub classifications" (not to be confused with the airline "hub and spoke" route models). The hub classifications used by the FAA are large hub primary, medium hub primary, small hub primary, and nonhub primary airports. Large hubs are those airports that account for at least 1 percent of the total annual passenger enplanements in the United States. Medium hubs account for at least 0.25 but less than 1 percent of the total passenger enplanements. Small hubs account for at least 0.05 percent but less than 0.25 percent, and non-hubs account for less than 0.05 percent but at least 10,000 annual enplaned passengers. The number of airports, by hub classification, is illustrated in Table 1-5.

Reliever airports are airports not currently serving regular commercial service but have been designated by the FAA as "general aviation-type airports that provide relief" when necessary to commercial service airports, typically by accommodating high volumes of general aviation activity within a metropolitan area and accommodating commercial service operations when the nearby commercial service airport is closed or otherwise cannot accommodate normal operations. Airports are typically given "reliever" status if they

Number of Airports	Airport Type	Percentage of 2006 Total Enplanements	Percentage of All Based Aircraft
30	Large hub primary	68.7	0.9
37	Medium hub primary	20.0	2.6
72	Small hub primary	8.1	4.3
244	Nonhub primary	3.0	10.9
139	Nonprimary commercial service	0.1	2.4
270	Relievers	0.0	28.2
2,564	General aviation	0.0	40.8
3,356 Existing NPIAS airports		99.9	89.8
16,459 Low-activity landing areas (Non-NPIAS)		0.1	10.2

TABLE 1-5 Number of NPIAS Airports by Hub Classification

are located within an SMSA of population of at least 5,000,000 or where passenger enplanements exceed 250,000 annually. In addition, the airport must have at least 100 aircraft based at the field or handle at least 25,000 itinerant operations annually. Reliever airports, although not serving regular commercial service operations, are among the busiest airports in the United States.

While most of the airports in the United States are privately owned and operated, the majority of public use airports are in fact publicly owned. Public use airports, and commercial service airports in particular, are typically owned and operated by local municipalities, counties, states, or some public “authority” typically overseen by representatives from a combination of local and regional jurisdictions. There are a few public use airports that are operated by private airport management companies but rarely do private firms actually own the property on which the airport is located. As such, in the United States, most planning and design programs at civil public use airports must go through extensive governmental processes for ultimate approval and often funding support.

While the United States has by far the greatest number of commercial service and general aviation airports in the world, many of the world’s largest and most important airports are located all over the globe. Table 1-6 lists the world’s busiest airports.

14 Airport Planning

Rank	Total Passenger Traffic 2008			Rank	Total Operations (Movements) 2008			Rank	Total Cargo (Metric Tons) 2008		
	City	Airport	Passengers		City	Airport	Movements		City	Airport	Metric Tons
1	Atlanta, GA	ATL	90,039,280	1	Atlanta, GA	ATL	978,824	1	Memphis, TN	MEM	3,695,561
2	Chicago, IL	ORD	69,353,654	2	Chicago, IL	ORD	881,566	2	Hong Kong, CN	HKG	3,656,724
3	London, GB	LHR	67,056,228	3	Dallas/Ft Worth, TX	DFW	656,310	3	Shanghai, CN	PVG	2,698,795
4	Tokyo, JP	HND	66,735,587	4	Denver, CO	DEN	615,573	4	Seoul, KR	ICN	2,423,717
5	Paris, FR	CDG	60,851,998	5	Los Angeles, CA	LAX	615,525	5	Anchorage, AK	ANC	2,361,088
6	Los Angeles, CA	LAX	59,542,151	6	Las Vegas, NV	LAS	579,949	6	Paris, FR	CDG	2,280,049
7	Dallas/Ft Worth, TX	DFW	57,069,331	7	Houston, TX	IAH	576,062	7	Frankfurt, DE	FRA	2,111,116
8	Beijing, CN	PEK	55,662,256	8	Paris, FR	CDG	559,812	8	Tokyo, JP	NRT	2,099,349
9	Frankfurt, DE	FRA	53,467,450	9	Charlotte, NC	CLT	536,253	9	Louisville, KY	SDF	1,973,965
10	Denver, CO	DEN	51,435,575	10	Phoenix, AZ	PHX	502,499	10	Singapore, SG	SIN	1,883,894
11	Madrid, ES	MAD	50,823,105	11	Philadelphia, PA	PHL	492,010	11	Dubai, AE	DXB	1,824,992
12	Hong Kong, CN	HKG	47,898,000	12	Frankfurt, DE	FRA	485,783	12	Miami, FL	MIA	1,806,769

13	New York, NY	JFK	47,790,485	13	London, GB	LHR	478,569	13	Los Angeles, CA	LAX	1,630,385
14	Amsterdam, NL	AMS	47,429,741	14	Madrid, ES	MAD	469,740	14	Amsterdam, NL	AMS	1,602,584
15	Las Vegas, NV	LAS	44,074,707	15	Detroit, MI	DTW	462,284	15	Taipei, TW	TPE	1,493,120
16	Houston, TX	IAH	41,698,832	16	Minneapolis/ St Paul, MN	MSP	446,840	16	London, GB	LHR	1,486,260
17	Phoenix, AZ	PHX	39,890,896	17	Amsterdam, NL	AMS	446,626	17	New York, NY	JFK	1,446,491
18	Bangkok, TH	BKK	38,604,009	18	New York, NY	JFK	435,750	18	Chicago, IL	ORD	1,324,820
19	Singapore, SG	SIN	37,694,824	19	Newark, NJ	EWR	433,463	19	Beijing, CN	PEK	1,303,258
20	Dubai, AE	DXB	37,441,440	20	Munich, DE	MUC	432,296	20	Bangkok, TH	BKK	1,173,131
21	San Francisco, CA	SFO	37,405,467	21	Beijing, CN	PEK	431,675	21	Indianapolis, IN	IND	1,025,895
22	Orlando, FL	MCO	35,622,252	22	Toronto, ON, CA	YYZ	429,829	22	Newark, NJ	EWR	889,121
23	Newark, NJ	EWR	35,299,719	23	San Francisco, CA	SFO	387,710	23	Tokyo, JP	HND	849,378

TABLE 1-6 The World's Busiest Airports

Rank	Total Passenger Traffic 2008			Rank	Total Operations (Movements) 2008			Rank	Total Cargo (Metric Tons) 2008		
	City	Airport	Passengers		City	Airport	Movements		City	Airport	Metic Tons
24	Detroit, MI	DTW	35,144,841	24	Salt Lake City, UT	SLC	387,695	24	Osaka, JP	KIX	845,496
25	Rome, IT	FCO	35,132,879	25	Los Angeles, CA	VNY	386,706	25	Luxembourg, LU	LUX	788,223
26	Charlotte, NC	CLT	34,732,584	26	New York, NY	LGA	377,940	26	Guangzhou, CN	CAN	685,866
27	Munich, DE	MUC	34,530,593	27	Phoenix, AZ	DVT	376,210	27	Kuala Lumpur, MY	KUL	661,212
28	London, GB	LGW	34,214,474	28	Miami, FL	MIA	372,635	28	Dallas/Pt Worth, TX	DFW	660,465
29	Miami, FL	MIA	34,063,531	29	Boston, MA	BOS	371,604	29	Atlanta, GA	ATL	655,277
30	Minneapolis/ St Paul, MN	MSP	34,032,710	30	Mexico City, MX	MEX	366,561	30	Brussels, BE	BRU	616,423

Source: Airports Council International.

TABLE 1-6 The World's Busiest Airports (*Continued*)

Contrary to the ownership structures in the United States, airports worldwide have traditionally been owned and operated by their respective federal governments through their ministries of transport, however many are increasingly becoming privatized to operate as for-profit business entities.

Historical Review of the Legislative Role in Aviation

Both in the United States and internationally, legislative actions taken by federal and state governments have had a profound impact on the growth of civil aviation and the planning and design of its airports.

As early as 1911, the Post Office Department showed an interest in civil aviation, particularly the transportation of mail by air, and from then on the department did much to encourage civil aviation. Attempts to obtain federal appropriations for air-mail began in 1912 but met with little success until 1916, when an appropriation for experimental purposes was made. In 1918, the first air-mail route in the United States was established between Washington, D.C. and New York City. At the start of this service the flying operations were conducted by the War Department but later that year the Post Office Department took over the entire operation with its own equipment and pilots. Service was inaugurated between New York City and Chicago in 1919 and was extended to San Francisco in 1920.

The Post Office Department, having demonstrated the practicality of moving mail by air, desired to turn over the operation to private enterprise. By 1925, the development work of the government had reached a stage where private operation seemed feasible. Accordingly, legislation permitting the Post Office Department to contract with private operators for the carriage of mail by air was provided by the Air Mail Act of 1925 (Kelly Act). However, it was not until 1926 that a number of contract routes were opened. Some of the early contractors were the Ford Motor Company, Boeing Air Transport (predecessor of United Airlines), and National Air Transport.

Air Commerce Act of 1926

The first year of the carriage of mail also saw the passage of the first federal law dealing with air commerce, the Air Commerce Act of 1926 (Public Law 64-254). Although this law provided regulatory measures, it did more to aid and encourage civil aviation than to regulate. The principal provisions of this act were as follows:

1. All aircraft owned by United States citizens operating in common-carrier service or in connection with any business must be registered.
2. All aircraft must be certificated and operated by certified airmen.

3. Authority was given to the Secretary of Commerce to establish air traffic rules.
4. The Secretary of Commerce was authorized to establish, operate, and maintain lighted civil airways.

In drafting the legislation, Congress relied considerably upon the precedents in maritime law. An analogy was utilized between the role of the government in meeting water navigation needs and the role of the government in air navigation. In water navigation these services included the signing, lighting, and marking of channels, safety inspection of ships and operating personnel, assistance in the development and improvement of ports and waterways, and laws concerning the operations of the industry. The provision of docks and terminal facilities were the responsibility of local government or the private sector of the economy. Therefore, the legislation was adopted in such a framework which held that airports were analogous to the docks of waterborne transportation [30].

Under the Air Commerce Act, control of air transportation was divided among several government agencies. The air-mail contracts were let by the Post Office Department, air-mail rates were subject to regulation by the Interstate Commerce Commission, and matters having to do with registration, certification, and airways were vested in the Bureau of Air Commerce in the Department of Commerce. The result of this divided jurisdiction was a lack of coordination in the efforts of government to develop the air transportation industry. In addition, the Act specifically prohibited any direct federal funding for airport development.

Civil Aeronautics Act of 1938

The Air Commerce Act of 1926 had been passed before the carriage of mail and passengers had developed into a substantial business enterprise. The failure of this legislation to provide adequate economic control led to wasteful and destructive competitive practices. The carriers had little security in their routes and therefore could not attract private investors and develop traffic volumes sufficient to achieve economic stability. These particular weaknesses in the existing legislation led to the enactment of the Civil Aeronautics Act of 1938 (Public Law 76-706). This act defined in a precise manner the role of the federal government in respect to the economic phases of air transport. It created one independent agency to foster and regulate air transport in lieu of the three agencies operating under the Air Commerce Act. This new agency was called the Civil Aeronautics Authority (not to be confused with the Civil Aeronautics Administration). It consisted of a five-member authority, a three-member air safety board, and an administrator. The five-member authority was principally concerned with the economic regulation of air carriers;

the safety board was an independent body for the investigation of accidents; and the concerns of the administrator dealt primarily with construction, operation, and maintenance of the airways.

During the first year and a half of its existence a number of organizational difficulties arose within the Civil Aeronautics Authority. As a result, President Franklin D. Roosevelt, acting within the authority granted to him in the Reorganization Act of 1939 (55 Stat. 561), reorganized the Civil Aeronautics Authority and created two separate agencies, the Civil Aeronautics Board and the Civil Aeronautics Administration. The five-member authority remained as an independent agency and became known as the Civil Aeronautics Board; the Air Safety Board was abolished and its functions given to the Civil Aeronautics Board; and the administrator became the head of an agency within the Department of Commerce known as the Civil Aeronautics Administration (CAA). The duties of the original five-member authority were unchanged, except that certain responsibilities, such as accident investigation, were added because of the abolition of the Air Safety Board. The administrator, in addition to retaining the functions of supervising construction, maintenance, and operation of the airways, was required to undertake the administration and enforcement of safety regulations and the administration of the laws with regard to aircraft operation. Subsequently, the administrator became directly responsible to the Secretary of Commerce (1950).

The Civil Aeronautics Act, like its predecessor the Air Commerce Act, authorized the federal government to establish, operate, and maintain the airways, but again, authorization for actively aiding airport development was lacking. The act, however, authorized the expenditure of federal funds for the construction of landing areas provided the administrator certified "that such landing area was reasonably necessary for use in air commerce or in the interests of national defense." The act also directed the administrator to make a survey of airport needs in the United States and report to Congress about the desirability of federal participation and the extent to which the federal government should participate.

In accordance with the requirements of the Act, the Civil Aeronautics Authority conducted a detailed survey of the airport needs of the nation. An advisory committee was appointed, composed of representatives of interested federal agencies (both military and civil), state aviation officials, airport managers, airline representatives, and others. A report was submitted to Congress on March 23, 1939 (House Document 245, 76th Congress, 1st Session). Some of the more important recommendations in this report were as follows:

1. Development and maintenance of an adequate system of airports and seaplane bases should be recognized in principle as a matter of national concern.

2. Such a system should be regarded, under certain conditions, as a proper object of federal expenditure.
3. In passing upon applications for federal expenditure on airport development or improvement, the highest preference should be given to airports which are important to the maintenance of safe and efficient operation of air transportation along the major trade routes of the nation; and to those rendering special service to the national defense.
4. At such times as the national policy includes the making of grants to local units of government for public-works purposes, or any work-relief activity, a proportion of the funds involved should be allocated to airport purposes. Such purposes should be given preference as rendering an important service to the localities concerned and at the same time being of particular importance to the nation's commerce and defense.
5. Whenever emergency public-works programs may be terminated, or when such programs may be curtailed to a degree not enabling adequate airport development to continue, or when the Congress for other reasons may determine federal assistance for airports should be continued through annual appropriations for that purpose, based upon annual reports which should include a review of the general status of the nation's airport system and of the work recently done or currently in course of being done, and recommendations for future work in the interest of developing and maintaining a system adequate to national needs, expenditures at these periods should be limited to projects of exceptional national interest.
6. In connection with such public-works or work-relief programs as normally involve joint contributions by the federal government and by local government, there should be a provision of supplementary funds to enable the federal government to increase its share of the total expense, in any proportion justified by the importance of the project.
7. All applications for federal airport grants from such a supplementary appropriation should be presented through agencies of state government.
8. In deciding upon the wisdom and propriety of granting any such applications, and the priority that should be given to them, consideration should be given to the aeronautical policy of the state in question, with reference to such matters as the state's policy in protecting the approaches to airports; the state's policy in respect to the employment of any taxes collected on the fuel used in aircraft; and any measures taken by the state to insure the proper maintenance of airports and the maintenance of reasonable charges for the services given them.

9. The detailed plans for the location and development of any airport with respect to which there is federal contribution of any kind, should be subject to the approval of the federal agency charged with the establishment of civil airways, landing areas, and necessary air navigation facilities.
10. There should be no direct federal contribution to the cost of maintaining airports, other than federal airports; except that the administrator of the Civil Aeronautics Authority may, in accordance with the Civil Aeronautics Act and so far as available funds permit, assume the cost of operating any lighting equipment and other air navigation facility as a part of the cost of operation of the federal airways system.

The airport survey submitted in 1939 was updated with new studies completed in 1940. Continuing studies were made through the war years. While first importance was attached to the military requirements, care was taken whenever possible to anticipate the needs of postwar civil aviation. During the war years the federal government, through the CAA, spent \$353 million for the development of military landing areas in the continental United States. This does not include funds spent by the military agencies. During the same period the CAA spent \$9.5 million for the development of landing areas in the United States solely for civil purposes.

Federal Airport Act of 1946

At the end of World War II, over 500 airports constructed for the military by the CAA were declared surplus and were turned over to cities, counties, and states for airport use. The interest in adequate airport facilities by various political subdivisions of government continued. The needs were made known to Congress by various interests. As a result, the House of Representatives passed a resolution (H.R. 598, 78th Congress) directing the CAA to make a survey of "need for a system of airports and landing areas throughout the United States" and report back to Congress.

The results of this survey were completed in 1944 (House Document 807, 78th Congress, 2nd Session) and contained the following principal recommendations:

1. That Congress authorize an appropriation to the Office of the Administrator of Civil Aeronautics not to exceed \$100 million annually to be used in a program of federal aid to public agencies for the development of a nationwide system of public airports adequate to meet the present and immediate future needs of civil aeronautics. The administrator be authorized to allocate such funds for any construction work involved in constructing, improving, or repairing an airport, including the construction, alteration, and repair of airport

buildings other than hangars, and the removal, lowering, marking, and lighting of airport obstructions; for the acquisition of any lands or property interest necessary either for any such construction or to protect airport approaches; for making field and specifications; supervising and inspecting construction work, and for any necessary federal expenses in the administration of this program.

2. That such a program can be conducted in cooperation with the state and other nonfederal public agencies on a basis to be determined by the Congress. That the federal contribution be determined by the Congress in passing the necessary enabling legislation. A good precedent for the proportionate sharing of costs exists in the public-roads program which has operated satisfactorily for many years on a 50-50 basis.
3. That any project for which federal aid is requested must meet with the approval of the administrator of Civil Aeronautics as to scope of development and cost, conform to Civil Aeronautics Administration standards for location, layout, grading, drainage, paving, and lighting and all work thereon be subject to the inspection and approval of the Civil Aeronautics Administration.
4. In order to participate in the federal-aid program, a state shall
 - a. Establish and empower an official body equipped to conduct its share of the program.
 - b. Have legislation adequate for the clearing and protection of airport approaches, and such other legislation as may be necessary to vest in its political subdivisions all powers necessary to enable them to participate through the state as sponsors of airport projects.
 - c. Have no special tax on aviation facilities, fuel, operations, or businesses, the proceeds of which are not used entirely for aviation purposes.
 - d. Ensure the operation of all public airports public interest, without unjust discrimination or unreasonable charges.
 - e. Ensure the proper operation and maintenance of all public airports within its jurisdiction.
 - f. Make airports developed with federal aid available for unrestricted use by United States government aircraft without charge other than an amount sufficient to cover the cost of repairing damage done by such aircraft.
 - g. Require the installation at all airports for which federal funds have been provided for a standard accounting and fiscal reporting system satisfactory to the administrator.

5. That sponsors of projects be required to enter into contracts with the Civil Aeronautics Administration ensuring the proper maintenance and protection of airports developed with federal aid and their operation in the public interest.

The recommendations contained in the airport needs survey report were written into an airport development bill, introduced into the House of Representatives (H.R. 5024) but no action was taken on it. After extensive hearings in both houses of Congress, the Senate passed an airport bill (S. 2) in 1945 and later that year the House passed a bill (H.R. 3615). The language in these two bills differed in several respects. One of the principal differences was the method employed in channeling funds to the municipalities. The Senate bill provided that funds be channeled to the municipalities through appropriate state aviation organizations unless a state did not have an appropriate agency to handle the matter. The House bill permitted channeling of funds either through the state or directly to a municipality or other political subdivision of government. The substitute bill agreed to in conference conformed more nearly to the House language. Another difference had to do with the size of the discretionary fund, which, instead of being apportioned among the states by a fixed formula, would be available for use by the administrator at his sole determination. The House bill provided 25 percent of the total appropriation for airport development as a discretionary fund, the Senate bill 35 percent. The compromise reached in conference retained the House version. Other differences which were worked out in conference concerned whether or not the costs of the acquisition of land and interest in airspace should be eligible for federal aid, project sponsorship requirements, and the reimbursement for damage to public airports caused by federal agencies.

The conference report was approved by the Congress and the Federal Airport Act of 1946 was enacted (Public Law 79-377). Known as the Federal Airport-Aid Program, appropriations of \$500 million over a 7-year period were authorized for projects within the United States plus \$20 million for projects in Alaska, Hawaii, Puerto Rico, and the Virgin Islands. In 1950, the 7-year period was extended an additional 5 years (Public Law 81-846). However, annual appropriations approved by Congress were much less than the amounts authorized by the act.

The original act provided that a project shall not be approved for federal aid unless "sufficient funds are available for that portion of the project which is not to be paid by the United States."

Local governments often required 2 to 3 years to make arrangements for raising funds. Most of the larger projects are financed locally through the sale of bonds. This method of financing requires legislation at the local level and, in some cases, also at the state level. General obligation bonds normally require approval by the electorate.

Programs to inform the public on the needs for airport improvement must be carefully planned and executed. Thus, after the completion of these events, local governments frequently found that sufficient federal funds were not appropriated to match local funds, and the projects were delayed. Another complaint of local governments had been that Congress failed to fulfill its obligation, since the amount appropriated by Congress fell far short of the amount authorized by the Federal Airport Act. These deficiencies as well as other matters were incorporated in a bill (S. 1855). Hearings on the bill were held before the subcommittee of the Committee on Interstate and Foreign Commerce of the United States Senate in 1955. Representatives of the Council of State Governments, the American Municipal Association, the National Association of State Aviation Officials, airport and industry trade associations, and individuals were unanimous in the feeling that air transportation had reached a stage of maturity where many airports were woefully inadequate and greater financial assistance from the federal government would be required to meet the current needs of aviation. After much debate, the bill was approved by the President (Public Law 84-211).

This amending act made no change in the basic policies and purposes expressed in the original act. There were no changes in the requirements with respect to the administration of the grants authorized, such as the distribution and apportionment of funds, the eligibility of the various types of airport construction, sponsorship requirements, etc. The primary purpose of the act was to substitute for the procedure of authorizing annual appropriations for airport projects, provisions granting substantial annual contract authorization in specific amounts over a period of four fiscal years. Airport sponsors were thus furnished assurance that federal funds would be available at the time projects were to be undertaken.

This law provided \$40 million for fiscal year 1956 and \$60 million for each of fiscal years 1957, 1958, and 1959 for airport construction in the continental United States. It also provided \$2.5 million in fiscal year 1956 and \$3 million for the three succeeding fiscal years for airport construction in Alaska, Hawaii, Puerto Rico, and the Virgin Islands. Besides the \$42.5 million made available in fiscal year 1956 by Public Law 84-211, Congress approved an additional appropriation of \$20 million for airport projects.

Federal Aviation Act of 1958

For a number of years there had been a growing concern about the division of responsibility in aviation matters among different agencies of the federal government. Unlike highway or other forms of transport, aviation is unique in its relation to the federal government. It was historically the only mode whose operations are conducted almost wholly within federal jurisdiction, and one subject to little or

no regulation by states or local authorities. Thus, the federal government bears virtually complete responsibility for the promotion and supervision of the industry in the public interest. The military interest and the entire national defense concept are also intimately related to aviation.

Recognizing that the demands on the federal government in the years ahead would be substantial, the director of the Bureau of the Budget requested a review of aviation-facilities problems in 1955. A report was issued later that year recommending that a study of "long-range needs for aviation facilities and aids be undertaken" and that such a study be made under the direction of an individual of national reputation.

President Dwight D. Eisenhower accepted these recommendations and appointed Edward P. Curtis as his Special Assistant for Aviation Matters in 1957. Curtis was charged with the responsibility of preparing a comprehensive aviation-facilities plan which would "provide the basis for the timely installation of technically adequate aids, for optimum coordination of the efforts of the civil and military departments, and for effective participation by state and local authorities and the aircraft operators in meeting facilities requirements." Curtis completed his report and submitted it to the President. In this report Curtis stated that "it has become evident that the fundamental reason for our previous failures lies with the inability of our governmental organizations to keep pace with the tremendous growth in private, commercial, and military aviation which has occurred in the last 20 years." Curtis recommended the consolidation of all aviation functions, other than military, into one independent agency responsible only to the President. However, the report recognized that to "develop new management structures and policy, to coordinate proposals within the executive branch and to obtain legislation implementing a new permanent organization might be as long as 2 or 3 years." The most urgent matter requiring attention was in the area of air traffic control. The collision of two aircraft over the Grand Canyon in 1956 provided the impetus for rapid legislative action for remedying midair collisions. Curtis recommended that, as an interim measure, there be created an Airways Modernization Board whose function was to "develop, modify, test, and evaluate systems, procedures, facilities, and devices, as well as define the performance characteristics thereof, to meet the needs for safe and efficient navigation and traffic control of all civil and military aviation except for those needs of military agencies which are peculiar to air warfare and primarily of military concern, and select such systems, procedures, facilities, and devices which will best serve such needs and will promote maximum coordination of air traffic control and air defense systems." The board was to consist of the Secretary of Commerce, the Secretary of Defense, and an independent chairman.

Congress was receptive to this recommendation and passed the Airways Modernization Act of 1957 (Public Law 85-133) establishing the board for a 3-year term.

In the meantime, there were more midair collisions and reports of near misses were given wide circulation. Costly disagreements between the CAA and the military on the type of navigational aids to be used on the airways no doubt also spurred Congressional action. As a result, instead of taking 2 or 3 years to create a single aviation agency as was predicted, the Congress acted favorably on the legislation within a year of the passage of the Airways Modernization Act. This legislation is known as the Federal Aviation Act of 1958 (Public Law 85-726). This law superseded the Civil Aeronautics Act of 1938 but not the Federal Airport Act of 1946.

The principal provisions of the law insofar as organizational changes were concerned are as follows:

1. The Federal Aviation Agency was created as an independent agency with an administrator directly responsible to the President. The agency incorporated the functions of the Civil Aeronautics Administration and the Airways Modernization Board, both of which were abolished.
2. The Civil Aeronautics Board was retained as an independent agency including all its functions except its safety rule-making powers, which were transferred to the Federal Aviation Agency.

Creation of the U.S. Department of Transportation

For many years it had been argued that there had been a proliferation of federal activities with regard to transportation. For example, the Bureau of Public Roads was part of the Department of Commerce whereas the Federal Aviation Agency was an independent agency. It was felt by different transport interests that there was a lack of coordination and effective administration of the transportation programs of the federal government resulting in a lack of a sound national transportation policy. It is interesting to note that the first legislative proposal in this direction dates back to 1874. However, in recent years, the involvement of the federal government in the development of the transportation systems of the nation has been enormous, requiring much more coordination among federal transport activities than ever before. With this as a background, a Cabinet-level Department of Transportation (DOT) was created headed by the Secretary of Transportation (Public Law 89-670). The department began to function on April 1, 1967.

The agencies and functions transferred to the Department of Transportation related to air transportation included the Federal Aviation Agency in its entirety and the safety functions of the Civil

Aeronautics Board, including the responsibility for investigating and determining the probable cause of aircraft accidents, and its appellate safety functions involving review on appeal of the suspension, modification, or denial of certificates or licenses. The name of the Federal Aviation Agency was changed to the Federal Aviation Administration (FAA). The administrator is still appointed by the President but reports directly to the Secretary of Transportation.

A National Transportation Safety Board (NTSB) was established by the same act which created the Department of Transportation to determine "the cause or probable cause of transportation accidents and reporting the facts, conditions, and circumstances relating to such accidents" for all modes of transportation. Although created by the act which established the DOT, the board in carrying out its functions is "independent of the secretary and other offices and officers of the department." The board consists of five members appointed by the President and annually reports directly to Congress.

The creation of the DOT did not alter any legislation in the Federal Aviation Act of 1958, with the exception of the transfer of the safety functions from the Civil Aeronautics Board to the National Transportation Safety Board. In the act of establishing the new department, however, there was a statutory requirement to establish an Office of Noise Abatement to provide policy guidance with respect to interagency activities related to the reduction of transportation noise. With the introduction of jet aircraft in 1958 the complaints against aircraft noise increased significantly. As a result, in 1968 the Federal Aviation Act was amended by Congress (Public Law 90-411) to require noise abatement regulations. Its principal purpose was to establish noise levels which aircraft manufacturers cannot exceed in the development of new aircraft.

Airport and Airway Development Act of 1970

In the mid-1960s, as air traffic was expanding at a fairly rapid pace, air traffic delays getting into and out of major airports began to increase rapidly. Along with the delays in the air, congestion was also taking place on the ground in parking areas, on access roads, and in terminal buildings. It was evident that to reduce congestion substantial financial resources would be required for investment in airway and airport improvements. For airports alone it was estimated that \$13 billion in new capital improvements would be required for public airports in the period 1970 to 1980. The amount of money authorized by the Federal Airport Act of 1946 was insufficient to assist in financing such a vast program. The normal and anticipated sources of revenue available to public airports were also not sufficient to raise the required funds for capital expenditures. It was argued that much of the congestion in the air at major airports was due to a lack of funds to modernize the airways system. Funds for airport development came from the budget of the FAA authorized by Congress each year

and not from the Federal Airport Act. The deficiencies in airport and airway development were documented in several reports. It was the consensus of industry and government that the only way to provide the funds needed for airports and airways was through increased or new taxes imposed upon the users of the air transport system. It was also argued that the revenues from these taxes should be specifically earmarked for aviation and not go to the general fund. The concept of establishing a trust fund similar to that of the national highways program was agreed upon. Finally after much debate in Congress, the Airport and Airway Development Act of 1970 and the Airport and Airway Revenue Act of 1970 were enacted (Public Law 91-258).

As finally passed, the act was divided into two sections: Title I detailed the airport assistance programs and established a financing program for airport grants, airways hardware acquisition, and research and development; Title II created the Airport and Airway Trust Fund and established the pattern of aviation excise taxes which would provide the resources upon which the Title I capital programs would depend through 1980. The excise taxes adopted consisted of a tax on domestic passenger tickets, a head tax on international passenger departures, a flowage tax on all fuel used by general aviation, and tax on all air cargo waybills. In addition, an annual aircraft registration tax was levied on all aircraft (commercial and general aviation) plus an annual weight surtax for all aircraft weighing in excess of 2500 lb. Finally, revenues from existing taxes on aircraft tires and tubes were transferred from the Highway Trust Fund to the Airport and Airway Trust Fund.

The amount of these excise taxes were changed in the Omnibus Budget Reconciliation Act of 1990 (Public Law 101-508) and currently consists of a 10 percent tax on domestic passenger tickets, a \$6 head tax on international passenger departures, a 17.5 cents a gallon flowage tax on all fuel used by general aviation, and a 6.25 percent tax on all air cargo waybills.

Significant changes from the Federal Airport Act were as follows:

1. The provision of funds to local agencies for airport system planning and master planning
2. The emphasis on airports served by air carriers and general aviation airports to relieve congested air carrier airports
3. The provision of funds for commuter service airports
4. The requirement that the FAA issue airport operating certificates to ensure that airports were adequately equipped for safe operations
5. Provision of requirements to ensure that airport projects did not adversely affect the environment and were consistent with long-range development plans of the area in which the project was proposed

6. Provision for terminal facility development in non-revenue producing public areas
7. The requirement that the FAA develop a National Airport System Plan (NASP)

To be eligible for federal aid, airport ownership was required to be vested in a public agency and the airport must be included in the NASP. This plan was reviewed and revised as necessary to keep it current. It was prepared by the FAA and submitted to Congress by the Secretary of Transportation. The plan specified, in terms of general location and type of development, the projects considered necessary to provide a system of public airports adequate to anticipate and meet the needs of civil aeronautics. These projects included all types of airport development eligible for federal aid under the act and were not limited to any classes or categories of public airports. The plan was based on projected needs over 5- and 10-year periods.

Because of the mounting public concern for the enhancement of the environment, the act specifically stated that authorized projects provide for the protection and enhancement of the natural resources and the quality of the environment of the nation. The Secretary of Transportation was required to consult with the Secretary of the Interior and the Secretary of Health, Education and Welfare regarding the effect of certain projects on natural resources and whether "all possible steps have been taken to minimize such adverse effects." The Act required that airport sponsors provide the "opportunity for public hearings for the purpose of considering the social, economic, and environmental effect on any project involving the location of an airport, the location of a runway, and a runway extension." In addition, the National Environmental Policy Act of 1969 (Public Law 91-190), supported by a Presidential Executive Order (11514, March 5, 1970), required the preparation of detailed environmental impact statements for all major airport development actions significantly affecting the quality of the environment. The environmental impact statement was required to include the probable impact of the proposed project on both the human and natural environment, including impact on ecological systems such as wildlife, fish, and marine life, and any probable adverse environmental effects which could not be avoided if the project was implemented. The Act also stipulated that no airport project involving "airport location, a major runway extension, or runway location" could be approved for federal funding unless the governor of the state in which the project was located certified to the Secretary of Transportation that there was reasonable assurance that the project would comply with applicable air and water quality standards. Finally the project had to be consistent with the plans of other agencies for development of the area, and the airport sponsor had to assure the government that adequate housing was available for any displaced people. The Aviation Safety and Noise Abatement

Act of 1979 (Public Law 96-193) amended this legislation to place increased emphasis on reducing the noise impacts of airports. Thus one of the principal differences between the Federal Airport Aid Program and the Airport Development Aid Program was the emphasis on environmental protection in the latter.

The Airport and Airway Development Act made no mention concerning specific standards for determining airports to be included in the National Airport System Plan. It did state, however, that

The Plan shall set forth, for at least a ten-year period, the type and estimated cost of an airport development considered by the Secretary to be necessary to provide a system of public airports adequate to anticipate and meet the needs of civil aeronautics, to meet the requirements in support of the national defense as determined by the Secretary of Defense, and to meet the special needs of the Postal Service. In formulating and revising the plan, the Secretary shall take into consideration, among other things, the relationship of each airport to the rest of the transportation system in the particular area, to the forecasted technological developments in aeronautics, and to developments forecasted in the other modes of transportation.

With this and other policy guidelines, the FAA developed entry criteria which described a broad and balanced airport system. The 1980 NASP, for example, included about 3600 airport locations, indicating that the federal interest in developing a basic airport system extended well beyond the major airports with scheduled airline service. In an effort to provide a safe and adequate airport for as many communities as possible, NASP criteria were developed to include the general aviation airports which serve smaller cities and towns.

The NASP airport entry criteria evolved from both policy and legislative considerations and focused on two broad categories of airports, those with scheduled service and those without significant scheduled service in the general aviation and reliever category. Airports with scheduled service were included in NASP because of their use by the general public, legislative provisions which specifically designated airports to receive development funds, and their use by CAB certified carriers. Commuter airports were identified in NASP starting in 1976, when legislation was enacted which designated them as a type of air carrier airport and provided them with special development funds. About 70 percent of the airports in the NASP were general aviation locations which met the criteria of viability because of the number of based aircraft or aircraft activity, and which provided reasonable access for aircraft owners and users to their community. Reliever airports have been included as a separate NASP category since the 1960s when Congress designated special funding for the purpose of relieving congestion in large metropolitan areas by providing additional general aviation capacity.

Airline Deregulation Act of 1978

The Airline Deregulation Act (Public Law 95-504) was passed by Congress in October 1978. This legislation eliminated the statutory authority for the economic regulation of the passenger airline industry in the United States. It provided that the Civil Aeronautics Board would be abolished in 1985. The legislation was intended to increase competition in the passenger airline industry by phasing out federal authority to exercise regulatory controls during the period of time between 1978 and 1985. The principal provisions of this legislation:

1. Required the CAB to place maximum reliance on competition in its regulation of interstate airline service, while continuing to ensure the safety of air transportation; to maintain service to small communities; and to prevent practices which were deemed anticompetitive in nature.
2. Required CAB approval of airline acquisitions, consolidations, mergers, purchases, and operating contracts; the burden to prove that an action was anticompetitive in nature was placed upon the party challenging that action.
3. Permitted carriers to change rates within a range of reasonableness from the standard industry fare without prior CAB approval; the CAB was authorized to disallow a fare change if it considered the change predatory.
4. Provided interstate carriers an exemption from state regulation of rates and routes.
5. Required the CAB to authorize new routes and services that were consistent with the public convenience and necessity.
6. Allowed carriers to be granted operating rights to any route on which only one other carrier was providing service and on which other airlines were authorized to provide service but were not actually providing a minimum level of service. If more than one airline was providing service on this route, the CAB was required to determine if the granting of additional route authority was consistent with the public convenience and necessity before allowing additional carriers to service the route. An airline not providing the specified minimum level of service on a route (dormant authority) could begin providing such service and retain its authority. Otherwise, the CAB was required to revoke the unused authority.
7. Provided for an automatic market entry program, whereby airlines could begin service on one additional route each year during the period 1979 to 1981 without formal CAB approval. Each carrier was also permitted to protect one of its existing routes each year by declaring it as ineligible for automatic market entry by another airline.

8. Authorized the CAB to order an airline to continue to provide “essential air transportation service” and, for a 10-year period, to provide subsidies or seek other willing carriers to ensure the continuation of essential service.
9. Required the CAB to determine within 1 year of the enactment of the legislation what it considered to be “essential air transportation service” for each point being serviced at the time of enactment of the legislation.
10. Required the CAB and the Department of Transportation to determine mechanism by which the state and local governments should share the cost of subsidies from the federal government to preserve small community air service and to make policy recommendations to Congress on this matter.
11. Exempted from most CAB regulation commuter aircraft weighing less than 18,000 lb and carrying fewer than 56 passengers.
12. Made commuter and intrastate air carriers eligible for the federal loan guarantee program.
13. Provided that the domestic route authority of the board would cease in 1981; its authority over domestic fares, acquisitions, and mergers would cease in 1983; and the board would be abolished (sunset) in 1985.
14. Provided that after the board was abolished, the local service carrier subsidy program was to be transferred to the Department of Transportation; the foreign air transportation authority of the board was to be transferred to the Transportation and Justice Departments, in consultation with the State Department; and the mail subsidy program was to be transferred to the U.S. Postal Service.

Impact of Airline Deregulation

In the United States prior to 1978, air carriers applied to the Civil Aeronautics Board (CAB) for permission to serve markets. The CAB granted air carriers service to markets with a determined operating and fare schedule. Upon deregulation, air carriers freely entered new markets, increasing the number of markets served, increasing competition and lowering overall airfares. To maximize their market share in the industry, several air carriers concentrated their route structures on one or more “hub” airports. The early years of this hub and spoke route system resulted in the greatest growth in commercial aviation in its history. The beginning of the twenty-first century exposed many of the weaknesses in the airline hub and spoke model, including the increased costs of operating through congested hub airports, increasing fuel and other operating expenses, combined with the ability for

the public to use the Internet to avoid extraordinarily high full fares thereby reducing the overall revenue stream of the airlines. The air carriers dependent on their traditional “legacy” business models fell into serious financial distress, and combined with the short-term drop in air travel demand following the terrorist attacks of September 11, 2001, many fell into bankruptcy to emerge under more efficient business models years later.

These more efficient business models were based on the emergence of “low-cost carriers” or LCCs, that operate primarily on a market-based origin to destination route network and price fares relative to operating costs, rather than solely by passenger demand. The LCC airline model has been the largest growth segment of the domestic airline industry in the United States.

The Airport and Airway Improvement Act of 1982

In 1982, Congress enacted the Airport and Airway Improvement Act (Title V of the Tax Equity and Fiscal Responsibility Act of 1982, Public Law 97-248). This act continued to provide funding for airport planning and development under a single program called the Airport Improvement Program (AIP). The Act also authorized funding for noise compatibility planning and implementing noise compatibility programs contained in the Noise Abatement Act of 1979 (Public Law 96-193). It required that to be eligible for a grant the airport must be included in the National Plan of Integrated Airport Systems (NPIAS). The NPIAS, the successor to the National Airport System Plan (NASP), is prepared by the FAA and published every 2 years and identifies public use airports considered necessary to provide a safe, efficient, and integrated system of airports to meet the needs of civil aviation, national defense, and the postal service.

The Airport and Airway Improvement Act has been amended several times resulting in significant changes in the provisions of the act and in the appropriations authorized under the Act. These amendments are included in the Continuing Appropriations Act of 1982 (Public Law 97-276), the Surface Transportation Assistance Act (Public Law 97-424), the Airport and Airway Safety and Capacity Expansion Act of 1987 (Public Law 100-223), the Airway Safety and Capacity Expansion Act of 1990 (Public Law 101-508), and the Airport and Airway Safety, Capacity, Noise Improvement and Intermodal Transportation Act of 1992 (102nd Congress H.R. 6168).

This legislation, as amended, significantly increased the level of federal funding for airports to an aggregate total of more than \$14 billion for the period from 1982 through 1992.

The Aviation Safety and Capacity Act of 1990

In response to issues concerning the provision of limited AIP funds to the largest airports, thereby leaving the smaller airports with little in

capital improvement funding support, in 1990, Congress passed the Aviation Safety and Capacity Expansion Act. This act established the policy of allowing airports to impose a passenger facility charge (PFC) to supplement their capital improvement programs, while allowing greater amounts of AIP funding to be allocated to smaller airports with capital improvement needs. Under this Act, an airport applied to collect a \$1, \$2, or \$3 charge, on any passenger enplaning at the airport. The fee would be collected by the air carriers, upon purchase of a ticket. Revenues generated by PFCs would then be spent by the airport that generated the revenue on allowable costs associated with certain capital improvement projects approved by the FAA that enhance safety, security, or capacity, or increase air carrier competition. In 2001, the maximum allowable PFC was raised to \$4.50. As of June 2007, approximately \$58.6 billion in PFCs have been collected at 367 airports nationwide. More than 1500 projects utilizing PFC revenues have been approved since the 1990 Aviation Safety and Capacity Expansion Act introduced the PFC program.

AIR-21: The Wendell Ford Aviation Investment Act for the 21st Century

In April 2000, funding for airport planning and design through the AIP and PFC programs was increased with the Wendell H. Ford Aviation Investment and Reform Act for the Twenty-First Century, known as AIR-21 (Public Law 106-181). This funding increase was designed to assist larger airports which have become highly congested, as well as smaller airports struggling to preserve commercial air service.

The AIR-21 Act was introduced at a time when the nation's air carriers were coming off record profits and growth in air transportation was at its highest in history. As part of the act, AIP funding was increased, on the order of 300 percent to many airports to allow for capital improvement projects designed to relieve the increased congestion and delays encountered at the nation's largest airports at the end of the 1990s.

The Aviation and Transportation Security Act of 2001

In response to the terrorist attacks involving the hijacking of four U.S. airliners used in suicide attack missions on Washington, D.C. and New York City, on September 11, 2001, The Aviation and Transportation Security Act (Public Law 107-071) was signed into law. This Act created the Transportation Security Administration (TSA), which took authority over aviation security and imposed a series of requirements for screening air carrier passengers and luggage including mandatory electronic inspection of all checked luggage. This has had profound effects on airport terminal planning and design. To fund these policies, the Act authorized a passenger surcharge of \$2.50 per flight segment and a fee imposed to air carriers equivalent to each

airline's costs of providing passenger security screening in the year 2000. These funds are collected by air carriers through ticket purchases and are used to fund the operation of the TSA and to contribute to airport development to accommodate enhanced security policies and procedures.

Vision 100 Century of Aviation Act of 2003

The AIR-21 Act authorized AIP funding through 2003 at which time reauthorization legislation was to occur. This reauthorization of AIP funding was accomplished with the Vision 100 Century of Aviation Act (Public Law 108-176) in December 2003. The purpose of the Vision 100 Act was to further increase, yet diversify, federal funding for airport and airspace improvements as the commercial air carrier industry recovers, and restructures, from the severe economic industry downturns following the September 11, 2001 attacks, and other economic and geopolitical issues. The act increased annual AIP authorizations to approximately \$3.4 billion in 2004, up to \$3.7 billion in 2007, the last year of the act's term and broadened the use of AIP and PFC funds to include airport improvements that have certain environmental benefits, investments to attract air service to underutilized airports, and to fund debt-service for projects previously funded through bond issuances.

NextGen Financing Reform Act of 2007/ FAA Reauthorization Act of 2009

The financial recovery the nation's airlines combined with increases in general aviation activity have begun to put increased strains on an aging air traffic control system and debates in Congress ensued regarding how to appropriately reauthorize funding for civil aviation as the terms of Vision 100 were due to expire in 2007. The Congress debate focused around a complete restructuring of the current funding programs, including major AIP reform. Rather than a system of funding airport and air traffic management through airline passenger, cargo, and fuel taxes, an aircraft-based user fee system was introduced to Congress for debate. This new legislation will be the first to implement fees directly on commercial and general aviation operations using the busiest areas of the national airspace system.

The NextGen Financing Reform Act focused funding on creating the Next Generation Air Traffic Management System to replace nearly 50-year-old air traffic control technology.

As of the end of 2007, the NextGen Financing Reform Act of 2007 had yet to be signed into law. Two versions of the act are being debated in Congress. The version supported by the House of Representatives (H.R. 2881) supports reauthorizing funding by increasing fuel taxes on general aviation fuel to between 24.1 and 35.9 cents per gallon, while maintaining the remaining tax structure implemented

in Vision 100. The version of the act in the Senate (S. 1300) proposes a user-based approach, including a per-flight surcharge of \$25 for "air traffic control costs," for general aviation jet and turbo-prop aircraft operating inside of controlled airspace.

The NextGen Financing Reform Act of 2007 was ultimately discarded with the new presidential administration in 2009. In March 2009, a new reauthorization bill, the FAA Reauthorization Act of 2009, was introduced as H.R. 915. As of publication of this text there is continued debate on how the federal government of the United States will fund the modernization of the national airspace system, particularly in the face of the economic downturn of beginning in late 2007. The focus of debate has continued to be around the potential implementation of user-based fees and other such taxes that may prohibit growth. The funding of the nation's aviation system continues to be a critical and highly debated topic.

State Roles in Aviation and Airports

State interest in aviation began as early as 1921, when the state of Oregon established an agency to handle matters concerning aviation. Since that time virtually all states have established aeronautical agencies either as commissions, departments, bureaus, boards, or divisions. Their responsibilities vary considerably and include channeling federal aid funds, planning state airport systems, providing state aid to local airport authorities, constructing and maintaining navigational equipment, investigating small aircraft accidents, enforcing safety regulations, and licensing airports.

Despite the growing concern of the states in airport development and aviation planning, their participation in the past has had little resemblance to the pattern associated with highway development. The states have always played a leading role in the development of roads and streets within their boundaries, whereas in airport development this has not been true. The reason for this pattern can best be explained by looking into the background of the entry of the states into aviation.

The majority of airports for civil aviation served by air carriers are municipally owned and operated. In a large number of states these airports were in operation prior to the formation of a state aeronautical agency. From its inception air transportation, because it is inherently of an interstate nature, became a matter of federal concern. The federal government provided much technical and financial assistance to municipalities. During World War II a great number of municipalities were the recipients of federal aid from the Civil Aeronautics Administration through the Defense Landing Area Program. Thus, in the early stages of airport development in this country, a fairly close relationship was established between the federal government

and the municipalities. This relationship was furthered when the Federal Airport Act of 1946 authorized the CAA to issue grants directly to municipalities, as long as such a procedure was not opposed to state policy. In the meantime, the majority of states were doing very little in the way of providing funds to municipalities for airports. While significant increases in state aid for airport development have occurred in the last several years, the amount of federal aid has been substantially higher.

States such as Alaska, Rhode Island, and Hawaii directly own and operate many of the airports within their respective boundaries. Other states support municipality-owned airports through state block grant funding programs. In those states where monetary aid is made available for airport development, the plans, specifications, and design for airport construction are generally reviewed by the state aeronautical agency.

There is no doubt that participation by the states in airport development is assuming a more significant role with the emergence of recent legislation in Congress to channel funds directly to state aeronautical agencies through block grant programs. The public concern for environmental control has resulted in legislation being passed at the state level, in addition to federal statutes, aimed at the control of aircraft noise and pollution. As general aviation and commuter activities continue to grow, the states will have to share the burden with the federal government in providing facilities for these activities, enforcing safety regulations, and other matters.

Aviation Organizations and Their Functions

The organizations directly involved in United States and international air carrier transportation and general aviation activity have an important influence on airport development as well as aircraft operations. These organizations and their functions can be classified into four groups, namely, federal agencies, state agencies, international government agencies, and industry or trade organizations.

Federal Agencies of the United States Government

There are several agencies at the federal level which dictate policy of direct and indirect effects on air transportation. The Federal Aviation Administration (FAA), the Transportation Security Administration (TSA), and the Environmental Protection Agency (EPA) are those agencies with the most direct influence on civil aviation policy, and airport planning and design.

Federal Aviation Administration

The Federal Aviation Administration is the agency within the U.S. Department of Transportation responsible for the safe and efficient

operation of the nation's civil aviation system. The FAA is headed by the chief executive known as the administrator who is appointed by the President. The FAA performs the following functions:

1. Encourages the establishment of civil airways, landing areas, and other air facilities
2. Designates federal airways and acquires, establishes, operates, and conducts research and development and maintains air navigation facilities along such civil airways
3. Makes provision for the control and protection of air traffic moving in air commerce
4. Undertakes or supervises technical development work in the field of aeronautics and the development of aeronautical facilities
5. Prescribes and enforces the civil air regulations for safety standards, including:
 - a. Effectuation of safety standards, rules, and regulations
 - b. Examination, inspection, or rating of airmen, aircraft engines, air navigation facilities, aircraft, and air agencies
 - c. Issuance of various types of safety certificates
6. Provides for aircraft registration
7. Requires notice and issues orders with respect to hazards to air commerce
8. Issues airport operating certificates to airports serving air carriers

The FAA develops, directs, and fosters the coordination of a national system of airports, the National Plan of Integrated Airport Systems [22], an aviation system capacity enhancement plan, the Aviation System Capacity Plan [9], a plan to modernize and significantly upgrade the air traffic control system, the National Airspace System Plan [21], and oversees funding for airports through the Airport Improvement Program and PFC Program. In this connection it performs the following functions:

1. Provides consultation and advisory assistance on airport planning, design, construction, management, operation, and maintenance to governmental, professional, industrial, and other individuals and agencies.
2. Develops and establishes standards, government planning methods and procedures; airport and seaplane base design and construction; and airport management, operation, and maintenance.
3. Collects and maintains an accurate record of all available airport facilities in the United States.

4. Directs, formulates, and keeps current a national plan (NPIAS) for the development of an adequate system of airports in cooperation with federal, state, and local agencies, and determines and recommends the extent to which portions or units of that system should be developed or improved.
5. Develops and recommends principles, for incorporation in state and local legislation, to permit or facilitate airport development, regulation, and protection of approaches through zoning or property acquisition.
6. Secures compliance with statutory and contractual requirements relative to airport operation practices, conditions, and arrangements.
7. Develops and recommends policies, requirements, and procedures governing the participation of states, municipalities, and other public agencies in federal-aid airport projects and secures adherence to such policies, requirements, and procedures.

As illustrated in Fig. 1-6, the FAA is organized into a number of offices within its headquarters in Washington, D.C. The offices most directly related to airport planning and development are the Office of Airport Planning and Programming, the Office of Airport Safety and Standards. In addition to Headquarters Offices, the FAA is divided into nine Airports Regional Offices, as illustrated in Fig. 1-7. Within these Regional Offices are Airports District Offices (ADOs). It's within these ADOs where specific consultation between the FAA and airport planners on airport planning and design programs are primarily discussed, analyzed, and ultimately approved.

The FAA publishes most of the regulations of concern to civil aviation and these—are found in Title 14—“Aeronautics and Space” of the United States Code of Federal Regulations. The Federal Aviation Regulations (FARs) are made up of more than 100 chapters, known as “parts,” regulating various aspects of the civil aviation system, including pilots, aircraft, the airspace system, and airports. The FARs of most concern to airport planning and design that will be further discussed in this text include

FAR Part 1: Definitions and Abbreviations

FAR Part 11: General Rule Making Procedures

PAR Part 36: Noise Standards: Aircraft Type and Airworthiness Certification

FAR Part 71: Designation of Class A, Class B, Class C, Class D, and Class E Airspace Areas, Airways, Routes, and Reporting Points

FAR Part 73: Special Use Airspace

FAR Part 77: Objects Affecting Navigable Airspace

FAR Part 91: General Operating and Flight Rules

As of October 23, 2006

Federal Aviation Administration

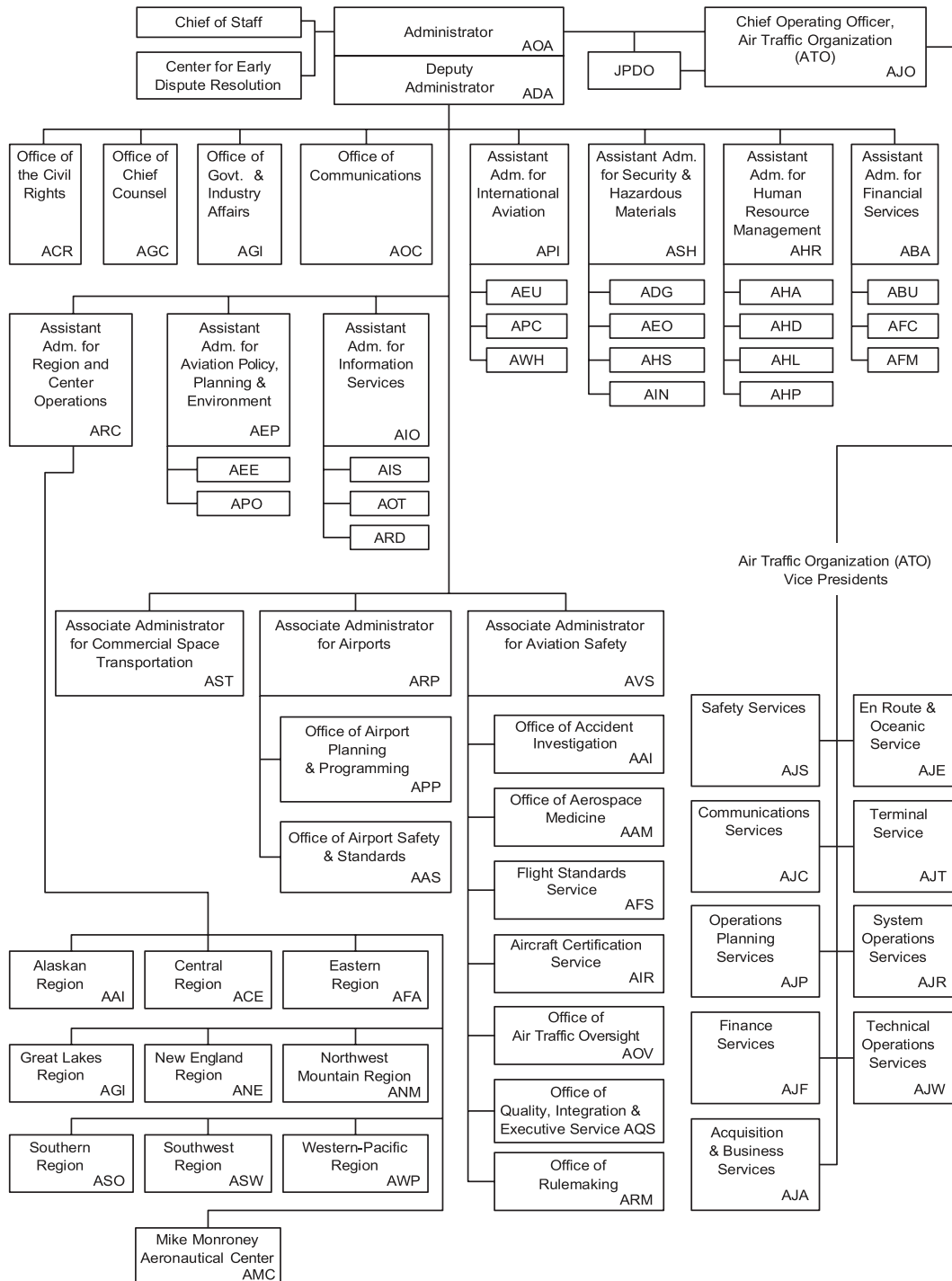


FIGURE 1-6 FAA headquarters organizational chart.

FAR Part 121: Operating Requirements: Domestic, Flag, and Supplemental Air Carrier Operations

FAR Part 139: Certification of Airports

FAR Part 150: Airport Noise Compatibility Planning

FAR Part 151: Federal Aid to Airports



FIGURE 1-7 FAA regions.

FAR Part 152: Airport Aid Program

FAR Part 156: State Block Grant Pilot Program

FAR Part 157: Notice of Construction, Alteration, Activation, and Deactivation of Airports

FAR Part 158: Passenger Facility Charges

FAR Part 161: Notice and Approval of Airport Noise and Access Restrictions

A complete list of FARs may be found at the Federal Aviation Administration website <http://www.faa.gov>.

In addition to federal regulations, the FAA publishes a series of Advisory Circulars (ACs) to provide guidance into the application of the regulations. The “150 series” of Advisory Circulars are focused on guiding airport managers and planners. There are more than 100 current and historical Advisory Circulars within the 150 series. Those of most direct application to airport planning and design include

AC 150/5020-1: Noise Control and Compatibility Planning for Airports

AC 150/5060-5: Airport Capacity and Delay

AC 150/5070-6B: Airport Master Plans

AC 150/5070-7: The Airport System Planning Process

AC 150/5300-13: Airport Design

AC 150/5325-4B: Runway Length Requirements for Airport Design

AC 150/5340-1J: Standards for Airport Markings

AC 150/5360-12D: Airport Signing and Graphics

AC 150/5360-13: Planning and Design Guidelines for Airport Terminal Facilities

AC 150/5360-14: Access to Airports by Individuals with Disabilities

Advisory Circulars are in a constant state of updating. The latest available ACs may be found at the FAA website at <http://www.faa.gov>.

Transportation Security Administration

The Transportation Security Administration (TSA) is the agency within the U.S. Department of Homeland Security responsible for the security nation's transportation systems, including civil aviation. The TSA was formed in 2001 in response to the terrorist attacks of September 11, 2001. In 2003, the TSA moved from the Department of Transportation to become part of the newly formed Department of Homeland Security. The TSA is led by an administrator appointed by the President.

With the formation of the TSA, all federal regulations pertaining to the security of the civil aviation system were moved from Title 14 of the Code of Federal Regulations to Title 49—Transportation, and have become commonly known as Transportation Security Regulations (TSRs). TSRs of specific importance to airport planners are those within Subsection C of the TSRs including

49 CFR Part 1500: Applicability, Terms, and Abbreviations

49 CFR Part 1510: Passenger Civil Aviation Security Service Fees

49 CFR Part 1540: Civil Aviation Security: General Rules

49 CFR Part 1542: Airport Security

49 CFR Part 1544: Aircraft Operator Security: Air Carriers and Commercial Operators

49 CFR Part 1546: Foreign Air Carrier Security

49 CFR Part 1550: Aircraft Security under General Operating and Flight Rules

Since 2001, the focus of the TSA has been on airport and commercial aviation security. In the future, it is expected that the TSA will further expand its regulatory role in air cargo and general aviation security, as well as into other modes of transportation, such as transit, rail, shipping, and the nation's highways. More information on the TSA may be found at its website at <http://www.tsa.gov>.

Environmental Protection Agency

Established in 1970 as part of the National Environmental Policy Act (NEPA), the Environmental Protection Agency (EPA) is responsible for preserving the environment with the goal of protecting human health.

The EPA has directed many of its efforts to minimizing environmental damage resulting from civil aviation activities, with focus on aircraft noise levels, emissions, air quality, and water runoff.

Most of the EPA requirements pertaining to civil aviation are incorporated into the FAA's Federal Aviation Regulations and policies regarding mandatory environmental impact evaluation of any proposed airport planning projects. More information on the EPA may be found at its website at <http://www.epa.gov>.

National Transportation Safety Board

The National Transportation Safety Board consists of five members appointed by the President. The NTSB performs the following functions:

1. Investigates certain aviation, highway, marine, pipeline, and railroad accidents, and reports publicly on the facts, conditions and circumstances, and the cause or probable cause of such accidents.
2. Recommends to Congress and federal, state, and local agencies measures to reduce the incidence of transportation accidents.
3. Initiates and conducts transportation safety studies and investigations.
4. Establishes procedures for reporting accidents to the board.
5. Assesses accident investigation techniques and issues recommendations for improving accident investigation procedures.
6. Evaluates the adequacy of the procedures and safeguards used for the transportation of hazardous materials.
7. Reviews, on appeal, the suspension, amendment, modification, revocation, or denial of certain operating certificates, documents, or licenses issued by the Federal Aviation Administration or the U.S. Coast Guard.

Information about the activities of the NTSB, including all records of civil aviation accidents and incidents, many of which impact airport planning and design, may be found on the NTSB website at <http://www.ntsbt.org>.

State Agencies

As mentioned earlier, the states are involved in varying degrees in the many aspects of aviation including airport financial assistance, flight safety, enforcement, aviation education, airport licensing, accident investigation, zoning, and environmental control. Because of the interstate nature of air transportation, the federal government has

preempted the legislative and administrative controls since the early days of aviation. However, for those aviation activities which occur wholly within the borders of a state, there have been formed regulatory agencies at the state level to oversee that these activities are operated in the best interests of the state.

Many state aviation agencies are participant members in the National Association of State Aviation Officials (NASAO). The mission of NASAO is to provide representation in Washington, D.C. on behalf of state aviation departments. Links to individual state aviation departments as well as a host of informational materials may be found on the NASAO website at <http://www.nasao.org>.

The International Civil Aviation Organization

Perhaps the most important international agency concerned with airport development is the International Civil Aviation Organization (ICAO), which is now a specialized agency of the United Nations with headquarters in Montreal, Canada. One hundred and eighty-eight nations were members of ICAO in 2009.

The ICAO concept was formed during a conference of 52 nations held in Chicago in 1944. This conference was by the invitation of the United States to consider matters of mutual interest in the field of air transportation. The objectives of ICAO as stated in its charter are to develop the principles and techniques of international air transportation so as to

1. Ensure the safe and orderly growth of international civil aviation throughout the world
2. Encourage the arts of aircraft design and operation for peaceful purposes
3. Encourage the development of airways, airports, and air navigation facilities for international aviation
4. Meet the needs of the peoples of the world for safe, regular, efficient, and economical air transport
5. Prevent economic waste by unreasonable competition
6. Ensure that the rights of contracting states are fully respected and that every contracting state has a fair opportunity to operate international airlines
7. Avoid discrimination between contracting states
8. Promote safety of flight in international air navigation
9. Promote generally the development of all aspects of international civil aeronautics

The ICAO has two governing bodies, the Assembly and the Council. The Council is a permanent body responsible to the Assembly

and is composed of representatives from 30 countries. The Council is the working group for the organization. It carries out the directives of the Assembly and discharges the duties and obligations specified in the ICAO charter.

To the airport planner and designer perhaps the most important document issued by ICAO is "Aerodromes," Annex 14 to the Convention on International Civil Aviation [1]. Annex 14 contains the international design standards and recommended practices which are applicable to nearly all airports serving international air commerce. In addition to Annex 14, ICAO publishes a great deal of technical and statistical information relative to international air transport [19] and is available on its website at <http://www.icao.org>.

Industry and Trade Organizations

There are many groups involved in the technical and promotional aspects of aviation. The following is a partial list of those groups which are primarily concerned with the airport aspects of aviation, most of these professional organizations serve as lobbying groups promoting the perspectives of the industry groups they represent.

1. *Aerospace Industries Association of America (AIA)*. The national trade association of companies in the United States engaged in research, development, and manufacture of aerospace systems.
2. *Aircraft Owners and Pilots Association (AOPA)*. An association of owners and pilots of general aviation aircraft. It is headquartered in Frederick, MD, a suburb of Washington, D.C.
3. *Air Line Pilots Association, International (ALPA)*. An association of airline pilots. It is headquartered in Herndon, VA, a suburb of Washington, D.C.
4. *Airports Council International (ACI)*. An association of over 400 large airports and airport authorities throughout the world. It is based in Geneva, Switzerland. The North American region of this organization (ACI-NA) is headquartered in Washington, D.C. Other regions include Europe, Africa, Asia/Pacific, and Latin America/Caribbean.
5. *Air Transport Association of America (ATA)*. An association of scheduled domestic and international airlines in the United States. The headquarters are in Washington, D.C.
6. *American Association of Airport Executives (AAAE)*. An association of the managers of public and private airports. It is located in Alexandria, VA, a suburb of Washington, D.C.
7. *General Aviation Manufacturers Association (GAMA)*. An association promoting the interests of general aviation. It is located in Washington, D.C.

8. *Helicopter Association International (HAI)*. An association which represents the interests of manufacturers and users of helicopters and promotes the use of helicopters. It is located in Alexandria, VA, a suburb of Washington, D.C.
9. *International Air Transport Association (IATA)*. An association of scheduled carriers in international air transportation. This organization is headquartered in Montreal, Canada.
10. *Regional Airline Association (RAA)*. An association of small regional and commuter aircraft operators promoting the needs of this segment of the air transportation industry. It was formerly called the Commuter Airline Association of America (CAAA). It is located in Washington, D.C.

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International Civil Aviation Organization: <http://www.icao.org>

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CHAPTER 2

Aircraft Characteristics Related to Airport Design

One of the great challenges for airport planning and design is creating facilities that accommodate a very wide variety of aircraft. Aircraft vary widely in terms of their physical dimensions and performance characteristics, whether they be operated for commercial air service, cargo, or general aviation activities.

There are a large number of specifications for which aircraft may be categorized. Depending on the portion of the area of the airport, certain aircraft specifications become more critical. For example, aircraft weight is important for determining the thickness and strengths of the runway, taxiway, and apron pavements, and affects the takeoff and landing runway length requirements at an airport, which in turn to a large extent influences planning of the entire airport property. The wingspan and the fuselage length influence the size of parking aprons, which in turn influences the configuration of the terminal buildings. Wingspan and turning radii dictate width of runways and taxiways, the distances between these traffic ways, and affects the required turning radius on pavement curves. An aircraft's passenger capacity has an important bearing on facilities within and adjacent to the terminal building.

Since the initial success of the Wright Flyer in 1903, fixed-wing aircraft have gone through more than 100 years of design enhancements, resulting in vastly improved performance, including the ability to fly at greater speeds and higher altitudes over larger ranges with more revenue generating carrying capacity (known as *payload*) at greater operating efficiencies. These improvements are primarily the results of the implementation of new technologies into aircraft

specifications, ranging from materials from which the airframes are built, to the engines that power the aircraft. Of great challenge to airport planning and design, historically has been to adapt the airport environment to accommodate changes in aircraft physical and performance specifications. For example:

- The introduction of “cabin-class” aircraft, such as the Douglas DC-3, in the mid-1930s resulted in the need for airports to construct longer, paved runways from the shorter grass strips that previously existed.
- The introduction of aircraft equipped with turbofan and turbojet engines in the late 1950s added requirements for longer and stronger runways, facilities to mitigate jet-blast, and policies to reduce the impact of aircraft noise at and around the airport.
- The introduction of “jumbo-jet” or “heavy” aircraft, such as the Boeing-747, in the late 1960s added new requirements for runway specifications, as well as terminal area design requirements for accommodating large volumes of passengers and cargo.
- The proliferation of regional jet aircraft, introduced because of more efficient engine technologies, resulted in the need for airports to modify many terminal areas that had accommodated larger jets or smaller turbo-prop aircraft.

Most recently, the introduction of the world’s largest passenger aircraft, the Airbus A-380, as well as the smallest of certified general aviation jet aircraft, continues to affect design specifications of airport airfield and terminal areas.

Table 2-1 provides a summary of some of the important aircraft characteristics of some of the aircraft that make up the world’s commercial airline fleet. Many regional airlines use smaller aircraft with less than 50 seats, while the world’s major airlines use very large aircraft, with potential configurations for more than 800 seats.

Table 2-2 provides a summary of important aircraft characteristics for common general aviation aircraft. While it should be noted that aircraft designed primarily for air carrier purposes are also often used for general aviation activity (e.g., the Boeing 737 is often configured for personal or business use and marketed as the Boeing Business Jet), most general aviation aircraft are smaller than typical commercial airline aircraft. Some of the aircraft listed in Table 2-2 are part of the fleet of “very light jets” that have emerged into the market since 2007.

Many of the values provided in Tables 2-1 and 2-2 are only approximate and tend to vary by specific model, as well as by each individual operation. For more precise values appropriate references, such

Turboprop Aircraft									
Aircraft	Wingspan	Length	MSTOW [†] (lb)	# Engines	Avg. # Seats	Runway Required (ft) [*]			
Beech 1900c	54'06"	57'10"	16,600	2	19	3,300			
Shorts 360	74'10"	70'10"	27,100	2	35	4,300			
Dornier 328-100	68'10"	68'08"	27,557	2	30	3,300			
SAAB 340B	70'04"	64'09"	28,500	2	37	4,200			
AT-42-300	80'06"	74'05"	36,815	2	45	3,600			
EMB 120	64'11"	65'7"	26,433	2	30	5,200			
Jet Aircraft Less than 100,000 lb MSTOW [†] (Regional Jets)									
Aircraft	Manufacturer	Wingspan	Length	MSTOW [†] (lb)	# Engines	Avg. # Seats	Runway Required (ft) [*]		
ERJ 135	Embraer	65'9"	86'5"	41,887	2	35	5,800		
ERJ 140	Embraer	65'9"	93'4"	44,313	2	40	6,100		
ERJ 145	Embraer	65'9"	98'0"	46,275	2	50	7,500		
CRJ 200	Bombardier	69'7"	87'10"	51,000	2	50	5,800		
CRJ 700	Bombardier	76'3"	106'8"	72,750	2	70	5,500		
CRJ 900	Bombardier	81'6"	119'4"	80,500	2	90	5,800		

TABLE 2-1 Characteristics of Commercial Service Aircraft

Jet Aircraft Less than 100,000 lb MSTOW [†] (Regional Jets)									
Aircraft	Manufacturer	Wingspan	Length	MSTOW [†] (lb)	# Engines	Avg. # Seats	Runway Required (ft) [*]		
BAe-RJ70	British	86'00"	78'9"	89,999	2	95	4,700		
BAe-RJ85	Aerospace								
	British	86'00"	86'11"	92,999	2	110	5,400		
Bae-RJ100	Aerospace								
	British	86'00"	94'10"	97,499	2	110	6,000		
	Aerospace								
Jet Aircraft between 100,000 and 250,000 lb MSTOW [†] (Narrow Body Jets)									
Aircraft	Manufacturer	Wingspan	Length	Wheel Base	Wheel Track	MSTOW [†] (lb)	# Engines	Avg. # Seats	Runway Required (ft) [*]
A-319	Airbus	111'25"	111'02"	41'33"	24'93"	141,095	2	140	5,800
MD-87	Industrie								
	McDonnell-	107'10"	130'05"	62'11"	16'08"	149,500	2	135	7,600
MD-90-30	Douglas								
	McDonnell-	107'10"	152'07"	77'02"	16'08"	156,000	2	165	6,800
	Douglas								
A-320-200	Airbus	111'03"	123'03"	41'05"	24'11"	158,730	2	160	5,700
	Industrie								
B-737-800	Boeing	112'06"	124'11"	50'09"	18'8"	172,445	2	175	
B-727-200	Boeing	108'00"	153'03"	63'03"	18'09"	184,800	3	165	8,600
B-757-200	Boeing	124'10"	155'03"	60'00"	24'00"	220,000	2	210	5,800

Jet Aircraft Greater than 250,000 lb MSTOW [†] (Wide Body Jets)									
A310-300	Airbus Industrie	144'00"	153'01"	49'11"	31'06"	330,690	2	240	7,500
B-767-300	Boeing	156'01"	180'03"	74'08"	30'06"	345,000	2	275	8,000
A-300-600	Airbus Industrie	147'01"	175'06"	61'01"	31'06"	363,765	2	310	7,600
L-1011-500	Lockheed	164'04"	164'03"	61'08"	36'00"	510,000	3	290	9,200
B-777-200	Boeing	199'11"	209'01"	84'11"	36'00"	535,000	2	375	8,700
DC-10-40	McDonnell-Douglas	165'04"	182'03"	72'05"	35'00"	555,000	3	325	9,500
A-340-200	Airbus Industrie	197'10"	195'00"	62'11"	16'09"	558,900	4	320	7,600
DC-10-30	McDonnell-Douglas	165'04"	182'03"	72'05"	35'00"	572,000	3	320	9,290
MD-11	McDonnell-Douglas	170'06"	201'04"	80'09"	35'00"	602,500	3	365	9,800
B-747SP	Boeing	195'08"	184'09"	67'04"	36'01"	630,000	4	315	7,000
B-747-400	Boeing	213'00"	231'10"	84'00"	36'01"	800,000	4	535	8,800

TABLE 2-1 Characteristics of Commercial Service Aircraft (*Continued*)

Jet Aircraft between 100,000 and 250,000 lb MSTOW [†] (Narrow Body Jets)									
Aircraft	Manufacturer	Wingspan	Length	Wheel Base	Wheel Track	MSTOW [†] (lb)	# Engines	Avg. # Seats	Runway Required (ft) [*]
B-787-8 Dreamliner	Boeing	197'04"18	'02"74'09	"32'07"242,	00 2	230			9,600
A-380	Airbus Industrie	261'08"23	'03"99'08	"46'11"1,23	,000 4	525			10,000

*Runway lengths are takeoff runway length estimates based on sea level elevation, temperature 20°C at maximum takeoff weight. It should be noted that required runway length varies considerably based on aircraft weight and local atmospheric conditions.

[†] MSTOW is maximum structural takeoff weight.

TABLE 2-1 Characteristics of Commercial Service Aircraft (*Continued*)

Piston and Turbo-Prop Engine Aircraft							
Aircraft	Manufacturer	Wingspan	Length	MSTOW (lb)	# Engines	Avg. # Seats	Runway Required*
PA28-Archer	Piper	35'00"23'0	"2,550	1		4	1,660
DA-40	Diamond	39'06"26'0	"2,645	1		4	1,198
PA28-Arrow	Piper	35'05"24'0	"2,750	1		4	1,525
C-182 Skylane	Cessna	35'10"28'0	"2,950	1		4	1,350
SR20-G2	Cirrus	35'07"26'0	"3,000	1		4	1,446
SR-22	Cirrus	38'04"26'0	"3,400	1		4	1,028
PA-32 Saratoga	Piper	36'02"27'0	"3,600	1		6	1,760
Corvalis 400	Cessna	36'01"25'0	"3,600	1		4	2,600
DA-42 Twin Star	Diamond	44'06"28'0	"3,748	2		4	1,130
C-310	Cessna	37'06"29'0	"5,500	2		6	1,790
BN2B-Islander	Britten-Norman	49'00"35'0	"6,600	2		9	1,155
C-402c	Cessna	44'01"36'0	"6,850		2	10	2,195
Cheyenne IIIA	Piper Aircraft	47'08"43'0	"11,200	2		10	2,400
Super KingAir	Beechcraft	54'06"43'0	"12,500	2		12	2,600
C-208 Grand Caravan	Cessna	52'01"41'0	"8,750	1		14	1,500

TABLE 2-2 Characteristics of General Aviation Aircraft

Very Light Jet Aircraft							
Aircraft	Manufacturer	Wingspan	Length	MSTOW (lb)	# Engines	Avg. # Seats	Runway Required*
Mustang	Cessna	43'2"40"	7"8,645	2		5	3,100
Eclipse 500	Eclipse	33'6"33'6"	5,995	2		5	2,400
Hondajet	Honda	39'10"41'8"	9,200	2		5	3,100
Business Jet Aircraft							
Citation CJ1	Cessna	46'11"42'7"	10,800	2		5	3,300
Citation X	Cessna	56'4"52'6"	36,400	2		10	3,560
Lear 45 XR	Bombardier	47'9"57'6"	21,500	2		9	5,040
Lear 60 XR	Bombardier	43'9"58'6"	23,500	2		9	3,400
Hawker 850 XP	Beechcraft	54'04"51'0"	28,000	2		8	5,200
G-IV	Gulfstream	77'10"88'0"	73,200	2		19	5,000
G-550	Gulfstream	93'06"96'0"	85,100	2		19	5,150

*Runway lengths are takeoff runway length estimates based on sea level elevation, temperature 20°C at maximum takeoff weight. It should be noted that required runway length varies considerably based on aircraft weight and local atmospheric conditions.

TABLE 2-2 Characteristics of General Aviation Aircraft (*Continued*)

as an airplane's characteristics and performance handbook, should be consulted. In particular, the runway length required to operate a particular aircraft, whether it be a takeoff or a landing, can vary considerably based on aircraft engine performance and total operating weight, as well as by the local environmental and atmospheric conditions. Calculation of required runway length is often performed prior to each operation as part of aircraft flight planning, often using tables, charts, or formulas provided by the aircraft manufacturer.

While there have certainly been recent breakthroughs in the introduction of very large aircraft such as the Airbus A-380, the overall trend in aircraft manufactured for civil air transport has focused design on efficiency, rather than the historical goals of increased size. More efficient aircraft may be smaller than older generation aircraft, but their increased efficiencies allow operators to focus on increasing service frequencies. This increase in operating efficiency has also shifted the focus of increasing aircraft speeds, at least in the realm of producing supersonic aircraft (i.e., those that travel at speeds greater than the speed of sound), to more efficient subsonic aircraft. As such production and operation of supersonic aircraft, such as the Concorde, was retired in the early part of the twenty-first century.

Dimensional Standards

Figure 2-1 illustrates some of the terms related to aircraft dimensions that are important to airport planning and design.

The *length* of an aircraft is defined as the distance from the front tip of the *fuselage*, or main body of the aircraft, to the back end of the tail section, known as the *empennage*. The length of an aircraft is used to determine the length of an aircraft's parking area, hangars. In addition for a commercial service airport, the length of the largest aircraft to perform at least five departures per day determines the required amount of aircraft rescue and firefighting equipment on the airfield.

The *wingspan* of an aircraft is defined as the distance from wingtip to wingtip of the aircraft's main wings. The wingspan of an aircraft is used to determine the width of aircraft parking areas and gate spacing, as well as determining the width and separations of runways and taxiways on the airfield.

The *maximum height* of an aircraft is typically defined as the distance from the ground to the top of the aircraft's tail section. Only in rare cases is an aircraft's maximum height found elsewhere on the aircraft, for example, the Airbus Beluga's maximum height is noted as the distance from the ground to the top of the forward fuselage entry door when it is fully extended upward in the open position.

The *wheelbase* of an aircraft is defined as the distance between the center of the aircraft's *main landing gear* and the center of its *nose gear*, or *tail-wheel*, in the case of a tail-wheel aircraft. An aircraft's *wheel track* is defined as the distance between the outer wheels of an aircraft's

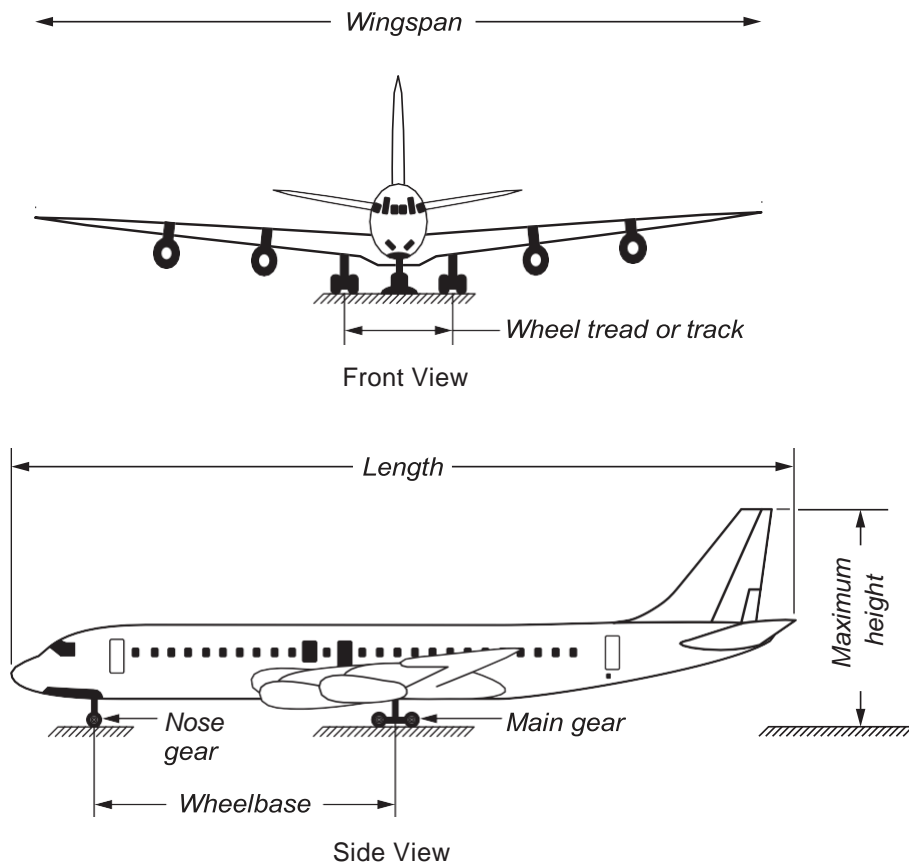


FIGURE 2-1 Aircraft dimensions.

main landing gear. The wheelbase and wheel track of an aircraft determine its *minimum turning radius*, which in turn plays a large role in the design of taxiway turnoffs, intersections, and other areas on an airfield which require an aircraft to turn.

Turning radii are a function of the nose gear steering angle. The larger the angle, the smaller the radii. From the center of rotation the distances to the various parts of the aircraft, such as the wingtips, the nose, or the tail, result in a number of radii. The largest radius is the most critical from the standpoint of clearance to buildings or adjacent aircraft. The minimum turning radius corresponds to the maximum nose gear steering angle specified by the aircraft manufacturer. The maximum angles vary from 60° to 80° , although for design purposes a steering angle of approximately 50° is often applied.

The turning radius of an aircraft may be expressed using the following formula:

$$R_{180^\circ \text{ turn}} = b \tan(90 - \beta) + t/2 \quad (2-1)$$

where b = wheelbase of an aircraft

t = wheel track of the aircraft

β = maximum steering angle

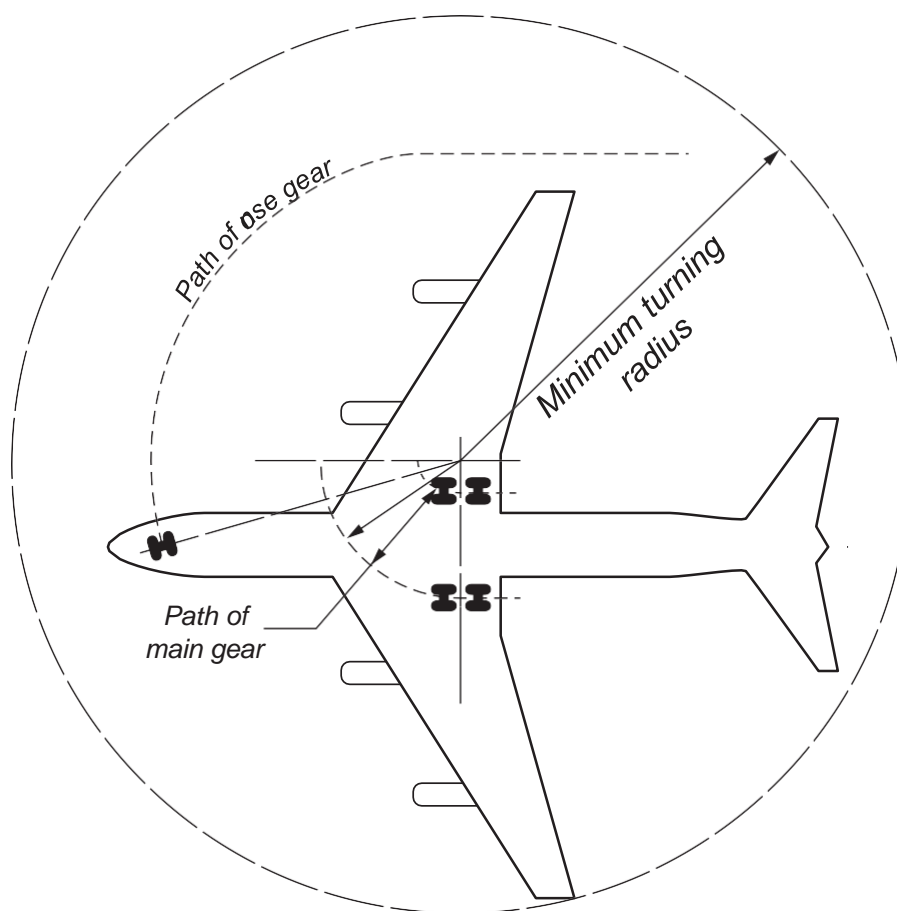


FIGURE 2-2 Turning radius.

The center of rotation can be easily determined by drawing a line through the axis of the nose gear at whatever steering angle is desired. The intersection of this line with a line drawn through the axes of the two main gears is the center of rotation. Some of the newer large aircraft have the capability of swiveling the main gear when making sharp turns. The effect of the swivel is to reduce the turning radius (Fig. 2-2). Minimum turning radii for some typical transport aircraft are given in Table 2-3.

Landing Gear Configurations

Aircraft currently operating in the world's civil use airports have been designed with various configurations of their landing gear. Most aircraft are designed with one of three basic landing gear configurations; the *single-wheel* configuration, defined as a main gear of having a total of two wheels, one on each strut, the *dual-wheel* configuration, defined as a main gear of having a total of four wheels, two on each strut, and the *dual-tandem* configuration, defined as two sets of wheels on each strut. These configurations are illustrated in Fig. 2-3.

Aircraft	Max. Steering Angle, deg	Radius, ft		
		Wingtips	Nose	Tail
MD-81/83/88	82	65.9	80.7	74.3
MD-90	82	66.5	85.5	74.6
B-737-800	78	69.4	65.4	73.6
B-727-200	78	71	79.5	80
A-320	70	72.2	60	71.9
B-757-200	65	92	84	91
A-310	65	98	75.6	94.9
A-300-600	65	104.9	87.7	108.4
B-767-200	65	112	85	98
B-747-200	70	113	110	125
B-747-SP	70	113	93	97
B-767-300	65	116.4	96.1	108.4
DC-10-30	68	118.1	105	100.8
MD-11	70	121.5	113.8	10.2
B-767-400	65	129.5	108.2	119.6
A-340	78	130.6	109.9	120.4
B-777-300	70	132	125	142
B-787-8	70	132	96.4	111
B-747-400	70	157	117	96

TABLE 2-3 Minimum Turning Radii for Typical Passenger Aircraft

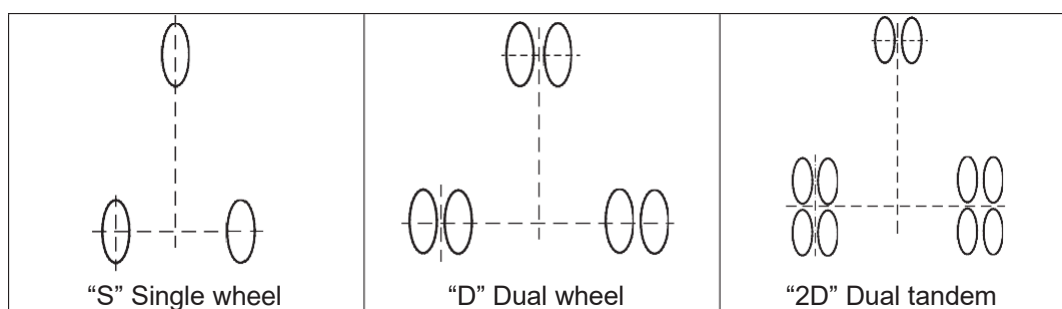


FIGURE 2-3 Traditional landing gear configurations (*Federal Aviation Administration*).

The landing configurations of the largest of commercial service aircraft have become more complex than the simple configurations illustrated in Fig. 2-3. For example, the Boeing 747, Boeing 777, and Airbus A-380 landing gear configurations are illustrated in Fig. 2-4.

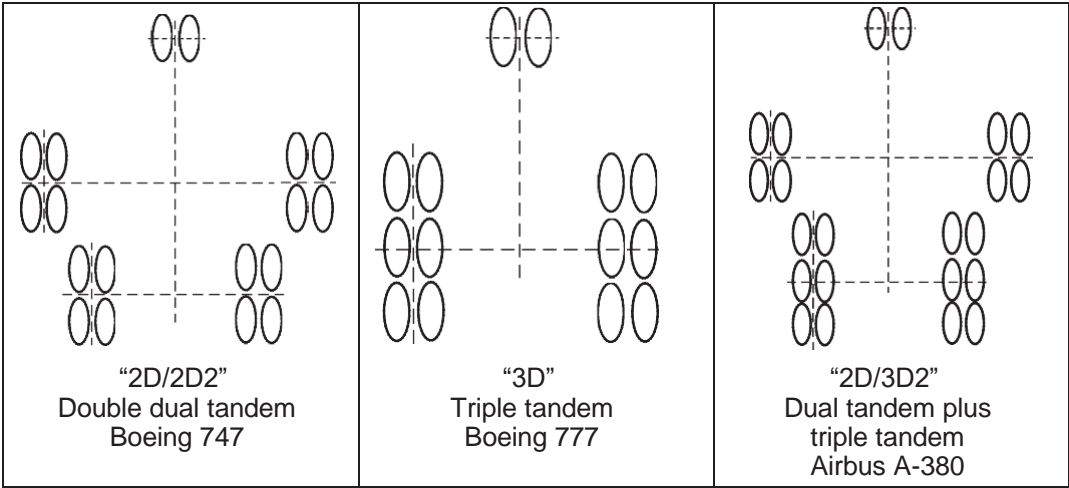


FIGURE 2-4 Complex landing gear configurations (*Federal Aviation Administration*).

The complexity of landing configurations prompted the FAA to adopt standard naming conventions for aircraft landing gear configurations [60]. Examples of this naming convention are represented in quotes in Figs. 2-3 and 2-4.

The landing gear configuration plays a critical role in distributing the weight of an aircraft on the ground it sits on, and thus in turn has a significant impact on the design of airfield pavements. Specifically, the more wheels on a landing gear, the heavier an aircraft can be and still be supported on a ramp, taxiway, or runway of a given pavement strength.

Aircraft Weight

While the concept of aircraft weight may be thought to be a simple one, the measurement of the weight of a given aircraft is actually relatively complex. An aircraft will in fact be measured with a certain number of weight measurements, depending on its level of loading with fuel, payload, and crew, and assigned maximum allowable weight values for takeoff, landing, and at rest.

These various measurements of aircraft weight are important to airport planning and design, in particular the facilities such as ramps, taxiways, and runways that are designed to support the aircraft.

While it is rare that any two aircraft, even those of the same model and configuration, have the same weight measurements (as there are almost always variations between aircraft in equipment, seating configurations, galleys, and other objects), most manufactures will assign typical weights to their aircraft for planning and design purposes. These weights are as follows.

The "lightest" measure of an aircraft's weight is known as the *operating empty weight* (OEW), the basic weight of the aircraft including crew and all the necessary gear required for flight but not including

payload and fuel. The OEW of an aircraft is considered for the design of aircraft that may occupy maintenance hangars, aircraft storage facilities, or any other areas that are not intended to support the weight of an aircraft when loaded with fuel or payload.

The *zero fuel weight* (ZFW) is the OEW of an aircraft plus the weight of its payload. The ZFW is the weight of the aircraft at which all additional weight must be fuel, so that when the aircraft is in flight, the bending moments at the junction of the wing and fuselage do not become excessive. The *payload* is a term which refers to the total revenue-producing load. This includes the weight of passengers and their baggage, mail, express, and cargo. The *maximum structural payload* is the maximum load which the aircraft is certified to carry, whether this load be passengers, cargo, or a combination of both. Theoretically, the maximum structural payload is a difference between the zero fuel weight and the operating empty weight. The maximum payload actually carried is usually less than the maximum structural payload because of space limitations. This is especially true for passenger aircraft, in which seats and other items consume a considerable amount of space.

The *maximum ramp weight* is the maximum weight authorized for ground maneuver including taxi and run-up fuel. As the aircraft taxis between the apron and the end of the runway, it burns fuel and consequently loses weight.

The *maximum gross takeoff weight* is the maximum weight authorized at brake release for takeoff. It excludes taxi and run-up fuel and includes the operating empty weight, trip and reserve fuel, and payload. The difference between the maximum structural takeoff weight and the maximum ramp weight is very nominal, only a few thousand pounds for the heaviest aircraft. The maximum gross landing weight actually varies with certain atmospheric conditions (namely, air density, which is a function of field elevation and ambient air temperature). This is due to the fact that at times of low air density (such as at high elevations and/or high temperatures), an airplane of a given weight may simply not have the engine power to get takeoff, while at the same weight it may be able to at a higher air density, found at lower elevations and/or lower air temperatures.

The *maximum structural takeoff weight* (MSTOW), is typically designed as the maximum gross takeoff weight for an aircraft operating at sea level elevation at a temperature of 59°F (15°C). It is also the maximum weight that the aircraft's landing gear can support. The MSTOW is the standard design weight measurement used in airport planning and design.

The *maximum structural landing weight* (MLW) is the structural capability of the aircraft in landing. The main gear is structurally designed to absorb the forces encountered during landing; the larger the forces, the heavier must be the gear. Normally the main gears of transport category aircraft are structurally designed for a landing at a

weight less than the maximum structural takeoff weight. This is so because an aircraft loses weight en route by burning fuel. This loss in weight is considerable if the journey is long, being in excess of 80,000 lb for large jet transports. It is therefore not economical to design the main gear of an aircraft to support the maximum structural takeoff weight during landing, since this situation will rarely occur. If it does occur, as in the case of aircraft malfunction just after takeoff, the pilot must jettison or burn off sufficient fuel prior to returning to the airport so as not to exceed the maximum landing weight. For short range aircraft, the main gear is designed to support, in a landing operation, a weight nearly equal to the maximum structural takeoff weight. This is so because the distances between stops are short, and therefore a large amount of fuel is not consumed between stops.

On landing, the weight of an aircraft is the sum of the operating weight empty, the payload, and the fuel reserve, assuming that the aircraft lands at its destination and is not diverted to an alternate airport. This landing weight cannot exceed the maximum structural landing weight of the aircraft. The takeoff weight is the sum of the landing weight and the trip fuel. This weight cannot exceed the maximum structural takeoff weight of the aircraft.

Engine Types

Perhaps the most significant contributor to increased aircraft performance has historically come from improvements in aircraft engine technology, from early twentieth-century piston engines to twenty-first-century high-performance jet engine technology.

While there are many makes and models of aircraft engines produced by a number of engine manufacturers, aircraft engine types can generally be placed into three categories, piston engines, turbo-props, and turbofan (or jet) engines.

The term *piston engine* applies to all propeller-driven aircraft powered by high-octane gasoline-fed reciprocating engines. Most small general aviation aircraft are powered by piston engines. The term *turboprop* refers to propeller-driven aircraft powered by turbine engines. The term *turbofan* or *jet* has reference to those aircraft which are not dependent on propellers for thrust, but which obtain the thrust directly from a turbine engine. Jet engines are typically powered using a form of diesel fuel, known as Jet-A. While historically jet engines have been used to power larger general aviation and commercial service aircraft, jet engines recently have been increasingly produced for smaller “regional jet” commercial service aircraft, and even smaller “very light jet” general aviation aircraft.

In the early part of the twenty-first century, most of the transport category aircraft in service are equipped with jet engines, and as such, much of the planning and design of airports serving commercial service and business general aviation are based around jet engine aircraft.

Jet engines can be classified into two general categories, turbojet and turbofan. A turbojet engine consists of a compressor, a combustion chamber, and a turbine at the rear of the engine. The early jet airline aircraft, particularly the Boeing 707 and the DC-8, were powered by turbojet engines, but these were discarded in favor of turbofan engines principally because the latter are far more economical.

A turbofan is essentially a turbojet engine to which has been added large-diameter blades, usually located in front of the compressor. These blades are normally referred to as the fan. A single row of blades is referred to as single stage, two rows of blades as multistage. In dealing with turbofan engines reference is made to the bypass ratio. This is the ratio of the mass airflow through the fan to the mass airflow through the core of the engine or the turbojet portion. In a turbofan engines the air flow through the core of the engine, the inner flow, is hot and very compressed and is burned in it. The air flow through the fan, the outer flow, is compressed much less and exits from the engine without burning into an annulus around the inner core. Fan engines are quieter than turbojet engines and the development of quiet propulsive integrated power plants in modern turbofan has included extensive acoustic lining development both in the inlet and the fan exhaust [38].

Most fans are installed in front of the main engine. A fan can be thought of as a small diameter propeller driven by the turbine of the main engine. Nearly all airline transport aircraft are now powered by turbofan. Current technological advances in engines are concentrated toward the development of *propfan* engines for short and medium haul aircraft and ultrahigh bypass ratio turbofan engines for long haul aircraft. These engine technologies reduce fuel consumption by 25 to 35 percent. These engines, which are variously termed *unducted fan* (UDF) engines and *ultrahigh bypass ratio* (UHB) turbofan engines, have brought on the emergence of very light jet aircraft.

Jet engine performance is made in measured both in terms of power and efficiency. The power of an aircraft engine is typically measured in pounds of forward moving force, or "thrust." Table 2-4 lists a sample of jet engines, and their measurements of thrust installed on historical and current transport category aircraft.

Aircraft engine power efficiency is measured in terms of the thrust-to-weight ratio, defined simply as the pounds of thrust provided by the engine, divided by the weight of the engine. Early jet engines were produced with thrust-to-weight ratios of approximately 3:1. In the early part of the twenty-first century, new light but powerful jet engines with thrust to weight ratios nearing 5:1 have significantly improved the operating efficiency of air transport aircraft and have made the emergence of the very light jet market feasible.

One important measure of engine performance efficiency is that of specific fuel consumption, expressed in terms of pounds of fuel per

Engine Family	Manufacturer	Max. Thrust (lb) Aircraft
PW610F	Pratt and Whitney	900 Eclipse 500
PW615F	Pratt and Whitney	1,350 Cessna Mustang
PW617F	Pratt and Whitney	1,700 Embraer Phenom 100
JT8D	Pratt and Whitney	21,000 DC-9, MD-80, SUPER 27
PW6000	Pratt and Whitney	24,000 A318
V2500	Pratt and Whitney	32,000 A-319, A-320, A-321, MD-90
PW2000	Pratt and Whitney	43,000 B-757, C -17, IL-96
JT9D	Pratt and Whitney	56,000 B-747, B-767, A-300, A-310, DC-10
PW4000-94	Pratt and Whitney	62,000 B-747-400, B767-200/300, MD-11, A-300, A-310
PW4000-100	Pratt and Whitney	69,000 A-300-200/300
GP7000	Pratt and Whitney	70,000 A-380
PW4000-112	Pratt and Whitney	98,000 B-777-200/300
RB211-535	Rolls-Royce	43,000 B-757-200/300, Tu-204
Trent 500	Rolls-Royce	56,000 A-340-500/600
RB211-524	Rolls-Royce	61,000 L-1011, B-747-200/400/400/SP/F, B-767-300
Trent 700	Rolls-Royce	71,000 A-330

TABLE 2-4 Turbojet Aircraft Engines

Engine Family	Manufacturer	Max. Thrust (lb) Aircraft
Trent 900	Rolls-Royce	76,000 A-380
Trent 800	Rolls-Royce	95,000 B-777-200/300
CT7	General Electric	2,100 Bell-214ST, Saab 340a
CF34	General Electric	20,000 CRJ-100-200/700/900, ARJ21, EMBRAER 170,175,190,195
CF6	General Electric	72,000 A-300, A-310, A-330
Genx	General Electric	75,000 8787, B-747-800
GE90	General Electric	115,000 B-777-200/ER/LR/300ER
CFM56-5B	GE/International Aerospace	33,000 A-318, A-319, A-320, A-321
CFM56-3	GE/International Aerospace	24,000 A-737-300/400/500
CFM56-2	GE/International Aerospace	24,000 B-707, KC-135
CFM56-7B	GE/International Aerospace	27,000 B-737-600/700/800/900, BBJ
CFM56-5A	GE/International Aerospace	27,000 A-319, A-320
CFM56-5C	GE/International Aerospace	34,000 A-340-200/300
V2500	International Aero	33,000 A-319, A-320, A-321, ACJ, MD-90

TABLE 2-4 Turbojet Aircraft Engines (*Continued*)

Aircraft	Engine	Bypass Ratio	Specific Fuel Consumption*
A340	CFM56-5C2	6.4	0.32
B-757	PW2037	6.0	0.33
A-330-300	CF6-80E1A2	5.1	0.33
A320	CFM56-5A1	6.0	0.33
B737-400/500	CFM56-3Ca	6.0	0.33
A-310	PW4152	4.9	0.348
B-767-200	CF6-80A2	4.7	0.35
B-747-400	PW4056	4.9	0.359
B-737-600	CFM56-7B20	5.5	0.36
A-321-200	V2533-A5	4.6	0.37
BA-146-300	LF507	5.6	0.406
MD-80	JT8D-219	1.8	0.519

*Specific Fuel Consumption is the amount of fuel required, in pounds, to create 1 lb of thrust.

TABLE 2-5 Performance Characteristics of Typical Jet Aircraft Engines

hour per pound of thrust. Fuel consumption of jet aircraft engines tends to be expressed in pounds rather than in gallons. This is because the volumetric expansion and contraction of fuel with changes in temperature can be misleading in the amount of fuel which is available. Each gallon of jet fuel weighs about 6.7 lb.

Specific fuel consumption for a particular type of aircraft, defined as the amount of fuel required (in pounds) to create 1 lb of thrust, is a function of its weight, altitude, and speed. Some typical values are given in Table 2-5 merely to illustrate the fuel economy of a turbofan engine particularly at high bypass ratios (a jet engine's *bypass ratio* is defined as the ratio between the mass flow rate of air drawn in by the fan but bypassing the engine core to the mass flow rate passing through the engine core). Significant gains in specific fuel consumption have been made with modern aircraft. Table 2-6 gives the approximate average consumption of fuel for typical aircraft.

Fuel consumption improvements in the last two decades have been significant. New engines, such as the CFM56, CF6, RB211-524D, and PW4000, as well as derivatives of current engines, have resulted in significant fuel economy gains.

An indication of the differences in fuel consumption attained by the various types of passenger aircraft in the different trip modes is given in Table 2-5. It should be pointed out, however, that the data are only indicators of fuel consumption and not productivity. Those aircraft which burn the higher rates of fuel generally are capable of greater speeds and have greater passenger capacity.

Aircraft	Engine	Fuel Consumption, lb/h	Fuel Consumption per Engine lb/h
EMB-145	AE3007A	2,253	1,127
A320-200	CFM56-5A3	4,054	2,027
A-319-100	CFM56-5A4	6,966	3,483
B-737-500	FM56-3B1R	7,879	3,940
B-737-200	JT8B-15A	8,829	4,415
B-757-200	RB211-535E4B	11,109	5,555
B-767-300	CF6-802C2B2F	11,893	5,947
A340-300	CFM-56-5C4	16,093	4,023
B-747-200	RB211-524D4	28,638	7,160

TABLE 2-6 Average Fuel Consumption of Typical Jet Aircraft

As observed in Fig. 2-5, the fuel consumption in gallons per available seat mile decreases with increasing route segment length. This ratio has become increasingly significant to aircraft operations as the price of fuel has increased dramatically in the early part of the twenty-first century. Most significantly for airport planning and

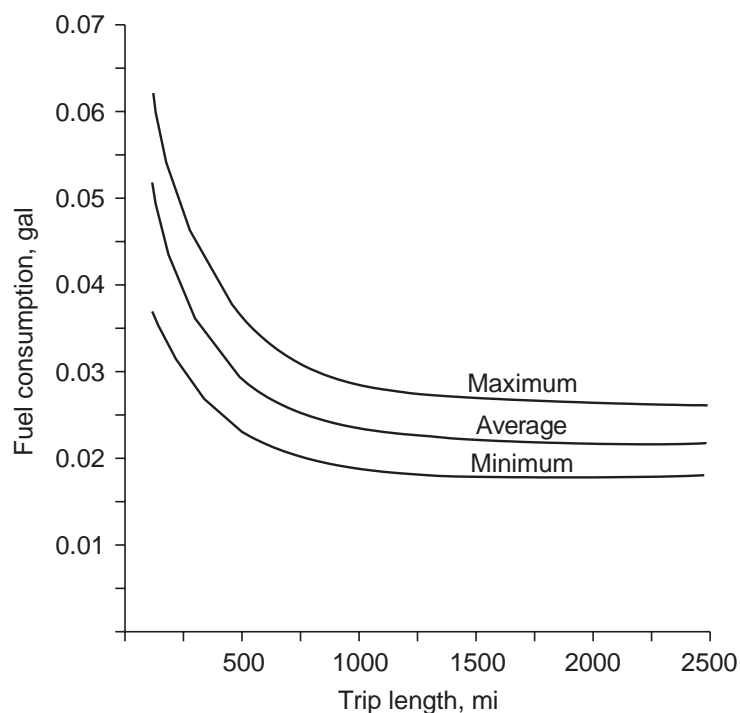


FIGURE 2-5 Fuel consumption in gallons per seat-mile as a function of route distance.

Air Pressure and Temperature

Since aircraft are designed to operate in the altitudes of the earth's atmosphere from sea level to nearly 50,000 ft above sea level, it is important to understand the characteristics of the atmosphere at these altitudes and how altitudes, as well as other atmospheric characteristics, affect aircraft performance.

The performance of all aircraft is affected significantly by the atmospheric conditions in which they operate. These conditions are constantly varying, based simply on the daily heating and cooling of the earth by the sun, and the associated winds and precipitation that occur.

In general, the performance of aircraft depends primarily on the density of the air through which it is operating. The greater the density of the air, the more air molecules flow over the wings, creating more lift, allowing the aircraft to fly. As air density decreases, aircraft require larger airspeed to maintain lift. For airport design, for example, this translates to longer runway length requirements when air is less dense. The density of the air is primarily a function of the air pressure, measured in English units as inches of mercury (inHg) and in metric units as millibars (mb) or hectopascals.

Air density is affected by air pressure and air temperature. As air pressure decreases, there are less air molecules per unit volume and thus air density decreases. As air temperature increases, the velocity and thus spacing between air molecules increases, thus reducing air density.

While these characteristics of the atmosphere vary from day to day and from place to place, for practical convenience for comparing the performance of aircraft, as well as for planning and design of airports, a *standard atmosphere* has been defined. A *standard atmosphere* represents the average conditions found in the actual atmosphere in a particular geographic region. Several different standard atmospheres are in use, but the one most commonly used is the one proposed by ICAO.

In the standard atmosphere it is assumed that from sea level to an altitude of about 36,000 ft, known as the *troposphere*, the temperature decreases linearly. Above 36,000 to about 65,000 ft, known as the *stratosphere*, the temperature remains constant; and above 65,000 ft, the temperature rises. Many conventional jet aircraft fly as high as 41,000 ft. The supersonic transports flew at altitudes on the order of 60,000 ft or more.

In the troposphere the standard atmosphere is defined as follows:

1. The temperature at sea level is 59°F or 15°C. This is known as the *standard temperature* at sea level.
2. The pressure at sea level is 29.92126 inHg or 1015 mb. This is known as the *standard pressure* at sea level.
3. The temperature gradient from sea level to the altitude at which the temperature becomes -69.7°F is 3.566°F per thousand feet. That is, for every increase in altitude of 1000 ft, the temperature decreases by approximately 3.5°F or 2°C.

Both standard pressure and standard temperature decrease with increasing altitude above sea level. The following relation establishes the standard pressure in the troposphere up to a temperature of -69.7°F .

$$\frac{P_0}{P} = \frac{T_0^{5.2561}}{T} \quad (2-2)$$

where P_0 = standard pressure at sea level (29.92 inHg)

P = standard pressure at a specified altitude

T_0 = standard temperature at sea level (59°F)

T = standard temperature at a specified altitude

In the above formula, the temperature is expressed in “absolute” or Rankine units. Absolute zero is equal to -459.7°F , 0°F is equal to 459.7°R , and 59°F is equal to 518.7°R .

Using these criteria, the standard temperature at an altitude of 5000 ft is 41.2°F , and the standard pressure is 24.90 inHg. Table 2-7 contains a partial listing of standard temperatures and pressures. It is common to refer to *standard conditions* or *standard day*. A standard

Altitude, ft	Temperature, $^{\circ}\text{F}$	Pressure, inHg	Speed of Sound, kn
0	59.0	29.92	661.2
1,000	55.4	28.86	658.9
2,000	51.9	27.82	656.6
3,000	48.3	26.82	654.3
4,000	44.7	25.84	652.0
5,000	41.2	24.90	649.7
6,000	37.6	23.98	647.7
7,000	34.0	23.09	645.1
8,000	30.5	22.23	642.7
9,000	26.9	21.39	640.4
10,000	23.3	20.58	638.0
20,000–12.2	16.89	626.2	
30,000–47.8	13.76	614.1	
40,000–69.7	8.90		589.2
50,000–69.7	7.06		576.3
60,000–69.7	6.41		573.3

TABLE 2-7 Table of Standard Atmospheres

condition is one in which the actual temperature and pressure correspond to the standard temperature and pressure at a particular altitude. When reference is made to the temperature being “above standard” it means that the temperature is higher than the standard temperature.

As aircraft takeoff performance data is typically related to the local barometric pressure and ambient air temperature, which in turn affects the density of the air, a defined value known as *density altitude* is often used to estimate the density of the air at any given time. Density altitude is a function of the effect of barometric pressure on air density, defined through the measurement known as pressure altitude, and the ambient temperature.

Assuming that at a standard day at sea level, where the elevation above sea level is effectively 0, the density altitude on a standard day would also be 0. If the barometric pressure was less than the standard pressure of 29.92 inHg, the pressure altitude would be greater than 0. Conversely, if the barometric pressure was greater than standard pressure, the pressure altitude would be less than 0. This relates to the fact that, when the atmospheric pressure drops, the air becomes less dense, requiring a longer run on the ground to obtain the same amount of lift as on a day when the pressure is high. Thus a reduction in atmospheric pressure at an airport has the same effect on its air density as if the airport had been moved to a higher elevation. *Pressure altitude* is defined as the altitude corresponding to the pressure of the standard atmosphere. Thus if the atmospheric pressure is 29.92 inHg, the pressure altitude is 0. If the pressure drops to 28.86 inHg, the pressure altitude is 1000 ft. This can be obtained from the formula relating pressure and temperature. If this lower pressure occurred at a sea level airport, the geographic altitude would be 0, but the pressure altitude would be 1000 ft. For airport planning purposes, it is satisfactory to assume that the geographic and pressure altitudes are equal unless the barometric pressures at a particular site are unusually low a great deal of the time.

Density altitude is defined as pressure altitude adjusted for temperature. Similar to the effect of barometric pressure on aircraft performance, if the temperature of the air was greater than standard temperature, the density of the air would be lower and the density altitude would increase, and if the temperature were lower than standard, the density altitude would decrease. It is because of the effect of both barometric pressure and ambient air temperature on aircraft performance that airports located at high elevations, where air pressure is generally lower than at sea level, and in locations where the ambient air temperature often rises well above 59°F, are airports constructed with longer runways, as longer runways are required for aircraft to reach needed airspeeds to get sufficient lift for takeoff, than at sea level elevations, or when temperatures are lower.

Wind Speed and Direction

Since aircraft depend on the velocity of air flowing over their wings to achieve lift, and fly through streams of moving air, similar to ships moving along water with currents, the direction and speed of wind, both near the surface of airports and at altitudes have great effect on aircraft performance.

As winds primarily affect the speed at which aircraft operate at an airport, it is important to understand the basic difference between two ways of measuring speed in an aircraft, *groundspeed* and *airspeed*. The groundspeed is the speed of the aircraft relative to the ground. True airspeed is the speed of an aircraft relative to the air flowing over the airfoil, or wing. For example, if an aircraft is flying at a groundspeed of 500 kn in air where the wind is blowing in the opposite direction, known as a *headwind*, at a speed of 100 kn, the true airspeed is 600 kn. Likewise, if the wind is blowing in the same direction, a *tailwind*, and the aircraft maintained a groundspeed of 500 kn, the true airspeed would be 400 kn.

On the airport surface, the speed and direction of winds directly affect aircraft runway utilization. For takeoff and landings, for example, aircraft perform best when operating with the wind blowing directly toward them, that is, with a direct *headwind*. Headwinds allow an aircraft to achieve lift at slower groundspeeds, and thus allow takeoffs and landings with slower groundspeeds and shorter runway lengths. While wind blowing from behind an aircraft, that is, a *tailwind* is preferable for aircraft flying at altitude, as they achieve greater groundspeeds at a given airspeed, it is not preferable for takeoff or landing, for precisely the same reason. As such, airports tend to plan and design runways so that aircraft may operate most often with direct headwinds, and orient their primary runways in the direction of the prevailing winds.

It is not very often the case that aircraft fly into a direct *headwind* or *tailwind*. Moreover, it is quite common for an aircraft to takeoff or land from an airport at such a time when the runways are not oriented directly into the existing wind. When this situation occurs, aircraft performance takes into consideration any effect of what are known as *crosswinds*.

While operating in direct headwind, tailwind, or calm conditions, the direction toward which an aircraft is pointing, or *heading*, is the same direction as the aircraft is actually traveling, or *tracking* over the ground. However, when operating with a crosswind, the aircraft heading is different than its track. A common analogy to this situation is the swimmer swimming across a river with a swift current. Even though such a swimmer may be pointing directly to the opposite shore of the river, he or she may end up farther downstream than simply straight across the river, and to end up directly across the river, the swimmer would have to point, or head, at some angle upstream.

Aircraft navigating a route at altitude operate in precisely the same manner. A heading is calculated, based on the speed and direction of the wind, and the speed of the aircraft itself, that will give the aircraft the desired track. The angle between the desired track and the calculated heading is known as the *crab angle*. The magnitude of this angle can be obtained from the following relation:

$$\sin x = \frac{V_c}{V_h} \quad (2-3)$$

where V_c is the crosswind in miles per hour or knots and V_h is the true airspeed in miles per hour or knots.

The crosswind, V_c , is defined as the component of the wind, V_w , that is at a right angle to the track. The angle x is referred to as the *crab angle*. It will be noted that the magnitude of the angle is directly proportional to the speed of the wind and inversely proportional to the speed of the aircraft.

As an aircraft approaches a runway, its heading (direction in which the nose is pointing) is of course also dependent on the strength of the wind traveling across the path of the aircraft (crosswind). The approach flight path to the runway is an extension of the centerline of the runway. An aircraft must fly along this track to safely reach the runway. The relation between track, heading, and crosswind is illustrated in Fig. 2-7. In order not to be blown laterally off the track by the wind, the aircraft must fly at an angle x from the track. This means that when the aircraft is moving slowly, as it does when it approaches a runway, and there is a strong crosswind, the angle x will be large. The term V_t is the true airspeed along the track and is equal to $V_h \cos x$. To obtain the groundspeed along the track, the component of the wind along the track must be subtracted from V_t . In the diagram the groundspeed along the track is equal to V_t minus the wind along the track, $V_w \sin x$. For example, assume that an aircraft was approaching a runway at a speed of 135 kn and the crosswind was 25 kn. The

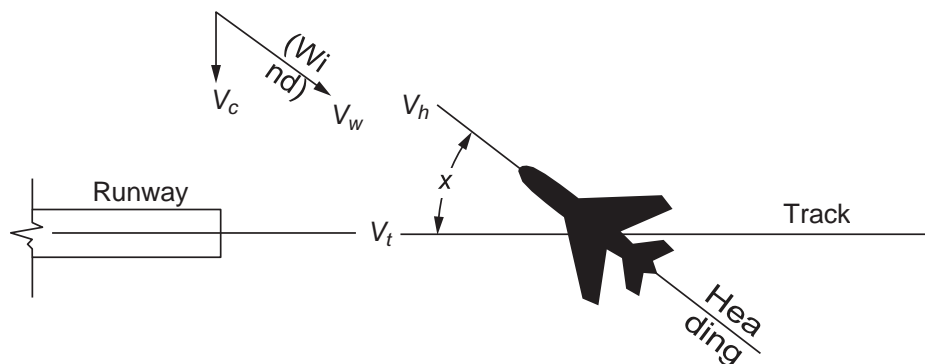


FIGURE 2-7 Crosswind correction.

crab angle x would be $10^\circ 10'$. This crab angle is reduced to 0 just prior to touchdown, so that the aircraft is appropriately pointed straight down the center of the runway.

While aircraft operators are trained to safely operate aircraft in these crosswind conditions, it is clearly desirable to minimize this occurrence. Furthermore, the physical ability of an aircraft to properly land in crosswind conditions is limited by the aircraft's weight, landing speed, and existing winds. Often times, small aircraft cannot safely land if crosswinds on a runway are too great. For this reason, airports accommodating smaller, slower aircraft are often designed with runways in several directions, to accommodate varying wind conditions. As opposed to the *primary* runways that are oriented into the prevailing winds, *crosswind* runways are oriented into the direction of winds occurring less frequently.

The FAA categorizes aircraft by the airspeeds at which they make approaches to land at an airport, known as the Aircraft Approach Category, and provides requirements to airports that runways be provided that allow for safe operation of the aircraft that use the airport for at least 95 percent of the annual wind conditions at the airport. The design process for estimating the number and orientation of primary, as well as crosswind runways based on the approach category of selected aircraft is detailed in Chap. 6 of this book.

Aircraft Performance Characteristics

Aircraft Speed

Reference is made to aircraft speed in several ways. Aircraft performance data is typically made reference two airspeeds, namely, *true airspeed* (TAS) and *indicated airspeed* (IAS). The pilot obtains his speed from an airspeed indicator. This indicator works by comparing the dynamic air pressure due to the forward motion of the aircraft with the static atmospheric pressure. As the forward speed is increased so does the dynamic pressure. The airspeed indicator works on the principle of the pitot tube. From physics it is known that the dynamic pressure is proportional both to the square of the speed and to the density of the air. The variation with the square of the speed is taken care of by the mechanism of the airspeed indicator, but not the variation in density. The indicator is sensitive to the product of the density of the air and the square of the velocity. At high altitudes the density becomes smaller and thus the indicated airspeed is less than the true airspeed.

If the true airspeed is required, it can be found with the aid of tables. As a very rough guide, one can add 2 percent to the indicated speed for each 1000 ft above sea level to obtain true airspeed.

The indicated airspeed is of more importance to the pilot than is the true airspeed. The concern is with the generation of lift, specifically

the stall speed, the speed at which there is not enough airflow over the wings to sustain lift, which is dependent on speed and air density. At high altitudes an aircraft will stall at a higher speed than it does at sea level. At higher altitudes, however, the airspeed indicator is indicating speeds lower than true speeds; consequently this is on the safe side and no corrections are necessary. Thus, an aircraft with a stalling speed of 90 kn will stall at the same indicated airspeed regardless of altitude. This is why aircraft manufacturers always report stalling speeds in terms of indicated airspeed rather than true airspeed. With the introduction of jet transports and high speed military aircraft, the reference datum for speed is often the speed of sound. The speed of sound is defined as *Mach 1* (after Ernst Mach, Austrian scientist). Thus Mach 3 means three times the speed of sound. Most of our current jet transports are *subsonic* (slower than the speed of sound) and cruise at a speed in the neighborhood of 0.8 to 0.9 Mach. Many military aircraft are *supersonic* (faster than the speed of sound). Again the reader is reminded that when the maximum speed of an aircraft is quoted as 0.9 Mach, this is in terms of true airspeed and not groundspeed. Such an aircraft can conceivably be traveling at a groundspeed higher than the speed of sound, depending on the magnitude of the tailwind.

The speed of sound is not a fixed speed; it depends on temperature and not on atmospheric pressure. As the temperature decreases, so does the speed of sound. The speed of sound at 32°F (0°C) is 742 mi/h (1090 ft/s), at -13°F (-25°C) it is 707 mi/h, and at 86°F (30°C) it is 785 mi/h. In fact, the speed of sound varies 2 ft/s for every change in temperature of 1°C above or below the speed at 0°C. The speed of sound at the altitudes at which jets normally fly is less than 700 mi/h, but at altitudes at which small aircraft normally fly (20,000 ft or less) it is greater than 700 mi/h.

The speed of sound may be computed from the formula

$$V_{sm} = 33.4T^{0.5} \quad (2-4)$$

$$V_{sf} = 49.04T^{0.5} \quad (2-5)$$

where V_{sm} = speed of sound in miles per hour at some temperature
 V_{sf} = speed of sound in feet per second at some temperature
 T = temperature in degrees Rankine

For convenience in navigation, aircraft distances and speeds are measured in nautical miles and knots, just like measurement on the high seas. One nautical mile (6080 ft) is practically equal to 1 min of arc of the earth's circumference. One knot is defined as 1 nmi/h. One nautical mile is approximately 1.15 land miles.

The performances of aircraft are, in part, defined by the various speeds at which they can safely liftoff, cruise, maneuver, and approach

to land. These speeds are defined in aircraft performance manuals as *V-speeds*. Such *V-speeds* include:

V_{ne} : Do-Not-Exceed Speed, the fastest an aircraft may cruise in smooth air to maintain safe structural integrity.

V_a : Design Maneuvering Speed, the recommended speed for an aircraft performing maneuvers (such as turns) or operating in turbulent air.

V_{lo} : Liftoff Speed, the recommended speed at which the aircraft can safely liftoff.

V_r : Rotate Speed, the recommended speed at which the nose wheel may be lifted off the runway during takeoff.

V_1 : Decision Speed, the speed at which, during a takeoff run, the pilot decides to continue with the takeoff, even if there might be an engine failure from this point before takeoff. If an aircraft develops an engine issue prior to reaching V_1 , the pilot will abort the takeoff.

V_{so} : Stall Speed (landing configuration), the minimum possible speed for an aircraft in landing configuration (landing gear down, flaps extended) to maintain lift. If the aircraft's airspeed goes below V_{so} , the airplane loses all lift and is said to *stall*. This speed is also typically the speed at which an aircraft will touch down on a runway during landing.

V_{ref} : Reference Landing Approach Speed, the speed at which an aircraft travels when on approach to landing. V_{ref} is typically calculated as $1.3 \times V_{so}$.

For airport planning and design, many of these speeds contribute to determining required runway lengths for takeoff and landing, as well as in determining the maximum number of operations (i.e., the capacity) that can be performed on runways over a given period of time.

Payload and Range

The maximum distance that an aircraft can fly, given a certain level of fuel in the tanks is known as the aircraft's *range*. There are a number of factors that influence the range of an aircraft, among the most important is payload. Normally as the range is increased the payload is decreased, a weight trade-off occurring between fuel to fly to the destination and the payload which can be carried.

The relationship between payload and range is illustrated in Fig. 2-8. The point A, the range at maximum payload, designates the farthest distance, R_a , that an aircraft can fly with a maximum structural payload. To fly a distance of R_a and carry a payload of P_a the aircraft has to take off at its maximum structural takeoff weight; however, its fuel tanks are not completely filled. Point B, the range at maximum

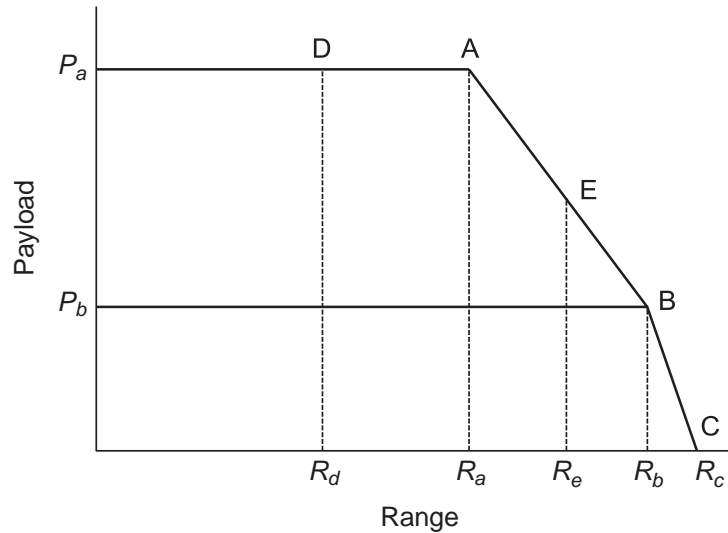


FIGURE 2-8 Typical relationship between payload and range.

fuel, represents the farthest distance, R_b , an aircraft can fly if its fuel tanks are completely filled at the start of the journey. The corresponding payload that can be carried is P_b . To travel the distance R_b , the aircraft must take off at its maximum structural takeoff weight. Therefore to extend the distance of travel from R_a to R_b the payload has to be reduced in favor of adding more fuel. Point C represents the maximum distance an aircraft can fly without any payload. Sometimes this is referred to as the *ferry range* and is used, if necessary, for delivery of aircraft. To travel this distance R_c , the maximum amount of fuel is necessary, but since there is no payload, the takeoff weight is less than maximum. In some cases the maximum structural landing weight may dictate how long an aircraft can fly with a maximum structural payload. If this is the case, the line DE represents the trade-off between payload and range which must occur since the payload is limited by the maximum structural landing weight. The shape of the payload versus range curve would then follow the line DEBC instead of ABC. Payload versus range depends on a number of factors such as meteorological conditions en route, flight altitude, speed, fuel, wind, and amount of reserve fuel. For performance comparison of different aircraft in an approximate way the payload range curves are usually shown for standard day, no wind, and long range cruise.

The actual payload, particularly in passenger aircraft, is normally less than the maximum structural payload even when the aircraft is completely full. This is due to the limitation in the use of space when passengers are carried. For computing payload, passengers and their baggage are normally considered as 200 lb units.

The aircraft manufacturers publish payload versus range diagrams in aircraft characteristics manuals for each aircraft which may be used for airport planning purposes. These diagrams are most

useful in airport planning for determining the most probable weight characteristics of aircraft flying particular stage lengths between airports.

The distribution of the load between the main gears and the nose gear depends on the type of aircraft and the location of the center of gravity of the aircraft. For any gross weight there is a maximum aft and forward center of gravity position to which the aircraft can be loaded for flight in order to maintain stability. Thus the distribution of weight between the nose and main gears is not a constant. For the design of pavements it is normally assumed that 5 percent of the weight is supported on the nose gear and the remainder on the main gears. Thus if there are two main gears, each gear supports 47.5 percent of the total weight. For example, if the takeoff weight of an aircraft is 300,000 lb, each main gear is assumed to support 135,000 lb. If the main gear has four tires, it is assumed that each tire supports an equal fraction of the weight on the gear, in this example, 33,750 lb. As will be discussed in Chap. 7, pavement strengths are designed based on the maximum structural takeoff weights, as well as the landing gear and loading configurations, of the aircraft of intended use.

Runway Performance

One of the most critical elements of aircraft performance is how such characteristics, along with local atmospheric conditions, affect the runway length for an aircraft to safely takeoff and land.

For any given operation, whether it be a takeoff or landing, an aircraft will require a certain amount of runway. Required runway length may vary widely for a specific aircraft, as a result of the aircraft's weight at the time of the operation, as well as the local atmospheric conditions. For the airport planner and designer, such variations have less direct impact on the design length of runways, and more to aircraft operators who must determine whether the length of a runway at a given time is safe for a particular operation. Nevertheless, the airport planner and designer should be aware of how an aircraft's performance characteristics specifically affect its runway length requirements.

The factors which have a bearing on an aircraft's runway length requirements for a given operation may be grouped into two general categories:

1. The physical capabilities of the aircraft under given environmental conditions
2. Requirements set by the government to protect for safe operations

An aircraft's performance capabilities and hence runway length requirements are often significantly affected by certain natural environmental conditions at the airport. The more important of these

conditions are temperature, surface wind, runway gradient, altitude of the airport, and condition of the runway surface.

Field Elevation

All other things being equal, the higher the field elevation of the airport, the less dense the atmosphere, requiring longer runway lengths for the aircraft to get to the appropriate groundspeed to achieve sufficient lift for takeoff. This increase is not linear but varies with the weight of the aircraft and with the ambient air temperature.

At higher altitudes the rate of increase is higher than at lower altitudes. For planning purposes, it can be estimated that between sea level and 5000 ft above sea level, runway lengths required for a given aircraft increases approximately 7 percent for every 1000 ft of increase in elevation, and greater under very hot temperatures those that experience very hot temperatures or are located at higher altitudes, the rate of increase can be as much as 10 percent. Thus, while an aircraft may require 5000 ft of runway to takeoff at an airport at sea level, the same aircraft may require 7500 ft or more at an airport 5000 ft above sea level, especially during periods of high temperatures.

Surface Wind

Wind speed and direction at an airport also have a significance on runway length requirements. Simply, the greater the headwind the shorter the runway length required, and the greater the tailwind the longer the runway required. Further, the presence of crosswinds will also increase the amount of runway required for takeoff and landing. From the perspective of the planner, it is often estimated that for every 5 kn of headwind, required runway length is reduced by approximately 3 percent and for every 7 kn of tailwind, runway length requirements increase by approximately 7 percent. For airport planning purposes runway lengths are often designed assuming calm wind conditions.

Runway Gradient

To accommodate natural topographic or other conditions, runways are often designed with some level of slope or *gradient*. As such, aircraft operating for takeoff on a runway with an uphill gradient requires more runway length than a level or downhill gradient, the specific amount depending on elevation of the airport and temperature. Conversely, landing aircraft require less runway length when landing on a runway with an uphill gradient, and more length for a downhill gradient.

Studies that have been made indicate that the relationship between uniform gradient and increase or decrease in runway length is nearly linear [55]. For turbine-powered aircraft this amounts to 7 to 10 percent for each 1 percent of uniform gradient. Airport design

criteria limit the gradient to a maximum of $1\frac{1}{2}$ percent. Information provided by aircraft manufacturers in flight manuals is based on uniform gradient, yet most runway profiles are not uniform. In the United States aircraft operators are allowed to substitute an average uniform gradient, which is a straight line joining the ends of the runway, as long as no intervening point along the actual path profile lies more than 5 ft above or below the average line. Fortunately most runways meet this requirement. For airport planning purposes only, the FAA uses an effective gradient. The effective gradient is defined as the difference in elevation between the highest and lowest points on the actual runway profile divided by the length of the runway. Studies indicate that within the degree of accuracy required for airport planning there is not very much difference between the use of the average uniform gradient and effective gradient.

Condition of Runway Surface

Slush or standing water on the runway has an undesirable effect on aircraft performance. Slush is equivalent to wet snow. It has a slippery texture which makes braking extremely poor. Being a fluid, it is displaced by tires rolling through it, causing a significant retarding force, especially on takeoff. The retarding forces can get so large that aircraft can no longer accelerate to takeoff speed. In the process slush is sprayed on the aircraft, which further increases the resisting forces on the vehicle and can cause damage to some parts. Considerable experimental work has been conducted by NASA and the FAA on the effect of standing water and slush. As a result of these tests, jet operations are limited to no more than $\frac{1}{2}$ in of slush or water. Between $\frac{1}{4}$ and $\frac{1}{2}$ in depth, the takeoff weight of an aircraft must be reduced substantially to overcome the retarding force of water or slush. It is therefore important to provide adequate drainage on the surface of the runway for removal of water and means for rapidly removing slush. Both water and slush result in a very poor coefficient of braking friction. When tires ride on the surface of the water or slush the phenomenon is known as hydroplaning. When the tires hydroplane, the coefficient of friction is on the order of wet ice and steering ability is completely lost. Hydroplaning is primarily a function of tire inflation pressure and to some extent the condition and type of grooves in the tires. According to tests made by NASA, the approximate speed at which hydroplaning develops may be determined by the following formula:

$$V_p = 10 p^{0.5} \quad (2-6)$$

where V_p is the speed in miles per hour at which hydroplaning develops and p is the tire inflation pressure in pounds per square inch.

The range of inflation pressures for commercial jet transports varies from 120 to over 200 lb/in². Therefore the hydroplaning speeds

would range from 110 to 140 mi/h or more. The landing speeds are in the same range. Therefore hydroplaning can be a hazard to jet operations. Hydroplaning can develop when the depth of water or slush is on the order of 0.2 in or less, the exact depth depending on tire tread design, condition of the tires, and the texture of the pavement surface. Smooth tread operating on a smooth pavement surface requires the least depth of fluid for hydroplaning.

To reduce the hazard of hydroplaning and to improve the coefficient of braking friction, runway pavements have been grooved in a transverse direction. The grooves form reservoirs for the water on the surface. The FAA is conducting extensive research to establish standards for groove dimensions and shape [54]. In the past the grooves were normally $\frac{1}{4}$ in wide and deep and spaced 1 in apart [44].

Declared Distances

Transport category aircraft are licensed and operated under the code of regulations known as the Federal Aviation Regulations (FAR). This code is promulgated by the federal government in coordination with industry. The regulations govern the aircraft gross weights at takeoff and landing by specifying performance requirements, known as *declared distances* which must be met in terms related to the runway lengths available. The regulations pertaining to turbine aircraft consider three general cases in establishing the length of a runway necessary for safe operations. These three cases are

1. A normal takeoff where all engines are available and sufficient runway is required to accommodate variations in liftoff techniques and the distinctive performance characteristics of these aircraft
2. Takeoff involving an engine failure, where sufficient runway is required to allow aircraft to continue the takeoff despite the loss of power, or else brake to a stop
3. Landing, where sufficient runway is required to allow for normal variation in landing technique, overshoots, poor approaches, and the like

The regulations pertaining to piston-engine aircraft retain in principal the above criteria, but the first criterion is not used. This particular regulation is aimed toward the everyday, normal takeoff maneuver, since engine failure occurs rather infrequently with turbine-powered aircraft. The runway length needed at an airport by a particular type and weight of turbine-powered aircraft is established by one of the foregoing three cases, whichever yields the longest length.

In the regulations for both piston-engine aircraft and turbine-powered aircraft, the word runway refers to *full-strength pavement* (FS).

Thus, in the discussion which follows, the terms runway and full-strength pavement are synonymous. In any discussion of the effect of the regulations on the length of the runway, however, it is important to note that the current regulations for turbine-powered aircraft do not require a runway for the entire takeoff distance, while the regulations for piston-engine aircraft normally do.

To indicate why there is a difference in the two regulations with regard to the length of full-strength pavement, it is necessary to examine in more detail the regulations pertaining to turbine-powered transports.

These three criteria as defined by the current turbine-powered transport regulations, FAR Part 25 [12] and Part 121 [23], are illustrated in Fig. 2-9.

Figure 2-9a illustrates the required *landing distance*. The regulations state that the *landing distance* (LD) required for an aircraft landing on a given runway must be sufficient to permit the aircraft to come to a full stop, *stop distance* (SD), within 60 percent of this distance, assuming that the pilot makes an approach at the proper speed and crosses the threshold of the runway at a height of 50 ft. The landing distance must be of full-strength pavement. The landing distance for piston-engine aircraft is defined in exactly the same manner.

Figure 2-9c, illustrates the runway length requirements for a normal takeoff with all engines fully operating, Fig. 2-9c defines a *takeoff distance* (TOD), which, for a specific weight of aircraft, must be 115 percent of the actual distance the aircraft uses to reach a height of 35 ft (D35). Not all of this distance has to be of full-strength pavement. What is necessary is that all this distance be free from obstructions to protect against an overshooting takeoff. Consequently the regulations permit the use of a *clearway* (CL) for part of this distance. A clearway is defined as a rectangular area beyond the runway not less than 500 ft wide and not longer than 1000 ft in length, centrally located about the extended centerline of the runway, and under the control of the airport authorities. The clearway is expressed in terms of a clearway plane, extending from the end of the runway with an upward slope not exceeding 1.25 percent above which no object nor any portion of the terrain protrudes, except that threshold lights may protrude above the plane if their height above the end of the runway is not greater than 26 in and if they are located to each side of the runway. Up to one-half the difference between 115 percent of the distance to reach the point of liftoff, *liftoff distance* (LOD), and the takeoff distance may be clearway. The remainder of the takeoff distance must be full-strength pavement and is identified as the *takeoff run* (TOR).

Figure 2-9b illustrates the engine-failure case, described as the case where one engine fails at a critical point during an aircraft takeoff roll, and the pilot makes an immediate judgmental decision whether or not to continue with a takeoff, or perform an emergency stop.

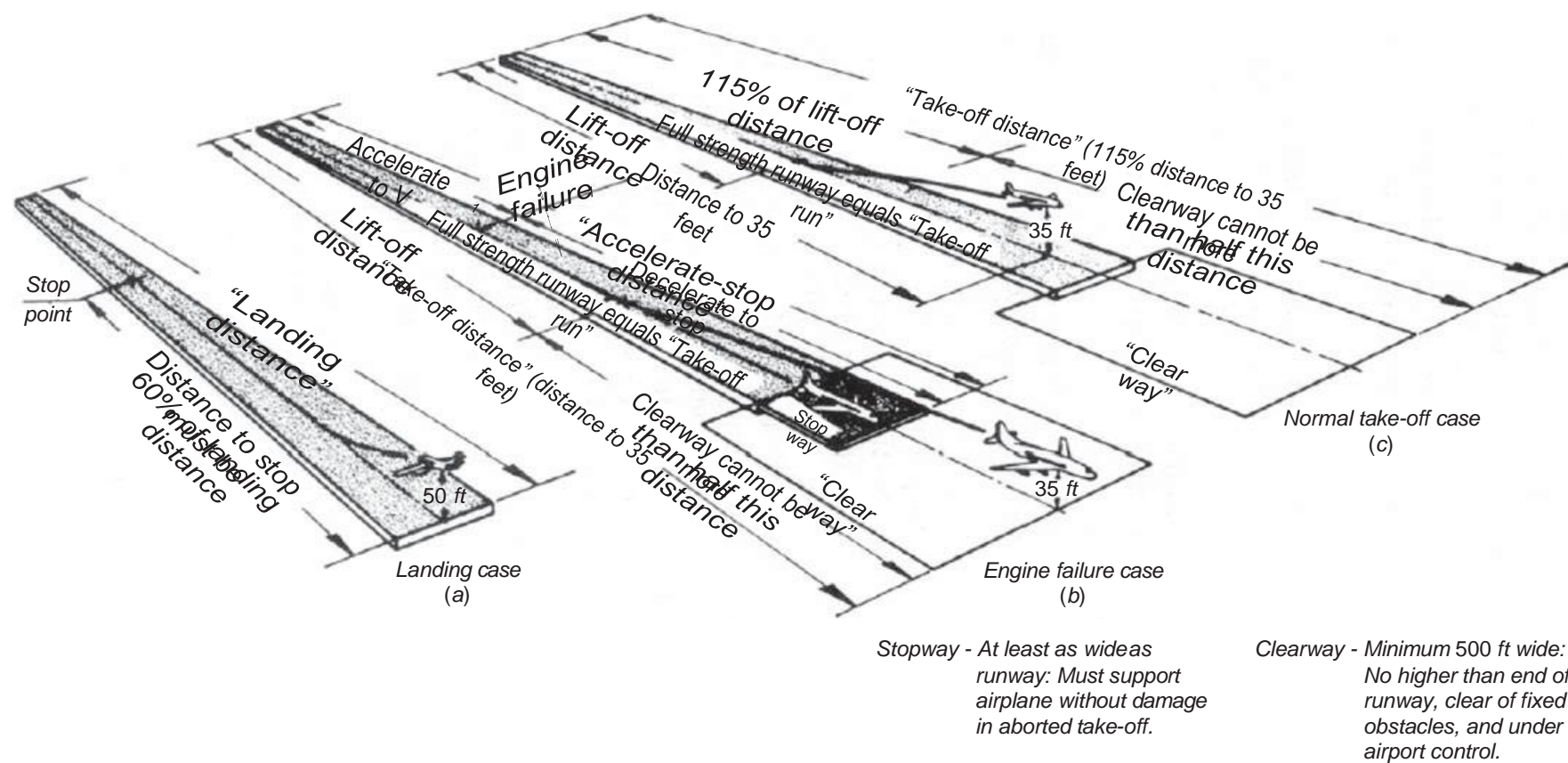


FIGURE 2-9 Declared distances, balanced field concept.

Federal regulations specify that the takeoff distance required during the engine-failure case is the actual distance to reach a height of 35 ft (D35) with no percentage applied as in the all-engine takeoff case. This recognizes the infrequency of occurrence of engine failure. The regulations again permit the use of a clearway, in this case up to one-half the difference between the liftoff distance and the takeoff distance, the remainder being full-strength pavement. The regulations for piston-engine aircraft normally require full-strength pavement for the entire takeoff distance.

The engine-failure case also requires that sufficient distance must also be available to stop the airplane rather than continue the takeoff. The speed at which engine failure is assumed to occur is selected by the aircraft manufacturer and is referred to as the *critical engine-failure speed* or *decision speed*, V_1 . If the engine fails at a speed greater than this speed, the pilot has no choice but to continue the takeoff. If an engine actually fails at or prior to this selected speed, the pilot brakes to a stop. This distance required, from beginning of the takeoff roll to the emergency stop is referred to as the *accelerate-stop distance* (DAS). For piston-engine aircraft only full-strength pavement is normally used for this purpose. The regulations for turbine-powered aircraft, however, recognize that an aborted takeoff is relatively rare and permit use of lesser strength pavement, known as *stopway* (SW), for that part of the accelerate-stop distance beyond the takeoff run. The stopway is defined as an area beyond the runway, not less in width than the width of the runway, centrally located about the extended centerline of the runway, and designated by the airport authorities for use in decelerating the aircraft during an aborted takeoff. To be considered as such, the stopway must be capable of supporting the airplane during an aborted takeoff without inducing structural damage to the aircraft. Engineered material arresting systems (EMAS) are being used as for this purpose with increasing frequency.

Based on the above requirements, aircraft operators estimate a required *field length* (FL) for each operation. The field length is generally made up of three components, namely, the full-strength pavement (FS), the partial strength pavement or stopway (SW), and the clearway (CL).

The preceding regulations for turbine-powered aircraft may be summarized for each of the cases in equation form to find the required field length.

Normal takeoff case:

$$FL_1 = FS_1 + CL_{1\max} \quad (2-7)$$

where

$$TOD_1 = 1.15(D35_1) \quad (2-7a)$$

$$CL_{1\max} = 0.50[TOD_1 - 1.15(LOD_1)] \quad (2-7b)$$

$$\text{TOR}_1 = \text{TOD}_1 - \text{CL}_{1\max} \quad (2-7c)$$

$$\text{FS}_1 = \text{TOR}_1 \quad (2-7d)$$

Engine-failure takeoff case:

$$\text{FL}_2 = \text{FS}_2 + \text{CL}_{2\max} \quad (2-8)$$

where

$$\text{TOD}_2 = \text{D35}_2 \quad (2-8a)$$

$$\text{CL}_{2\max} = 0.50(\text{TOD}_2 - \text{LOD}_2) \quad (2-8b)$$

$$\text{TOR}_2 = \text{TOD}_2 - \text{CL}_{2\max} \quad (2-8c)$$

$$\text{FS}_2 = \text{TOR}_2 \quad (2-8d)$$

Engine-failure aborted takeoff:

$$\text{FL}_3 = \text{FS} + \text{SW} \quad (2-9)$$

where

$$\text{FL}_3 = \text{DAS} \quad (2-9a)$$

Landing case:

$$\text{FL}_4 = \text{LD} \quad (2-10)$$

where

$$\text{LD} = \frac{\text{SD}}{0.60} \quad (2-10a)$$

$$\text{FS}_4 = \text{LD} \quad (2-10b)$$

To determine the required field length and the various components of length which are made up of full-strength pavement, stopway, and clearway, the above equations must each be solved for the critical design aircraft at the airport. This will result in finding each of the following values:

$$\text{FL} = \max [(\text{TOD}_1), (\text{TOD}_2), (\text{DAS}), (\text{LD})] \quad (2-11)$$

$$\text{FS} = \max [(\text{TOR}_1), (\text{TOR}_2), (\text{LD})] \quad (2-12)$$

$$\text{SW} = [(\text{DAS}) - \max (\text{TOR}_1, \text{TOR}_2, \text{LD})] \quad (2-13)$$

where SW_{\min} is zero.

$$CL = \min [(FL - DAS), (CL_{1\max}), (CL_{2\max})] \quad (2-14)$$

where CL_{\min} is zero and CL_{\max} is 1000 ft.

If operations are to take place on the runway in both directions, as is the usual case, the field length components must exist in each direction. Example Problem 2-1 illustrates the application of these requirements for a hypothetical aircraft.

Example Problem 2-1 Determine the runway length requirements according to the specifications of FAR 25 and FAR 121 for a turbine-powered aircraft with the following performance characteristics:

Normal takeoff:

Liftoff distance = 7000 ft

Distance to height of 35 ft = 8000 ft

Engine failure:

Liftoff distance = 8200 ft

Distance to height of 35 ft = 9100 ft

Engine-failure aborted takeoff:

Accelerate-stop distance = 9500 ft

Normal landing:

Stop distance = 5000 ft

From Eq. (2-4) for a normal takeoff

$$TOD_1 = 1.15 D_{35_1} = (1.15)(8000) = 9200 \text{ ft}$$

$$CL_{1\max} = 0.50[TOD_1 - 1.15(LOD_1)] = (0.50)[9200 - 1.15(7000)] = 575 \text{ ft}$$

$$TOR_1 = TOD_1 - CL_{1\max} = 9200 - 575 = 8625 \text{ ft}$$

From Eq. (2-5) for an engine-failure takeoff

$$TOD_2 = D_{35_2} = 9100 \text{ ft}$$

$$CL_{2\max} = 0.50(TOD_2 - LOD_2) = 0.50(9100 - 8200) = 450 \text{ ft}$$

$$TOR_2 = TOD_2 - CL_{2\max} = 9100 - 450 = 8650 \text{ ft}$$

From Eq. (2-6) for an engine-failure aborted takeoff $DAS = 9500$ ft

From Eq. (2-7) for a normal landing

$$LD = \frac{SD}{0.60} = \frac{5000}{0.60} = 8333 \text{ ft}$$

Using the above quantities in Eqs. (2-8) through (2-11), the actual runway component requirements become

$$FL = \max [(TOD_1), (TOD_2), (DAS), (LD)]$$

$$= \max [(9200), (9100), (9500), (8333)] = 9500 \text{ ft}$$

$$FS = \max [(TOR_1), (TOR_2), (LD)]$$

$$= \max [(8625), (8650), (8333)] = 8650 \text{ ft}$$

$$\begin{aligned}
SW &= [(DAS) - \max (TOR_1, TOR_2, LD)] \\
&= (9500) - \max [(8625), (8650), (8333)] = (9500 - 8650) = 850 \text{ ft} \\
CL &= \min [(FL - DAS), CL_{1\max}, CL_{2\max}] \\
&= \min [(9500 - 9500), 575, 450] = 0 \text{ ft}
\end{aligned}$$

The above regulations, as illustrated in Example Problem 2-1, are applied at all airports, in the form of *declared distances* for each runway [1, 9]. Declared distances are the distances that are declared available and suitable for satisfying the takeoff run, takeoff distance, accelerate-stop distance and landing distance requirements of aircraft. Four declared distances are commonly reported for each runway. They are the takeoff run available (TORA), takeoff distance available (TODA), accelerate-stop distance available (ASDA), and landing distance available (LDA).

The *takeoff run available* (TORA) is the runway length declared available and suitable for the ground run of an aircraft during takeoff. For Example Problem 2-1, the TORA would be 8650 ft. The *takeoff distance available* (TODA) is the takeoff run available plus the length of any remaining runway and clearway beyond the far end of the takeoff run available. For Example Problem 2-1, the TODA would be 9500 ft. The *accelerate-stop distance available* (ASDA) is the amount of runway plus stopway declared available and suitable for the acceleration and deceleration of an aircraft during an aborted takeoff. For Example Problem 2-1, the ASDA would also be 9500 ft. The *landing distance available* (LDA) is the runway length available and suitable for landing an aircraft. For Example Problem 2-1, the LDA would be 8650 ft.

It is apparent that both the takeoff distance and accelerate-stop distance will depend on the speed the aircraft has achieved when an engine fails.

Since, for piston-engine aircraft, full-strength pavement was normally used for the entire accelerate-stop distance and the takeoff distance, it was the general practice to select V_1 , so that the distance required to stop from the point where V_1 was reached was equal to the distance (from the same point) to reach a specified height above the runway. The runway length established on this basis is referred to as the balanced field concept or balanced runway and results in the shortest runway. For turbine-powered aircraft, the selection of V_1 on this basis will not necessarily result in the shortest runway if a clearway or a stopway is provided.

From an airport planning perspective, it is not typical to design a runway's full-strength pavement, stopway, and clearway based on a given aircraft. Rather, for each individual aircraft operation, a V_1 speed is selected which best accommodates the runway on which it will be operating.

For example, for an aircraft operating on a relatively short runway, a lower V_1 may be selected, which will allow for a shorter accelerate-stop distance, but would require at least some clearway to allow for the aircraft to safely climb out to 35 ft. Conversely, for relatively long runways that may have obstacles near the runway's end, or for runways with less full-strength pavement but a stopway at the runway end, a higher V_1 may be selected, to allow for steeper climb-out under engine-failure conditions, and the ability to accommodate a longer accelerate-stop distance.

Thus, one can see that the regulations pertaining to turbine-powered aircraft offer a number of alternatives to the aircraft operator. It should be emphasized that the takeoff distance and the takeoff run for the engine-failure case must be compared with the corresponding distance for the normal all engine takeoff case. The longer distance always governs. A further discussion of these concepts is presented by ICAO [1].

Both aircraft operators and airport planners are interested in clearways, because clearways will, for a fixed available length of runway, allow the operator additional gross takeoff weight with less expense to airport management than building full-strength pavement would require.

Wingtip Vortices

Whenever the wings lift an aircraft, vortices form near the ends of the wings. The vortices are made up of two counter-rotating cylindrical air masses about a wingspan apart, extending aft along the flight path. The velocity of the wind within these cylinders can be hazardous to other aircraft encountering them in flight. This is particularly true if a lighter aircraft encounters a vortex generated by a much heavier aircraft. The tangential velocities in a vortex are directly proportional to the weight of the aircraft and inversely proportional to the speed. The more intense vortices are therefore generated when the aircraft is flying slowly near an airport [52]. The winds created by vortices are often referred to as wake turbulence or wake vortex.

Once vortices are generated they move downward and drift laterally in the direction of the wind. The rate at which vortices settle toward the ground is dependent to some extent on the weight of an aircraft, the heavier the vehicle the faster the vortex will settle. About one wingspan height above the ground the vortices begin to move laterally away from the aircraft, as shown in Fig. 2-10. The duration of a vortex is dependent to a great extent on the velocity of the wind. When there is very little or no wind they can persist for longer than 2 min. As a result of these tests, the FAA and ICAO divide aircraft into three classes for the purposes wake-turbulence separation minima.

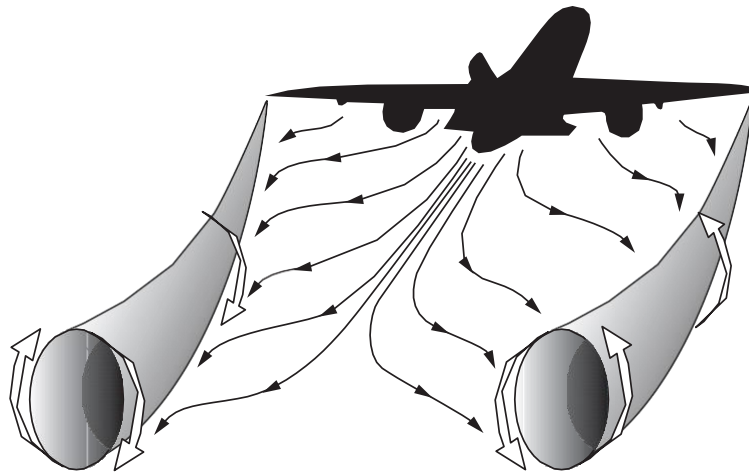


FIGURE 2-10 An illustration of wake turbulence.

FAA Wake Turbulence Classifications by Aircraft Weight (MSTOW)		ICAO Wake Turbulence Classifications by Aircraft Weight (MSTOM)	
Category	Weight	Category	Weight
Small	≤41,000 lb	Light	≤7,000 kg
Large	41,000–255,000 lb	Medium	7,000–136,000 kg
Heavy	>255,000 lb	Heavy	>136,000 kg

TABLE 2-8 FAA and ICAO Wake Turbulence Classification

For airport planning and design, as well as air traffic safety purposes, aircraft have been categorized into wake-turbulence classifications, based primarily their maximum structural takeoff weights, as illustrated in Table 2-8. Operating aircraft of varying wake-turbulence classifications in the same vicinity has significant effects on the safe and efficient operation of an airfield.

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UNIT – II – AIRPORT PLANNING – SAE1605

CHAPTER 4

Airport Planning Studies

Introduction

The planning of an airport is such a complex process that the analysis of one activity without regard to the effect on other activities will not provide acceptable solutions. An airport encompasses a wide range of activities which have different and often conflicting requirements. Yet they are interdependent so that a single activity may limit the capacity of the entire complex. In the past airport master plans were developed on the basis of local aviation needs. In more recent times these plans have been integrated into an airport system plan which assessed not only the needs at a specific airport site but also the over-all needs of the system of airports which service an area, region, state, or country. If future airport planning efforts are to be successful, they must be founded on guidelines established on the basis of comprehensive airport system and master plans.

The elements of a large airport are shown in Fig. 4-1. It is divided into two major components, the airside and the landside. The aircraft gates at the terminal buildings form the division between the two components. Within the system, the characteristics of the vehicles, both ground and air, have a large influence on planning. The passenger and shipper of goods are interested primarily in the overall door-to-door travel time and not just the duration of the air journey. For this reason access to airports is an essential consideration in planning.

The problems resulting from the incorporation of airport operations into the web of metropolitan life are complex. In the early days of air transport, airports were located at a distance from the city, where inexpensive land and a limited number of obstructions permitted flexibility in airport operations. Because of the nature of aircraft and the infrequency of flights, noise was not a problem to the

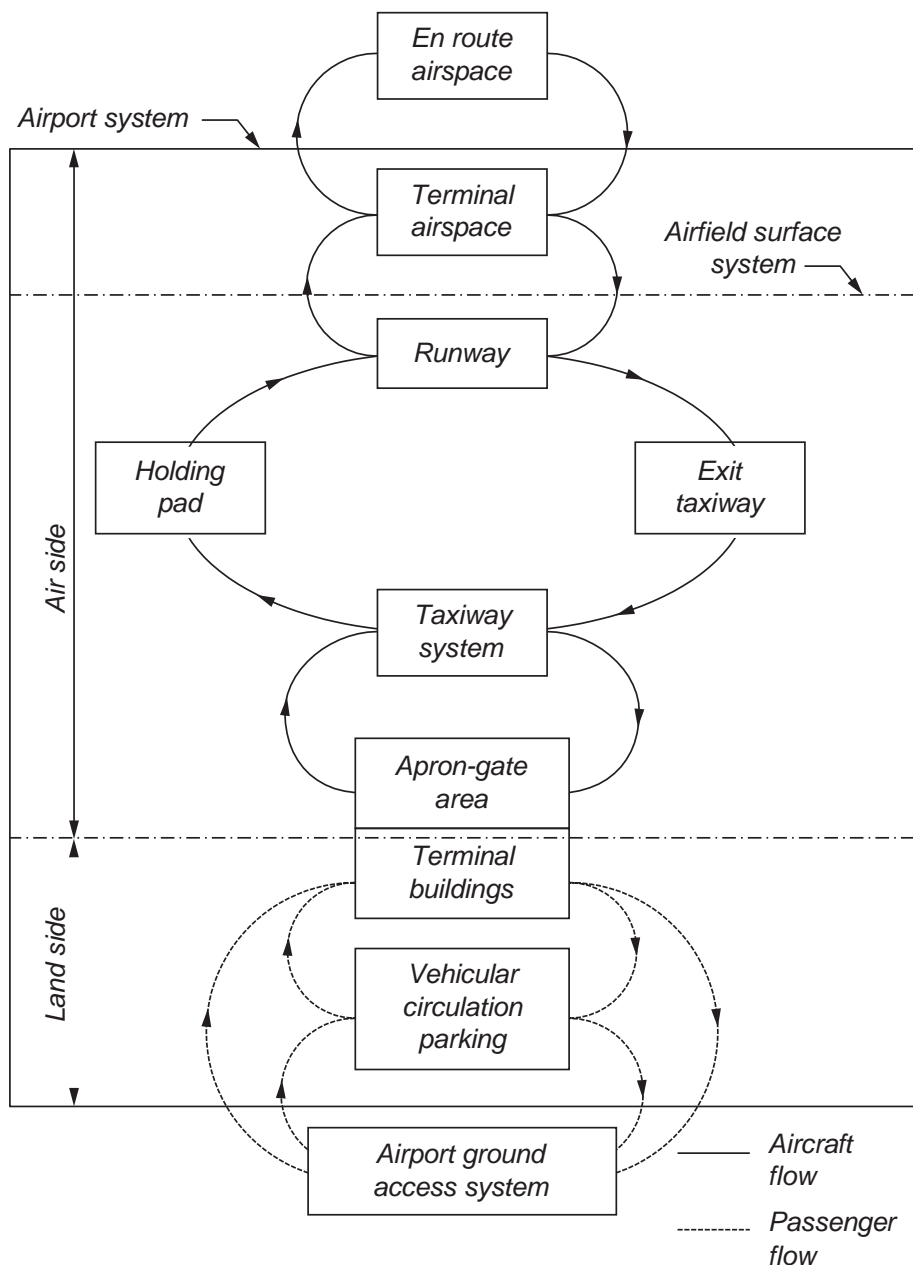


FIGURE 4-1 Components of the airport system for a large airport.

community. In many cases the arrival and departure of passenger and cargo planes was often a source of local pride. In addition, low population density in the vicinity of the airport and light air traffic prevented occasional accidents from alarming the community. In spite of early lawsuits, the relationship between airport and community was relatively free of strife resulting from problems of nuisance or hazard.

Airport operations have been increasingly hampered by obstructions resulting from industrial development related to the airport and from industry attracted by adjacent inexpensive land and access to the transportation afforded by the airfield and its associated highways.

While increasingly dense residential development has resulted from this economic stimulation, one must not overlook the effects of the unprecedented suburban spread during the post-World War II era, resulting from the backlog of housing needs and a period of economic prosperity.

Radical developments in the nature of air transport have produced new problems. The phenomenal growth of air traffic has increased the probability of unfavorable community reaction, but developments in the aircraft themselves have had the most profound effect on airport community relations. The greater size and speed of aircraft have resulted in increases in approach and runway requirements, while increases in the output of power plants have brought increases in noise. Faced with these problems the airport must cope with the problems of securing sufficient airspace for access to the airport, sufficient land for ground operations, and, at the same time, adequate access to the metropolitan area.

Types of Studies

Many different types of studies are performed in airport planning. These include studies related to facility planning, financial planning, traffic and markets, economics, and the environment. However, each of these studies can usually be classified as being performed at one of three levels: the system planning level, the master planning level, or the project planning level.

The Airport System Plan

An airport system plan is a representation of the aviation facilities required to meet the immediate and future needs of a metropolitan area, region, state, or country. The National Plan of Integrated Airport Systems (NPIAS) [11] is an example of a system plan representing the airport development needs of the United States. The Michigan Aviation System Plan [10] is an example of a system plan representing the airport development needs of the state of Michigan, and the Southeast Michigan Regional Aviation System Plan [13] is a system plan representing the airport development needs of a seven county region comprising the Detroit Metropolitan area.

The system plan presents the recommendations for the general location and characteristics of new airports and heliports and the nature of expansion for existing ones to meet forecasts of aggregate demand. It identifies the aviation role of existing and recommended new airports and facilities. It includes the timing and estimated costs of development and relates airport system planning to the policy and objectives of the relevant jurisdiction. Its overall purpose is to determine the extent, type, nature, location, and timing of airport development needed to establish a viable, balanced, and integrated system of

airports [1, 8]. It also provides the basis for detailed airport planning such as that contained in the airport master plan.

The airport system plan provides both broad and specific policies, plans, and programs required to establish a viable and integrated system of airports to meet the needs of the region. The objectives of the system plan include

1. The orderly and timely development of a system of airports adequate to meet present and future aviation needs and to promote the desired pattern of regional growth relative to industrial, employment, social, environmental, and recreational goals.
2. The development of aviation to meet its role in a balanced and multimodal transportation system to foster the overall goals of the area as reflected in the transportation system plan and comprehensive development plan.
3. The protection and enhancement of the environment through the location and expansion of aviation facilities in a manner which avoids ecological and environmental impairment.
4. The provision of the framework within which specific airport programs may be developed consistent with the short- and long-range airport system requirements.
5. The implementation of land-use and airspace plans which optimize these resources in an often constrained environment.
6. The development of long-range fiscal plans and the establishment of priorities for airport financing within the governmental budgeting process.
7. The establishment of the mechanism for the implementation of the system plan through the normal political framework, including the necessary coordination between governmental agencies, the involvement of both public and private aviation and nonaviation interests, and compatibility with the content, standards, and criteria of existing legislation.

The airport system planning process must be consistent with state, regional, or national goals for transportation, land use, and the environment. The elements in a typical airport system planning process [8] include the following:

1. Exploration of issues that impact aviation in the study area
2. Inventory of the current system
3. Identification of air transportation needs
4. Forecast of system demand
5. Consideration of alternative airport systems

6. Definition of airport roles and policy strategies
7. Recommendation of system changes, funding strategies, and airport development
8. Preparation of an implementation plan

Although the process involves many varied elements, the final product will result in the identification, preservation, and enhancement of the aviation system to meet current and future demand. The ultimate result of the process will be the establishment of a viable, balanced, and integrated system of airports.

Airport Site Selection

The emphasis in airport planning is normally on the expansion and improvement of existing airports. However if an existing airport cannot be expanded to meet the future demand or the need for a new airport is identified in an airport system plan, a process to select a new airport site may be required. The scope of the site selection process will vary with size, complexity, and role of the new airport, but there are basically three steps—identification, screening, and selection.

Identification—criteria is developed that will be used to evaluate different sites and determine if a site can function as an airport and meets the needs of the community and users. One criterion will be to identify the land area and basic facility requirements for the new airport. Part of this analysis will be a definition of airport roles if more than two airports serve the region. Other criteria might be that sites are within a certain radius or distance from the existing airport or community, or that sites should be relatively flat. Several potential sites that meet the criteria are identified.

Screening—once sites are identified, a screening process can be applied to each site. An evaluation of all potential sites that meet the initial criteria should be conducted, screening out those with the most obvious shortcomings. Screening factors might include topography, natural and man-made obstructions, airspace, access, environmental impacts, and development costs. If any sites are eliminated from further consideration, thorough documentation of the reasons for that decision is recommended. The remaining potential sites should then undergo a detailed comparison using comprehensive evaluation criteria. While the criteria will vary, the following is typically considered:

Operational capability—airspace considerations, obstructions, weather

Capacity potential—available land, suitability for construction, weather

Ground access—distance from the demand for aviation services, regional highway infrastructure, available public transportation modes

Development costs—terrain, land costs, land values, soil conditions, availability of utilities

Environmental consequences—aircraft noise, air quality, groundwater runoff, impact on flora and fauna, existence of endangered species or cultural artifacts, historical features, changes in local land use, relocation of families and businesses, changes in socioeconomic characteristics

Compatibility with area-wide planning—impact on land use, effect on comprehensive land-use plans and transportation plans at the local and regional levels

Selection—the final step is selecting and recommending a preferred site. While a weighting of the evaluation criteria and weighted ratings or ranking of the alternative sites is often used in selecting a site, caution must be used in applying this technique since it introduces an element of sensitivity into the analysis. The process should focus on providing decision makers with information on the various sites in a manner that is understandable and unbiased.

The Airport Master Plan

An airport master plan is a concept of the ultimate development of a specific airport. The term development includes the entire airport area, both for aviation and nonaviation uses, and the use of land adjacent to the airport [1, 4, 9]. It presents the development concept graphically and contains the data and rationale upon which the plan is based. Figure 4-2 shows a simple flowchart of the steps for preparing an airport master plan. Master plans are prepared to support expansion and modernization of existing airports and guide the development of new airports.

The overall objective of the airport master plan is to provide guidelines for future development which will satisfy aviation demand in a financially feasible manner and be compatible with the environment, community development, and other modes of transportation. More specifically it is a guide for

1. Developing the physical facilities of an airport
2. Developing land on and adjacent to the airport
3. Determining the environmental effects of airport construction and operations
4. Establishing access requirements
5. Establishing the technical, economic and financial feasibility of proposed developments through a thorough investigation of alternative concepts

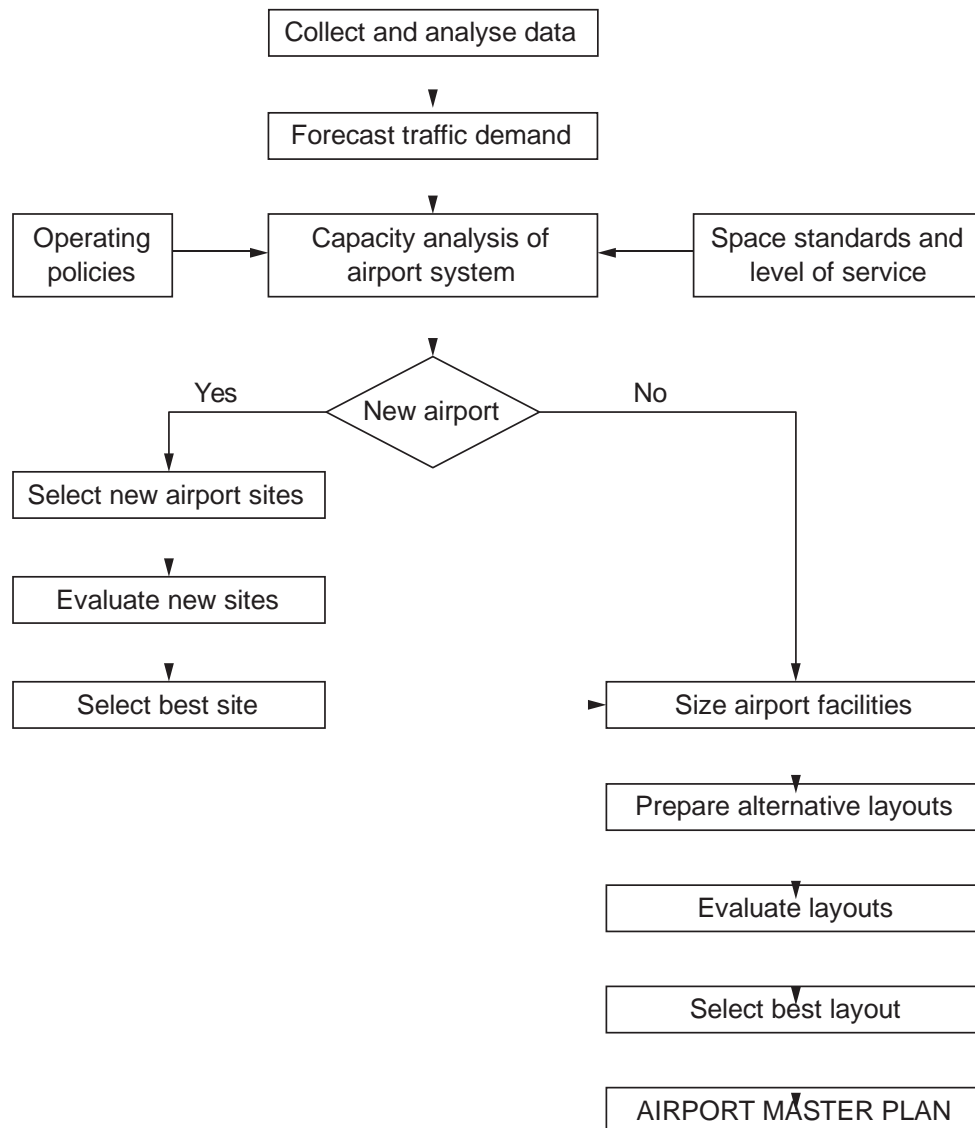


FIGURE 4-2 Flowchart for preparing an airport master plan.

6. Establishing a schedule of priorities and phasing for the improvements proposed in the plan
7. Establishing an achievable financial plan to support the implementation schedule
8. Establishing a continuing planning process which will monitor conditions and adjust plan recommendations as circumstances warrant

Guidelines for completing an airport master plan are described by ICAO [4] and in the United States by the FAA [1]. A master plan report is typically organized as follows:

Master plan vision, goals, and objectives—establishes the vision and overarching goals for the master plan as well as objectives that

will guide the planning process and help ensure that the goals are achieved and the vision is realized.

Inventory of existing conditions—provides an overview of the airport's history, role in the region and nation, growth and development over time, description of its physical assets (airfield and airspace, terminal, ground access, and support facilities), and key industry trends.

Forecast of aviation demand—future levels of aircraft operations, number of passengers, and volume of cargo are forecasted for short, intermediate, and long-range time periods. Typically forecasts are made for 5, 10, and 20 years on both annual as well as daily and busiest hours of the day.

Demand/capacity analysis and facility requirements—compares the future demand with the existing capacity of each airport component and identifies the facility requirements necessary to accommodate the demand.

Alternatives development—identifies, refines, and evaluates a range of alternatives for accommodating facility requirements. If the existing site cannot accommodate the anticipated growth, a selection process to find a new site may be necessary.

Preferred development plan—identifies, describes, and defines the alternative that best achieves the master plan goals and objectives. Figure 4-3 illustrates the development plan for the Chicago O'Hare International Airport.

Implementation plan—provides a comprehensive plan for the implementation of the preferred development plan, including the definition of projects, construction sequence and timeline, cost estimates, and financial plan.

Environmental overview—provides an overview of the anticipated environmental impacts associated with the preferred development plan in order to understand the severity and to help expedite subsequent environmental processing at the project specific stage.

Airport plans package—documents that show the existing as well as planned modifications are prepared and the more notable is the airport layout plan (ALP). It comprises drawings that include the airfield's physical facilities, obstruction clearance and runway approach profiles, land-use plans, terminal area and ground access plans, and a property map. Specific guidelines for the airport layout plan in the United States are identified by FAA [1].

Stakeholder and public involvement—documents the coordination efforts that occur among the stakeholders throughout the study.



FIGURE 4-3 Development plan for Chicago O'Hare International Airport.

The Airport Project Plan

A project plan focuses on a specific element of the airport master plan which is to be implemented in the short term and may include such items as the addition of a new runway, the modification of existing of runways, the provision of taxiways or taxiway exits, the addition of gates, the addition to or the renovation of terminal building facilities, or the modification of ground access facilities.

The overall objective of the airport project plan is to provide the specific details of the development which will satisfy immediate aviation needs and be consistent with the objectives and constraints identified in the airport master plan. More specifically it is a detailed plan for

1. Developing the specific physical facilities at an airport including the architectural and engineering design for these facilities
2. Determining the environmental effects of this development through the construction and operational phases
3. Determining the detailed costs and financial planning for the development
4. Establishing a schedule for the construction and phasing of the specific items of development in the plan

Land-Use Planning

A land-use plan for property within the airport boundary and in areas adjacent to the airport is an essential part of an airport master plan. The land-use plan on and off the airport is an integral part of an area wide comprehensive planning program, and therefore it must be coordinated with the objectives, policies, and programs for the area which the airport is to serve. Incompatibility of the airport with its neighbors stems primarily from the objections of people to aircraft noise. A land-use plan must therefore project the extent of aircraft noise that will be generated by airport operations in the future. Contours of equal intensity of noise can be drawn and overlaid on a land-use map and from these contours an estimate can be made of the compatibility of existing land use with airport operations. If the land outside the airport is underdeveloped, the contours are the basis for establishing comprehensive land-use zoning requirements.

Although zoning is used as a method for controlling land use adjacent to an airport, it is not effective in areas which are already built-up because it is usually not retroactive. Furthermore jurisdictions having zoning powers may not take effective zoning action. Aircraft operations into and out of the airport may be made unnecessarily complex to minimize noise encroachment on incompatible land uses. Despite these shortcomings the planner should utilize zoning as a vehicle to achieve compatibility wherever this approach is feasible.

Airports become involved in two types of zoning. One type is height and hazard zoning, which is mainly to protect the approaches to the airport from obstructions. The other type is land-use zoning.

The extent of land use in the airport depends a great deal on the amount of acreage available. Land uses can be classified as either

closely related to aviation or remotely related to aviation. Those closely related to aviation use include the runways, taxiways, aprons, terminal buildings, parking, and maintenance facilities. Nonaviation uses include space for recreational, industrial, and commercial activities. When considering commercial or industrial activities, care should be taken to ensure that they will not interfere with aircraft operations, communications equipment, and aids to navigation on the ground. Recreational facilities such as golf courses may be suitable within the immediate proximity of the airport boundary or certain agricultural uses are also appropriate as long as they do not attract birds. When there is acreage within the airport boundary in excess of aviation needs, it is sound fiscal planning to provide the greatest financial return from leases of the excess property. Thus the land-use plan within the airport is a very effective tool in helping airport management make decisions concerning requests for land use by various interests and often airports delineate areas on the airport property for the development of industrial parks.

The principal objective of the land-use plan for areas outside the airport boundary is to minimize the disturbing effects of noise. As stated earlier the delineation of noise contours is the most promising approach for establishing noise-sensitive areas. The contours define the areas which are or are not suitable for residential use or other use and, likewise, those which are suitable for light industrial, commercial, or recreational activity. Although the responsibility for developing land uses adjacent to the airport lies with the governing bodies of adjacent communities, the land-use plan provided by the airport authority will greatly influence and assist the governing bodies in their task of establishing comprehensive land-use zoning.

Environmental Impact Assessment

Environmental factors must be considered carefully in the development of a new airport or the expansion of an existing one. In the United States, this is a requirement of the Airport and Airway Improvement Act of 1982 and the Environmental Policy Act of 1969. Studies of the impact of the construction and operation of a new airport or the expansion of an existing one upon acceptable levels of air and water quality, noise levels, ecological processes, and demographic development of the region must be conducted to determine how the airport requirements can best be met with minimal adverse environmental and social consequences.

Aircraft noise is the severest environmental problem to be considered in the development of airport facilities. Much has been done to quiet engines and modify flight procedures, resulting in substantial reductions in noise. Another effective means for reducing noise is through proper planning of land use for areas adjacent to the airport. For an existing airport this may be difficult as the land may have

already been built up. Every effort should be made to orient air traffic away from noise-sensitive land development.

Other important environmental factors include air and water pollution, industrial wastes and domestic sewage originating at the airport, and the disturbance of natural environmental values. In regard to air pollution, the federal government and industry have worked jointly toward alleviating the problem, and there is a reason to believe that it will probably be eliminated in the near future as an environmental factor. An airport can be a major contributor to water pollution if suitable treatment facilities for airport wastes are not provided. Chemicals used to deice aircraft are a major source of potential ground water pollution and provisions need to be made to safely dispose of this waste product. The environmental study must include a statement detailing the methods for handling sources of water pollution.

The construction of a new airport or the expansion of an existing one may have major impacts on the natural environment. This is particularly true for large developments where streams and major drainage courses may be changed, the habitats of wildlife may be disrupted, and wilderness and recreational areas may be reshaped. The environmental study should indicate how these disruptions might be alleviated.

In the preparation of an environmental study, or an environmental impact statement, the findings must include the following items [12]:

1. The environmental impact of the proposed development
2. Any adverse environmental effects which cannot be avoided should the development be implemented
3. Alternatives to the proposed development
4. The relationship between local short-term uses of the environment and the maintenance and enhancement of long-term productivity
5. Any irreversible environmental and irretrievable commitments of resources which would be involved in the proposed development should it be implemented
6. Growth inducing impact
7. Mitigation measures to minimize impact

In the application of these guidelines attention must be directed to the following questions. Will the proposed development

1. Cause controversy
2. Noticeably affect the ambient noise level for a significant number of people

3. Displace a significant number of people
4. Have a significant aesthetic or visual effect
5. Divide or disrupt an established community or divide existing uses
6. Have any effect on areas of unique interest or scenic beauty
7. Destroy or derogate important recreational areas
8. Substantially alter the pattern of behavior for a species
9. Interfere with important wildlife breeding, nesting, or feeding grounds
10. Significantly increase air or water pollution
11. Adversely affect the water table of an area
12. Cause excessive congestion on existing ground transportation facilities
13. Adversely affect the land-use plan for the region

The preparation of an environmental impact statement based upon an environmental assessment study is an extremely important part of the airport planning process. The statement should clearly identify the problems that will affect environmental quality and the proposed actions to alleviate them. Unless the statement is sufficiently comprehensive, the entire airport development may be in jeopardy.

Economic and Financial Feasibility

The economic and financial feasibility of alternative plans for a new airport or expansion of an existing site must be clearly demonstrated by the planner. Even if the selected alternative is shown to be economically feasible, then also it is necessary to show that the plan will generate sufficient revenues to cover annual costs of capital investment, administration, operations, and maintenance. This must be determined for each stage or phase of development detailed in the airport master plan.

An evaluation of economic feasibility requires an analysis of benefits and costs. A comparison of benefits and costs of potential capital investment programs indicates the desirability of a project from an economic point of view. The economic criterion used in evaluating an aviation investment is the total cost of facilities, including quantifiable social costs, compared with the value of the increased effectiveness measured in terms of total benefits. The costs include capital investment, administration, operation, maintenance, and any other costs that can be quantified. The benefits include a reduction in aircraft and passenger delays, improved operating efficiency, and other benefits. The costs and benefits are usually determined on an annual basis.

There are a number of techniques for comparing benefits with costs. Most of them consider the time value of money based on an appropriate discount rate which reflects the opportunity cost of capital. The discount rate is a value by which a unit of money received in the future is multiplied to obtain its present value or present worth. In other words a cost incurred in 2010 has a different economic value from that of the same item incurred in 2015.

If the time value of money is not considered, the ratio of benefits to costs is made for each year by merely dividing the benefits in a particular year by the cost of the project in that year. A project is considered economically feasible when the ratio of the benefits to costs is greater than unity, that is, the benefits exceed the costs. The larger the ratio, the more attractive is the project from an economic standpoint. A ratio can also be obtained by comparing the present value of benefits with the present value of costs. This approach recognizes the time value of money. Another approach is to plot the *net present value* (NPV) for each year against time. The net present value is defined as the present value of benefits minus the present value of costs.

The financing of capital improvements for airports is discussed in Chap. 13. In the early years of airport development, substantial capital improvement programs were financed at the local level by sale of general obligation bonds backed by the taxing power of the community. As air transportation became mature and the requirements of the community for capital spending programs increased, airports began to utilize revenue bonds as a source of financing. A financial feasibility study is therefore an analysis to determine if bonds are marketable at reasonable interest rates. It also includes the feasibility of other forms of financing. The analysis requires a thorough evaluation of the revenues to be developed by a proposed improvement and the corresponding costs. Usually this is done in a traffic and earnings study performed over the planning horizon. In such a study, the forecast of demand is utilized and rates and charges established for the various revenue categories. This results in annual revenue projections. To make revenue bonds attractive to buyers a typical airport revenue bond should show an expected coverage by net revenues (gross revenues minus costs) of at least 1.25 times the debt service requirements. If the analysis indicates that the revenues will be insufficient, revisions in the scheduling or scope of the proposed development may have to be made or the rates and charges to the users of the airport may require adjustment.

Continuing Planning Process

A continuous airport planning process is necessary in order to respond to the needs of air transportation in a changing environment [8]. Changes in aviation demand, community policies, new technology,

financial constraints, and other factors can alter the need for and the timing of facility improvements. Current data must be continually collected and assessed relative to airport needs, operations and utilization, environmental impact, and financial capabilities. The staging of airport improvements assists in the reevaluation of continuing needs at the points in time when implementation decisions are required.

In the airport planning process, the overall objective of establishing and maintaining a continuous process is to ensure that the airport system plan and airport master plan remain responsive to public needs. As a result, the airport system plan and airport master plan should be formally reviewed and updated at least every 5 years. Specific objectives associated with the continuous airport planning process include [8]

1. Surveillance, maintenance, inventory, and update of the basic data such as aviation activity and socioeconomic and environmental factors relating to the existing airport system and master plan
2. Review and validation of data affecting the airport system and master plan
3. Reappraisal of the airport system and master plan in view of changing conditions
4. Modification of the airport system and master plan to retain its viability
5. Development of a continuous mechanism for ensuring the interchange of information between the system planning and master planning processes
6. Provision of a means for receiving and considering public comment in order to maintain and ensure the public awareness of the role airports play in the transportation system of an area
7. Redefinition of air transportation goals and policies
8. Integration of airport system planning into a multimodal planning process
9. Analysis of special issues
10. Publication of interim reports and formal plan updates

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CHAPTER 5

Forecasting for Airport Planning

Introduction

Plans for the development of the various components of the airport system depend to a large extent on the activity levels which are forecast for the future. Since the purpose of an airport is to process aircraft, passengers, freight, and ground transport vehicles in an efficient and safe manner, airport performance is judged on the basis of how well the demand placed upon the facilities within the system is handled. To adequately assess the causes of performance breakdowns in existing airport systems and to plan facilities to meet future needs, it is essential to predict the level and distribution of demand on the various components of the airport system. Without a reliable knowledge of the nature and expected variation in the loads placed upon a component, it is impossible to realistically assess the physical and operational requirements of such a component. For example, a forecast to project the mix of aircraft and the types of aviation activity at an airport site is necessary to identify the critical aircraft which dictates the elements of geometric and structural design, the type and extent of physical facilities, the navigational aid requirements, and any special or unique facility needs at the airport [15, 17].

An understanding of future demand patterns allows the planner to assess future airport performance in light of existing and improved facilities, to evaluate the impact of various quality of service options on the airlines, travelers, shippers, and community, to recommend development programs consistent with the overall objectives and policies of the airport operator, to estimate the costs associated with these facility plans, and to project the sources and level of revenues to support the capital improvement program.

It is essential in the planning and design of an airport to have realistic estimates of the future demand to which airports are likely to be subjected. This is a basic requirement in developing either an

airport master plan or an airport system plan. These estimates determine the future needs for which the physical facilities are designed. A financial plan to achieve the recommended staged development along with required land-use zoning usually accompanies the plan. It should be apparent that an airport is designed for a projected level and pattern of demand and changes in the magnitude or characteristics of this demand may require facility modifications or operational measures to meet changing needs. Facility planning is necessary to provide adequate levels of service to airport users.

The development of accurate forecasts requires a considerable expense of time and other resources because of the complex methodologies which must be used and the extensive data acquisition that is often required. The usual justification for a demand forecast in an aviation plan is that the expected level of uncertainty associated with the estimation of essential variables will be reduced, thereby reducing the probability of errors in the planning process and enhancing the decision-making process. The implication, of course, is that the benefits gained due to a better knowledge of the magnitude and fluctuation in demand variables will outweigh the costs incurred in performing the forecast.

To assess the characteristics of future demand, the development of reliable predictions of airport activity is necessary. There are numerous factors that will affect the demand and planners who are preparing forecasts of demand or updating existing forecasts should consider local and regional socioeconomic data and characteristics, demographics, geographic attributes, and external factors such as fuel costs and quality of service parameters. Political developments, including rising international tensions, changes in security, airline delays and congestion, and travel attitudes, will impact demand. Actions taken by local airport authorities, such as changes in user charges, can also stimulate or hinder the demand and investment decisions made as the result of the planning process itself can also produce change by removing physical constraints to growth [2].

Over the years, certain techniques have evolved which enable airport planners and designers to forecast future demand. The principal items for which estimates are usually needed include

- The volume and peaking characteristics of passengers, aircraft, vehicles, and cargo
- The number and types of aircraft needed to serve the above traffic
- The number of based general aviation aircraft and the number of movements generated
- The performance and operating characteristics of ground access systems

Using forecasting techniques, estimates of these parameters and a determination of the peak period volumes of passengers and aircraft

movements can be made. From these estimates concepts for the layout and sizing of terminal buildings, runways, taxiways, apron areas, and ground access facilities may be examined.

Forecasting demand in an industry as dynamic as aviation is an extremely difficult matter, and if it could be avoided it undoubtedly would. Nonetheless estimates of traffic must be made as a prelude to the planning and design of facilities. It is very important to remember that forecasting is not a precise science and that considerable subjective judgment must be applied to any analysis no matter how sophisticated the mathematical techniques involved. By anticipating and planning for variations in predicted demand, the airport designer can correct projected service deficiencies before serious deficiencies in the system occur.

Levels of Forecasting

Demand estimates are prepared for a variety of reasons. Broad large-scale aggregated forecasts are made by aircraft and equipment manufacturers, aviation trade organizations, governmental agencies, and others to determine estimates of the market requirements for aviation equipment, trends in travel, personnel needs, air traffic control requirements, and other factors. Similarly, forecasts are made on a smaller scale to examine these needs in particular regions of an area and at specific airports.

In economics, forecasting is done on two levels, aggregate forecasting and disaggregate forecasting, and the same holds true in aviation. From the inception of the planning process for an airport consideration is given at both levels. In airport planning, the designer must view the entire airport system as well as the airport under immediate consideration. Aggregate forecasts are forecasts of the total aviation activity in a large region such as a country, state, or metropolitan area. Typical aggregate forecasts are made for such variables as the total revenue passenger-miles, total enplaned passengers, and the number of aircraft operations, aircraft in the fleet, and licensed pilots in the country. Disaggregate forecasts deal with the activity at individual airports or on individual routes. Disaggregate forecasts for airport planning determine such variables as the number of originations, passenger origin-destination traffic, the number of enplaned passengers, and the number of aircraft operations by air carrier and general aviation aircraft at an airport. Separate forecasts are usually made, depending on the need in a particular study, for cargo movements, commuter service, and ground access traffic. These forecasts are normally prepared to indicate annual levels of activity and are then disaggregated for airport planning purposes to provide forecasts of the peaking characteristics of traffic during the busy hours of the day, days of the week, and months of the year. As appropriate to the requirements of an airport planning study, forecasts of such quantities as the number

of general aviation aircraft based at an individual airport and the number of general aviation and military operations are also prepared.

In aggregate forecasting, the entire system of airports is examined relative to the geographic, economic, industrial, and growth characteristics of a region to determine the location and nature of airport needs in a region. The disaggregate forecast then examines the expected demand at local airports and identifies the necessary development of the airside, landside, and terminal facilities to provide adequate levels of service. Within the two levels of forecasting there are certain techniques which enable the planner to project such parameters as annual, daily, and hourly aircraft operations, passenger enplanements, cargo, and general aviation activity. In disaggregate forecasting, there are many significant variables. The forecast of each variable is quite important because it ultimately determines the size requirements of the facilities which will be necessary to accommodate demand. Often, the forecasts of the different variables are linked by a series of steps, that is, originating passengers are forecast first, this then becomes a component in the forecast of enplaned passengers, which leads to a forecast of annual operations, and so on.

The type of forecast and the level of effort depend on the purpose for which the forecast is being used. Forecasts are typically prepared for short-, medium-, and long-term periods. Short-term forecasts, up to 5 years, are used to justify near-term development and support operational planning and incremental improvements or expansion of facilities. Medium-term forecasts, a 6- to 10-year time frame, and long-term forecasts of 10 to 20 years are used to plan major capital improvements, such as land acquisition, new runways and taxiways, extensions of a runway, a new terminal, and ground access infrastructure. Forecasts beyond 20 years are used to assess the need for additional airports or other regional aviation facilities [1].

Forecasting Methods

There are several forecasting methods or techniques available to airport planners ranging from subjective judgment to sophisticated mathematical modeling. The selection of the particular methodology is a function of the use of the forecast, the availability of a database, the complexity and sophistication of the techniques, the resources available, the time frame in which the forecast is required and is to be used, and the degree of precision desirable. There are four major methods:

- Time series method
- Market share method
- Econometric modeling
- Simulation modeling

Time series analysis essentially involves extrapolating or projecting existing historical activity data into the future. Market share forecasting is a simple top-down approach, where current activity at an airport is calculated as a share of some other more aggregate measure for which a forecast has been made (typically a regional, state, or national forecast of aviation activity). Econometric modeling is a multistep process in which a casual relationship is established between a dependent variable (the item to be forecast) and a set of independent variables that influence the demand for air travel. Once the relationship is established, forecasts of independent variables are input to determine a forecast of the dependent variable. These techniques can also be referred to as a bottom-up forecast. Simulation models are often used when one needs very detailed estimates of aircraft, passengers, or vehicles. These models impose precise rules that govern how passengers, aircraft, or vehicles are routed, and then aggregates the results so that planners can assess the needs of the network or a component of the airport to handle the estimated demand. Typically the outputs from the other forecasting methods are used as inputs to simulation models. Forecasts from simulation models represent snapshots of how a given amount of traffic flows across a network or through an airport, rather than a monthly or annual estimate of total traffic.

An important element which should be utilized in any forecasting technique is the use of professional judgment. A forecast prepared through the use of mathematical relationships must ultimately withstand the test of rationality. Frequently a group of professionals knowledgeable about aviation and the factors influencing aviation trends are assembled to examine forecasts from several different sources, and composite forecasts are prepared in accordance with the information in these sources and the collective judgment of the group. In some cases, judgment becomes the principal approach used with or without an evaluation of economic and other factors that are believed to affect aviation activity. A common approach being utilized more often today for preparing forecasts by judgment is known as the Delphi method. In this method a panel of experts on a particular subject matter is asked to rate or otherwise prioritize a series of questions or projections through a survey technique. The results of the survey are then distributed to the members of the panel and an opportunity is provided for each member to reevaluate the original rating based upon the collective ratings of the group. The reevaluation process is often sent through several iterations in order to arrive at a better result. In the Delphi method, the results of the technique do not have to represent a consensus of the panel and, in fact, it is often quite useful to have a forecast which indicates the spread of the panel in reaching conclusions on a particular issue.

The preparation of judgmental forecasts which reflect the collective wisdom of a broad range of professionals has proven to be very

successful in many instances principally due to the large number of factors which may be considered in such a process. Though there is often a lack of mathematical sophistication in the process, the knowledge and consideration of the many diverse factors influencing aviation forecasts usually improves the results. The disadvantages of this forecasting technique include the absence of statistical measures on which to base the results and the inability, except in the most obvious cases, to gain a significant consensus relative to the expected performance of the explanatory factors in the future.

Time Series Method

Time series analysis or extrapolation is based upon an examination of the historical pattern of activity and assumes that those factors which determine the variation of traffic in the past will continue to exhibit similar relationships in the future. This technique utilizes times series type data and seeks to analyze the growth and growth rates associated with a particular aviation activity. In practice, trends appear to develop in situations in which the growth rate of a variable is stable in either absolute or percentage terms, there is a gradual increase or decrease in growth rate, or there is a clear indication of market saturation trend over time [11]. Statistical techniques are used to assist in defining the reliability and the expected range in the extrapolated trend. The analysis of the pattern of demand generally requires that upper and lower bounds be placed upon the forecast and statistics are used to define the confidence levels within which specific projections may be expected to be valid. From the variation in the trends and the upper and lower bounds placed on the forecast a preferred forecast is usually developed. Quite often smoothing techniques are incorporated into the forecast to eliminate short run, or seasonal, fluctuations in a pattern of activity which otherwise demonstrates a trend or cyclical pattern in the long run [16].

An illustration of the application of a trend line analysis to forecast annual enplanements at an airport is shown in Example Problem 5-1.

Example Problem 5-1 The historical data shown in Table 5-1 have been collected for the annual passenger enplanements in a region and one of the commercial service airports in this region. It is necessary to prepare a forecast of the annual passenger enplanements at the study airport in the design years 2010 and 2015 using a trend line analysis.

In applying the trend line analysis to these data, a forecast technique of the annual enplanements at the study airport will be made by forecasting the historical trend to the design years. A plot of the trend in the annual passengers enplaned at the study airport is given in Fig. 5-1.

By extrapolating the trend into the future, an estimate of the annual enplaned passengers at the study airport in 2010 is found to be 2,100,000 passengers and in the year 2015 is found to be 2,900,000 passengers.

Year	Annual Enplanements		Area Population
	Regional	Airport	
1998	13,060,000	468,900	250,000
1999	14,733,000	514,300	260,000
2000	16,937,000	637,600	270,000
2001	21,896,000	758,200	280,000
2002	24,350,000	935,200	290,000
2003	28,004,000	995,500	300,000
2004	31,658,000	1,139,700	310,000
2005	37,226,000	1,360,700	320,000
2006	40,753,000	1,488,900	330,000
2007	44,018,000	1,650,600	340,000

TABLE 5-1 Enplanement Data for Airport Demand Forecast for Use in Example Problems 5-1 through 5-3

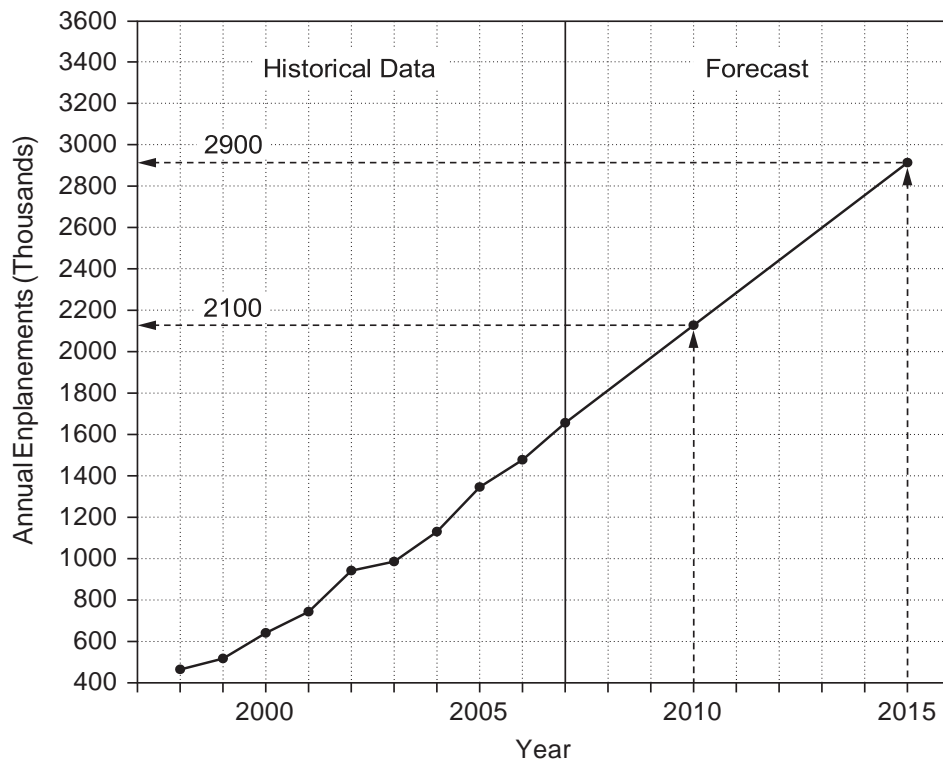


FIGURE 5-1 Trend line forecast of annual enplanements for Example Problem 5-1.

The inability of time series techniques to show a causal relationship between the dependent and independent variables is a serious disadvantage. This is particularly true because, in the absence of such relationships, the degree of uncertainty in such forecasts increases with time. However, the time series method is useful for short-term forecasts in which the response to changes in those factors which stimulate the dependent variables is usually less dynamic. In those cases where cyclic variations may be expected to occur, time series methods may also be quite beneficial.

Market Share Method

Forecasting techniques which are utilized to proportion a large-scale aviation activity down to a local level are called market share, ratio, or top-down models. Inherent to the use of such a method is the demonstration that the proportion of the large-scale activity which can be assigned to the local level is a regular and predictable quantity. This method has been the dominant technique for aviation demand forecasting at the local level and its most common use is in the determination of the share of total national traffic activity which will be captured by a particular region, traffic hub, or airport. Historical data are examined to determine the ratio of local airport traffic to total national traffic and the trends are ascertained. From exogenous sources the projected levels of national activity are determined and these values are then proportioned to the local airport based upon the observed and projected trends. The ratio method is most commonly used in the development of microforecasts for regional airport system plans or for airport master plans.

These methods are particularly useful in applications in which it can be demonstrated that the market share is a regular, stable, or predictable parameter. For example, the number of annual enplaned passengers at major air traffic hubs has been shown to be a consistent and relatively stable factor and, therefore, this method is often used to predict this parameter.

Quite often the application of the market share technique is a two-step process in which a ratio is applied to disaggregate activity forecasts from a national to a regional level and then another ratio is applied to apportion the regional share among the airports in the region.

The most compelling advantage of the market share method is its dependence on existing data sources which minimizes forecasting cost. However, its principal disadvantages lie in its dependence on the stability and predictability of the ratios from which the forecasts are made and the uncertainty which may surround market shares in specific applications. Several forecasts may be required under a differing set of assumptions which are deemed appropriate to the determination of market shares. An illustration of the application of a market share analysis is given in Example Problem 5-2.

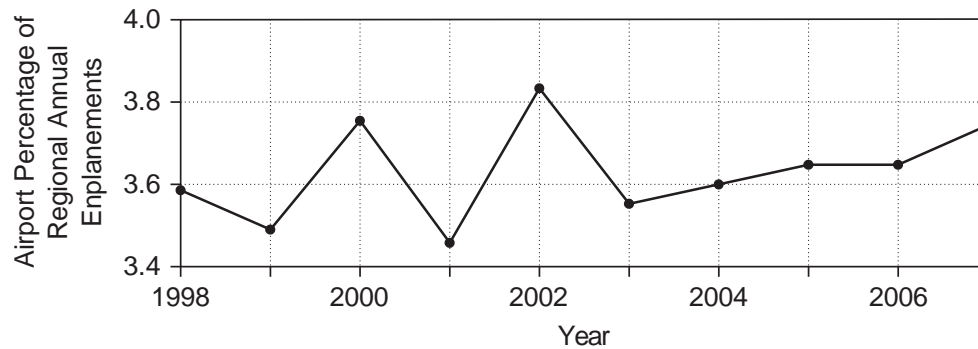


FIGURE 5-2 Trend line of airport percentage of regional annual enplanements for Example Problem 5-2.

Example Problem 5-2 The historical data shown in Table 5-1 could also be used to prepare a forecast of the annual passenger enplanements at the study airport in the design years 2010 and 2015 using a market share method.

In applying the market share method to these data, a top-down forecast technique will be used. The implicit assumption in such a technique is that the same relationship between regional annual enplanements and the annual enplanements at the study airport will be maintained in the future. To prepare such a forecast, a projection of the percentage of the regional annual enplanements captured at the study airport is performed and then a forecast is made of the regional annual enplanements. The study airport forecast percentage is applied to the regional forecast to arrive at the forecast of the study airport annual enplanements in the design years. A plot of the trend in the percentage of regional annual passengers enplaned at the study airport is given in Fig. 5-2.

Because the variations shown in Fig. 5-2 often make it difficult to determine if trends may exist, a smoothing function is often applied to the data to assist in identifying trends which may not be obvious. In this case, a smoothing of the data was obtained by computing a running 3-year average of the data points. As is shown in Fig. 5-3, this tends to smooth out the variations in the original data and more readily identifies trends in these data.

Though it may not be apparent in the original plot, the smoothing mechanism does indicate a very slight upward trend in the percentage of regional annual passengers captured by the study airport. This trend is shown by the dashed line in Fig. 5-3. This trend line, when projected to the design years, indicates a forecast of

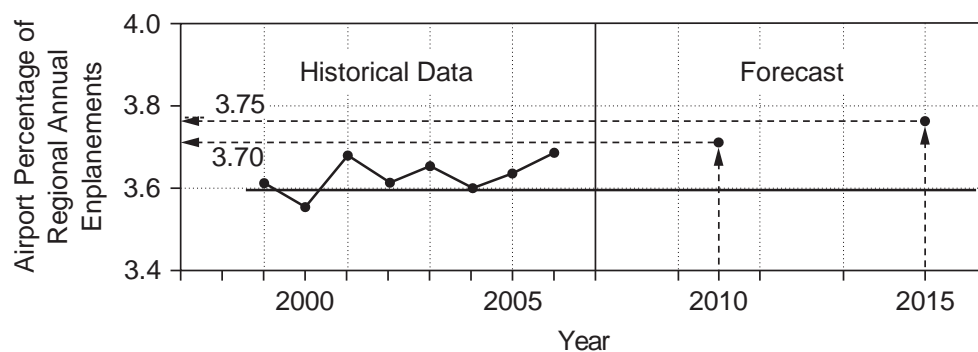


FIGURE 5-3 Trend line forecast of airport percentage of regional annual enplanements by applying smoothing function to trend data for Example Problem 5-2.

3.70 percent in the year 2010 and 3.75 percent in the year 2015 as the proportion of regional annual enplanements forecast to be captured by the study airport.

To complete the forecast by the market share method, a determination of the regional annual enplanements must be made. This is done by extrapolating the trend in regional annual enplanements as shown in Fig. 5-4. This extrapolation indicates regional annual enplanements in the year 2010 of 52,500,000 and in the year 2015 of 63,500,000.

Therefore, the forecast for the annual enplanements at the study airport becomes $0.0370 \times 52,500,000 = 1,942,500$ passengers in the year 2010 and $0.0375 \times 63,500,000 = 2,381,250$ passengers in the year 2015.

Econometric Modeling

The most sophisticated and complex technique in airport demand forecasting is the use of econometric models. Trend extrapolation methods do not explicitly examine the underlying relationships between the projected activity descriptor and the many variables which affect its change. There are a wide range of economic, social, market, and operational factors which affect aviation. Therefore, to properly assess the impact of predicted changes in the other sectors of society upon aviation demand and to investigate the effect of alternative assumptions on aviation, it is often desirable to use mathematical techniques to study the correlations between dependent and independent variables. Econometric models which relate measures of aviation activity to economic and social factors are extremely valuable techniques in forecasting the future.

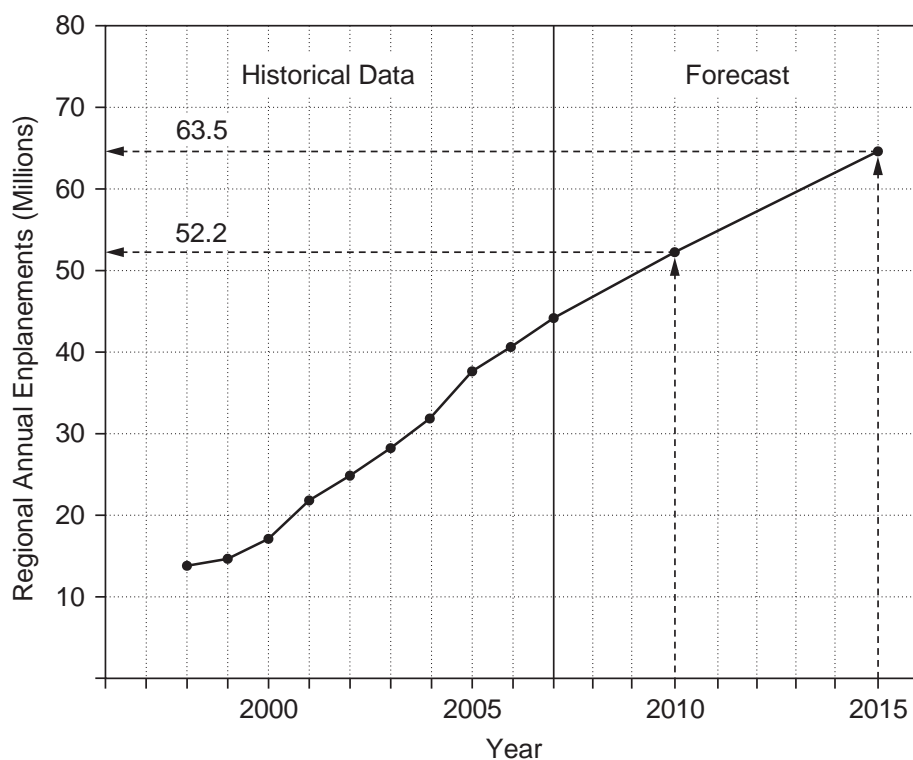


FIGURE 5-4 Trend line forecast of regional annual enplanements for Example Problem 5-2.

There are a great variety of techniques which are used in econometric modeling for airport planning. Classical trip generation and gravity models are quite common in forecasting passenger and aircraft traffic. Simple and multiple regression analysis techniques, both linear and nonlinear, are often applied to a great variety of forecasting problems to ascertain the relationships between the dependent variables and such explanatory variables as economic and population growth, market factors, travel impedance, and intermodal competition. The form of the equations used in multiple linear regression analysis is given in Eq. (5-1).

$$Y_{\text{est}} = a_0 + a_1 X_1 + a_2 X_2 + a_3 X_3 + \cdots + a_n X_n \quad (5-1)$$

where Y_{est} = dependent variable or variable which is being forecast
 $X_1, X_2, X_3, \dots, X_n$ = independent variables or variables being used to explain the variation in the dependent variable
 $a_0, a_1, a_2, a_3, \dots, a_n$ = regression coefficients or constants used to calibrate the equation

There are many statistical tests which can be performed to determine the validity of econometric models in accurately portraying historical phenomena and to reliably project demand. Though the constants may be found to define the general equation of the model, it is possible that the range of error associated with the equation may be large or that the explanatory variables chosen do not directly determine the variation in the dependent variable.

There may be a tendency when performing sophisticated mathematical modeling to become disassociated from the significance of the results. It is incumbent upon the analyst to consider the reasonableness as well as the statistical significance of the model. Adequate consideration must be given to the rationality of the functional form and variables chosen for the analysis, and to the logic associated with calibrated constants.

In many cases it is essential to determine the sensitivity of forecasts to changes in the explanatory variables. If a particular design parameter being forecast varies considerably with a change in a dependent variable, and there is a significant degree of unreliability in this independent variable, then a great deal of confidence cannot be placed upon the forecast and, more importantly, the design based upon the forecast. Tests are usually performed to determine the explanatory power of the independent variables and their interrelationships. The analyst should carefully investigate the sensitivity of projections within the expected variation of explanatory variables. It is also possible that certain explanatory variables do not significantly affect the modeling equation and the need for collecting the data associated with these variables required for projections could be

eliminated. An illustration of the application of simple linear regression analysis is presented in Example Problem 5-3.

Example Problem 5-3 The historical data shown in Table 5-1 could also be used to prepare a forecast of the annual passenger enplanements at the study airport in the design years 2010 and 2015 using a simple regression analysis.

In applying simple regression analysis to these data, let us assume that a relationship between the study airport annual enplanements (ENP) and the study area population (POP) is to be examined. Therefore, it is assumed that a linear relationship of the form shown in Eq. (5-1) exists between the variables.

$$\text{ENP} = a_0 + a_1(\text{POP})$$

Using a standard regression analysis computer program the relationship is found to be

$$\text{ENP} = -3,047,032 + 13.8633(\text{POP})$$

where the coefficient of determination R^2 is 0.983815, the coefficient of correlation is 0.991874, and the standard error of the estimate, σ_{yest} is 55,520.9. The regression line and the data points upon which this regression line is based are shown in Fig. 5-5.

The coefficient of determination indicates that there is an extremely good relationship between the annual enplanements at the study airport and the study area population, that is, 98.4 percent of the variation in the study airport annual enplanements is explained by the variation in the study area population.

The standard error of the estimate, however, indicates that there is a large range of error associated with forecasting with this equation, that is, there is a 68 percent probability that the forecast of annual enplanements at the study airport will have an error range of $\pm 55,520.9$ annual enplanements. This may

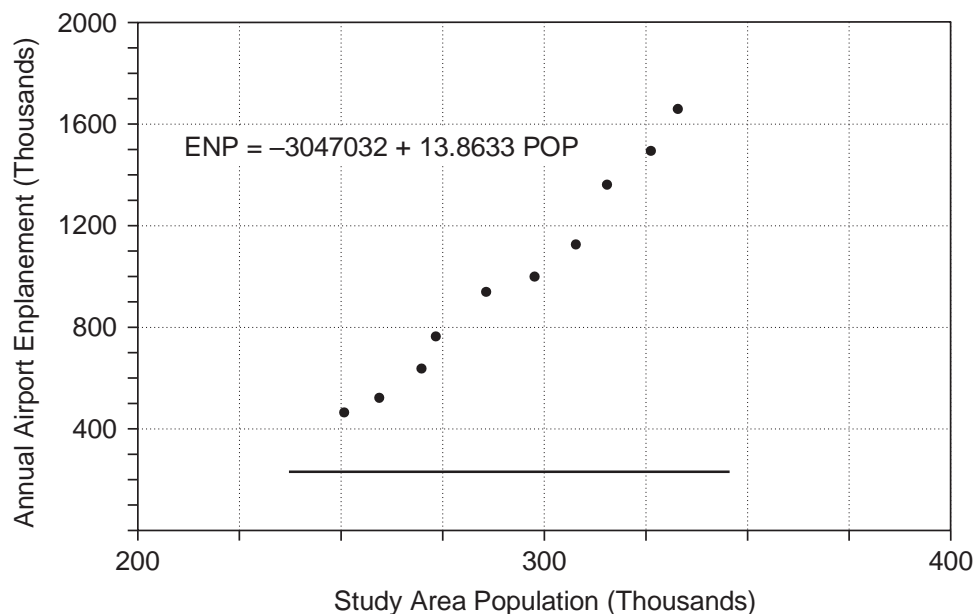


FIGURE 5-5 Trend line forecast of study area population for Example Problem 5-3.

or may not be too high depending on the level of annual operations forecast in the future and the sensitivity of various components of the airport system to such variations.

Using a trend projection, it is forecast that the area population in the year 2010, as shown on Fig. 5-6, is expected to be 363,000. The forecast of the annual enplanements at the airport in the year 2010 can be found by substitution into the regression equation yielding 1,985,300 annual enplanements. Similarly, if the forecast of the area population in the year 2015 is expected to be 410,000, then the forecast of the annual enplanements at the airport in the year 2015 is found to be 2,636,900.

Given the range in the standard error of the estimate, it could be expected that in the year 2010 there is a probability of 68 percent that the forecast could range between $1,985,300 \pm 55,500$, or from 1,929,800 to 2,040,800 annual enplanements about 68 percent of the time. Similarly, it could be expected that in the year 2015 the forecast could range between $2,636,900 \pm 55,500$, or from 2,581,900 to 2,692,400. It is likely that this range in the forecasts is acceptable since it represents about a 2 to 3 percent error.

It is interesting to compare the results found by the three different techniques used in Example Problems 5-1 through 5-3. The results compare very well and it gives one some degree of confidence in the results when the three forecasts compare well. This is called redundancy in forecasting.

Based upon the results found in these example problems, a preferred forecast would be developed. If there is no reason to suspect

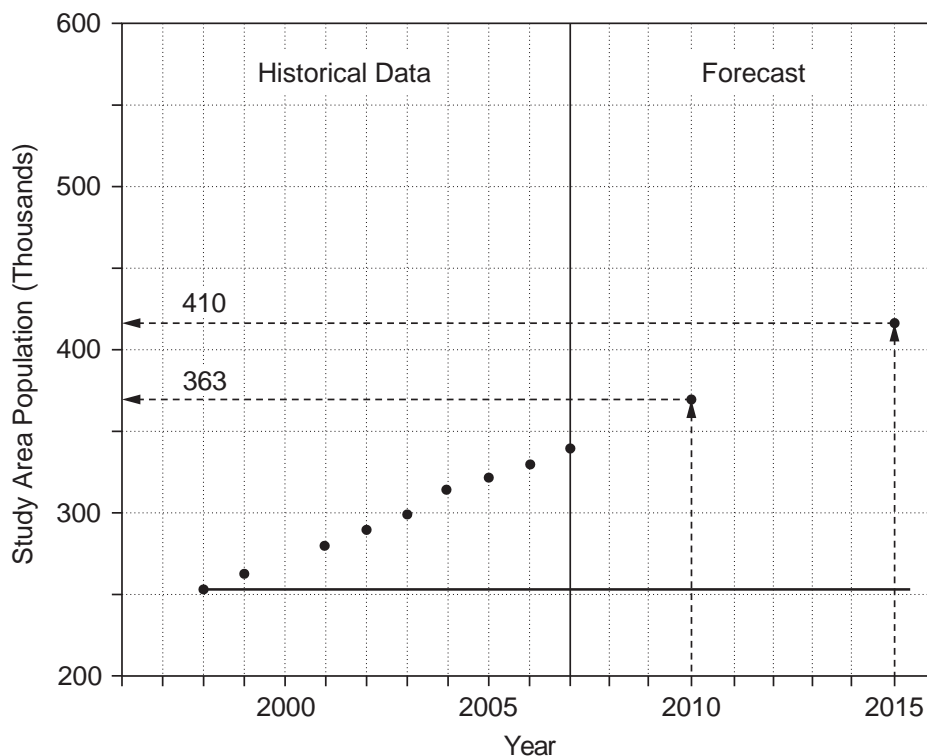


FIGURE 5-6 Trend line forecast of study area population for Example Problem 5-3.

that one technique is better than another, then a simple average might be used to develop the preferred forecast. If this is done in these examples, then the preferred forecast for the year 2010 is about 2,000,000 annual enplaned passengers and in the year 2015 is 2,600,000 annual enplaned passengers.

The Federal Aviation Administration utilizes econometric models to determine national forecasts of U.S. aviation demand. The FAA Aerospace Forecast [8] provides a 12-year outlook and view of the immediate future for aviation. It is updated in March each year and includes aggregate level forecasts of the following:

- Passenger enplanements, revenue passenger miles, fleet, and hours flown for large carriers and regional commuters
- Cargo revenue ton miles and cargo fleet for large air carriers
- Fleet, hours, and pilots for general aviation
- Activity forecasts for FAA and contract towers by major user category

The FAA Long Range Aerospace Forecasts [9] is a long range forecast that extends the 12-year forecast to a longer time horizon, typically for a period of 25 years. This forecast contains projections of aircraft, fleet and hours, air carrier and regional/commuter passenger enplanements, air cargo freight revenue ton-miles, pilots, and FAA workload measures.

The success in applying mathematical modeling techniques to ascertain the level of future activity depends to a large extent on the certainty associated with the independent variables and the relative influence of these variables on the dependent variable. Simple and multiple regression analysis methods are often applied to a great variety of forecasting problems to determine the relationships between transport related variables and such explanatory factors as economic and population growth, market factors, travel impedance, and competitive forces. Table 5-2 lists many of the variables required for various purposes in aviation planning studies.

Forecasting Requirements and Applications

The specific forecasting needs depend on the nature and scope of the study being undertaken. The requirements for a state aviation system plan are very different than those required for an airport master plan. Facility planning requires projections of the parameters which determine physical design whereas financial planning requires projections of the cost elements and revenue sources associated with physical development. This section outlines the general forecasting requirements for various types of airport studies and discusses the more common methodologies used to arrive at these requirements.

Application In Planning Studies	Forecast Variables Required
Macroforecast	
National airport system needs	Revenue passengers
State or regional airport needs	Revenue passenger-miles
Airlines	Aircraft fleet
Aircraft and equipment manufacturers	Air carrier
Investment planning	General aviation
Research and development needs	Composition
Route planning	Size
Workforce requirements	Capacity
	Enplaned passengers
	Aircraft operations
Microforecast	
Airport facilities	Aircraft operations
Airside	Air carrier
Runways	Fleet mix
Taxiways	Capacity
Apron areas	Peak hour
Navigational aids	General-aviation-based aircraft
General aviation needs	
Landside	Passenger traffic
Gates	Enplaned/deplaned
Terminal facilities	Originating/terminating
Cargo needs	Connecting/transferring
	Airlines
Curb frontage	Peaking characteristics
Parking	Cargo activity
Internal road network	Vehicle traffic
Ground access	
Regional road network	
Public transit	

TABLE 5-2 Typical Air Transportation Forecast Variables and Their Use in Aviation and Airport Planning

The Airport System Plan

The purpose of an airport system plan is to identify the aviation development required to meet the immediate and future aviation needs of a region, state, or metropolitan area [17]. It recommends the general location and characteristics of new and expanded airports, shows the timing, phasing, and estimated cost of development, and identifies revenue sources and legislation for the implementation of the plan. The aviation system plan also provides a basis for the definitive and detailed development of the individual airports in the system.

The primary forecasting requirement for the airport system plan is a projection of the level of aviation activity during the planning period. The forecasts are usually made on an annual basis for the planning entity as a whole and are then proportioned among the various individual airports within this entity. Specific projections are generally made for total aircraft operations, air carrier and general aviation operations, based aircraft, total air cargo, and passenger enplanements for the short-, medium-, and long-range time frames within the planning period. These demand projections are compared to the inventory of physical facilities to determine development needs. It should be emphasized that these projections are normally made in a very aggregate fashion and tend to examine overall determinates of regional activity rather than the specific factors affecting local activity.

The preparation of a forecast for a system plan is initiated by the collection of historical data indicative of the various components of aviation activity. These data normally include broad measures of socioeconomic activity as well as aviation activity statistics. Due to the fact that data collection is very expensive, most of the data collection in a system plan depends primarily on secondary or existing sources and very little survey work is performed. Many of the overall regional projections are made on the basis of trend projections or simple econometric models and these are then apportioned to the individual airports within the region on the basis of ratio methods. In the analysis of observed trends and the preparation of future forecasts, broad indicators are generally used including an examination of the consistency and realization of past trends and a comparison of growth rates and economic indicators.

The Airport Master Plan

The purpose of the airport master plan is to provide the specific details for the future development of an individual airport to satisfy aviation needs consistent with community objectives. The airport master plan requires detailed projections of the level of demand on the various facilities associated with the airport. Various concepts and alternatives for development are examined and evaluated, and recommendations are made relative to the prioritizing, scheduling,

and financing of the plan [3]. Though the basis for projections in a system plan is the aggregate level of annual demand, this is not sufficient for the master plan. Projections must be made for magnitude, nature, and variation of demand on a monthly, daily, and hourly basis on the many facilities located at the airport.

Annual forecasts of airport traffic during the planning period are the basis for the preparation of the detailed forecasts in the master plan. These forecasts are made for each type of major airport user including air carrier, commuter, and general aviation aircraft and are often expressed in terms of upper and lower bounds. The master plan forecasts are usually made under both constrained and unconstrained conditions. An unconstrained forecast is one which is made relative to the potential aviation market in which the basic factors which tend to create aviation demand are utilized without regard to any constraining factors that could affect aviation growth at the location. A constrained forecast is one which is made in the context of alternative factors which could limit growth at the specific airport. Constraining factors addressed in master plans include limitations on airport capacity due to land availability and noise restrictions, the development of alternative reliever airports to attract general aviation demand, policies which alter access to airports by general aviation and commuter aircraft operations, and the availability and cost of aviation fuel. The determination of the level of general aviation activity at an airport can be significantly changed when land availability is restricted, thereby placing limits on airside capacity. The available capacity may be utilized in the context of policies which favor commercial over general aviation growth.

Specific forecasts made for a master plan include the annual, daily, and peak hour operations by air carrier, commuter, general aviation, cargo, and military aircraft, passenger enplanements, and annual cargo tonnage, as well as daily and hourly ground access system and parking demand. Projections are also made for the mix or types of aircraft in each of the categories which will utilize the airport during the planning period. In the preparation of these forecasts some variables are projected directly and others are derived from these projections. For example, annual passenger enplanements might be forecasted from an econometric model and then, based upon exogenous estimates of average air carrier fleet passenger capacity and boarding seat load factors, annual air carrier operations could be derived from these data. Guidance is provided by FAA [3, 10] for forecasting the various elements of the master plan and FAA recommends a tabular format for presenting forecasts for review. An example is presented in Table 5-3. Although the activity elements shown in the table refer to annual estimates, master plan forecasts will also require peak period activity levels for the planning of many airport facilities, and depending on the situation, seasonal, monthly, daily, and/or time-of-day demands must be forecast.

A. Forecast Levels and Growth Rates (Sample Data Shown)									
Specify Base Year: 2007									
	Base Yr Level	Base Yr + 1 yr	Base Yr + 5 yr	Base Yr + 10 yr	Base Yr + 15 yr	Average Annual Compound Growth Rates			
						Base Yr to +1	Base Yr to +5	Base Yr to +10	Base Yr to +15
Passenger Enplanements									
Air Carrier	868,981	904,400	1,021,000	1,273,000	1,587,000	4.1%	3.3%	3.9%	4.1%
Commuter	136,184	143,000	179,000	234,000	306,000	5.0%	5.6%	5.6%	5.5%
Total	1,005,165	1,047,400	1,200,000	1,507,000	1,893,000	4.2%	3.6%	4.1%	4.3%
Operations									
Itinerant									
Air carrier	25,155	25,700	28,000	33,600	40,000	2.2%	2.2%	2.9%	3.1%
Commuter/air taxi	18,100	18,800	22,000	24,700	28,000	3.9%	4.0%	3.2%	3.0%
Total Commercial Operations	43,225	44,500	50,000	58,300	68,000	2.9%	2.9%	3.0%	3.1%
General aviation	40,124	41,600	47,000	52,000	57,500	3.7%	3.2%	2.6%	2.4%
Military	3,124	3,124	3,124	3,124	3,124	0.0%	0.0%	0.0%	0.0%
Local									
General aviation	16,167	16,700	17,500	18,500	19,500	3.3%	1.6%	1.4%	1.3%
Military	2,436	2,436	2,436	2,436	2,436	0.0%	0.0%	0.0%	0.0%
Total Operations	105,106	108,360	120,060	134,360	150,560	3.1%	2.7%	2.5%	2.4%
Instrument Operations	206,910	209,000	220,000	230,000	241,000	1.3%	1.3%	1.1%	1.0%
Peak Hour Operations	40	42	44	47	50	5.0%	1.9%	1.6%	1.5%
Cargo/mall (Enplaned + Deplaned Tons)	16,800	18,010	23,100	30,200	39,500	7.2%	6.6%	6.0%	5.9%

Based Aircraft									
Single Engine (Nonjet)	90	91	93	94	95	1.1%	0.7%	0.4%	0.4%
Multi Engine (Nonjet)	14	15	20	25	30	7.1%	7.4%	6.0%	5.2%
Jet Engine	10	11	15	19	23	10.0%	8.4%	6.6%	5.7%
Helicopter	2	2	3	3	4	0.0%	8.4%	4.1%	4.7%
Other	0	0	0	0	0	0.0%	0.0%	0.0%	0.0%
Total	116	119	131	141	152	2.6%	2.5%	2.0%	1.8%
B. Operational Factors									
	Base Yr Level	Base Yr + 1 yr	Base Yr + 5 yr	Base Yr + 10 yr	Base Yr + 15 yr				
Average Aircraft Size (Seats)									
Air carrier	105.0	106.0	108.0	111.0	115.0				
Commuter	36.0	38.0	40.0	46.0	52.0				
Average Enplaning Load Factor									
Air carrier	65.8%	66.4%	67.5%	68.2%	69.0%				
Commuter	41.8%	40.0%	40.6%	41.2%	42.0%				
GA Operations per Based Aircraft	485	490	492	500	507				

Note: Show base plus one year if forecast was done. If planning effort did not include all forecast years shown interpolate years as needed, using average annual compound growth rates.

TABLE 5-3 Template for Summarizing and Documenting Airport Planning Forecasts

For the most part, the methods used in the study of new airports are similar to those used for existing airports. However, the principal difference is the inability of the analyst to obtain a local historical database to generate extrapolation trends, market shares, or econometric models. To overcome this deficiency, an attempt is usually made to forecast by drawing an analogy between the subject airport and other existing airports which demonstrate similar traffic experience, and which are located in areas possessing similar socioeconomic, demographic, and geographic characteristics. Forecasts are then made for the airport under consideration by using these airports as surrogates and adjustments are performed to accommodate expected differences between the airports. In the past, the Air Transport Association and the Federal Aviation Administration have collected and tabulated a significant amount of data for many airports. These data have included the number and distribution of commercial air carrier operations, fleet mix, and passengers on a peak and average monthly, daily, and hourly basis. It is apparent that one may expect a rather high degree of uncertainty associated with forecasting through such an analogy.

The Future Aviation Forecasting Environment

Many of the forecasts made by the various aviation-related organizations become biased by the impact of recent events. Forecasts made in the early 1960s showed rather moderate growth, whereas those made in the late 1960s showed fairly ambitious growth. These forecasts were made in the context of expectations which reflected the behavior of aviation at the time when they were made. In the 1970s and 1980s, the overall economic conditions, the availability and cost of petroleum-based fuels, and airline deregulation considerably affected aviation. Forecasts made in this era attempted to analyze the impact of these factors in projecting the demand for aviation in the future.

The Federal Aviation Administration and numerous other transportation organizations are always looking into the future of aviation [2, 12]. As far as future trends in air travel are concerned, it is expected that there will be a greater growth of international air traffic which has been attributed to the globalization of the airline industry and to changing market forces in the United States. The factors which have contributed to the rise of the U.S. air transportation industry are changing. The steady decline in the real cost of air travel has reached a point in which the unit costs in the 1990s will remain steady or slowly rise. The increased quality of service from improvements in the speed, comfort, convenience, and safety of air travel has been largely realized. Past demographic and cultural factors, such as the baby boom, are declining in importance. The rise in family discretionary income has peaked and the use of discretionary income for air travel is meeting

competition from mortgages, savings, and luxury goods. Foreign travel by U.S. citizens will be adversely affected by the decline in the exchange value of the dollar. Furthermore, this factor will encourage foreign nationals to travel to the United States [12, 13, 14].

Conferences related to aviation forecasting methodologies [10] concluded that financial concerns and forces concerns are forcing the aviation industry to place more emphasis on short-term forecasting methods. This is not only the case with the airlines but airports are also shortening their planning horizons. Throughout the aviation industry there is a shift to simpler forecasting techniques requiring fewer variables and less detailed data. A wider use is being made of forecasting techniques which depend less on mathematical modeling and more on an analysis of different scenarios, judgment, and market segmentation. Scenarios are used to test basic assumptions and to explore alternatives. Although judgment has always played a significant role in demand forecasting it is becoming more important as a subjective test of the reality associated with forecasting outcomes. There is a growing recognition that airline management strategies are important forces shaping the future of aviation development.

To adequately cope with the uncertainties associated with the traditional air transportation forecasting process and to react in a timely manner to inaccuracies found in estimates, the planning process is emerging into a phase-oriented, continuing process. For example, the FAA prepares annual forecasts of aviation activity on a national and terminal area basis which extend several years into the future [8, 9, 18]. Due to the high costs associated with the traditional planning process and the implementation of physical design changes, and the apparent inability to forecast with any degree of certainty, it is essential that planning techniques be developed which can respond to changes in the demand parameters prior to the investment decision. Perhaps the key to such a process is the recognition of the interaction of demand to supply parameters. The knowledge of the sensitivity of a physical facility component to a variation in demand can lead to more informed decisions and an understanding of the flexibility in facility design. A continued monitoring of the need for physical facilities in light of changing demand requirements provides a sound basis for the investment decision. Recognition of the uncertainties in the demand forecasting process can prevent a wasteful commitment of valuable resources. Explicit treatment of the variability of demand projections and facility modification recommendations though the use of sensitivity and tradeoff analyses is warranted.

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PART 2

Airport Design

CHAPTER 6
Geometric Design
of the Airfield

CHAPTER 7
Structural Design of
Airport Pavements

CHAPTER 8
Airport Lighting,
Marking, and Signage

CHAPTER 9
Airport Drainage

CHAPTER 10
Planning and Design
of the Terminal Area

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CHAPTER 6

Geometric Design of the Airfield

Airport Design Standards

In order to provide assistance to airport designers and a reasonable amount of uniformity in the design of airport facilities for aircraft operations, design guidelines have been prepared by the FAA [6] and the ICAO [2, 3, 4]. Any design criteria involving the widths, gradients, separations of runways, taxiways, and other features of the aircraft operations area must necessarily incorporate wide variations in aircraft performance, pilot technique, and weather conditions.

The FAA design criteria provide uniformity at airport facilities in the United States and serve as a guide to aircraft manufacturers and operators with regard to the facilities which may be expected to be available in the future. The FAA design standards are published in Advisory Circulars which are revised periodically as the need arises [1]. The ICAO strives toward uniformity and safety on an international level. Its standards, which are very similar to the FAA standards, apply to all member nations of the Convention on International Civil Aviation and are published as Annex 14 to that convention [2]. Requirements for military services are so specialized that they are not included in this chapter.

The design standards prepared by the FAA and the ICAO are presented in the text which follows under the general headings of airport classification, runways, taxiways, and aprons. The material is organized so that the various criteria may be readily compared. It is incumbent upon airport planners to review the latest specifications for airport design at the time studies are undertaken due to the fact that changes are incorporated as conditions dictate.

The FAA presents guidelines for airfield design in a series of Advisory Circulars. There are more than 200 Advisory Circulars pertaining to different aspects of airport planning and design, a complete list of which may be found on the FAA's website at <http://www.faa.gov>.

Advisory Circular 150/5300-13 “Airport Design” is the primary source of most airfield design standards. Originally published in 1989, AC 150/5300-13 has been updated 15 times as of 2010. The reader is encouraged to visit the FAA’s website for the latest updates to this and any Advisory Circulars when performing airport planning and design work, as they are updated often.

Airport Classification

For the purpose of stipulating geometric design standards for the various types of airports and the functions which they serve, letter and numerical codes and other descriptors have been adopted to classify airports.

For design purposes, airports are classified based on the aircraft they accommodate. While at any airport, a wide variety of aircraft, from small general aviation piston-engine aircraft to heavy air transport aircraft, will use the airfield, airports are designed based on a series of “critical” or “design” aircraft. These aircraft are selected from the fleet using the airport as those most critical to airfield design. The FAA defines the term *critical aircraft* as the aircraft most demanding on airport design that operates at least 500 annual itinerant operations at a given airport. In many cases, more than one critical aircraft will be selected at an airport for design purposes. For example, it is often the smallest aircraft that is critical to the orientation of runways, while the largest aircraft determines most of the other dimensional specifications of an airfield.

As described in Chap. 2, certain dimensional and performance characteristics of the critical aircraft determine the airport’s *airport reference code*. The airport reference code is a coding system used to relate the airport design criteria to the operational and physical characteristics of the aircraft intended to operate at the airport. It is based upon the *aircraft approach category* and the *airplane design group* to which the aircraft is assigned. The aircraft approach category, as shown in Table 6-1, is determined by the aircraft approach speed, which is defined as 1.3 times the stall speed in the landing configuration of aircraft at maximum certified landing weight [6].

The airplane design group (ADG) is a grouping of aircraft based upon wingspan or tail height, as shown in Table 6-2. An airplane design group for a particular aircraft is assigned based on the greater (higher Roman numeral) of that associated with the aircraft’s wingspan or tail height.

The airport reference code is a two designator code referring to the aircraft approach category and the airplane design group for which the airport has been designed. For example, an airport reference code of B-III is an airport designed to accommodate aircraft

Category	Approach Speed, kn
A	<91
B	91–120
C	121–140
D	141–166
E	>166

1 kn is approximately 1.15 mi/h

TABLE 6-1 Aircraft Approach Categories

with approach speeds from 91 to less than 121 kn (aircraft approach category B) with wingspans from 79 to less than 118 ft or tail heights from 30 to less than 45 ft (airplane design group III). The FAA publishes a list of the airport reference codes for various aircraft in Advisory Circular 150/5300-13 “Airport Design” [6].

As an example, an airport designed to accommodate the Boeing 767-200 which has an approach speed of 130 kn (aircraft approach category C) and a wingspan of 156 ft 1 in (airplane design group IV) would be classified with an airport reference code C-IV.

The ICAO uses a two-element code, the *aerodrome reference code*, to classify the geometric design standards at an airport [2, 3]. The code elements consist of a numeric and alphabetic designator. The aerodrome code numbers 1 through 4 classify the length of the runway available, the *reference field length*, which includes the runway length and, if present, the stopway and clearway. The reference field length is the approximate required runway takeoff length converted to an equivalent length at mean sea level, 15°C, and zero percent gradient. The aerodrome code letters A through E classify the wingspan and outer main gear wheel span for the aircraft for which the airport has been designed.

Group Number	Tail Height, ft	Wingspan, ft
I<20<49		
II	20 - <30	49 - <79
III	30 - <45	79-<118
IV	45 -<60	118 - <171
V	60 - <66	171 - <214
VI	66 - <80	214 - <262

TABLE 6-2 Aircraft Design Groups

Code Number	Reference Field Length, m	Code Letter	Wingspan, m	Distance between Outside Edges of Main Wheel Gear, m
1<800	A<15<4.			
2	800-<1200	B	15-<24	4.5-<6
3	1200-<1800	C	24-<36	6-<9
4≥1800	D	36-<52	9-<14	
		E	52-<65	9-<14
		F	65-<80	14-<16

TABLE 6-3 ICAO Aerodrome Reference Codes

These aerodrome reference codes are given in Table 6-3. For example, an airport which is designed to accommodate a Boeing 767–200 with an outer main gear wheel span of width of 34 ft 3 in (10.44 m), a wingspan of 156 ft 1 in (48 m), at a maximum takeoff weight of 317,000 lb, requiring a runway length of about 6000 ft (1830 m) at sea level on a standard day, would be classified by ICAO with an aerodrome reference code of 4-D. It will be noted that this classification system does not explicitly include the function of the airport, the service it renders, or the type of aircraft accommodated.

There is an approximate correspondence between the airport reference code of the FAA and the aerodrome reference code of the ICAO [2, 3]. The FAA's aircraft approach category of A, B, C, and D are approximately the same as the ICAO aerodrome code numbers 1, 2, 3, and 4, respectively. Similarly the FAA's airplane design groups of I, II, III, IV, and V approximately correspond to ICAO aerodrome code letters A, B, C, D, and E.

Utility Airports

Utility airport is defined as one which has been designed, constructed, and maintained to accommodate approach category A and B aircraft [6]. The specifications for utility airports are grouped for *small aircraft*, those of maximum certified takeoff weights of 12,500 lb or less, and *large aircraft*, those with maximum certified takeoff weight in excess of 12,500 lb.

Design specifications for utility airports are governed by the airplane design group and the types of approaches authorized for the airport runway, that is, visual, nonprecision instrument or precision instrument approaches.

Utility airports for small aircraft are called *basic utility stage I*, *basic utility stage II*, and *general utility stage I*. Utility airports for large aircraft

are called *general utility stage II*. Utility airports are further grouped for either visual and nonprecision instrument operations or precision instrument operations. The visual and nonprecision instrument operation utility airports are the basic utility stage I, basic utility stage II, or general utility stage I airports. The precision instrument operation utility airport is the general utility stage II airport.

A basic utility stage I airport has the capability of accommodating about 75 percent of the single engine and small twin engine aircraft used for personal and business purposes. This generally means aircraft weighing on the order of 3000 lb or less is given the airport reference code B-I, which indicates that it accommodates aircraft in aircraft approach categories A and B and aircraft in airplane design group I. A basic utility stage II airport has the capability of accommodating all of the airplanes of a basic utility stage I airport plus some small business and air taxi-type airplanes. This generally means aircraft weighing on the order of 8000 lb or less is also given the airport reference code B-I. A general utility stage I airport accommodates all small aircraft. It is assigned the airport reference code of B-II. A general utility stage II airport serves large airplanes in aircraft approach categories A and B and usually has the capability for precision instrument operations. It is assigned the airport reference code of B-III.

Transport Airports

A *transport airport* is defined as an airport which is designed, constructed, and maintained to accommodate aircraft in approach categories C, D, and E [6]. The design specifications of transport airports are based upon the airplane design group.

Runways

A runway is a rectangular area on the airport surface prepared for the takeoff and landing of aircraft. An airport may have one runway or several runways which are sited, oriented, and configured in a manner to provide for the safe and efficient use of the airport under a variety of conditions. Several of the factors which affect the location, orientation, and number of runways at an airport include local weather conditions, particularly wind distribution and visibility, the topography of the airport and surrounding area, the type and amount of air traffic to be serviced at the airport, aircraft performance requirements, and aircraft noise [2].

Runway Configurations

The term “runway configuration” refers to the number and relative orientations of one or more runways on an airfield. Many runway configurations exist. Most configurations are combinations of several

basic configurations. The basic configurations are (1) single runways, (2) parallel runways, (3) intersecting runways, and (4) open-V runways.

Single Runway

This is the simplest of the runway configurations and is shown in Fig. 6-1. It has been estimated that the hourly capacity of a single runway in VFR conditions is somewhere between 50 and 100 operations per hour, while in IFR conditions this capacity is reduced to 50 to 70 operations per hour, depending on the composition of the aircraft mix and navigational aids available [4].

Parallel Runways

The capacities of parallel runway systems depend on the number of runways and on the spacing between the runways. Two, three, and four parallel runways are common. The spacing between parallel runways varies widely. For the purpose of this discussion, the spacing is classified as close, intermediate, and far, depending on the center-line separation between two parallel runways. Close parallel runways are spaced from a minimum of 700 ft (for air carrier airports) to less than 2500 ft [5]. In IFR conditions an operation of one runway is dependent on the operation of other runway. Intermediate parallel runways are spaced between 2500 ft to less than 4300 ft [5]. In IFR conditions an arrival on one runway is independent of a departure on the other runway. Far parallel runways are spaced at least 4300 ft apart [5]. In IFR conditions the two runways can be operated independently for both arrivals and departures. Therefore,

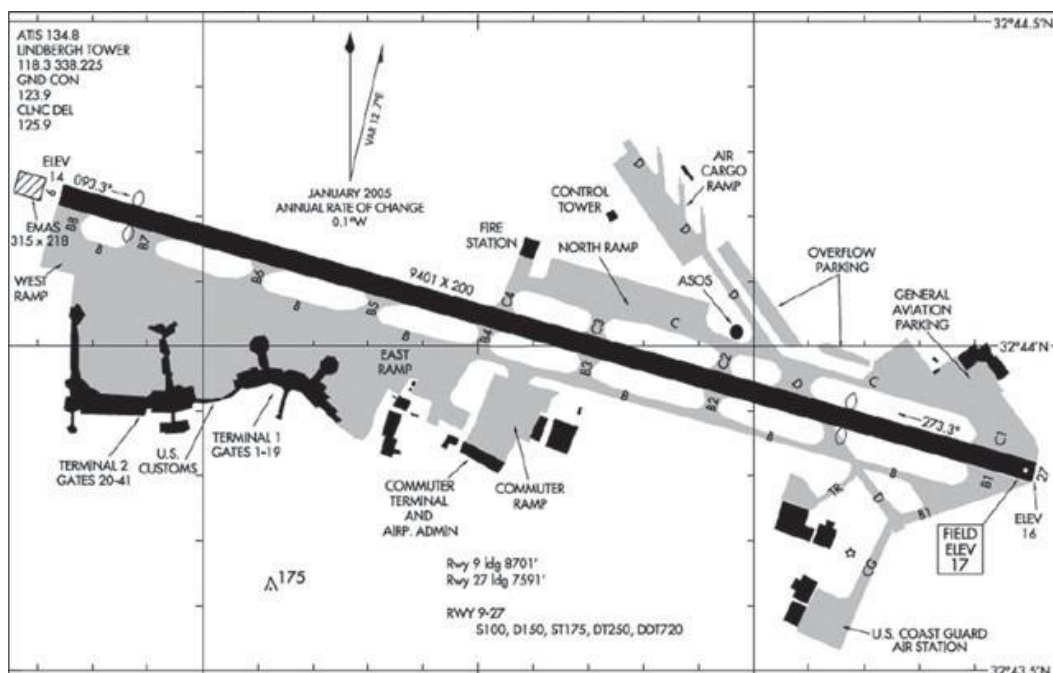


FIGURE 6-1 Single runway configuration: San Diego International Airport (NOAA Approach Charts).

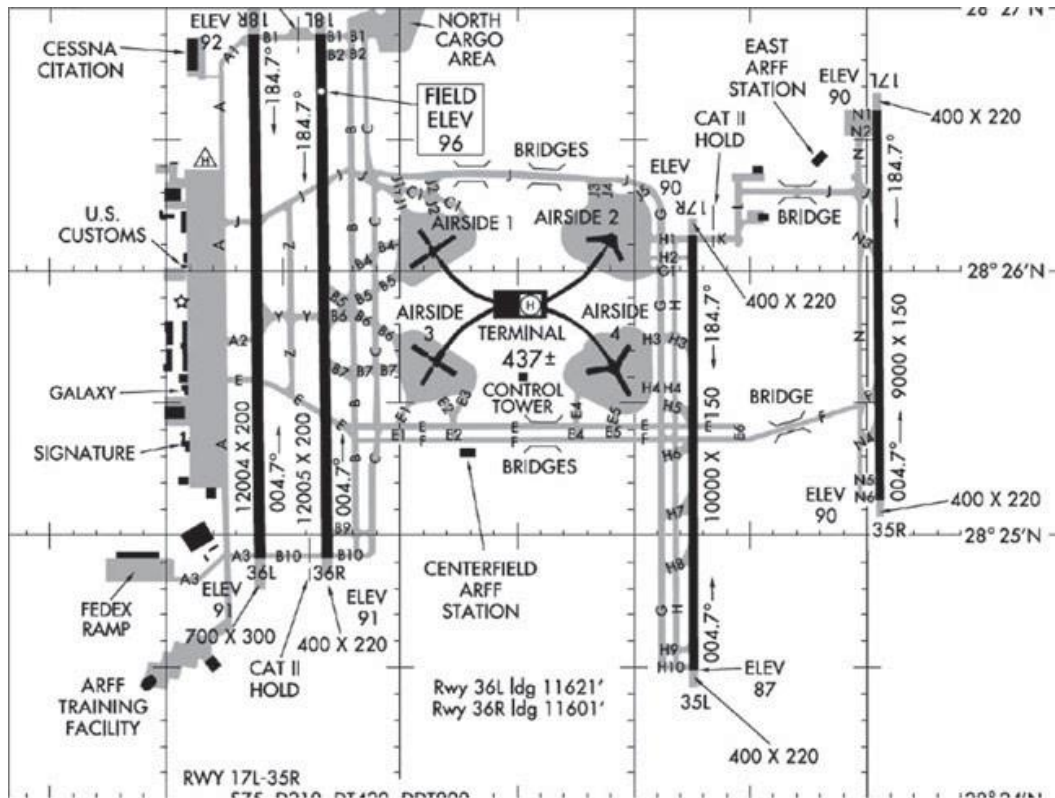


FIGURE 6-2 Example of parallel runway configuration: Orlando International Airport.

as noted earlier, the centerline separation of parallel runways determines the degree of interdependence between operations on each of the parallel runways. It should be recognized that in future the spacing requirements for simultaneous operations on parallel runways may be reduced. If this occurs, new spacing can be applied to the same classifications. Figure 6-2 illustrates an airport with multiple parallel runways with various spacing.

If the terminal buildings are placed between parallel runways, runways are always spaced far enough apart to allow room for the buildings, the adjoining apron, and the appropriate taxiways. When there are four parallel runways, each pair is spaced close, but the pairs are spaced far apart to provide space for terminal buildings.

In VFR conditions, close parallel runways allow simultaneous arrivals and departures, that is, arrivals may occur on one runway while departures are occurring on the other runway. Aircraft operating on the runways must have wingspans less than 171 ft (airplane design groups I through IV, see Table 6-2) for centerline spacing at the minimum of 700 ft [5]. If larger wingspan aircraft are operating on these runways (airplane design groups V and VI), the centerline spacing must be at least 1200 ft for such simultaneous operations [5]. In either case, wake vortex avoidance procedures must be used for simultaneous operations on closely spaced parallel runways. Furthermore, simultaneous arrivals to both runways or simultaneous departures

from both runways are not allowed in VFR conditions for closely spaced parallel runways. In IFR conditions, closely spaced parallel runways cannot be used simultaneously but may be operated as dual-lane runways.

Intermediate parallel runways may be operated with simultaneous arrivals in VFR conditions. Intermediate parallel runways may be operated in IFR conditions with simultaneous departures in a non-radar environment if the centerline spacing is at least 3500 ft and in a radar environment if the centerline spacing is at least 2500 ft [5]. Simultaneous arrivals and departures are also permitted if the centerline spacing is at least 2500 ft if the thresholds of the runways are not staggered [5]. There are times when it may be desirable to stagger the thresholds of parallel runways. The staggering may be necessary because of the shape of the acreage available for runway construction, or it may be desirable for reducing the taxiing distance of takeoff and landing aircraft. The reduction in taxiing distance, however, is based on the premise that one runway is to be used exclusively for takeoff and the other for landing. In this case the terminal buildings are located between the runways so that the taxiing distance for each type of operation (takeoff or landing) is minimized. If the runway thresholds are staggered, adjustments to the centerline spacing requirement are allowed for simultaneous arrivals and departures [5]. If the arrivals are on the near threshold then the centerline spacing may be reduced by 100 ft for each 500 ft of threshold stagger down to a minimum centerline separation of 1000 ft for aircraft with wingspans up to 171 ft and a minimum of 1200 ft for larger wingspan aircraft. If the arrivals are on the far threshold the centerline spacing must be increased by 100 ft for each 500 ft of threshold stagger. Simultaneous arrivals in IFR conditions are not permitted on intermediate parallel runways but are permitted on far parallel runways with centerline spacings of at least 4300 ft [5].

The hourly capacity of a pair of parallel runways in VFR conditions varies greatly from 60 to 200 operations per hour depending on the aircraft mix and the manner in which arrivals and departures are processed on these runways [4]. Similarly, in IFR conditions the hourly capacity of a pair of closely spaced parallel runways ranges from 50 to 60 operations per hour, of a pair of intermediate parallel runways from 60 to 75 operations per hour, and for a pair of far parallel runways from 100 to 125 operations per hour [4].

A dual-lane parallel runway consists of two closely spaced parallel runways with appropriate exit taxiways. Although both runways can be used for mixed operations subject to the conditions noted above, the desirable mode of operation is to dedicate the runway farthest from the terminal building (outer) for arrivals and the runway closest to the terminal building (inner) for departures. It is estimated that a dual-lane runway can handle at least 70 percent more traffic than a single runway in VFR conditions and about 60 percent more

traffic than a single runway in IFR conditions. It is recommended that the two runways be spaced not less than 1000 ft apart (1200 ft, where particularly larger wingspan aircraft are involved). This spacing also provides sufficient distance for an arrival to stop between the two runways. A parallel taxiway between the runways will provide for a nominal increase in capacity, but is not essential. The major benefit of a dual-lane runway is to provide an increase in IFR capacity with minimal acquisition of land [7, 14].

Intersecting Runways

Many airports have two or more runways in different directions crossing each other. These are referred to as intersecting runways. Intersecting runways are necessary when relatively strong winds occur from more than one direction, resulting in excessive crosswinds when only one runway is provided. When the winds are strong, only one runway of a pair of intersecting runways can be used, reducing the capacity of the airfield substantially. If the winds are relatively light, both runways can be used simultaneously. The capacity of two intersecting runways depends on the location of the intersection (i.e., midway or near the ends), the manner in which the runways are operated for takeoffs and landings, referred to as the runway use strategy, and the aircraft mix. The farther the intersection is from the takeoff end of the runway and the landing threshold, the lower is the capacity. The highest capacity is achieved when the intersection is close to the takeoff and landing threshold. Figure 6-3 provides an example of intersecting runways with the intersection closer to the runway thresholds.

Open-V Runways

Runways in different directions which do not intersect are referred to as open-V runways. This configuration is shown in Fig. 6-4. Like intersecting runways, open-V runways revert to a single runway when winds are strong from one direction. When the winds are light, both runways may be used simultaneously.

The strategy which yields the highest capacity is when operations are away from the V and this is referred to as a diverging pattern. In VFR the hourly capacity for this strategy ranges from 60 to 180 operations per hour, and in IFR the corresponding capacity is from 50 to 80 operations per hour [4]. When operations are toward the V it is referred to as a converging pattern and the capacity is reduced to 50 to 100 operations per hour in VFR and to between 50 and 60 operations per hour in IFR [4].

Combinations of Runway Configurations

From the standpoint of capacity and air traffic control, a single-direction runway configuration is most desirable. All other things being equal,

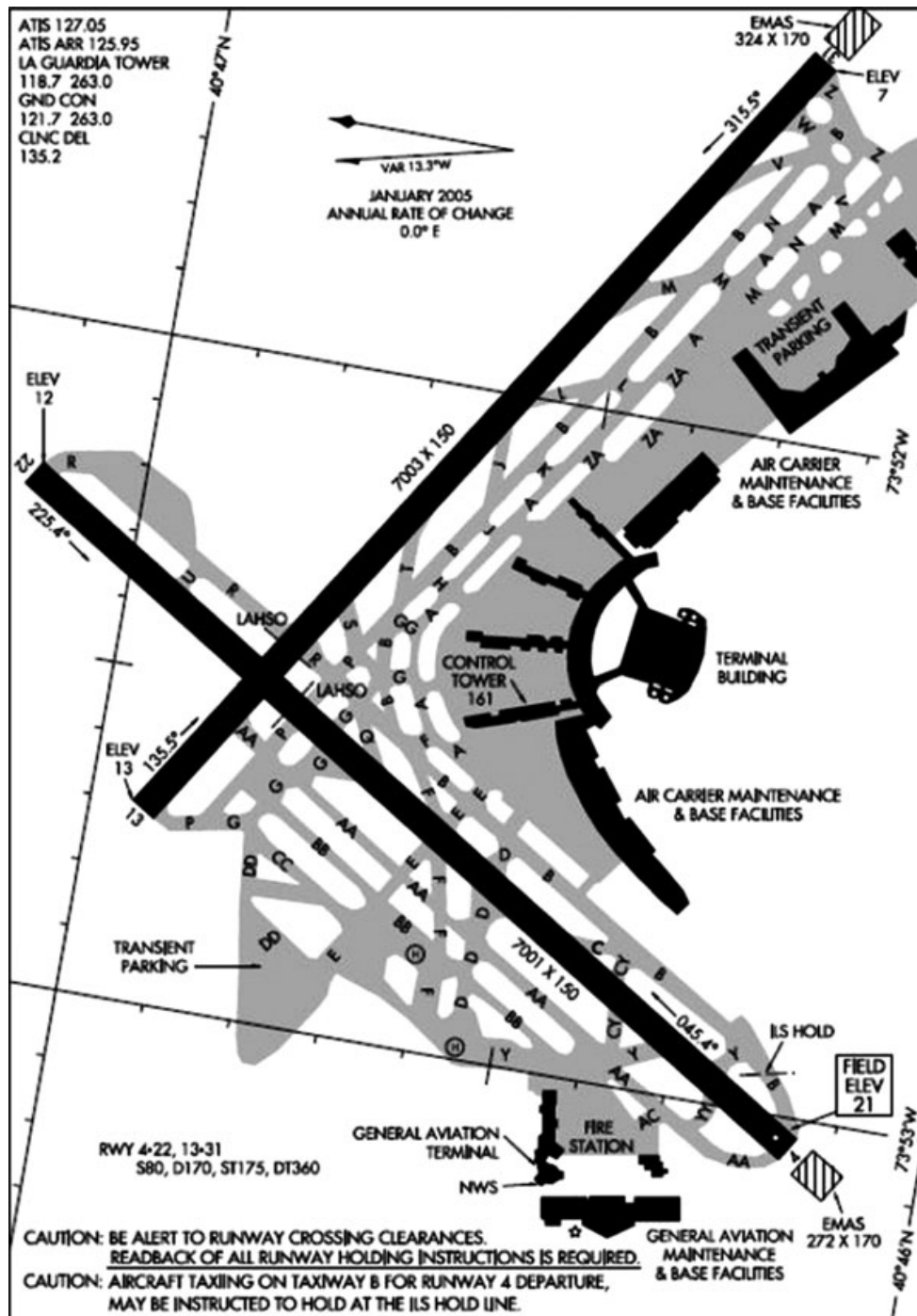


FIGURE 6-3 Example of intersecting runways: LaGuardia Airport, New York.

this configuration will yield the highest capacity compared with other configurations. For air traffic control the routing of aircraft in a single direction is less complex than routing in multiple directions. Comparing the divergent configurations, the open-V runway pattern is more desirable than an intersecting runway configuration. In the open-V configuration an operating strategy that routes aircraft away from the V will yield higher capacities than if the operations are reversed. If intersecting runways cannot be avoided, every effort

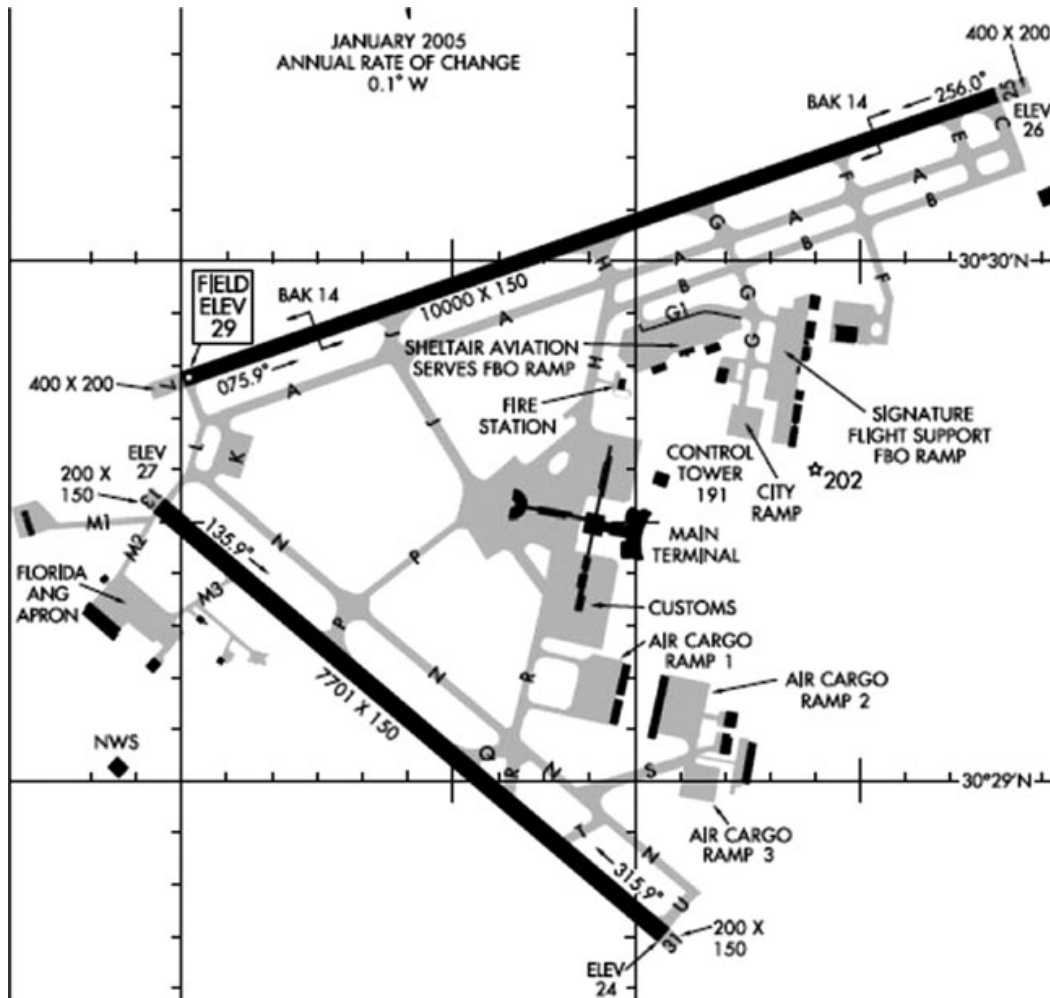


FIGURE 6-4 Example of open-V runways: Jacksonville International Airport.

should be made to place the intersections of both runways as close as possible to their thresholds and to operate the aircraft away from the intersection rather than toward the intersection.

Figure 6-5 illustrates the complex runway configuration of Chicago's O'Hare Field, with multiple parallel, intersecting, and non-intersecting runways. It should be noted that a large capital improvement program is being undertaken to simplify the runway configuration, by adding additional parallel runways and removing many intersecting runways. This runway redesign is being done with the intention of improving the capacity and efficiency of airport operations at the airport. The runway configuration redesign is illustrated in Fig. 6-6.

Runway Orientation

The orientation of a runway is defined by the direction, relative to magnetic north, of the operations performed by aircraft on the runway. Typically, but not always, runways are oriented in such a manner that they may be used in either direction. It is less preferred to

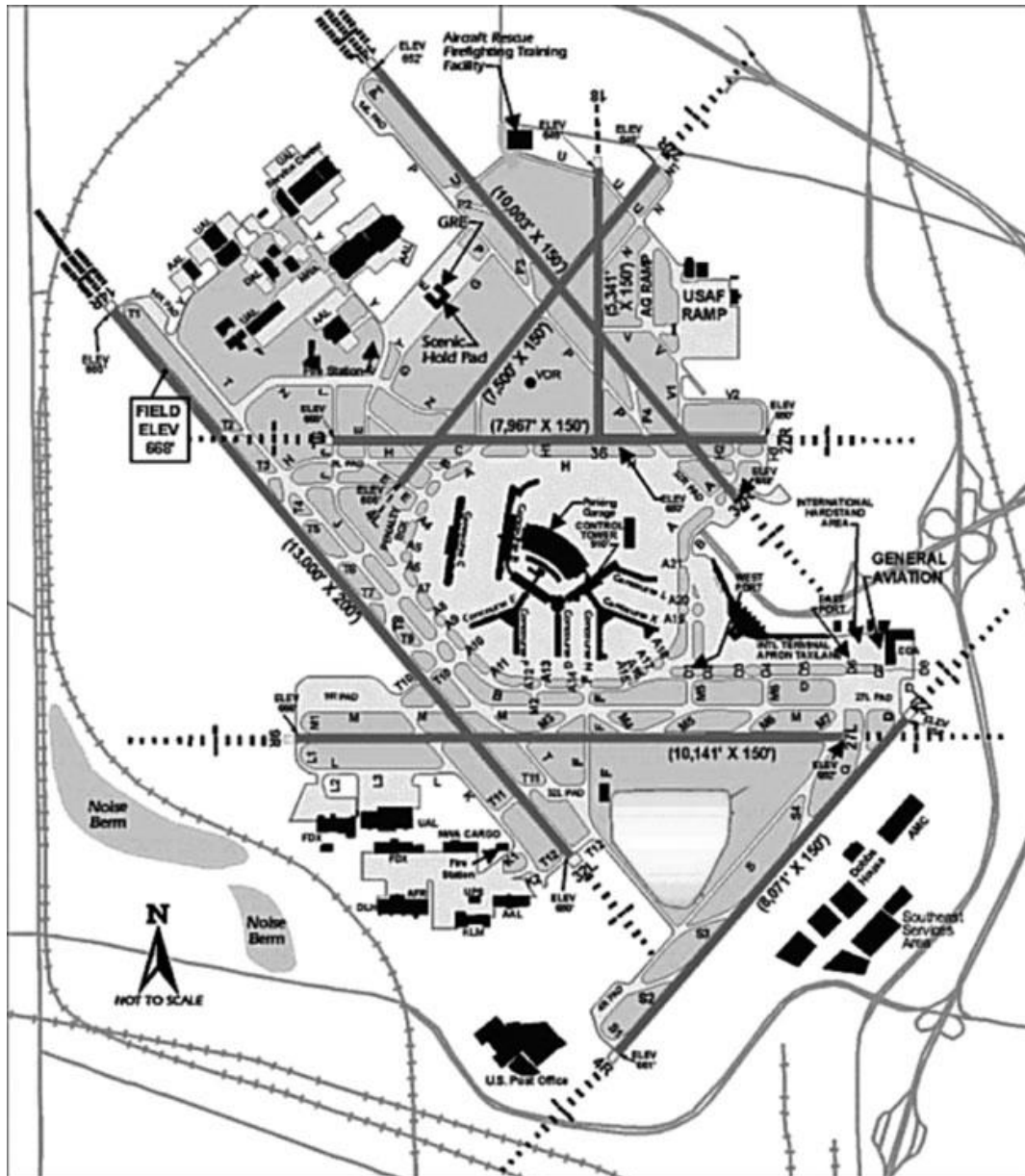


FIGURE 6-5 Example of complex runway system: Chicago O'Hare International Airport.

orient a runway in such a way that operating in one direction is precluded, normally due to nearby obstacles.

In addition to obstacle clearance considerations, which will be discussed later in this chapter, runways are typically oriented based on the area's wind conditions. As such, an analysis of wind is essential for planning runways. As a general rule, the primary runway at an airport should be oriented as closely as practicable in the direction of the prevailing winds. When landing and taking off, aircraft are able to maneuver on a runway as long as the wind component at right angles to the direction of travel, the crosswind component, is not excessive.

The FAA recommends that runways should be oriented so that aircraft may be landed at least 95 percent of the time with

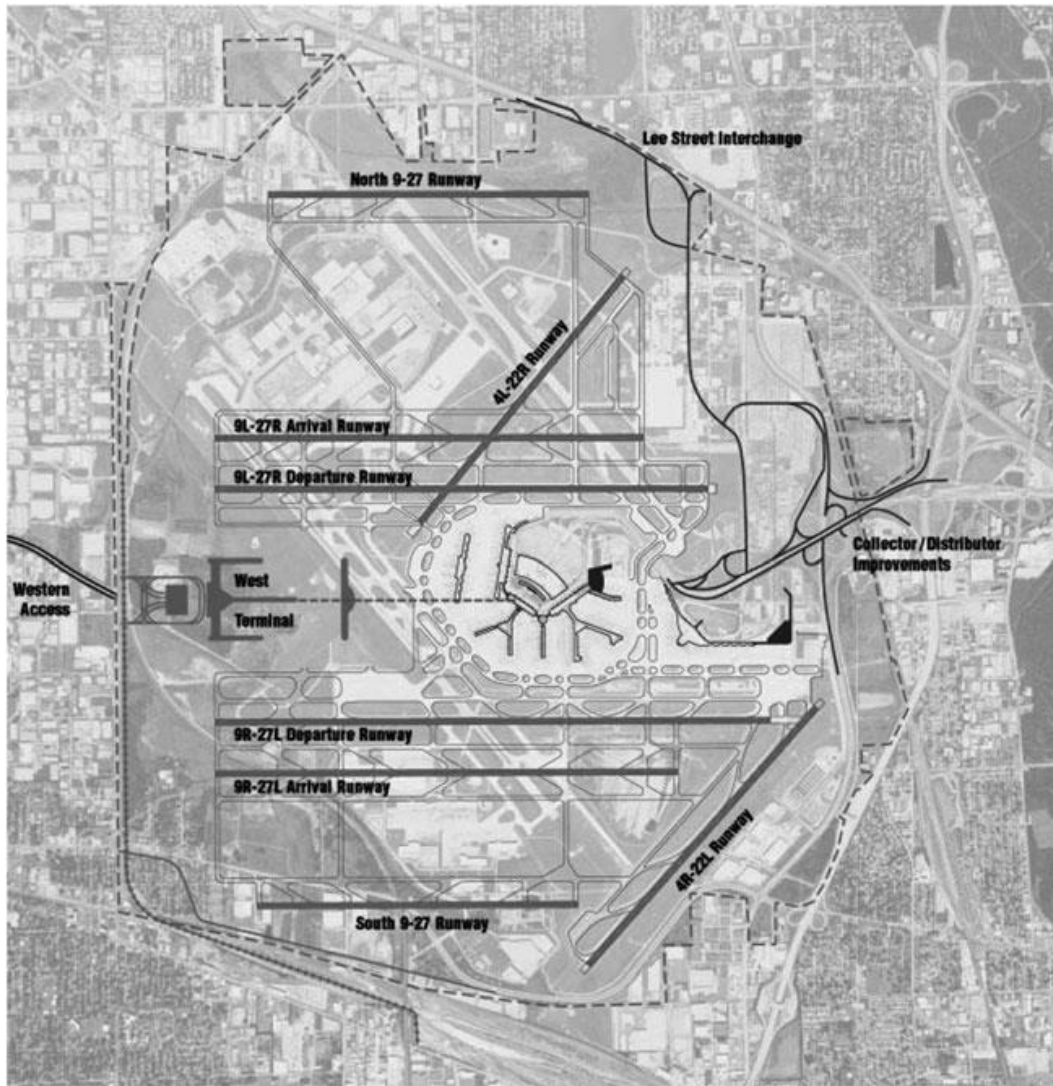


FIGURE 6-6 Planned simplified runway configuration: Chicago O'Hare International Airport (Courtesy Chicago O'Hare Modernization program).

allowable crosswind components not exceeding specified limits based upon the airport reference code associated with the critical aircraft that has the shortest wingspan or slowest approach speed. When the wind coverage is less than 95 percent a crosswind runway is recommended.

The allowable crosswind is 10.5 kn (12 mi/h) for Airport Reference Codes A-I and B-I, 13 kn (15 mi/h) for Airport Reference Codes A-II and B-II, 16 kn (18.5 mi/h) for Airport Reference Codes A-III, B-III, C-I, C-II, C-III and C-IV, and 20 knots (23 mph) for Airport Reference Codes A-IV through D-VI [5].

ICAO also specifies that runways should be oriented so that aircraft may be landed at least 95 percent of the time with crosswind components of 20 kn (23 mph) for runway lengths of 1500 m more, 13 kn (15 mi/h) for runway lengths between 1200 and 1500 m, and 10 kn (11.5 mi/h) for runway lengths less than 1200 m [1, 2].

Once the maximum permissible crosswind component is selected, the most desirable direction of runways for wind coverage can be determined by examination of the average wind characteristics at the airport under the following conditions:

1. The entire wind coverage regardless of visibility or cloud ceiling
2. Wind conditions when the ceiling is at least 1000 ft and the visibility is at least 3 mi
3. Wind conditions when ceiling is between 200 and 1000 ft and/or the visibility is between $\frac{1}{2}$ and 3 mi.

The first condition represents the entire range of visibility, from excellent to very poor, and is termed the all weather condition. The next condition represents the range of good visibility conditions not requiring the use of instruments for landing, termed visual meteorological condition (VMC). The last condition represents various degrees of poor visibility requiring the use of instruments for landing, termed instrument meteorological conditions (IMC).

The 95 percent criterion suggested by the FAA and ICAO is applicable to all conditions of weather; nevertheless it is still useful to examine the data in parts whenever this is possible.

In the United States, weather records can be obtained from the Environmental Data and Information Service of the National Climatic Center at the National Oceanic and Atmospheric Administration located in Ashville, N.C., or from various locations found on the Internet.

Weather data are collected from weather stations throughout the United States on an hourly basis and recorded for analysis. The data collected include ceiling, visibility, wind speed, wind direction, storms, barometric pressure, the amount and type of liquid and frozen precipitation, temperature, and relative humidity. A report illustrating the tabulation and representation of some of the data of use in airport studies was prepared for the FAA [15]. The weather records contain the percentage of time certain combinations of ceiling and visibility occur (e.g., ceiling, 500 to 900 ft; visibility, 3 to 6 mi), and the percentage of time winds of specified velocity ranges occur from different directions (e.g., from NNE, 4 to 7 mi/h). The directions are referenced to true north.

The Wind Rose

The appropriate orientation of the runway or runways at an airport can be determined through graphical vector analysis using a wind rose. A standard wind rose consists of a series of concentric circles cut by radial lines using polar coordinate graph paper. The radial lines are drawn to the scale of the wind magnitude such that the area between each pair of successive lines is centered on the wind direction.

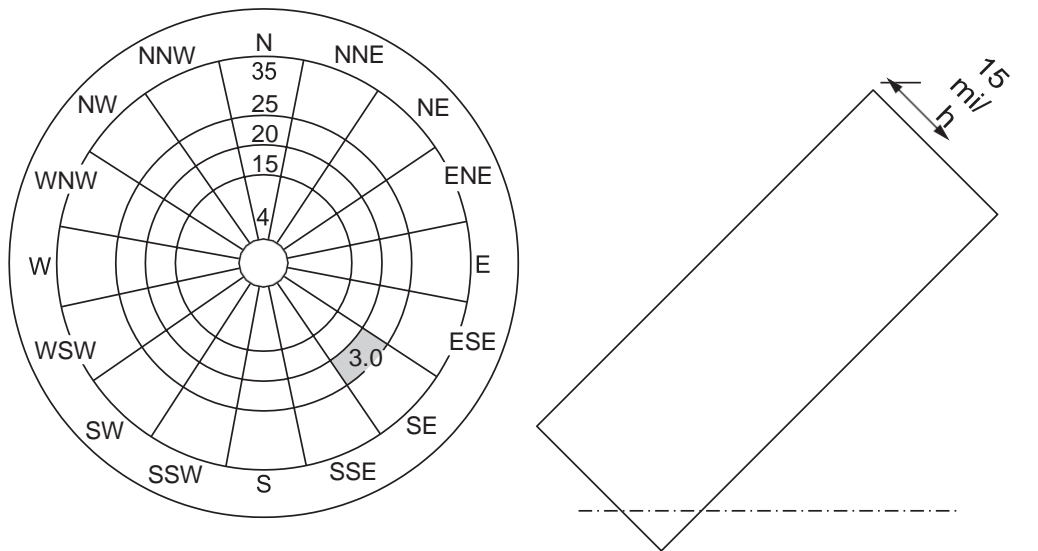


FIGURE 6-7 Wind rose coordinate system and template.

A typical wind rose polar coordinate system is shown on the left side of Fig. 6-7. The shaded area indicates that the wind comes from the southeast (SE) with a magnitude between 20 and 25 mi/h. A template is also drawn to the same radial scale representing the crosswind component limits. A template drawn with crosswind component limits of 15 mi/h is shown on the right side of Fig. 6-7. On this template three equally spaced parallel lines have been plotted. The middle line represents the runway centerline, and the distance between the middle line and each outside line is, to scale, the allowable crosswind component (in this case, 15 mi/h). The template is placed over the wind rose in such a manner that the centerline on the template passes through the center of the wind rose.

By overlaying the template on the wind rose and rotating the centerline of the template through the origin of the wind rose one may determine the percentage of time a runway in the direction of the centerline of the template can be used such that the crosswind component does not exceed 15 mi/h. Optimum runway directions can be determined from this wind rose by the use of the template, typically made on a transparent strip of material. With the center of the wind rose as a pivot point, the template is rotated until the sum of the percentages included between the outer lines is a maximum. If a wind vector from a segment lies outside either outer line on the template for the given direction of the runway, that wind vector must have a crosswind component which exceeds the allowable crosswind component plotted on the template. When one of the outer lines on the template divides a segment of wind direction, the fractional part is estimated visually to the nearest 0.1 percent. This procedure is consistent with the accuracy of the wind data and assumes that the wind percentage within the sector is uniformly distributed within that sector. In practice, it is usually easier to add the percentages contained in the

sectors outside of the two outer parallel lines and subtract these from 100 percent to find the percentage of wind coverage.

Example Problem 6-1 As an example, assume that the wind data for all conditions of visibility are those shown in Table 6-4. This wind data is plotted to scale as indicated above to obtain a wind rose, as shown in Fig. 6-8.

The percentage of time the winds correspond to a given direction and velocity range is marked in the proper sector of the wind rose by means of a polar coordinate scale for both wind direction and wind magnitude. The template is rotated about the center of the wind rose, as explained earlier, until the direction of the centerline yields the maximum percentage of wind between the parallel lines.

Once the optimum runway direction has been found in this manner, the next step is to read the bearing of the runway on the outer scale of the wind rose where the centerline on the template crosses the wind direction scale. Because true north is used for published wind data, this bearing usually will be different

Sector	True Azlmuth	Wind Speed Range, ml/h				Total
		4-15	15-20	20-25	25-35	
		Percentage of Time				
N	0.0	2.4	0.4	0.1	0.0	2.9
NNE	22.5	3.0	1.2	1.0	0.5	5.7
NE	45.0	5.3	1.6	1.0	0.4	8.3
ENE	67.5	6.8	3.1	1.7	0.1	11.7
E	90.0	7.1	2.3	1.9	0.2	11.5
ESE	112.5	6.4	3.5	1.9	0.1	11.9
SE	135.0	5.8	1.9	1.1	0.0	8.8
SSE	157.5	3.8	1.0	0.1	0.0	4.9
S	180.0	1.8	0.4	0.1	0.0	2.3
SSW	202.5	1.7	0.8	0.4	0.3	3.2
SW	225.0	1.5	0.6	0.2	0.0	2.3
WSW	247.5	2.7	0.4	0.1	0.0	3.2
W	270.0	4.9	0.4	0.1	0.0	5.4
WNW	292.5	3.8	0.6	0.2	0.0	4.6
NW	315.0	1.7	0.6	0.2	0.0	2.5
NNW	337.5	1.7	0.9	0.1	0.0	2.7
Subtotal		60.4	19.7	10.2	1.6	91.9
Calms						8.1
Total						100.0

TABLE 6-4 Example Wind Data

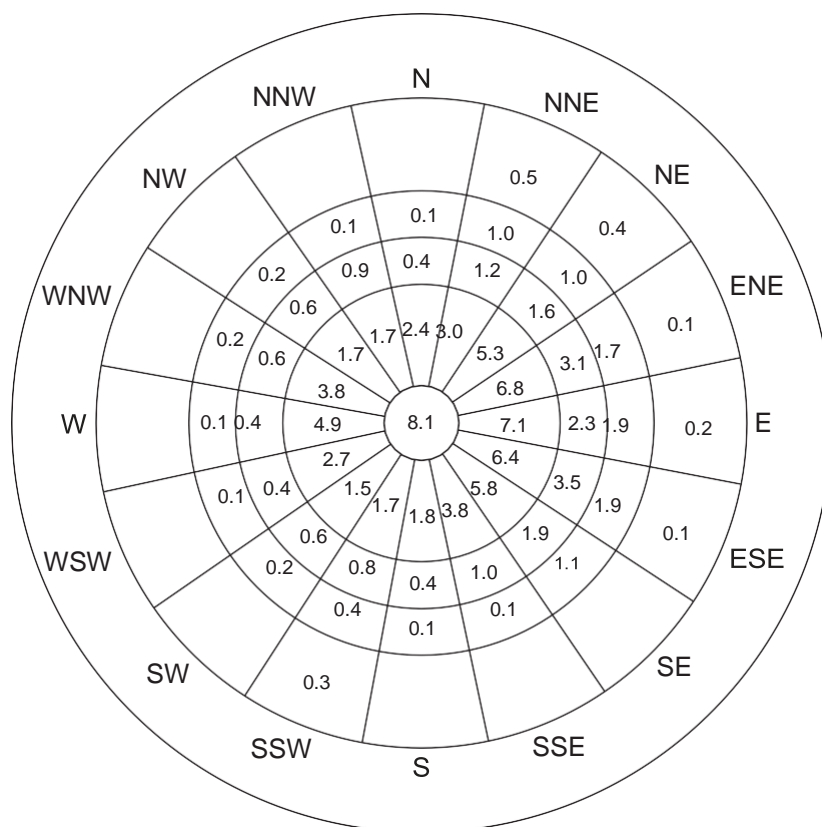


FIGURE 6-8 Wind data in wind rose format.

from that used in numbering runways since runway designations are based on the magnetic bearing. As illustrated in Fig. 6-9, a runway oriented on an azimuth to true north of 90° to 270° (N 90° E to S 90° W true bearing) will permit operations 90.8 percent of the time with the crosswind components not exceeding 15 mi/h.

Should the wind analysis not give the desired wind coverage, the template may then be used to determine the direction of a second runway, a crosswind runway, which would increase the wind coverage to 95 percent. This is done by blocking out the area between the two outer parallel lines for the direction of the primary runway (since this has already been counted in the wind coverage for the primary runway) and rotating the template until the percentages between the outer parallel lines for the remaining area for another direction is maximized.

If this is done in this problem it is found that the crosswind runway should be located in an orientation of 12° to 192° (N 12° E to S 12° W true bearing). This will permit an additional wind coverage of 6.2 percent above that provided by the runway oriented 90° to 270° for a total wind coverage for both runways of 97.0 percent.

Let us say that because of noise-sensitive land uses in the direction of the optimal crosswind runway, a crosswind runway will be located at the airport in the orientation of 30° to 210° direction which results in an additional wind coverage of 5.8 percent. This runway orientation, called runway 3-21, is shown in Fig. 6-10. The total wind coverage for both runways is then 96.6 percent. The total wind coverage for a runway in the orientation of 30° to 210° direction is found to be 84.8 percent from Fig. 6-11. The combined wind coverage of 96.6 percent for the use of either runway is shown in Fig. 6-11.

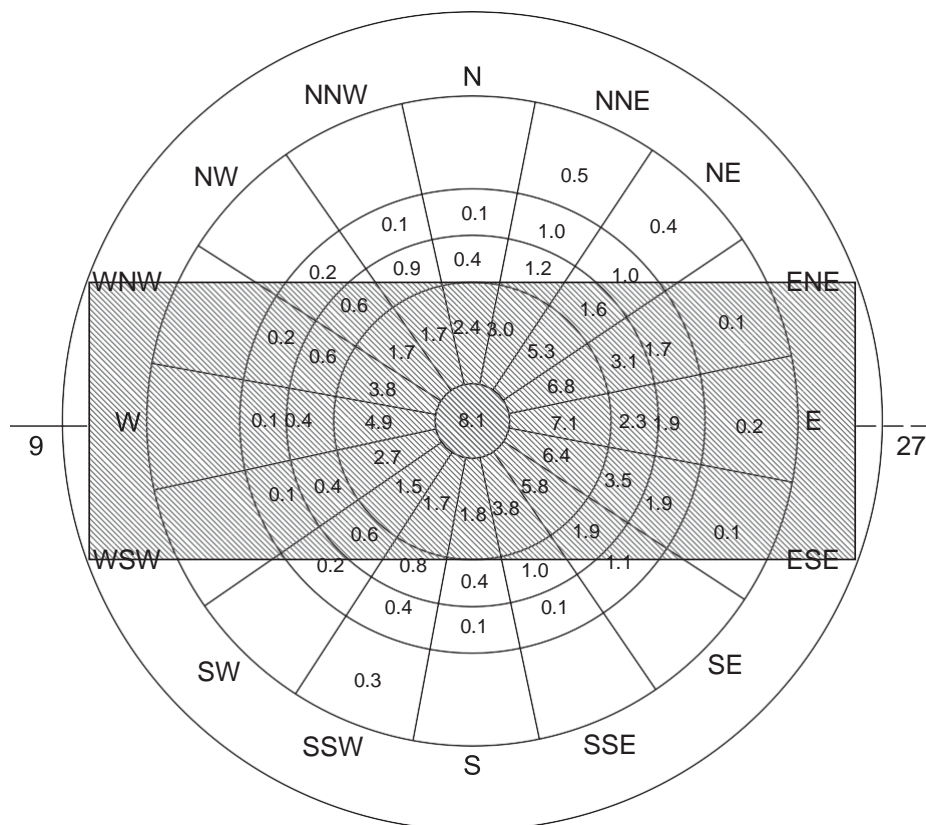


FIGURE 6-9 Wind coverage for runway 9-27, Example Problem 6-1.

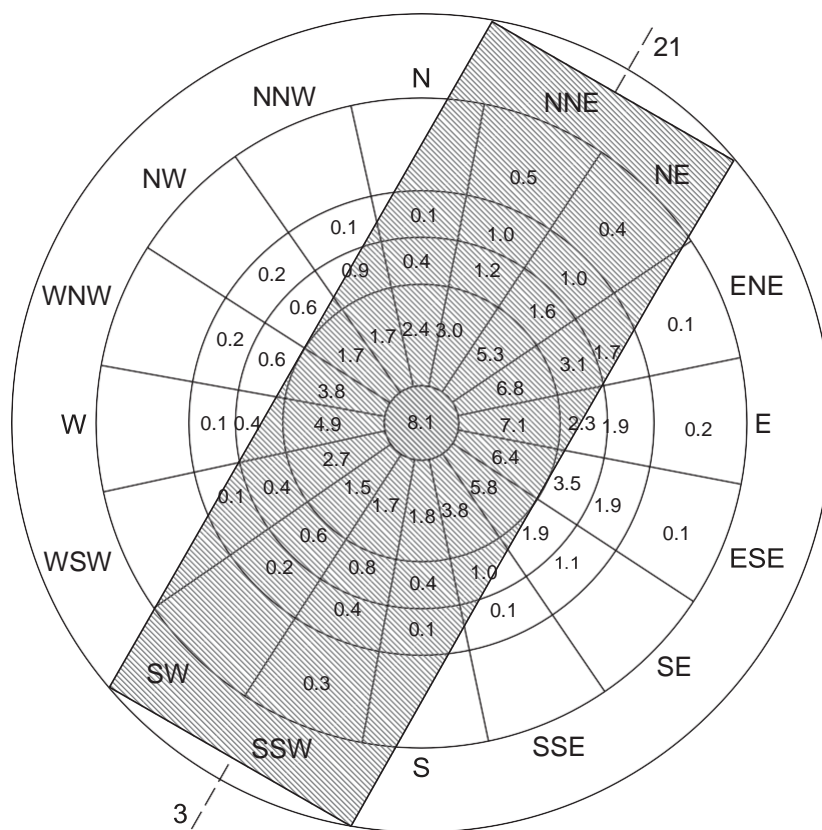


FIGURE 6-10 Wind coverage for runway 3-21, Example Problem 6-1.

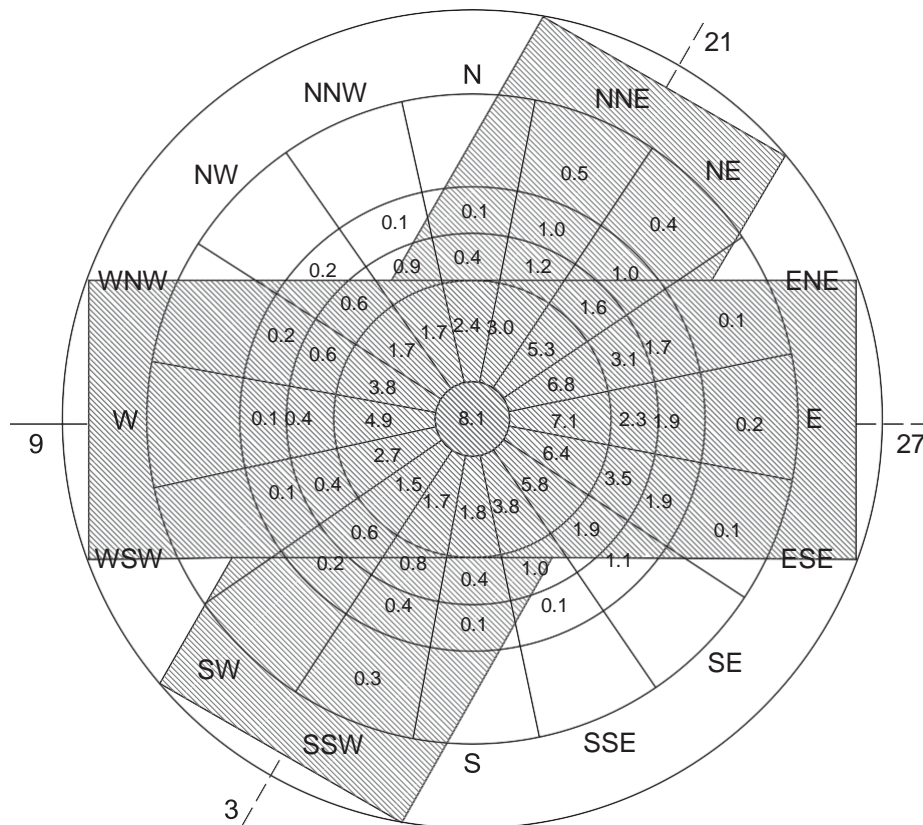


FIGURE 6-11 Wind coverage for runways 9-27 and 3-21, Example Problem 6-1.

Estimating Runway Length

Other than orientation, planning and designing the length of a runway is critical to whether or not a particular aircraft can safely use the runway for takeoff or landing. Furthermore, designing a runway to accommodate a given aircraft is a difficult task, given the fact that an aircraft's required runway length will vary based on aircraft weight, as well as on several ambient conditions.

As a guide to airport planners, the FAA has published Advisory Circular 150/5325-4b, "Runway Length Requirements for Airport Design" [17]. In this publication, procedures are defined for estimating the design runway length of aircraft, based on their maximum takeoff weights (MTOW), certain aircraft performance specifications, and the airport's field elevation and temperature. The airport design runway length is found for the critical aircraft, defined as the aircraft which flies the greatest nonstop route segment from the airports at least 500 operations per year and requires the longest runway. The FAA's procedure for estimating runway length is based on the following data:

1. Designation of a critical aircraft
2. The maximum takeoff weight of the critical aircraft at the airport

3. The airport elevation
4. The mean daily maximum temperature for the hottest month at the airport
5. The maximum different in elevation along the runway centerline.

For the purposes of estimating runway length requirements, the FAA groups aircraft by MGTOW. Based on the MGTOW of the critical aircraft, the following procedures are defined:

Aircraft Less than 12,500 lb MGTOW

Critical aircraft less than or equal to 12,500 lb MGTOW are considered “small airplanes” for the purposes of estimating runway length requirements. For these small aircraft, design runway length is based on the aircraft’s reference approach speed, V_{ref} .

Aircraft with $V_{ref} < 30$ kn are considered short takeoff and landing (STOL) aircraft. The design runway length for STOL aircraft is 300 ft (92 m) at sea level. For airports at elevation above sea level, the design runway length is 300 ft plus 0.03 ft for every foot above sea level.

For aircraft with $30 \leq V_{ref} < 50$ kn, the design runway length at sea level is 800 ft (244 m). For airports at elevation above sea level, the design runway length is 800 ft plus 0.08 ft for every foot above sea level.

For aircraft with $V_{ref} \geq 50$ kn, the design runway length is based on the number of passenger seats in the aircraft. For those aircraft with less than 10 passenger seats, Fig. 6-12 is referenced. This figure has two sets of curves, one representing “95 percent of fleet,” to be used at airports serving small communities, and one representing “100 percent of fleet,” to be applied at airports near larger metropolitan areas.

Figure 6-12 is illustrated with an example case where the mean daily maximum temperature at the hottest month at the airport is 59°F and elevation is sea level. A vertical line is drawn from the point on the horizontal axis associated with 59°F to the sea level field elevation curve. A horizontal line is then drawn from the associated location on the elevation curve to the right side of the figure. The value at the end of the horizontal line on the right side of the figure is the recommended design runway length. In this case, applying the 95 percent of fleet curve results in a design runway length of 2700 ft, while the 100 percent of fleet curve resulting in a design runway length of 3200 ft.

For those aircraft with 10 or more passenger seats at airports at elevation 3000 ft AMSL or less, Fig. 6-13 is referenced. At airports at elevation greater than 3000 ft AMSL, Fig. 6-12 “100 percent of fleet” is referenced.

Figure 6-13 is illustrated with an example case where the mean daily maximum temperature at the hottest month at the airport is

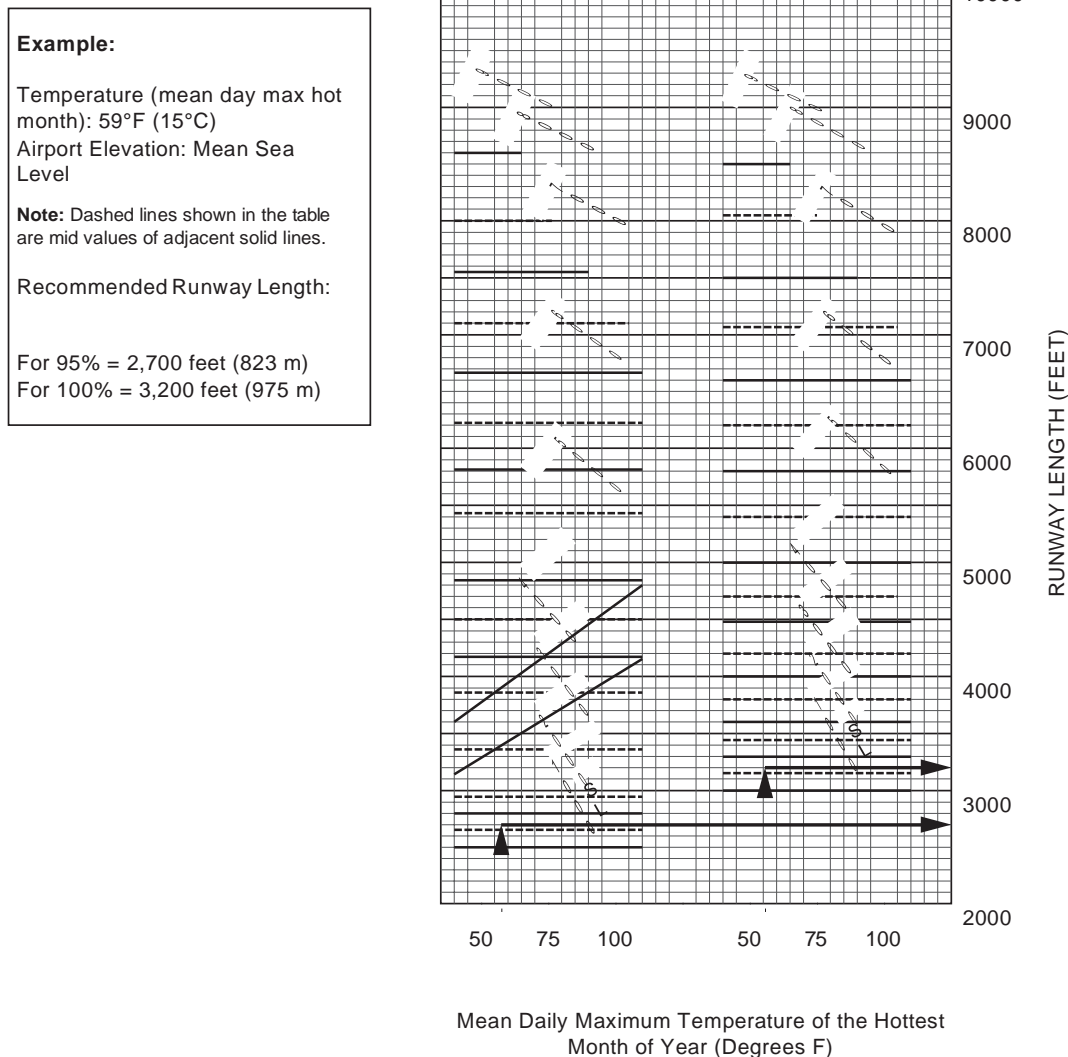


FIGURE 6-12 Small airplanes with fewer than 10 passenger seats (FAA AC 150/5325-4b).

90°F and elevation is 1000 ft AMSL. A vertical line is drawn from the point on the horizontal axis associated with 90°F to the 1000 ft AMSL field elevation curve. A horizontal line is drawn from the associated point on the field elevation curve to the right side of the figure, where the runway length is estimated. In this example, the design runway length is estimated to be 4400 ft.

Aircraft Greater than 12,500 lb but Less than or Equal to 60,000 lb MGTOW

For aircraft greater than 12,500 lb but less than or equal to 60,000 lb MGTOW, the critical aircraft is located on Table 6-5 "75 percent of fleet," or Table 6-6, "100 percent of fleet." Table 6-5 represents aircraft that generally require less than 5000 ft of runway, while Table 6-6 represents aircraft that generally require 5000 ft or more of runway.

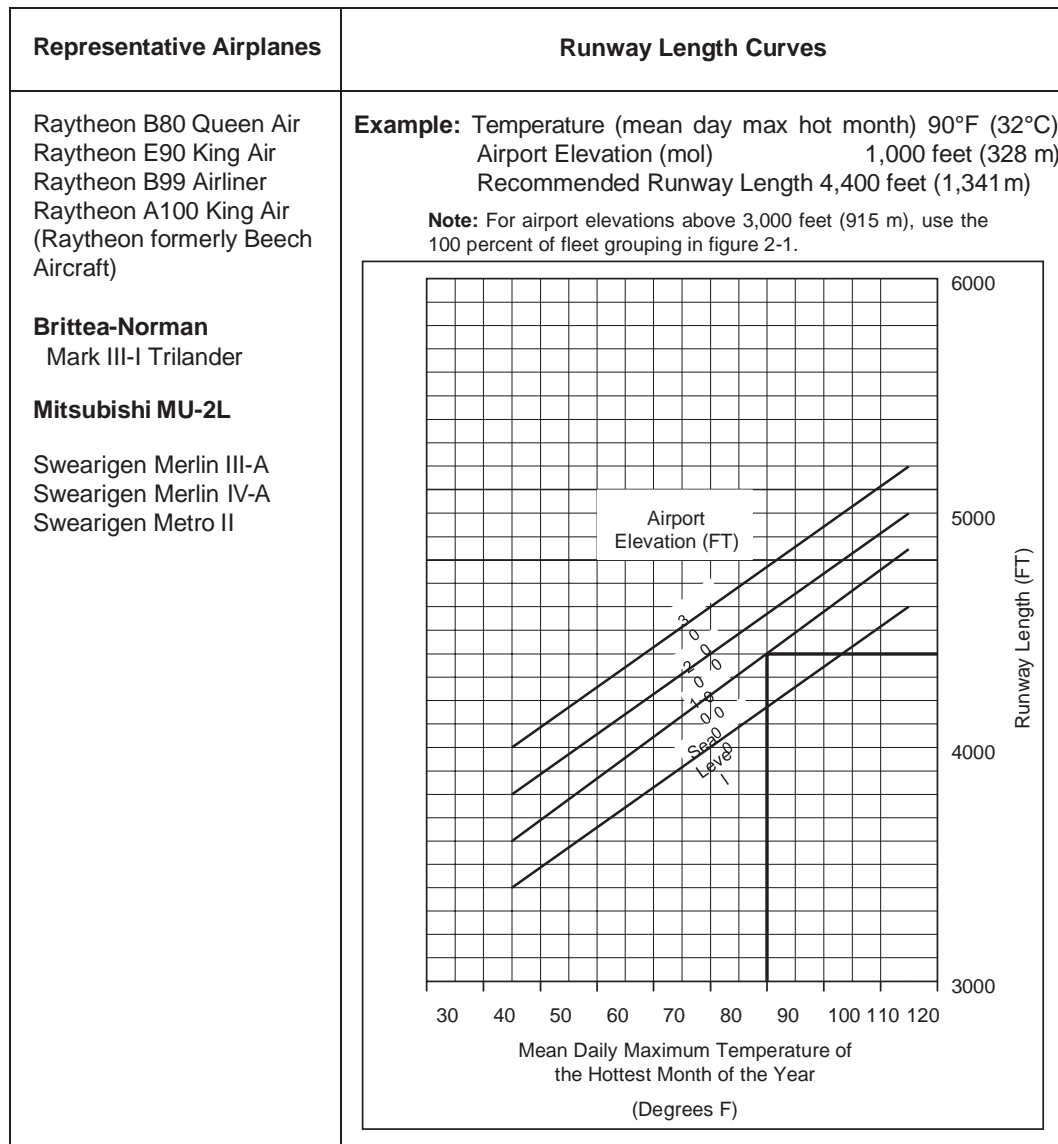


FIGURE 6-13 Small airplanes having 10 or more passenger seats (FAA AC 150/5325-4b).

As such, if the selected critical aircraft is found on Table 6-5, it is said that the runway length estimated will be able to accommodate 75 percent of the fleet. If the selected critical aircraft is found on Table 6-6, it is said that the runway length estimated will be able to accommodate 100 percent of this size fleet.

For the design aircraft, a “useful load” of either 60 or 90 percent is selected. A 60 percent useful load represents the condition where the critical aircraft typically operates at 60 percent load factors, or performs shorter range operations, requiring less fuel, while a 90 percent useful load represents the condition where the critical aircraft typically operates at 90 percent load factors, or performs longer range operations.

For aircraft falling within the “75 percent of fleet” group as identified in Table 6-5, Fig. 6-14 is then applied, selecting either the 60 or

Manufacturer	Model	Manufacturer	Model
Aerospatale	Sn-601 Corvette	Dassault	Falcon 10
Bae	125-700	Dassault	Falcon 20
Beech Jet	400A	Dassault	Falcon 50/50 EX
Beech Jet	Premier I	Dassault	Falcon 900/900B
Beech Jet	2000 Starship	Israel Aircraft Industries (LAI)	Jet Commander 1121
Bombardier	Challenger 300	IAI	Westwind 1123/1124
Cessua	500 Citation/ 501 Citation Sp	Learjet	20 Series
Cessna	Citation I/II/III	Learjet	31/31A/31A ER
Cessna	525A Citation II (CJ-2)	Learjet	35/35A/36/36A
Cessna	350 Citation Bravo	Learjet	40/45
Cessna	550 Citation II	Mitsubishi	Mu-300 Diamond
Cessna	551 Citation II/Special	Raytheon	390 Premier
Cessna	552 Citation	Raytheon Hawker	400/400 XP
Cessna	560 Citation Encore	Raytheon Hawker	600
Cessna	560/560 XL Citation Excel	Sabreliner	40/60
Cessna	560 Citation V Ultra	Sabreliner	75A
Cessna	650 Citation VII	Sabreliner	80
Cessna	680 Citation Sovereign	Sabreliner	T-39

Source: FAA AC 150/5235-4b.

TABLE 6-5 Airplanes that Make Up 75 Percent of the Fleet

90 percent useful load sides of the figure, and applied based on the mean daily maximum temperature of the hottest month (in Fahrenheit), and the elevation of the airfield (in feet AMSL).

Figure 6-14 illustrates two examples, one for an airport at sea level with average high temperature during the hottest month at 59°F and a critical aircraft falling within the 75 percent of fleet category at

Manufacturer	Model
Bae	Corporate 800/1000
Bombardier	600 Challenger
Bombardier	601/601-3A/3ER Challenger
Bombardier	604 Challenger
Bombardier	BD-100 Continental
Cessna	S550 Citation S/II
Cessna	650 Citation III/IV
Cessna	750 Citation X
Dassault	Falcon 900C/900EX
Dassault	Falcon 2000/2000EX
Israel Aircraft Industries (IAI)	Astra 1125
IAI	Galaxy 1126
Learjet	45 XR
Learjet	55/55B/55C
Learjet	60
Raytheon/Hawker	Horizon
Raytheon/Hawker	800/800 XP
Raytheon/Hawker	1000
Sabreliner	65/75

TABLE 6-6 Aircraft that (Including Those in Table 6-1) Make Up 100 Percent of the Fleet

60 percent useful load, and one for an airport at 1000 ft AMSL, average high temperature during the hottest month at 100°F, and a critical aircraft falling within the 75 percent of fleet category at 90 percent useful load. For aircraft falling within the “100 percent of fleet” group as identified in Table 6-6, Fig. 6-15 is similarly applied.

Figure 6-15 is illustrated with two examples, one illustrating an airport at 2000 ft AMSL with average high temperature during the hottest month at 59°F and a critical aircraft falling within the 100 percent of fleet category at 60 percent useful load, and one illustrating an airport at 3000 ft AMSL, average high temperature during the hottest month at 100°F, and a critical aircraft falling within the 100 percent of fleet category at 90 percent useful load.

Based on the runway lengths found in either Fig. 6-14 or Fig. 6-15, an adjustment is made for any nonlevel runway gradient. Specifically, the runway length found in Fig. 6-14 or Fig. 6-15 is increased by

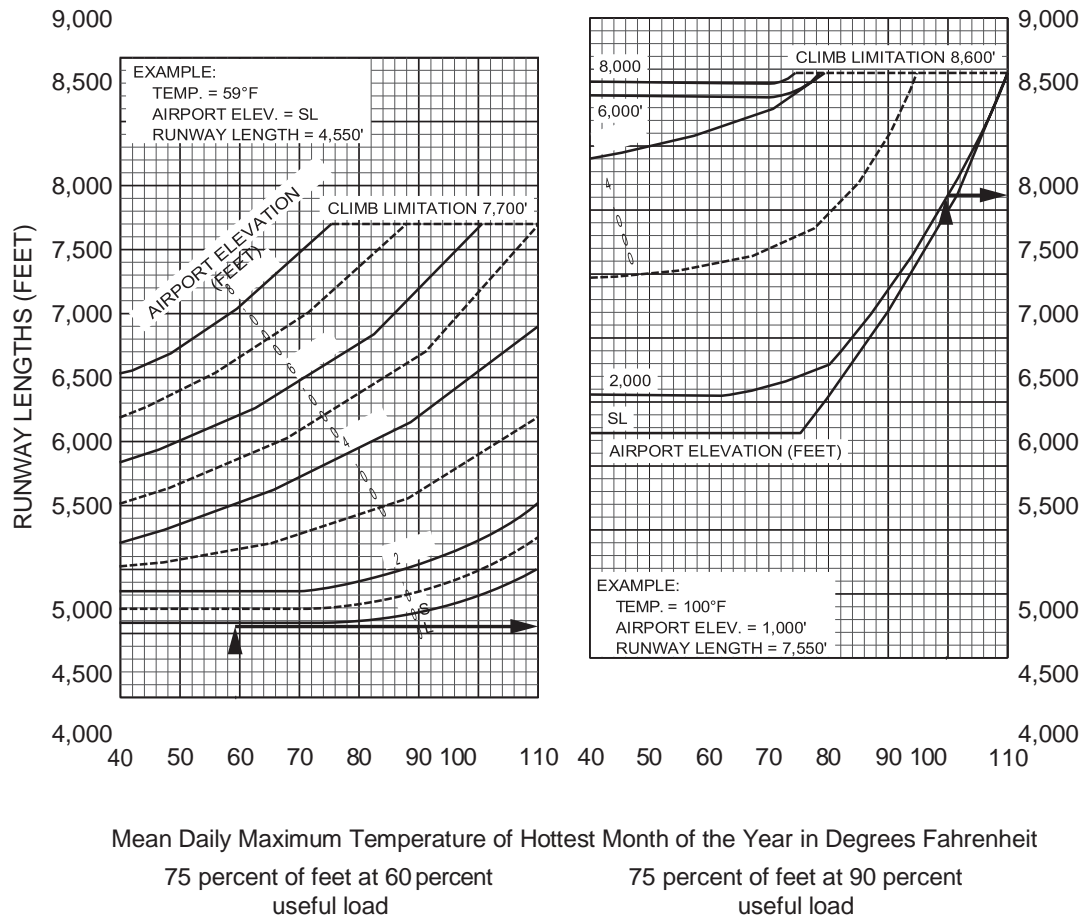
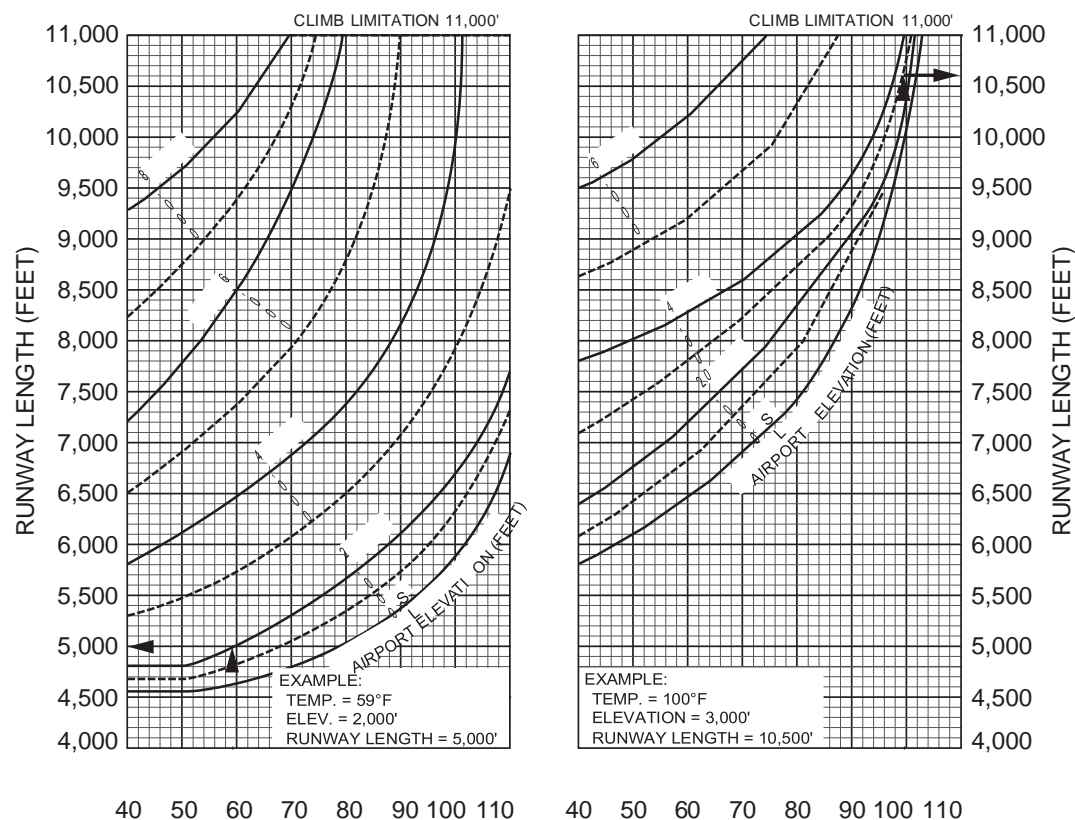


FIGURE 6-14 Seventy-five percent of feet at 60 or 90 percent useful load.



100 percent of feet at 60 percent
useful load

100 percent of feet at 90 percent
useful load

FIGURE 6-15 Hundred percent of fleet at 60 or 90 percent useful load.

10 ft for every foot in elevation difference between the lowest point and highest point on the runway.

At higher elevations, it is often the case that the required runway length is greater for aircraft less than 12,500 lb MGTOW than for aircraft greater than 12,500 lb. If this is the case, the design runway length would be that for the lighter aircraft. As such, airport planners estimating runway lengths at high elevation airports should perform runway length estimations for the smallest aircraft, in addition to that for the selected critical aircraft.

Aircraft Greater than 60,000 lb MGTOW

For aircraft greater than 60,000 lb MGTOW, runway lengths are estimated based on the specific performance specifications of the critical aircraft. These performance specifications may be found in the published aircraft “airport planning manuals.” These manuals may be found on the Internet sites of the major aircraft manufacturers.

Within the aircraft airport planning manuals are performance charts that are used to determine the aircraft’s required runway lengths for both takeoff and landing, based on the aircraft’s operating configuration, its estimated weights during takeoff and landing, as well as the airport elevation and average high temperature during the hottest month.

Example Problem 6-2 illustrates the procedure for estimating runway length using these charts.

Example Problem 6-2 Consider the situation where an airport with elevation 1000 ft AMSL and mean daily maximum temperature of the hottest month of 84°F, is planning for a new runway to be designed for the Boeing 737-900 aircraft, equipped with Pratt & Whitney CFM56-7B27 engines. At the airport a runway gradient of 20 ft is projected.

According to the performance specification chart, illustrated in Fig. 6-16, found in the Boeing 737-900 airport planning manual, the maximum design landing weight for the aircraft is 146,300 lb and the maximum design takeoff weight is 174,200 lb.

First, estimation of required runway length for landing is performed using the landing runway length performance chart for the aircraft. As with most landing performance charts, runway length requirements found landing may be found under both dry and wet runway conditions. For airport planning purposes, design runway length for landing is estimated by considering wet runway conditions. If a landing runway length performance chart does not include wet runway conditions, the design runway length is estimated as the runway length found under dry runway conditions, plus 15 percent.

Figure 6-17 illustrates this example. Applying the case example, a vertical line is drawn from the base of the horizontal axis at the location of the maximum design landing weight (146,300 lb), up to an interpolated point between the “sea level” and “2000 ft” (to represent the example airport’s 1000 ft elevation) wet-runway curves, and then a horizontal line is drawn to the vertical axis, where the estimated required runway length may be found. In this example, the estimated runway length for landing is approximately 6600 ft.

CHARACTERISTICS	UNITS	737-900	
MAX DESIGN TAXI WEIGHT	POUNDS	164,500	174,700
	KILOGRAMS	74,616	79,243
MAX DESIGN TAKEOFF WEIGHT	POUNDS	164,000	174,200
	KILOGRAMS	Takeoff Weight	79,016
MAX DESIGN LANDING WEIGHT	POUNDS		146,300
	KILOGRAMS	Landing Weight	66,361
MAX DESIGN ZERO FUEL WEIGHT	POUNDS		138,300
	KILOGRAMS		63,639
OPERATING EMPTY WEIGHT (1)	POUNDS		94,580
	KILOGRAMS	42,901	
MAX STRUCTURAL PAYLOAD	POUNDS	43,720	45,720
	KILOGRAMS	19,831	20,738
SEATING CAPACITY (1)	TWO-CLASS	177	177
	ALL-ECONOMY	189	189
MAX CARGO - LOWER DECK	CUBIC FEET	1,835	1,835
	CUBIC METERS	52.0	52.0
USABLE FUEL	US GALLONS	6875	6875
	LITERS	26,022	26,022
	POUNDS	46,063	46,063
	KILOGRAMS	20,894	20,894

NOTE: (1) OPERATING EMPTY WEIGHT FOR BASELINE MIXED CLASS CONFIGURATION. CONSULT WITH AIRLINE FOR SPECIFIC WEIGHTS AND CONFIGURATIONS.

FIGURE 6-16 Boeing 737-900 general airplane characteristics (*Boeing Corp. document #D6-58325-3 and FAA AC 150/5325-4B*).

These charts are designed for level runways. An adjustment for runway gradient must be made by adding 10 ft of runway length for every foot of runway gradient. In this example, an additional 200 ft of runway length is added, resulting in an adjusted runway length for landing of 6800 ft.

Second, estimation of required runway length for takeoff is performed using the takeoff runway length performance chart for the aircraft. Oftentimes, an aircraft will have multiple takeoff performance charts, typically for different average high temperatures. The chart associated with the temperature nearest the airport's average high during the hottest month is used.

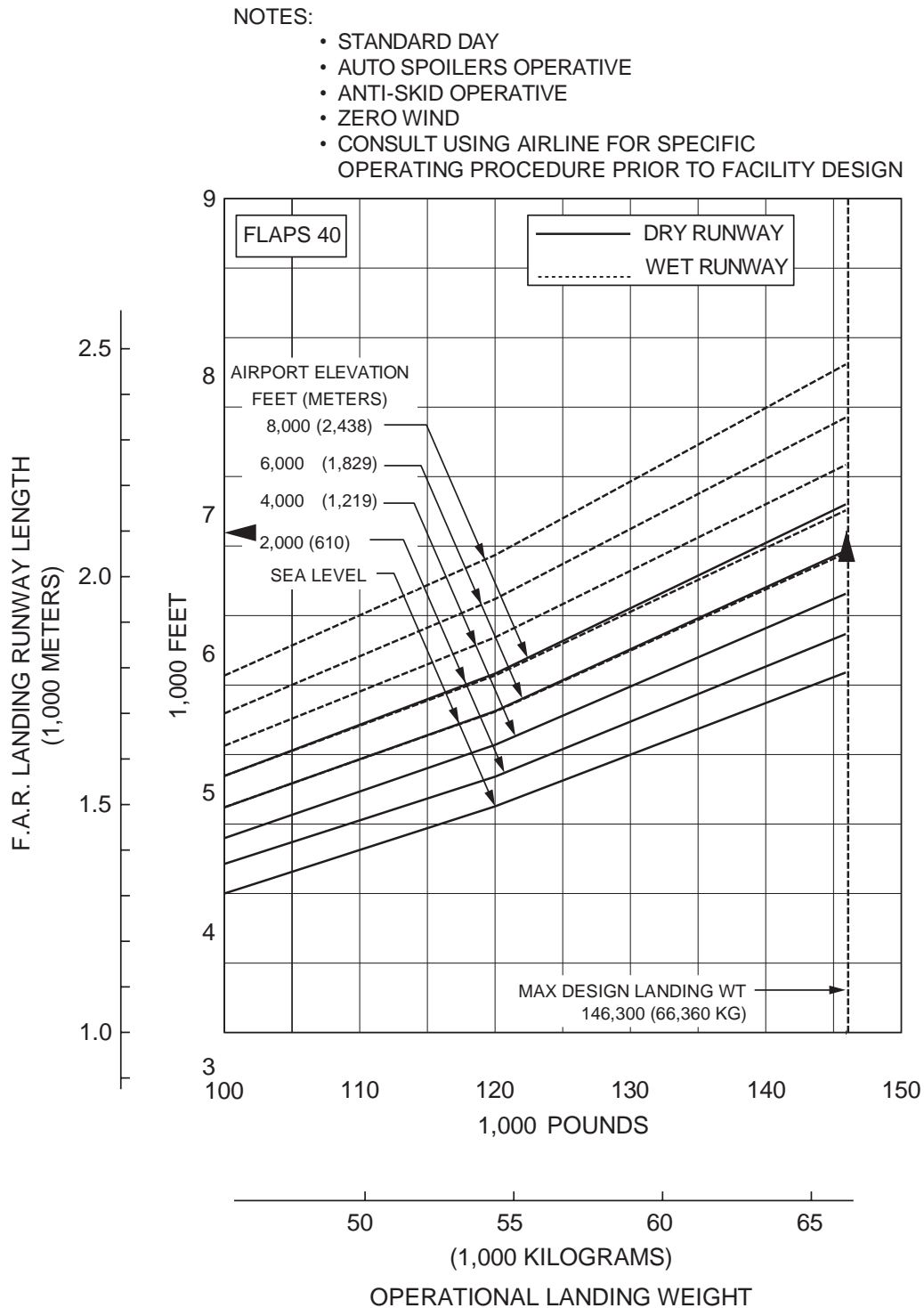


FIGURE 6-17 Landing runway length for Boeing 737-900 (CFM56-7B27 Engines, 40° Flaps) (Ref: Boeing Doc. D6-58325-3).

Figure 6-18 illustrates this example. Applying the case example, a vertical line is drawn from the base of the horizontal axis at the location of the maximum design takeoff weight (174,200 lb), up to an interpolated point between the “sea level” and “2000 ft” curves, and then a horizontal line is drawn to the vertical axis, where the estimated required runway length for takeoff may be found. In this example, the estimated runway length for takeoff is approximately 8800 ft. Considering the example’s runway gradient, an additional 200 ft of runway length is added, resulting in an adjusted runway length for takeoff of 9000 ft.

NOTES:

- CFM56-7827 ENGINES RATED AT 27,300 LB SLST
- NO ENGINE AIR BLEED FOR AIR CONDITIONING
- ZERO WIND, ZERO RUNWAY GRADIENT
- DRY RUNWAY SURFACE
- CONSULT WITH USING AIRLINE FOR SPECIFIC OPERATING PROCEDURE PRIOR TO FACILITY DESIGN
- LINEAR INTERPOLATION BETWEEN ALTITUDES INVALID
- LINEAR INTERPOLATION BETWEEN TEMPERATURES INVALID

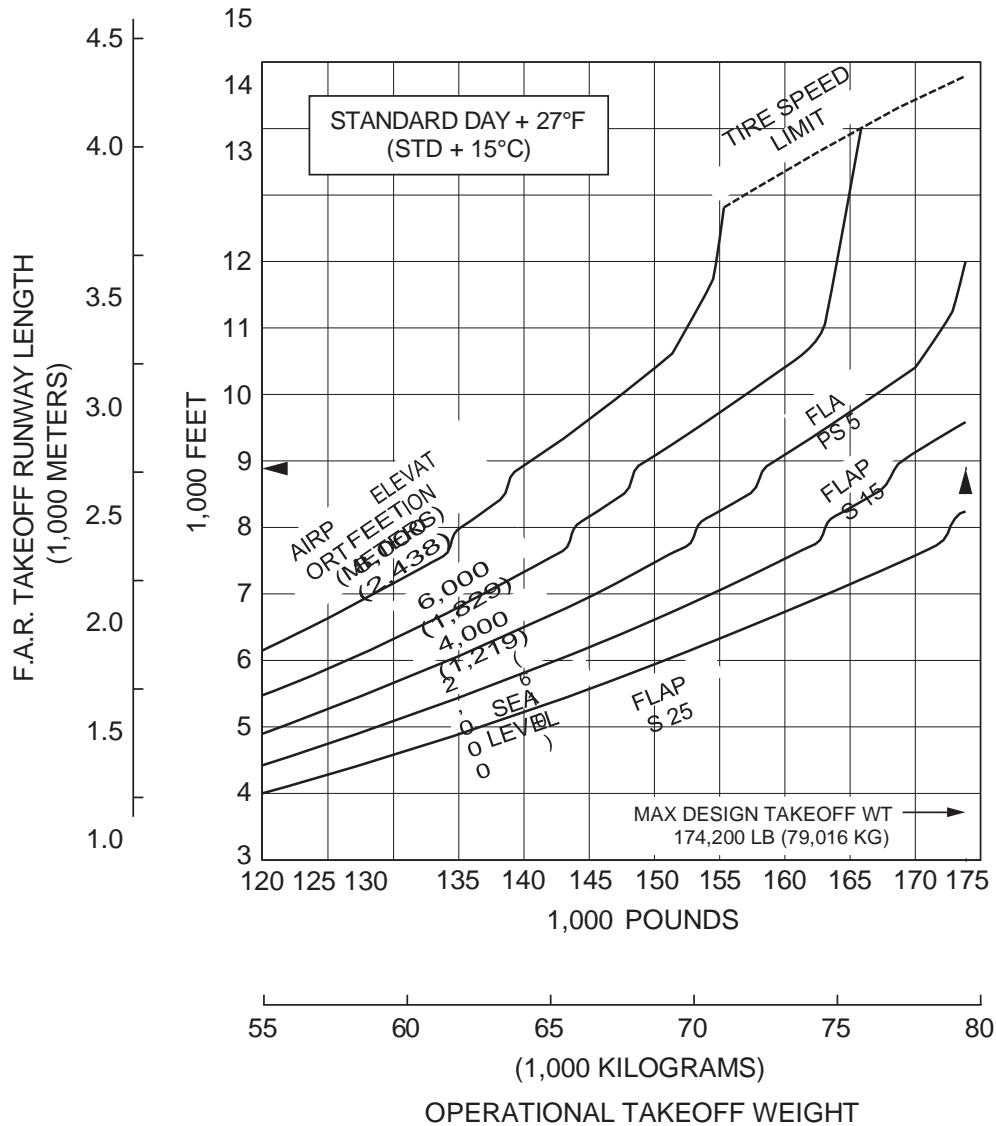


FIGURE 6-18 Takeoff runway length for Boeing 737-900 (CFM56-7B27 Engines)
(Ref: Boeing Doc. D6-58325-3).

For design purposes, the design runway length is the longer of the required runway lengths for landing and for takeoff. In this case, the design runway length for this example is 9000 ft.

Runway System Geometric Specifications

The runway system at an airport consists of the structural pavement, the shoulders, the blast pad, the runway safety area, various obstruction-free surfaces, and the runway protection zone, as shown in Figs. 6-19 and 6-20.

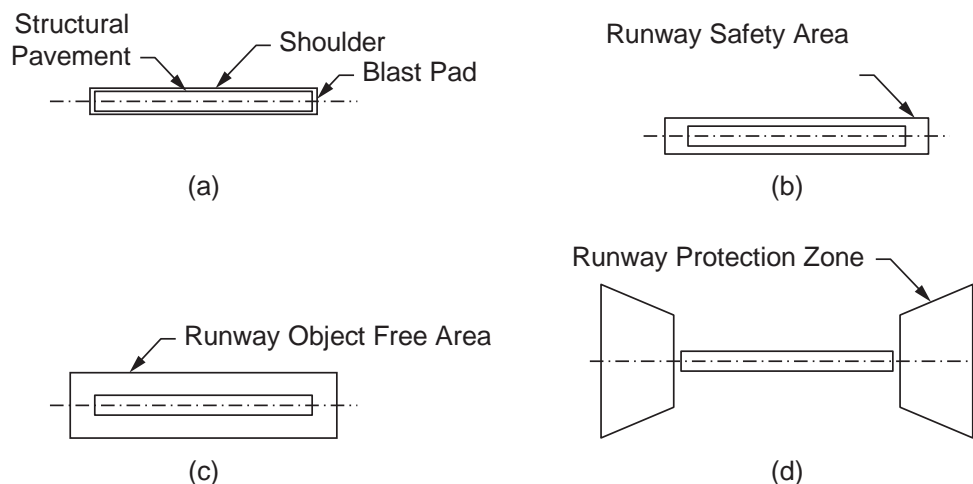


FIGURE 6-19 Runway system dimensions.

1. The runway *structural pavement* supports the aircraft with respect to structural load, maneuverability, control, stability, and other operational and dimensional criteria.
2. The *shoulder* adjacent to the edges of the structural pavement resists jet blast erosion and accommodates maintenance and emergency equipment.
3. The *blast pad* is an area designed to prevent erosion of the surfaces adjacent to the ends of runways due to jet blast or propeller wash.
4. The *runway safety area* (RSA) is an area surrounding the runway prepared or suitable for reducing the risk of damage to aircraft in the event of an undershoot, overshoot, or excursion from the runway. ICAO refers to an area similar to the runway safety area as the *runway strip* and the *runway end safety*

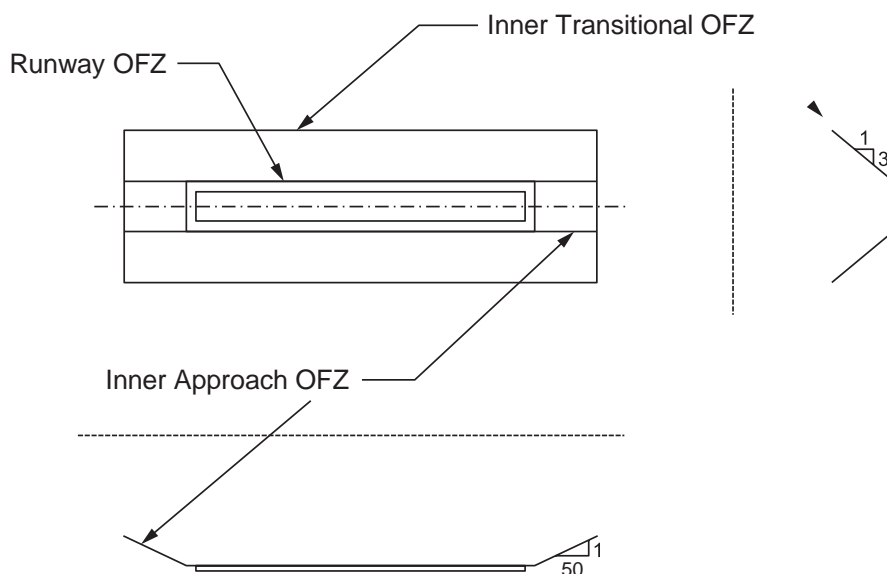


FIGURE 6-20 Object-free zone dimensions.

area. The runway safety area includes the structural pavement, shoulders, blast pad, and stopway, if provided. This area should be capable of supporting emergency and maintenance equipment as well as providing support for aircraft. The runway safety area is cleared, drained, and graded and should have no potentially hazardous ruts, humps, depressions, or other surface variations. It should be free of objects except for objects that are required to be located in the runway safety area because of their function. These objects are required to be constructed on frangible mounted structures at the lowest possible height with the frangible point no higher than 3 in above grade.

5. The runway *object-free area* (OFA) is defined by the FAA as a two-dimensional ground area surrounding the runway which must be clear of parked aircraft and objects other than those whose location is fixed by function.
6. The runway *obstacle-free zone* (OFZ) is a defined volume of airspace centered above the runway which supports the transition between ground and airborne operations. The FAA specifies this as the airspace above a surface whose elevation is the same as that of the nearest point on the runway centerline and extending 200 ft beyond each end of the runway.
7. The *inner approach obstacle-free zone*, which applies only to runways with approach lighting systems, is the airspace above a surface centered on the extended runway centerline beginning 200 ft beyond the runway threshold at the same elevation as the runway threshold and extending 200 ft beyond the last light unit on the approach lighting system. Its width is the same as the runway obstacle-free zone and it slopes upward at the rate of 50 horizontal to 1 vertical.
8. The *inner transitional obstacle-free zone*, which applies only to precision instrument runways, is defined by the FAA as the volume of airspace along the sides of the runway and the inner approach obstacle-free zone. The surface slopes at the rate of 3 horizontal to 1 vertical out from the edge of the runway obstacle-free zone and the inner approach obstacle-free zone until it reaches a height of 150 ft above the established airport elevation.
9. The *runway protection zone* (RPZ) is an area on the ground used to enhance the protection of people and objects near the runway approach.

The FAA runway standards related to the pavement and shoulder width, the safety area, the blast pad, and the obstacle-free surfaces are given in Tables 6-7 and 6-8. Similar data for the ICAO are given in Table 6-9.

	Approach Type									
	Visual and Nonprecision Instrument, Airplane Design Group					Precision Instrument, Airplane Design Group				
	I*	I	II	III	IV	I*	I	II	III	IV
Runway width	60	60	75	100	150	75	100	100	100	150
Shoulder width	10	10	10	20	25	10	10	10	20	25
Blast pad										
Width	80	80	95	140	200	95	120	120	140	200
Length	60	100	150	200	200	60	100	150	200	200
Safety area										
Width	120	120	150	300	500	300	300	300	400	500
Length†	240	240	300	600	1000	600	600	600	800	1000
Object-free area										
Width	250	400	500	800	800	800	800	800	800	800
Length†	300	500	600	1000	1000	1000	1000	1000	1000	1000
Obstacle-free zone										
Width‡	120§	250	250	250	250	300	300	300	300	300
Length¶	200	200	200	200	200	200	200	200	200	200

*Facilities for small airplanes only.

†From end of runway; with the declared distance concept, these lengths begin at the stop end of each ASDA and both ends of the LDA, whichever is greater.

‡For runways serving small aircraft only; for large aircraft the greater of 400 ft or 180 ft plus the wingspan of the most demanding aircraft plus 20 ft for each 1000 ft of airport elevation.

§For runways serving small aircraft with approach speeds of less than 50 kn; increase to 250 ft for runways serving aircraft with approach speeds greater than 50 kn.

¶Beyond the end of each runway.

TABLE 6-7 Runway Dimensional Standards, ft—Approach Category A and B Aircraft

	Airplane Design Group					
	I	II	III	IV	V	VI
Runway width	100	100	100 ^a	150	150	200
Shoulder ^b width	10	10	20 ^a	25	35	40
Blast pad						
Width	120	120	140 ^a	200	220	280
Length	100	150	200	200	400	400
Safety area						
Width ^c	500	500	500	500	500	500
Length ^d	1000	1000	1000	1000	1000	1000
Object-free area						
Width	800	800	800	800	800	800
Length ^d	1000	1000	1000	1000	1000	1000
Obstacle-free zone						
Width ^e	400	400	400	400	400	400
Length ^f	200	200	200	200	200	200

^aFor airplane design group III serving aircraft with maximum certified takeoff weight greater than 150,000 lb, the standard runway width is 150 ft, the shoulder width is 25 ft, and the blast pad width is 200 ft.

^bAirplane design groups V and VI normally require stabilized or paved shoulder surfaces.

^cFor Airport Reference Code C-I and C-II, a runway safety area width of 400 ft is permissible. For runways designed after 2/28/83 to serve aircraft approach category D aircraft, the runway safety area width increases 20 ft for each 1000 ft of airport elevation above mean sea level.

^dFrom end of runway; with the declared distance concept, these lengths begin at the stop end of each ASDA and both ends of the LDA, whichever is greater.

^eFor large aircraft the greater of 400 ft or 180 ft plus the wingspan of the most demanding aircraft plus 20 ft for each 1000 ft of airport elevation; for small aircraft 300 ft for precision instrument runways, 250 ft for all other runways serving small aircraft with approach speeds of 50 kn or more, and 120 ft for all other runways serving small aircraft with approach speeds less than 50 kn.

^fBeyond the end of each runway.

TABLE 6-8 Runway Dimensional Standards, ft—Approach Category C, D, and E Aircraft

Parallel Runway System Spacing

The spacing of parallel runways depends on a number of factors such as whether the operations are in VMC or IMC and, if in IMC, whether it is desired to have the capability of accommodating simultaneous arrivals or simultaneous arrivals and departures. At those airports serving both heavy and light aircraft simultaneous use of runways even in VMC conditions may be dictated by separation requirements to safeguard against wake vortices.

	Aerodrome Code Letter				
	A	B	C	D	E
Pavement width					
Aerodrome code number					
1*	18	18	23		
2*	23	23	30		
3	30	30	30	45	
4			45	45	45
Pavement and shoulder width ^{†,‡}			60	60	60
	Aerodrome Code Number				
	1	2	3	4	
Runway strip width [‡]					
Precision approach	150	150	300	300	
Nonprecision approach	150	150	300	300	
Visual approach	60	80	150	150	
Clear and graded area width [‡]					
Instrument approach	80	80	150 [§]	150 [§]	
Visual approach	60	80	150	150	

*The width of a precision approach runway should not be less than 30 m where the aerodrome code number is 1 or 2.

[†]Minimum width of pavement and shoulders when pavement width is less than 60 m.

[‡]Symmetrical about the runway centerline.

[§]It is recommended that this be provided for the first 150 m from each end of the runway and that it should be increased linearly from this point to a width of 210 m at a point 300 m from each end of the runway and remain at this width for the remainder of the runway.

TABLE 6-9 ICAO Runway and Runway Strip Dimensional Standards, m

Under VMC, the FAA requires parallel runway centerline separations of 700 ft for all aircraft when the operations are in the same direction and wake vortices are not prevalent. It also recommends increasing the separation to 1200 ft for airplane design group V and VI runways. If wake vortices are generated by heavy jets and it is desired to operate on two runways simultaneously in VMC when little or no crosswind is present, the minimum distance specified by the FAA is 2500 ft.

For operations under VMC, the ICAO recommends that the minimum separations between the centerlines of parallel runways for simultaneous use disregarding wake vortices be 120 m (400 ft) for aerodrome code number 1, 150 m (500 ft) for aerodrome code number 2, and 210 m (700 ft) for aerodrome code number 3 or 4 runways.

In IMC conditions, the FAA specifies 4300 ft and ICAO specifies 1525 m (5000 ft) as the minimum separation between centerlines of

parallel runways for simultaneous instrument approaches. However, there is evidence that these distances are conservative and steps are being taken to reduce it. The ultimate goal is to reduce this distance by about one-half. For dependent instrument approaches both the FAA and ICAO recommend centerline separations of 3000 ft (915 m). For triple and quadruple simultaneous instrument approaches, the FAA requires 5000-ft separation between runway centerlines, although will allow 4300 ft separations on a case-by-case basis.

Both the FAA and ICAO specify that two parallel runways may be used simultaneously for radar departures in IMC if the centerlines are separated by at least 2500 ft (760 m). The FAA requires a 3500-ft centerline separation for simultaneous nonradar departures. If two parallel runways are to be operated independently of each other in IMC under radar control, one for arrivals and the other for departures, both the FAA and ICAO specify that the minimum separation between the centerlines is 2500 ft (760 m) when the thresholds are even. If the thresholds are staggered, the runways can be brought closer together or must be separated farther depending on the amount of the stagger and which runways are used for arrivals and departures. If approaches are to the nearest runway, then the spacing may be reduced by 100 ft (30 m) for each 500 ft (150 m) of stagger down to a minimum of 1200 ft (360 m) for airplane design groups V and VI and 1000 ft (300 m) for all other aircraft. However, if the approaches are to the farthest runway, then the runway spacing must be increased by 100 ft (30 m) for each 500 ft (150 m) of stagger.

Sight Distance and Longitudinal Profile

The FAA requirement for sight distance on individual runways requires that the runway profile permit any two points 5 ft above the runway centerline to be mutually visible for the entire runway length. If, however, the runway has a full length parallel taxiway, the runway profile may be such that an unobstructed line of sight will exist from any point 5 ft above the runway centerline to any other point 5 ft above the runway centerline for one-half the runway length.

The FAA recommends a clear line of sight between the ends of intersecting runways. The terrain must be graded and permanent objects designed and sited so that there will be an unobstructed line of sight from any point 5 ft above one runway centerline to any point 5 ft above an intersecting runway centerline within the runway visibility zone. The runway visibility zone is the area formed by imaginary lines connecting the visibility points of the two intersecting runways. The runway visibility zone for intersecting runways is shown in Fig. 6-21. The visibility points are defined as follows:

1. If the distance from the intersection of the two runway centerlines is 750 ft or less, the visibility point is on the centerline at the runway end designated by point a in Fig. 6-21.

2. If the distance from the intersection of the two runway centerlines is greater than 750 ft but less than 1500 ft, the visibility point is on the centerline 750 ft from the intersection of the centerlines designated by point b in Fig. 6-21.
3. If the distance from the intersection of the two runway centerlines is equal to or greater than 1500 ft, the visibility point is on the centerline equidistant from the runway end and the intersection of the centerlines designated by points c and d in Fig. 6-21.

The ICAO requirement for sight distance on individual runways requires that the runway profile permit an unobstructed view between any two points at a specified height above the runway centerline to be mutually visible for a distance equal to at least one-half the runway length. ICAO specifies that the height of these two points be 1.5 m (5 ft) above the runway for aerodrome code letter A runways, 2 m (7 ft) above the runway for aerodrome code letter B runways, and 3 m (10 ft) above the runway for aerodrome code letter C, D, or E runways.

It is desirable to minimize longitudinal grade changes as much as possible. However, it is recognized that this may not be possible for reasons of economy. Therefore both the ICAO and FAA allow changes

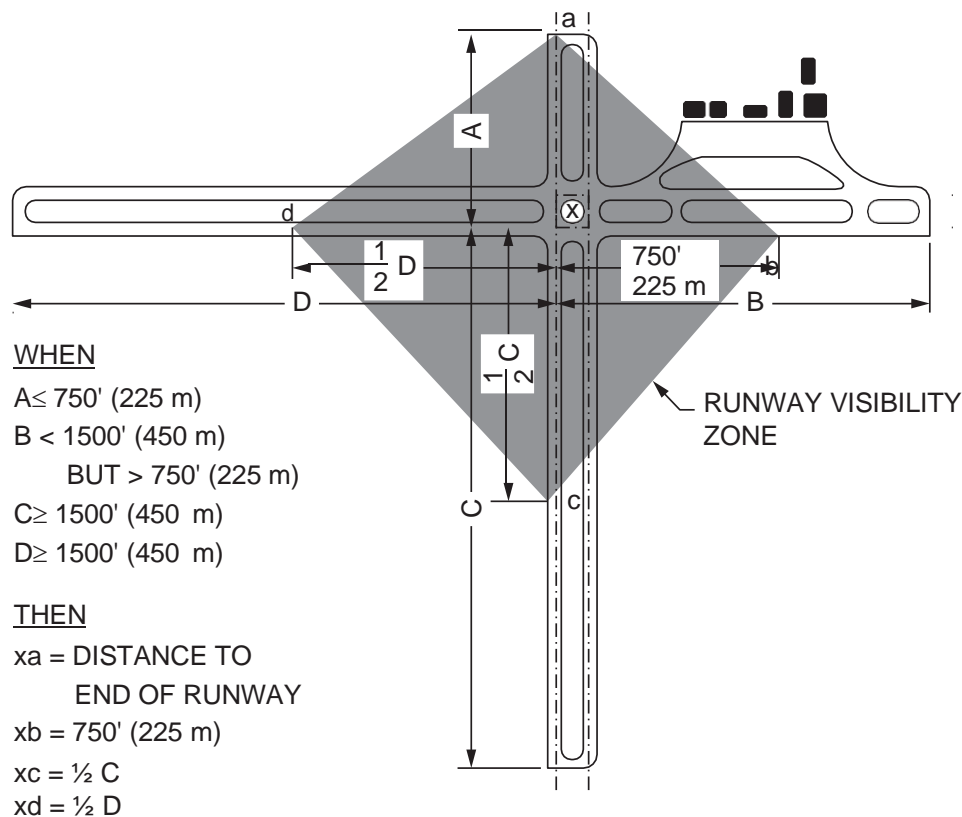


FIGURE 6-21 Runway visibility zone for intersecting runways (*Federal Aviation Administration*).

	Aircraft Approach Category				
	A	B	C	D	E
Gradient (%)					
Pavement longitudinal ^a					
Maximum	2.0	2.0	1.5 ^b	1.5 ^b	1.5 ^b
Maximum change	2.0	2.0	1.5	1.5	1.5
Pavement transverse					
Maximum	2.0	2.0	1.5	1.5	1.5
Shoulder transverse					
Minimum	3.0	3.0	1.5 ^c	1.5 ^c	1.5 ^c
Maximum ^d	5.0	5.0	5.0	5.0	5.0
Runway end safety area					
Maximum longitudinal ^e	3.0	3.0	3.0	3.0	3.0
Maximum longitudinal					
Grade change	2.0	2.0	2.0	2.0	2.0
Minimum transverse	1.5	1.5	1.5	1.5	1.5
Maximum transverse ^d	5.0	5.0	3.0	3.0	3.0
Vertical curve (ft)					
Minimum length ^{a,f}	300 ^g	300 ^g	1000	1000	1000
Minimum distance between points of intersection ^{a,h}	250	250	1000	1000	1000

^aApplies also to runway safety area adjacent to sides of the runway.

^bMay not exceed 0.8 percent in the first and last quarter of runway.

^cA minimum of 3 percent for turf.

^dA slope of 5 percent is recommended for a 10 ft width adjacent to the pavement areas to promote drainage.

^eFor the first 200 ft from the end of the runway and if it slopes it must be downward. For the remainder of the runway safety area the slope must be such that any upward slope does not penetrate the approach surface or clearway plane and any downward slope does not exceed 5 percent.

^fFor each 1 percent change in grade.

^gNo vertical curve is required if the grade change is less than 0.4 percent.

^hDistance is multiplied by the sum of the absolute grade changes in percent.

Source: Federal Aviation Administration [6].

TABLE 6-10 Runway Surface Gradient Standards

in grade but limit their number and size. The maximum longitudinal grade changes that are permitted by the FAA are listed in Table 6-10 and illustrated in Fig. 6-22. The maximum longitudinal grade changes that are permitted by the ICAO are listed in Table 6-11. Tables 6-10 and 6-11 also list the maximum longitudinal grade. The FAA limits both

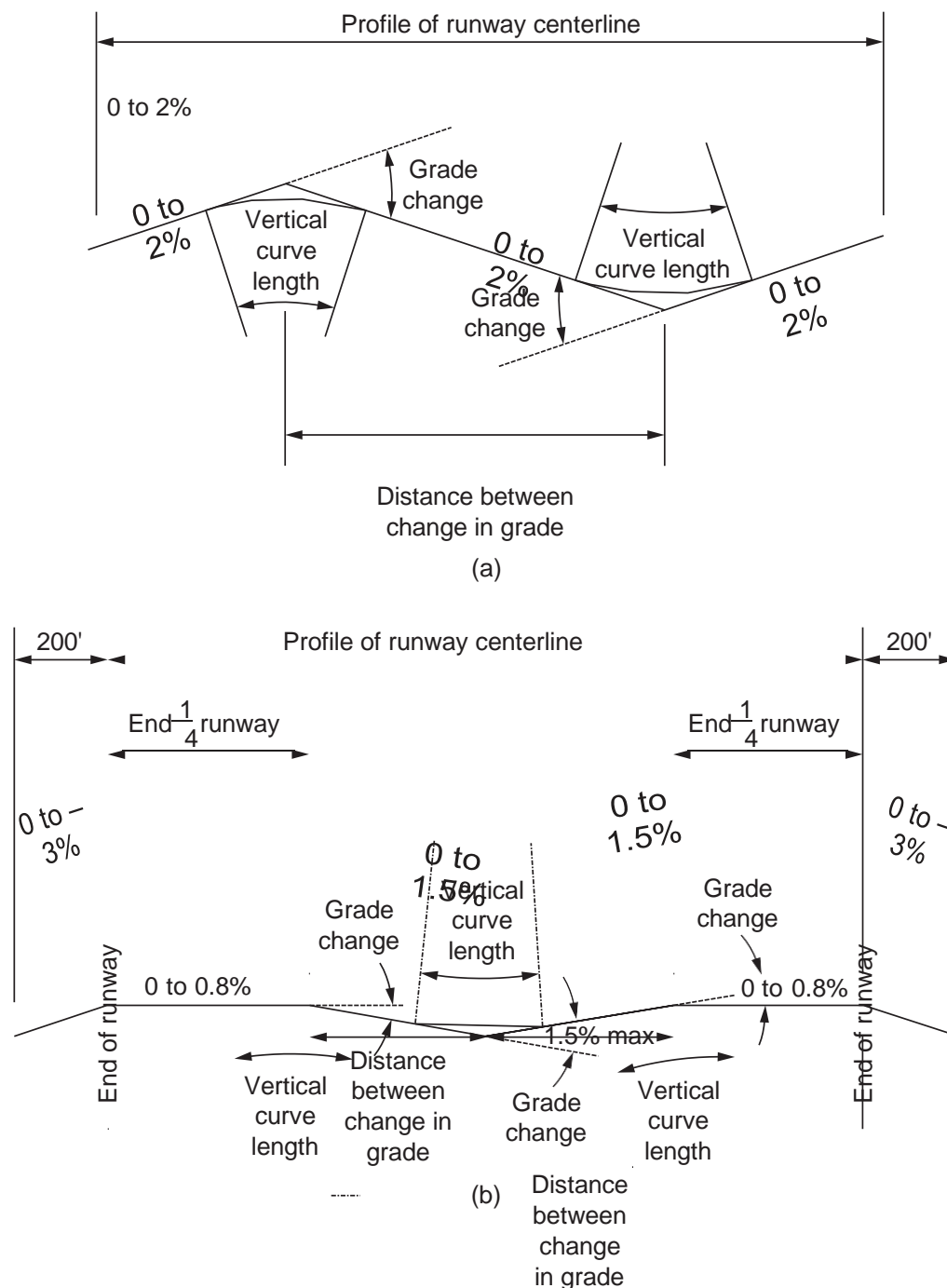


FIGURE 6-22 Runway longitudinal profile: (a) utility airports, (b) transport airports.

longitudinal gradient and longitudinal grade changes to 2 percent for runways serving approach category A and B aircraft and 1.5 percent for runways serving approach category C, D, and E aircraft. ICAO limits both longitudinal gradient and longitudinal grade changes to 2 percent for aerodrome code number 1 and 2 runways and 1.5 percent for aerodrome code number 3 runways. For aerodrome code number 4 runways the maximum longitudinal gradient is 1.25 percent and the

maximum change in longitudinal gradient is 1.5 percent. In addition, for runways that are equipped to be used in bad weather, the gradient of the first and last quarter of the length of the runway must be very flat for reasons of safety. Both the ICAO and the FAA require that this gradient not exceed 0.8 percent. In all cases it is desirable to keep both longitudinal grades and grade changes to a minimum.

Longitudinal slope changes are accomplished by means of vertical curves. The length of a vertical curve is determined by the magnitude of the changes in slope and the maximum allowable change in the slope of the runway. Both these values are also listed in Tables 6-11 and 6-12.

	Aerodrome Code Number				
	1	2	3	4	
Runway longitudinal Gradient (%)					
Maximum	2.0	2.0	1.5*	1.25*	
Maximum change	2.0	2.0	1.5	1.5	
Maximum effective [†]	2.0	2.0	1.0	1.0	
Vertical curve (m)					
Minimum length of curve [‡]	75	150	300	300	
Minimum distance between points of intersection [§]	50	50	150	300	
Runway strips Gradient (%)					
Maximum longitudinal	2.0	2.0	1.75	1.5	
Maximum transverse	3.0	3.0	2.5	2.5	
	Aerodrome Code Letter				
	A	B	C	D	E
Runway transverse gradient (%)					
Maximum	2.0	2.0	1.5	1.5	1.5
Minimum	1.0	1.0	1.0	1.0	1.0
Shoulder transverse gradient (%)					
Maximum	2.5	2.5	2.5	2.5	2.5

*May not exceed 0.8 percent in the first and last quarter of runway for aerodrome code number 4 or for a category II or III precision instrument runway for aerodrome code number 3.

[†]Difference in elevation between high and low point divided by runway length

[‡]For each 1 percent change in grade.

[§]Distance is multiplied by sum of absolute grade changes in percent minimum length is 45 m.

Source: International Civil Aviation Organization [3].

TABLE 6-11 Runway Surface Gradient Standards

	Airplane Design Group				
	I*	I	II	III	IV
Visual or nonprecision runway centerline to					
Taxiway or taxilane centerline [†]	150	225	240	300	400
Hold line [†]	125	200	200	200	250
Helicopter touchdown pad	400	400	400	400	400
Aircraft parking area	125	200	250	400	500
Precision instrument runway centerline to					
Taxiway or taxilane centerline [†]	200	250	300	350	400
Hold line [†]	175	250	250	250 [‡]	250 [‡]
Helicopter touchdown pad	400	400	400	400	400
Aircraft parking area	400	400	400	400	500

*For facilities for small aircraft only.

[†]Satisfies the requirement that no part of an aircraft at a holding an increase to these separations may be needed to achieve this result.

[‡]For sea level up to elevation 6000 ft. Increase by 1 ft for each 100 ft of airport elevation above 6000 ft.

Source: Federal Aviation Administration [6].

TABLE 6-12 Airfield Separation Criteria for Aircraft in Approach Categories A and B, ft

The number of slope changes along the runway is also limited. The FAA requires that the distance between the points of intersection of two successive curves should not be less than the sum of the absolute percentage values of change in slope multiplied by the 250 ft for airports serving aircraft approach category A and B aircraft and 1000 ft for airports serving aircraft approach category C, D, and E aircraft. The ICAO requires that the distance between the points of intersection of two successive curves should not be less than the sum of the absolute percentage values of change in slope multiplied by 50 m (165 ft) for aerodrome code number 1 and 2 runways, 150 m (500 ft) for aerodrome code number 3 runways, and 300 m (1000 ft) for aerodrome code number 4 runways. ICAO also specifies that the minimum distance in all cases is 45 m (150 ft).

For example, for an FAA runway serving transport aircraft, that is, approach category C, D, or E aircraft, if the change in slope was 1.5 percent, the required length of vertical curve would be 1500 ft. Vertical curves are normally not necessary if the change in slope is not more than 0.4 percent. The FAA specifies a minimum length of vertical transition curve of 300 for each 1 percent change in grade for runways

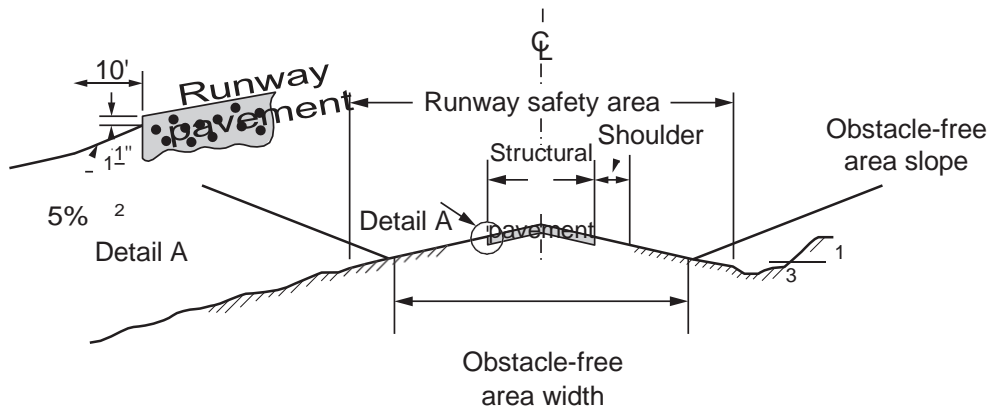


FIGURE 6-23 Runway gradient cross section.

serving approach category A and B aircraft and 1000 ft for each 1 percent change in grade for airport serving approach category C, D, and E aircraft. ICAO specifies a minimum length of vertical transition curve of 75 m for each 1 percent change in grade for aerodrome code number 1 runways, 150 m for each 1 percent change in grade aerodrome code number 2 runways, and 300 m for each 1 percent change in grade for aerodrome code number 4 runways.

Transverse Gradient

A typical cross section of a runway is shown in Fig. 6-23. The FAA and ICAO specifications for transverse slope on the runways are given in Tables 6-10 and 6-11, respectively. It is recommended that a 5 percent transverse slope be provided for the first 10 ft of shoulder adjacent to a pavement edge to ensure proper drainage.

Airfield Separation Requirements Related to Runways

The minimum distance from the runway centerline to parallel taxiways, taxilanes, aircraft holding lines, helicopter touchdown pads, and aircraft parking areas are also specified. These distances are given in Tables 6-12 and 6-13 for the FAA and Tables 6-14 and 6-15 for ICAO.

Obstacle Clearance Requirements

In addition to the geometric standards associated with the design of runways, there are specific requirements concerning the protection of airspace around airfields to provide for the safe navigation of aircraft to and from the airport.

In the United States, the FAA requires that protection zones be provided at the ends of runways. The runway protection zone is the area on the ground beneath the approach surface to a runway from the end of the primary surface to the point where the approach surface is 50 ft above the primary surface, as shown in Fig. 6-24. The dimensions of the runway protection zone are provided in Table 6-16.

	Airplane Design Group					
	I	II	III	IV	V	VI
Visual or nonprecision runway						
Centerline to Taxiway or taxilane centerline*	300	300	400	400	400 [¶]	600
Hold line*	250	250	250	250	250	250
Helicopter touchdown pad	400	400	400	400	400	400
Aircraft parking area	400	400	500	500	500	500
Precision instrument runway						
Centerline to Taxiway or taxilane centerline*	400	400	400	400	400 [¶]	600
Hold line* [§]	250	250	250 [†]	250 [†]	280 [‡]	325
Helicopter touchdown pad	400	400	400	400	400	400
Aircraft parking area	500	500	500	500	500	500

*Satisfies the requirement that no part of an aircraft at a holding location or on a taxiway centerline is within the runway safety area or penetrates the obstacle free zone. Accordingly, at higher elevations an increase to these separations may be needed to achieve this result.

[†]For aircraft in aircraft approach category C and airplane design groups III and IV increase by 1 ft for each 100 ft of airport elevation greater than 3200 ft.

[‡]For aircraft in aircraft approach category C and airplane design group V increase by 1 ft for each 100 ft of airport elevation above mean sea level.

[§]For aircraft in aircraft approach category D increase by 1 ft for each 100 ft of airport elevation above mean sea level.

[¶]For airports at or below an elevation of 1345 ft; increase to 450 ft for airports at elevations between 1345 and 6560 ft and to 500 ft for airports at an elevation above 6560 ft.

Source: Federal Aviation Administration [6].

TABLE 6-13 Airfield Separation Criteria for Aircraft in Approach Categories C and D, ft

When the runway protection zone begins at a location other than 200 ft beyond the end of the runway due to the application of the declared distance concept discussed in Chap. 2, two runway protection zones are usually required, an approach runway protection zone and a departure runway protection zone. The dimensions of the approach runway protection zone are given in Table 6-16 but the departure runway protection zone begins 200 ft beyond the far end of

	Aerodrome Code Letter				
	A	B	C	D	E
Runway centerline to parallel taxiway centerline					
Noninstrument runways					
Aerodrome code 1	37.5	42			
Aerodrome code 2	47.5	52			
Aerodrome code 3			93	101	
Aerodrome code 4				101	107.5
Instrument runways					
Aerodrome code 1	82.5	87			
Aerodrome code 2	82.5	87			
Aerodrome code 3			168	176	
Aerodrome code 4				176	182.5

Source: International Civil Aviation Organization [2, 3, 4].

TABLE 6-14 Runway to Taxiway Separation Criteria on the Airfield, m

	Type of Runway				
	Non-Instrument	Non-precision Approach	Precision Approach Category		Takeoff
			I	II & III	
Aerodrome code 1	30	40	60*	—	30
Aerodrome code 2	40	40	60*	—	40
Aerodrome code 3	75	75	90*. [†]	90*. [†]	75
Aerodrome code 4	75	75	90*. [†]	90*. [†]	75

*This distance may have to be increased to avoid interference with radio aids; for a precision instrument category III runway this increase may be in the order of 50 m.

[†]If a holding bay or a taxiway holding position is at a lower elevation compared to the runway threshold the distance may be decreased by 5 m for every meter the holding bay or holding position is lower than the threshold, contingent upon not interfering with the inner transitional surface; if a holding bay or a taxiway holding position is at a higher elevation compared to the runway threshold the distance should be increased by 5 m for every meter the holding bay or holding position is higher than the threshold.

Source: International Civil Aviation Organization [2, 3, 4].

TABLE 6-15 Runway to Holding Line Separation Criteria on the Airfield, m

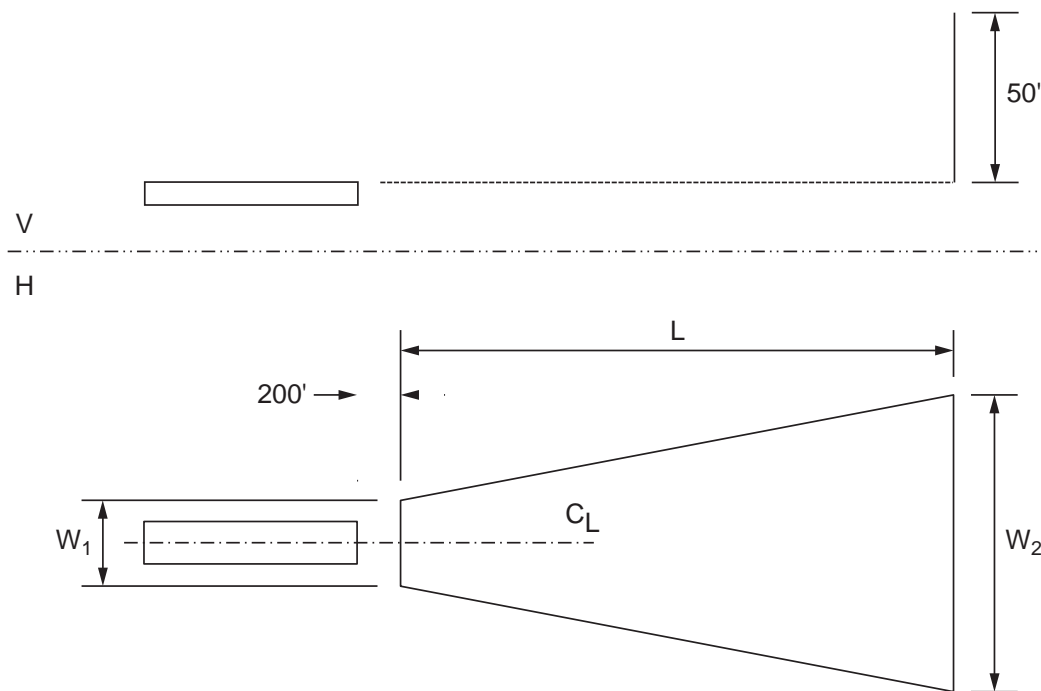


FIGURE 6-24 Runway protection zone.

the takeoff run available and the portion of the runway between the takeoff run available and the end of the runway is declared unavailable and unsuitable for the takeoff run. The dimensions of the departure runway protection zone are

1. For runways serving only small aircraft in aircraft approach categories A and B, the length is 1000 ft, the inner width is 250 ft and the outer width is 450 ft.
2. For runways serving large aircraft in aircraft approach categories A and B, the length is 1000 ft, the inner width is 500 ft and the outer width is 700 ft.
3. For runways serving aircraft in aircraft approach categories C, D, or E, the length is 1700 ft, the inner width is 500 ft and the outer width is 1010 ft.

FAR Part 77

Part 77 of the Federal Aviation Regulations establishes standards for determining what would be considered obstructions to navigable airspace, sets forth the requirements for notice to the FAA due to certain proposed construction or alteration activities, and provides for aeronautical studies of obstructions to air navigation to determine the effect of these obstructions on the safe and efficient use of airspace [8, 9]. The airport operator has the responsibility to ensure that the aerial approaches to the airport will be adequately cleared and protected and that the land adjacent to or in the immediate vicinity of the airport

Aircraft Served	Runway Approach End	Approach* Opposite End	Width			
			Length L, ft	Inner W ₁ , ft	Outer W ₂ , ft	Area, acres
Small	V	V	1000	250	450	8.035
		NP	1000	500	650	13.200
		NP+1000	1000	1050		23.542
		P	1000	1000	1050	23.542
	NP	V	1000	500	800	14.922
		NP	1000	500	800	14.922
		NP+1000	1000	1200		25.252
		P	1000	1000	1200	25.252
	NP+V	1700		1000	1510	48.978
		NP	1700	1000	1510	48.978
		NP+1700	1000	1510		48.978
		P	1700	1000	1510	48.978
	P	V	2500	1000	1750	78.914
		NP	2500	1000	1750	78.914
		NP+2500	1000	1750		78.914
		P	2500	1000	1750	78.914
Large	V	V	1000	500	700	13.770
		NP	1000	500	700	13.770
		NP+1000	1000	1100		24.105
		P	1000	1000	1100	24.105
	NP	V	1700	500	1010	29.465
		NP	1700	500	1010	29.465
		NP+1700	1000	1425		47.320
		P	1700	1000	1425	47.320
	NP+V	1700		1000	1510	48.978
		NP	1700	1000	1510	48.978
		NP+1700	1000	1510		48.978
		P	1700	1000	1510	48.978
	P	V	2500	1000	1750	78.914
		NP	2500	1000	1750	78.914
		NP+2500	1000	1750		78.914
		P	2500	1000	1750	78.914

*V= visual approach; NP = nonprecision instrument approach with visibility minimums more than $\frac{3}{4}$ statute mile; NP+ = nonprecision instrument approach with visibility minimums as low as $\frac{3}{4}$ statute mile; P = precision instrument approach.

Source: Federal Aviation Administration [5].

TABLE 6-16 Runway Protection Zone Dimensions

is reasonably restricted to the extent possible through the use of such measures as the adoption of zoning ordinances. A model zoning ordinance to limit the height of objects around airports is published by the FAA [6].

Subpart C of FAR Part 77 establishes standards for determining obstructions to air navigation. The standards apply to existing and man-made objects, objects of natural growth, and terrain.

In order to determine whether an object is an obstruction to air navigation, several imaginary surfaces are established with relation to the airport and to each end of a runway. The size of the imaginary surfaces depends on the category of each runway (e.g., utility or transport) and on the type of approach planned for that end of the runway (e.g., visual, nonprecision instrument, or precision instrument).

The principal imaginary surfaces are shown in Fig. 6-25. They are described as follows:

1. *Primary surface.* The primary surface is a surface longitudinally centered on a runway. When the runway is paved, the primary surface extends 200 ft beyond each end of the runway. When the runway is unpaved, the primary surface coincides with each end of the runway. The elevation of the primary surface is the same as the elevation of the nearest point on the runway centerline.
2. *Horizontal surface.* The horizontal surface is a horizontal plane 150 ft above the established airport elevation, the perimeter of which is constructed by swinging arcs of specified radii from the center of each end of the primary surface of each runway and connecting each arcs by lines tangent to those arcs.
3. *Conical surface.* The conical surface is a surface extending outward and upward from the periphery of the horizontal surface at a slope of 20 horizontal to 1 vertical for a horizontal distance of 4000 ft.
4. *Approach surface.* The approach surface is a surface longitudinally centered on the extended runway centerline and extending outward and upward from each end of a runway at a designated slope based upon the type of available or planned approach to the runway.
5. *Transitional surface.* Transitional surfaces extend outward and upward at right angles to the runway centerline plus the runway centerline extended at a slope of 7 to 1 from the sides of the primary surface up to the horizontal surface and from the sides of the approach surfaces. The width of the transitional surface provided from each edge of the approach surface is 5000 ft.

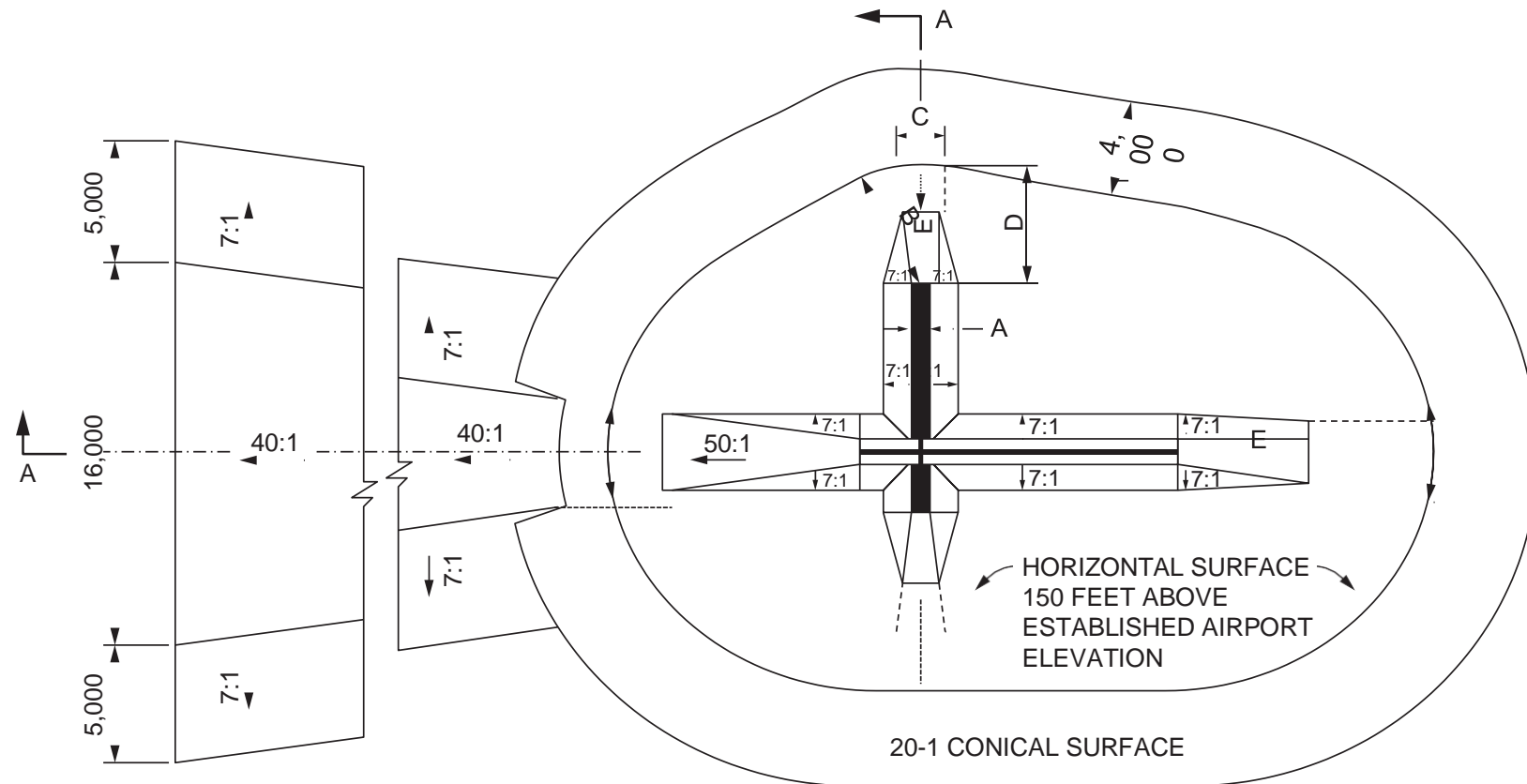


FIGURE 6-25 FAR part 77 imaginary surfaces.

OBSTRUCTION IDENTIFICATION SURFACES
FEDERAL AVIATION REGULATIONS PART 77

DIM	ITEM	DIMENSIONAL STANDARDS (FEET)					
		VISUAL RUNWAY		NON - PRECISION INSTRUMENT RUNWAY			PRECISION INSTRUMENT RUNWAY
		A	B	A	B		
A	WIDTH OF PRIMARY SURFACE AND APPROACH SURFACE WIDTH AT INNER END	250	500	500	500	1,000	1,000
B	RADIUS OF HORIZONTAL SURFACE	5,000	5,000	5,000	10,000	10,000	10,000
		VISUAL APPROACH		NON - PRECISION INSTRUMENT APPROACH			PRECISION INSTRUMENT APPROACH
		A	B	A	B		
					C	D	
C	APPROACH SURFACE WIDTH AT END	1,250	1,500	2,000	3,500	4,000	16,000
D	APPROACH SURFACE LENGTH	5,000	5,000	5,000	10,000	10,000	*
E	APPROACH SLOPE	20:1	20:1	20:1	34:1	34:1	*

- A - UTILITY RUNWAYS
- B - RUNWAYS LARGER THAN UTILITY
- C - VISIBILITY MINIMUMS GREATER THAN 3/4 MILE
- D - VISIBILITY MINIMUMS AS LOW AS 3/4 MILE
- * - PRECISION INSTRUMENT APPROACH SLOPE IS 50:1 FOR INNER 10,000 FEET AND 40:1 FOR AN ADDITIONAL 40,000 FEET

FIGURE 6-26 Part 77 Imaginary Surface Dimensions, ft.

Dimensions of the several imaginary surfaces are shown in Fig. 6-26.

In addition to the surfaces defined earlier, other standards for determining obstructions to air navigation are contained in FAR Part 77. Existing and future objects, whether stationary or mobile, are considered to be obstructions to air navigation if they are of greater height than any of the following heights or surfaces:

1. A height of 500 ft above ground level at the site of the object.
2. A height that is 200 ft above ground level or 200 ft above the established airport elevation, whichever is greater, within 3 nautical miles of the established reference point at an airport with its longest runway more than 3200 ft in actual length. This height increases in the ratio of 100 ft for each additional nautical mile of distance from the reference point up to a maximum of 500 ft.
3. A height within a terminal obstacle clearance area, including an initial approach segment, a departure area, and a circling approach area, which would result in the vertical distance between any point on the object and an established minimum instrument flight altitude within that area or segment to be less than the required obstacle clearance.
4. A height within an en route obstacle clearance area, including turn and termination areas, of a federal airway or approved off-airway route, that would increase the minimum obstacle clearance altitude.
5. The surface of a takeoff and landing area of an airport or any of the imaginary surfaces defined earlier.

6. Except for traverse ways on or near an airport with an operative ground traffic control service furnished by the air traffic control tower or by airport management and coordinated with the air traffic control service, the heights of traverse ways must be increased by 17 ft for interstate highways, 15 ft for any other public roadway, 10 ft or the height of the highest mobile object that would normally traverse the road, whichever is greater, for a private road, 23 ft or an amount equal to the height of the highest mobile object that would normally traverse it for railroads, waterways, or any other thoroughfare not previously mentioned.

Subpart B of FAR Part 77 identifies circumstances where notice is required to be given to the FAA when certain construction or alteration activities are proposed. These include the circumstances associated with the standards given above and also any construction or alteration of greater height than an imaginary surface extending outward and upward at one of the following slopes [9]:

1. A slope of 100 horizontal to 1 vertical for a horizontal distance of 20,000 ft from the nearest point of the nearest runway at an airport or seaplane base with at least one runway more than 3200 ft in actual length.
2. A slope of 50 horizontal to 1 vertical for a horizontal distance of 10,000 ft from the nearest point of the nearest runway at an airport or seaplane base with its longest runway no more than 3200 ft in actual length.
3. A slope of 25 horizontal to 1 vertical for a horizontal distance of 5000 ft from the nearest point of the nearest takeoff and landing area for a heliport.

FAR Part 77 imposes strict requirements on both airport sponsors and others associated with construction activities in the vicinity of airports which should be referenced prior to initiating construction activities.

ICAO Annex 14

The ICAO requirements are similar to FAR Part 77 with the following exceptions. ICAO separates arrivals and departures and specifies dimensions for approach surfaces and takeoff climb surfaces for departures. The horizontal surface specified by ICAO is a circle whose center is at the *airport reference point*, whereas in FAR Part 77 it is not a circle nor is the airport reference point used to determine the horizontal surface. The airport reference point is the geometric centroid of the runway system at the airport based upon the lengths of the runways. The height of this surface is 150 ft above the airport elevation, the same as in Part 77. In FAR Part 77 the conical surface extends horizontally 4000 ft at a slope of 20 to 1 irrespective of the type of runway and

visibility. In ICAO Annex 14 [1, 2, 3] the slope of the conical surface is the same, but the horizontal distance varies depending upon the aerodrome reference code.

In FAR Part 77 the slope of the transitional surface is a constant 7 to 1, whereas in ICAO Annex 14 this slope is specified for runway reference codes 3 and 4. For other runways the slope is 5 to 1.

TERPS

As defined in FAA Order 8260.3b, TERPS (which stands for terminal instrument approach procedures) is a compilation of criteria used to design published standard procedures for aircraft using instrument-based navigation to depart and approach to airport facilities. These procedures are designed based primarily on the performance characteristics of aircraft, the various types of instrument navigational aids that may be present at or around an airport, and currently existing natural and man-made objects surrounding the airport. As part of these procedures, minimum climb-out gradients for aircraft departures, and minimum descent gradients and safe operating altitudes for aircraft approaches are defined. While TERPS contain standards for creating such procedures, for any given runway at any given airport, one or more approach and departure procedures may be defined, each of which may be entirely unique, based on the airport environment.

With respect to airport design, TERPS defines a “required obstacle clearance” (ROC) value. For aircraft operating within the airport environment, this value is typically as low as 250 ft above the highest object near the runway. The required obstacle clearance values for a published procedure in turn define the TERPS obstacle clearance surface (OCS), as illustrated in Fig. 6-27. The typical slopes for obstacle clearance surfaces for aircraft on approach is on the order of 318 ft/nmi and for departures approximately 200 ft/nmi.

A typical TERPS procedure consists of a series of segments, including climb, en route, initial approach, intermediate approach, final approach, and missed approach segments, that are created based

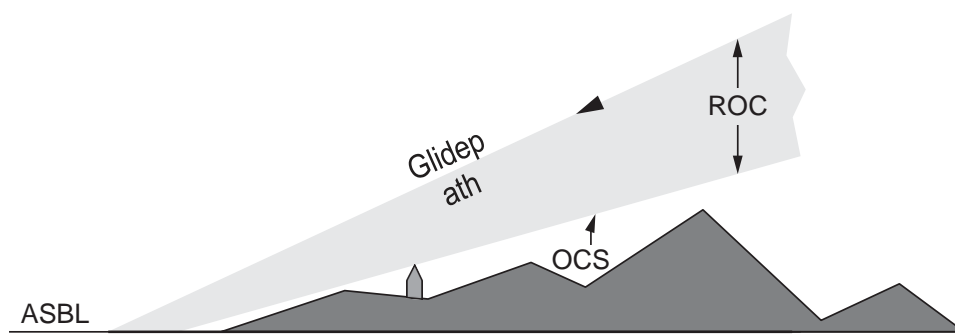


FIGURE 6-27 TERPS obstacle clearance surface.

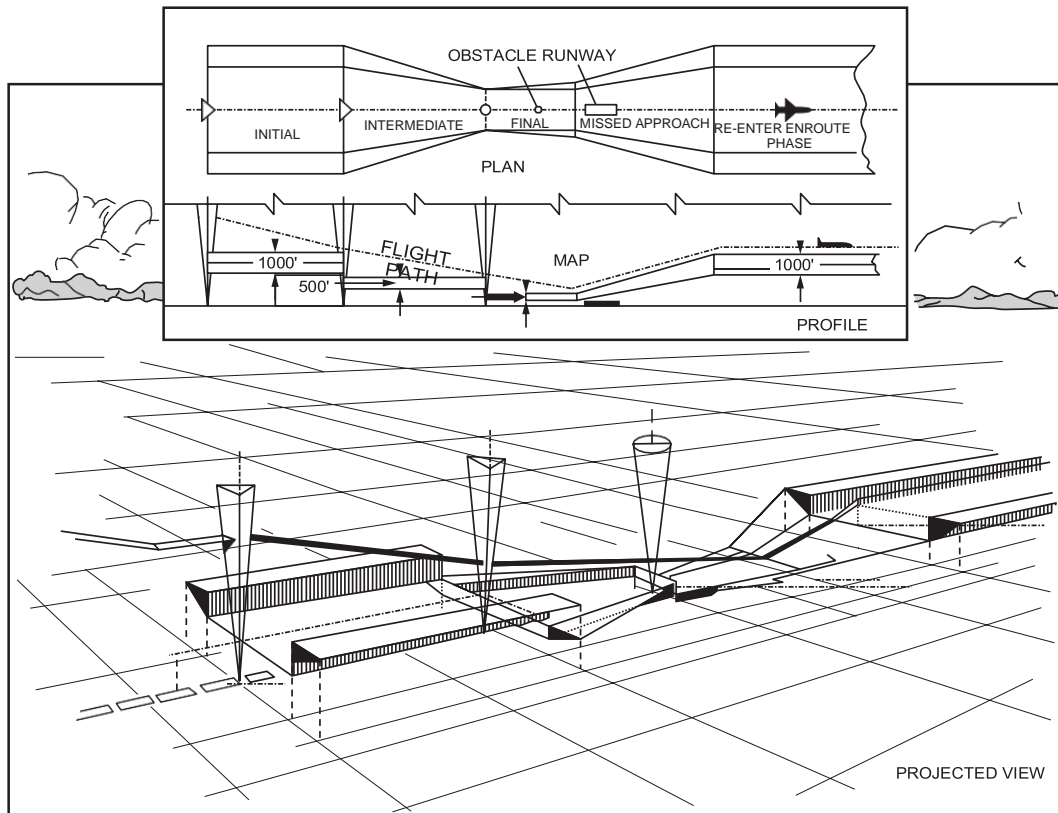


FIGURE 6-28 Typical TERPS procedure segments.

on the above standards and the existing terrain and obstacle environment for any given runway, as illustrated in Fig. 6-28.

It is widely understood that protecting airspace for TERPS is a complex process, often unique to each airport. For planning purposes, however, a slope with 40:1 gradient and 15° from the runway end should be considered as TERPS obstacle clearance surface criteria. Runways with the intention of being supported by published instrument procedures should be designed in such a manner to avoid any natural or man-made obstacles that penetrate this surface. Once a runway exists, airport planners should work to ensure that future development does not conflict with TERPS or FAR Part 77 obstacle clearance requirements.

Runway End Siting Requirements

The specifications for determining obstacles to safe air navigation to existing runways are described in FAR Part 77 and TERPS procedures. However, when locating, or siting a runway, the FAA prescribes a different, yet complimentary set of specifications. These specifications are published in Appendix 2 of Advisory Circular AC 150/5300-13, identified in Table 6-17, and illustrated in Figs. 6-29 through 6-31.

	Runway Type	Dimensional Standards ^a ft					Slope/ OCS
		A	B	C	D	E	
1	Approach end of runways expected to serve small airplanes with approach speeds less than 50 kn (visual runways only, day/night)	0	60	150	500	2,500	15:1
2	Approach end of runways expected to serve small airplanes with approach speeds of 50 kn or more (visual runways only, day/night)	0	125	350	2,250	2,750	20:01
3	Approach end of runways expected to serve large airplanes (visual day/night); or instrument minimums ≥ 1 statute mile (day only)	0	200	500	1,500	8,500	20:1
4	Approach end of runways expected to support instrument night circling ^a	200	200	1,700	10,000	0	20:1
5	Approach end of runways expected to support instrument straight in night operations. Serving approach category A and B aircraft only. ^a	200	200	1,900	10,000 ^b	0	20:1
6	Approach end of runways expected to support instrument straight in night operations serving greater than approach category B aircraft. ^a	200	400	1,900	10,000 ^b	0	20:1
7 ^{e,f,g,h}	Approach end of runways expected to accommodate approaches with positive vertical guidance (GQS)	0	$\frac{1}{2}$ width runway +100	760	10,000 ^b	0	30:1
8	Approach end of runways expected to accommodate instrument approaches having visibility minimums $\geq \frac{3}{4}$ but < 1 statute mile, day or night	200	400	1,900	10,000 ^b	0	20:1

9	Approach end of runways expected to accommodate instrument approaches having visibility minimums < ¾ statute mile or precision approach (ILS, GLS, or MLS) day or night	200	400	1,900	10,000 ^b	0	34:1
10	Approach runway ends having category II approach minimums or greater	The criteria are set forth in TERPS. Order 8260.3.					
11	Departure runway ends for all instrument operations	0 ^d	See Fig. 6-30				40:1
12	Departure runway ends supporting air carrier operations ^c	0 ^d	See Fig. 6-31				625:1

*Dimensional standards illustrated in Fig. 6-29

Notes:

^aLighting of obstacle penetrations to this surface or the use of a VGSI, as defined by the TERPS order, may avoid displacing the threshold.

^b10,000 ft is a nominal value for planning purposes. The actual length of these areas is dependent upon the visual descent point position for 20:1 and 34:1 and decision altitude point for the 30:1.

^cAny penetration to this surface will limit the runway end to nonprecision approaches. No vertical approaches will be authorized until the penetration(s) is/are removed except obstacles fixed by function and/or allowable grading.

^dDimension A is measured relative to departure end of runway (DER) or TODA (to include clearway).

^eData collected regarding penetrations to this surface are provided for information and use by the air carriers operating from the airport. These requirements do not take effect until January 1, 2009.

^fSurface dimensions/obstacle clearance surface (OCS) slope represent a nominal approach with 3° GPA, 50' TCH, < 500' HAT. For specific cases refer to TERPS. The obstacle clearance surface slope (30:1) represents a nominal approach of 3° (also known as the glide path angle). This assumes a threshold crossing height of 50 ft. Three degrees is commonly used for ILS systems and VGSI aiming angles. This approximates a 30:1 approach angle that is between the 34:1 and the 20:1 notice surfaces of Part 77. Surfaces cleared to 34:1 should accommodate a 30:1 approach without any obstacle clearance problems.

^gFor runways with vertically guided approaches the criteria in Row 7 is in addition to the basic criteria established within the table, to ensure the protection of the glide path qualification surface.

^hFor planning purposes, sponsors and consultants determine a tentative decision altitude based on a 3° glide path angle and a 50-ft threshold crossing height.

TABLE 6-17 Runway End Siting Requirements Dimensions Table

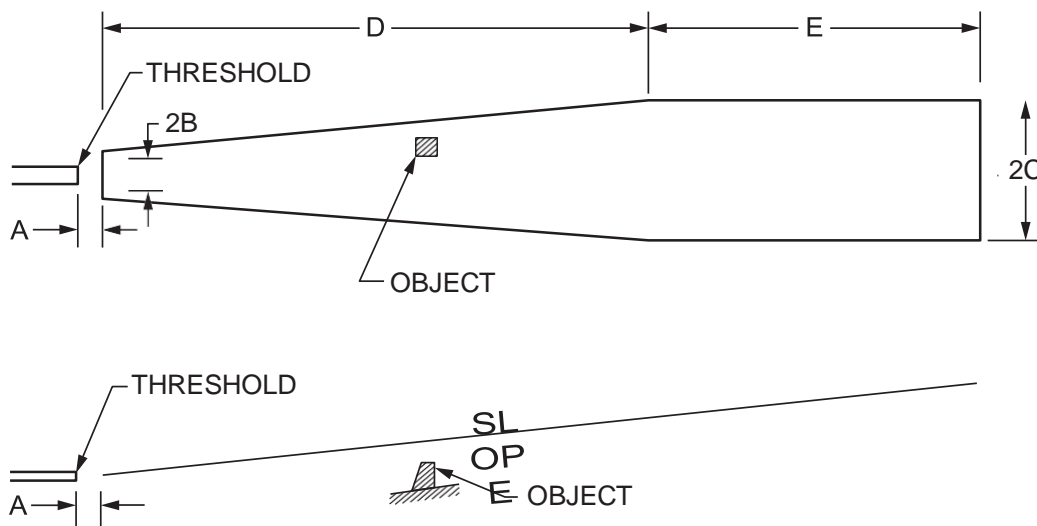
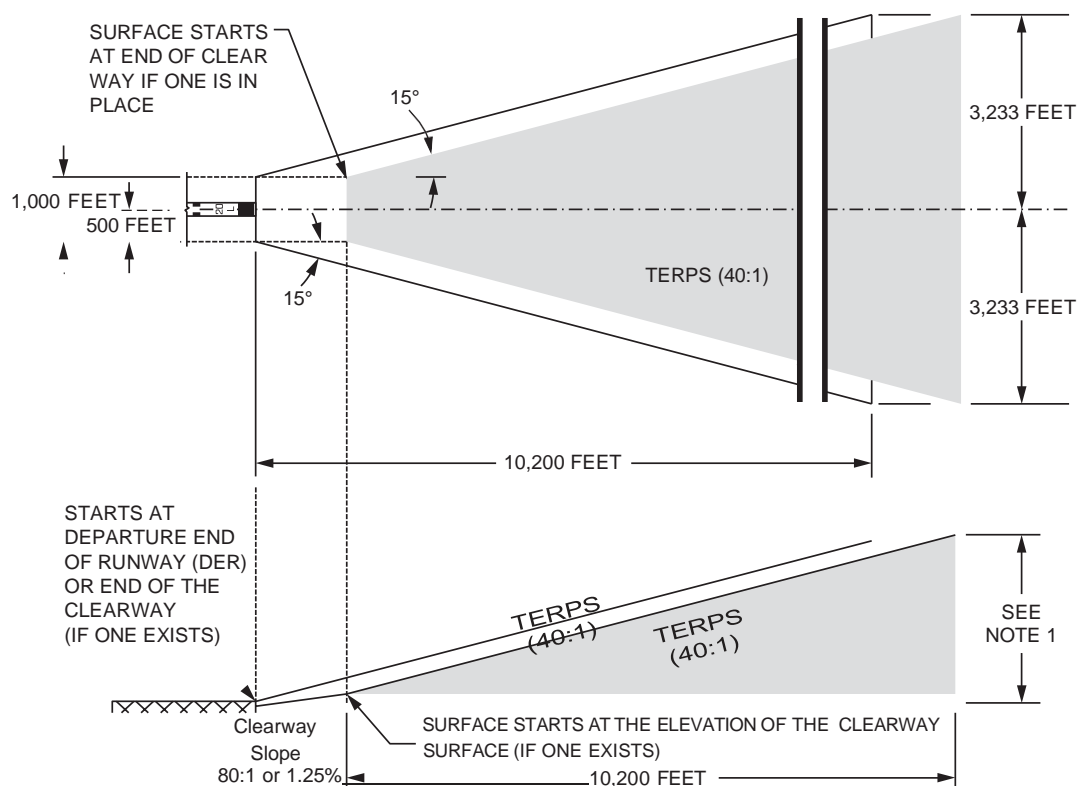


FIGURE 6-29 Runway end siting requirements.

These specifications are used to site the location of a runway's threshold so that approach and departure procedures associated with that runway are not adversely affected by existing obstacles or terrain. The siting specifications vary depending on a number of runway use conditions, including



NOTES:

1. THIS IS AN INTERPRETATION OF THE APPLICATION OF THE TERPS SURFACE ASSOCIATED WITH A CLEARWAY.

FIGURE 6-30 TERPS departure obstacle identification surfaces.

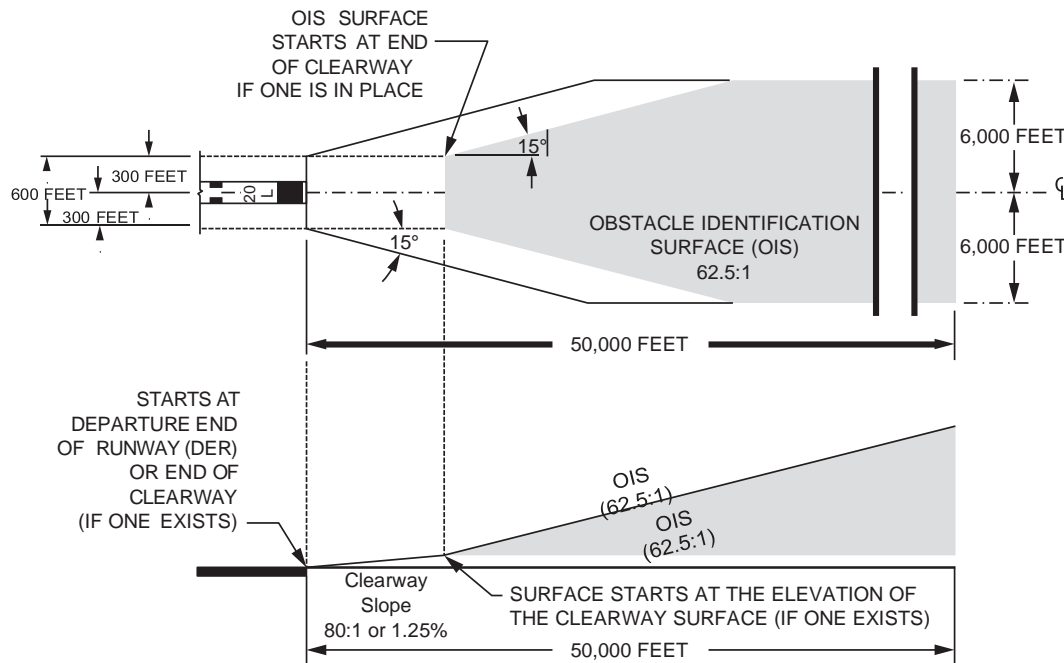


FIGURE 6-31 One engine inoperative obstacle identification surface (62.5:1).

- The approach speed of arriving aircraft
- The approach category of arriving aircraft
- Day versus night operations
- Types of instrument approaches
- The presence of published instrument departure procedures
- The use of the runway by air carriers

Runway end siting requirements are often the most confusing as well as overlooked element of runway planning. Care should be given to fully understand the purpose of the planned runway, the type of aircraft that will be using the runway, the current and future instrument approach procedures associated with the runway, and of course any terrain or obstacles in the vicinity.

Should an object penetrate any of the surfaces at the site of a runway, the airport planner has the option of displacing the runway threshold, as illustrated in Fig. 6-32. Displacing the threshold allows the airport planner to design runways with sufficient lengths to accommodate aircraft departures, while also allowing arrivals to safely approach the runway by maintaining sufficient clearance from upstream obstacles. Displacing the threshold does carry the penalty of reducing available runway lengths for landing. The FAA recommends avoiding the need for displaced thresholds when possible, but recognizes their benefits in the wake of no other alternatives.

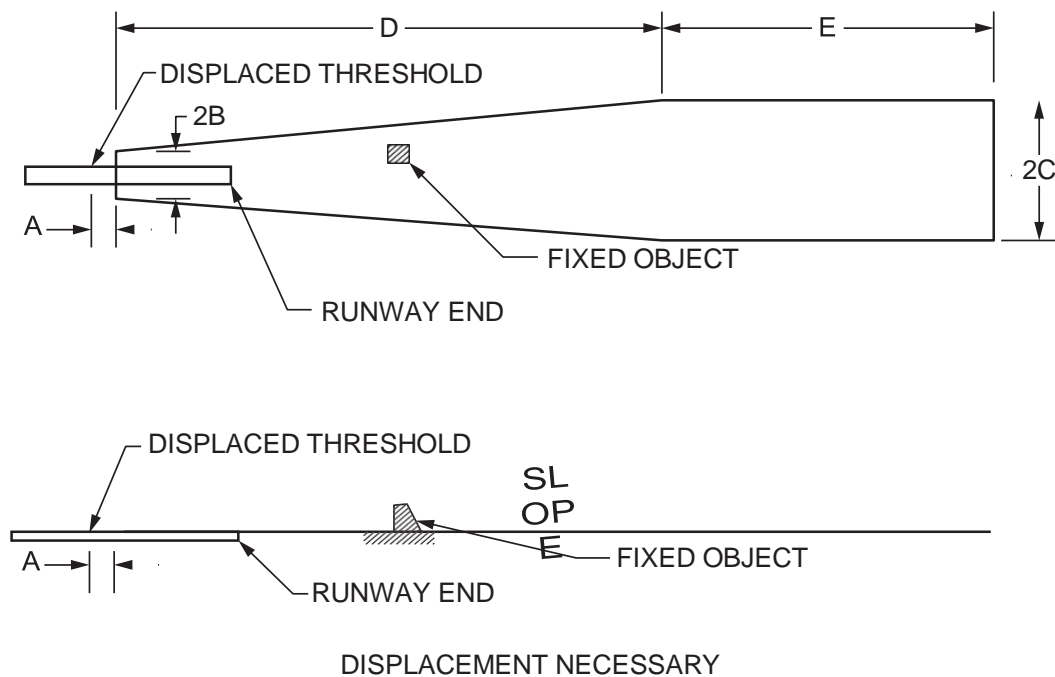


FIGURE 6-32 Use of displaced threshold, runway siting requirements.

Taxiways and Taxilanes

Taxiways are defined paths on the airfield surface which are established for the taxiing of aircraft and are intended to provide a linkage between one part of the airfield and another. The term “dual parallel taxiways” refers to two taxiways parallel to each other on which airplanes can taxi in opposite directions. An apron taxiway is a taxiway located usually on the periphery of an apron intended to provide a through taxi route across the apron. A taxilane is a portion of the aircraft parking area used for access between the taxiways and the aircraft parking positions. ICAO defines an aircraft stand taxilane as a portion of the apron intended to provide access to the aircraft stands only.

In order to provide a margin of safety in the airport operating areas, the trafficways must be separated sufficiently from each other and from adjacent obstructions. Minimum separations between the centerlines of taxiways, between the centerlines of taxiways and taxilanes, and between taxiways and taxilanes and objects are specified in order that aircraft may safely maneuver on the airfield.

Widths and Slopes

Since the speeds of aircraft on taxiways are considerably less than on runways, criteria governing longitudinal slopes, vertical curves, and sight distance are not as stringent as for runways. Also the lower speeds permit the width of the taxiway to be less than that of the runway. The principal geometric design features of interest are listed in Tables 6-18 and 6-19 for the FAA. ICAO standards are listed in Tables 6-20 and 6-21.

	Airplane Design Group					
	I	II	III	IV	V	VI
Width	25	35	50 ^a	75	75	100
Edge safety margin ^b	5	7.5	10 ^c	15	15	20
Shoulder width	10	10	20	25	35 ^d	40 ^d
Safety area width ^e	49	79	118	171	214	262
Object-free area width						
Taxiway ^f	89	131	186	259	320	386
Taxilane ^g	79	115	162	225	276	334
Separations						
Taxiway centerline to taxiway centerline ^h fixed or movable object ⁱ	69	105	152	215	267	324
	44.5	62.5	93	129.5	160	193
Taxilane centerline to taxilane centerline ^j fixed or movable object ^k	64	97	140	198	245	298
	39.5	57.5	81	112.5	138	167

^aFor airplanes in airplane design group III with a wheelbase equal to or greater than 60 ft, the standard taxiway width is 60 ft.

^bThe taxiway edge safety margin is the minimum acceptable between the outside of the airplane wheels and the pavement edge.

^cFor airplanes in airplane design group III with a wheelbase equal or greater than 60 ft, the taxiway edge safety margin is 15 ft.

^dAirplanes in airplane design groups V and VI normally stabilized or paved taxiway shoulder surfaces.

^eMay use aircraft wingspan in lieu of these values.

^fMay use 1.4 wingspan plus 20 ft in lieu of these values.

^gMay use 1.2 wingspan plus 20 ft in lieu of these values.

^hMay use 1.2 wingspan plus 10 ft in lieu of these values.

ⁱMay use 0.7 wingspan plus 10 ft in lieu of these values.

^jMay use 1.1 wingspan plus 10 ft in lieu of these values.

^kMay use 0.6 wingspan plus 10 ft in lieu of these values.

Source: Federal Aviation Administration [6].

TABLE 6-18 Taxiway Dimensional Standards, ft

Taxiway and Taxilane Separation Requirements

FAA Separation Criteria

The separation criteria adopted by the FAA are predicated upon the wingtips of the aircraft for which the taxiway and taxilane system have been designed and provide a minimum wingtip clearance on

	Aircraft Approach Category				
	A	B	C	D	E
Gradient (%)					
Taxiway, shoulder and safety area					
Longitudinal					
Maximum	2.0	2.0	1.5	1.5	1.5
Maximum change	3.0	3.0	3.0	3.0	3.0
Taxiway transverse					
Minimum	1.0	1.0	1.0	1.0	1.0
Maximum	2.0	2.0	1.5	1.5	1.5
Shoulder transverse					
Minimum	3.0	3.0	1.5*	1.5*	1.5*
Maximum [†]	5.0	5.0	5.0	5.0	5.0
Safety area transverse					
Minimum	3.0	3.0	1.5	1.5	1.5
Maximum	5.0	5.0	3.0	3.0	3.0
Vertical curve (ft)					
Minimum length [‡]	100	100	100	100	100
Minimum distance between points of intersection [§]	100	100	100	100	100

*A minimum of 3 percent for turf.

[†]A slope of 5 percent is recommended for a 10-ft width adjacent to the pavement areas to promote drainage.

[‡]For each 1 percent of grade change.

[§]Distance is multiplied by the sum of the absolute grade changes in percent.

Source: Federal Aviation Administration [6].

TABLE 6-19 Taxiway Gradient Standards

these facilities. The required separation between taxiways, between a taxiway and a taxilane, or between a taxiway and a fixed or movable object requires a minimum wingtip clearance of 0.2 times the wingspan of the most demanding aircraft in the airplane design group plus 10 ft. This clearance provides a minimum taxiway centerline to a parallel taxiway centerline or taxilane centerline separation of 1.2 times the wingspan of the most demanding aircraft plus 10 ft, and between a taxiway centerline and a fixed or movable object of 0.7 times the wingspan of the most demanding aircraft plus 10 ft. This

	Aerodrome Code Letter				
	A	B	C	D	E
Width					
Pavement	7.5	10.5	15*	18 [†]	23
Pavement and shoulder			25	38	44
Edge safety margin, U_1	1.5	2.25	3 [‡]	4.5	4.5
Strip	27	39	57	85	93
Graded portion of strip	22	25	25	38	44
Minimum separation					
Taxiway centerline to taxiway centerline	21	31.5	46.5	68.5	81.5
Object	13.5	19.5	28.5	42.5	49
Aircraft stand taxilane to object	12	16.5	24.5	36	42.5

*18 m if used by aircraft with a wheelbase equal to or greater than 18 m.

[†]23 m is used by aircraft with an outer main gear wheel span equal to or greater than 9 m.

[‡]4.5 m. if intended to be used by airplane with a wheelbase equal to or greater than 18 m.

Source: International Civil Aviation Organization [2, 3, 4].

TABLE 6-20 Taxiway Dimensional Standards, m

separation is also applicable to aircraft traversing through a taxiway on an apron or ramp. This separation may have to be increased to accommodate pavement widening on taxiway curves. It is recommended that a separation of at least 2.6 times the wheelbase of the most demanding aircraft be provided to accommodate a 180° turn when the pavement width is designed for tracking the nose wheel on the centerline.

The taxilane centerline to a parallel taxilane centerline or fixed or movable object separation in the terminal area is predicated on a wingtip clearance of approximately half of that required for an apron taxiway. This reduction in clearance is based on the consideration that taxiing speed is low in this area, taxiing is precise, and special guidance techniques and devices are provided. This requires a wingtip clearance or wingtip-to-object clearance of 0.1 times the wingspan of the most demanding aircraft plus 10 ft. Therefore, this establishes a minimum separation between the taxilane centerlines of 1.1 times the wingspan of the most demanding aircraft plus 10 ft, and between a taxilane centerline and a fixed or movable object of 0.6 times the wingspan of the most demanding aircraft plus 10 ft [6]. Therefore, when dual parallel taxilanes are provided in the terminal apron area, the taxilane object-free area becomes 2.3 the wingspan of the most demanding aircraft plus 30 ft.

	Aerodrome Code Letter				
	A	B	C	D	E
Gradient (%)					
Pavement longitudinal					
Maximum	3.0	3.0	1.5	1.5	1.5
Maximum change	4.0	4.0	3.33	3.33	3.33
Pavement transverse					
Maximum	2.0	2.0	1.5	1.5	1.5
Strip					
Maximum transverse					
Graded portion					
Upward	3.0	3.0	2.5	2.5	2.5
Downward	5.0	5.0	5.0	5.0	5.0
Ungraded portion					
Upward	5.0	5.0	5.0	5.0	5.0
Vertical curve (m)					
Minimum length*	25	25	30	30	30

*For each 1 percent of grade change.

Source: International Civil Aviation Organization [2, 3, 4].

TABLE 6-21 Taxiway Gradient Standards

The separation criteria are presented in Table 6-18. The values indicated in this table are based upon the specifications using the largest wingspan in each airplane design group. As noted in this table, the required separations may be reduced to those that would result using the actual wingspan of the design aircraft.

ICAO Separation Criteria

The separation criteria adopted by ICAO are also predicated upon the wingtips of the aircraft for which the taxiway and taxilane system have been designed and providing a minimum wingtip clearance on these facilities, but also consider a minimum clearance between the outer main gear wheel and the taxiway edge. The required separation between taxiways or between a taxiway and a taxilane requires a minimum wingtip clearance, C_1 , of 3 m for aerodrome code letter A and B runways, 4.5 m for aerodrome code letter C runways, and

7.5 m for aerodrome code letter D and E runways. The minimum clearance between the edge of each taxiway and the outer main gear wheels, the taxiway edge safety margin U_1 , is given in Table 6-20. This clearance provides a minimum taxiway centerline to a parallel taxiway centerline or taxilane centerline separation given by Eq. (6-1).

$$S_{TT} = WS + 2U_1 + C_1 \quad (6-1)$$

where S_{TT} = minimum taxiway-to-taxiway or taxiway-to-taxilane separation

WS = wingspan of the most demanding aircraft

U_1 = taxiway edge safety margin

C_1 = minimum wingtip clearance

Therefore, for example, an ICAO aerodrome code letter E runway, which accommodates aircraft with wingspans up to 65 m, requires a taxiway centerline to a taxiway centerline or a taxilane centerline separation from Eq. (6-1) of $65 + 2(4.5) + 7.5 = 81.5$ m.

The required separation between a taxiway centerline or an apron taxiway centerline and a fixed or movable object is found from Eq. (6-2).

$$S_{TO} = 0.5 WS + U_1 + C_2 \quad (6-2)$$

where S_{TO} is the minimum taxiway or apron taxiway to a fixed or movable object separation and C_2 is the required clearance between a wingtip and an object.

The required clearance between a wingtip and an object C_2 is 4.5 m for aerodrome code letter A runways, 5.25 m for aerodrome code letter B runways, 7.5 m for aerodrome code letter C runways, and 12 m for aerodrome code letter D and E runways.

The required separation between an aircraft stand taxilane centerline and a fixed or movable object is found from Eq. (6-3).

$$S_{ATO} = 0.5 WS + U_2 + C_1 \quad (6-3)$$

where S_{ATO} is the minimum aircraft stand taxilane to fixed or movable object separation and U_2 is the aircraft stand safety margin.

Since aircraft moving on the aircraft stand taxilane are moving at low speed and are often under positive ground guidance, the aircraft stand safety margin is less than on the taxiway system. The value for this safety margin U_2 is 1.5 m for aerodrome code letter A and B airports, 2 m for aerodrome code letter C airports, and 2.5 m for aerodrome code letter D or E airports. The taxiway and taxilane separation criteria adopted by ICAO are given in Table 6-20.

Sight Distance and Longitudinal Profile

As in the case of runways, the number of changes in longitudinal profile for taxiways is limited by sight distance and minimum distance between vertical curves.

The FAA does not specify line of sight requirements for taxiways other than those discussed earlier related to runway and taxiway intersections. However, the sight distance along a runway from an intersecting taxiway needs to be sufficient to allow a taxiing aircraft to enter or cross the runway safely. The FAA specifies that from any point on the taxiway centerline the difference in elevation between that point and the corresponding point on a parallel runway, taxiway, or apron edge is 1.5 percent of the shortest distance between the points.

ICAO requires that the surface of the taxiway should be seen for a distance of 150 m from a point 1.5 m above the taxiway for aerodrome code letter A runways, for a distance of 200 m from a point 2 m above the taxiway for aerodrome code letter B runways, and for a distance of 300 m from a point 3 m above the taxiway for aerodrome code letter C, D, or E runways.

In regard to longitudinal profile of taxiways, the ICAO does not specify the minimum distance between the points of intersection of vertical curves. The FAA specifies that the minimum distance for both utility and transport category airports should be not less than the product of 100 ft multiplied by the sum of the absolute percentage values of change in slope.

Exit Taxiway Geometry

The function of exit taxiways, or runway turnoffs as they are sometimes called, is to minimize runway occupancy by landing aircraft. Exit taxiways can be placed at right angles to the runway or some other angle to the runway. When the angle is on the order of 30°, the term high-speed exit is often used to denote that it is designed for higher speeds than other exit taxiway configurations. In this chapter, specific dimensions for high-speed exit, right-angle exit (low-speed) taxiways are presented. The dimensions presented here are the results obtained from research conducted many years ago [13] and subsequent research conducted by the FAA.

The earlier tests [13] were conducted on wet and dry concrete and asphalt pavement with various types of civil and military aircraft in order to determine the proper relationship between exit speed and radii of curvature and the general configuration of the taxiway. A significant finding of the tests was that at high speeds a compound curve was necessary to minimize tire wear on the nose gear and, therefore, the central or main curve radius R_2 should be preceded by a much larger radius curve R_1 .

Aircraft paths in the test approximated a spiral. A compound curve is relatively easy to establish in the field and begins to approach

the shape of a spiral, thus the reason for suggesting a compound curve. The following pertinent conclusions were reached as a result of the tests [13]:

1. Transport category and military aircraft can safely and comfortably turn off runways at speeds on the order of 60 to 65 mi/h on wet and dry pavements.
2. The most significant factor affecting the turning radius is speed, not the total angle of turn or passenger comfort.
3. Passenger comfort was not critical in any of the turning movements.
4. The computed lateral forces developed in the tests were substantially below the maximum lateral forces for which the landing gear was designed.
5. Insofar as the shape of the taxiway is concerned, a slightly widened entrance gradually tapering to the normal width of taxiway is preferred. The widened entrance gives the pilot more latitude in using the exit taxiway.
6. Total angles of turn of 30° to 45° can be negotiated satisfactorily. The smaller angle seems to be preferable because the length of the curved path is reduced, sight distance is improved, and less concentration is required on the part of the pilots.
7. The relation of turning radius versus speed expressed by the formula below will yield a smooth, comfortable turn on a wet or dry pavement when f is made equal to 0.13.

$$R = \frac{V^2}{15f} \quad (6-4)$$

where V is the velocity in mi/h and f is the coefficient of friction.

8. The curve expressed by the equation for R_2 should be preceded by a larger radius curve R_1 at exit speeds of 50 to 60 mi/h. The larger radius curve is necessary to provide a gradual transition from a straight tangent direction section to a curved path section. If the transition curve is not provided tire wear on large jet transports can be excessive.
9. The length of the transition curve can be roughly approximated by the relation

$$L = \frac{V^3}{CR_2} \quad (6-5)$$

where V is in feet per second, R_2 is in feet, and C was found experimentally to be on the order of 1.3.

10. Sufficient distance must be provided to comfortably decelerate an aircraft after it leaves the runway. It is suggested that for the present this distance be based on an average rate of deceleration of 3.3 ft/s^2 . This applies only to transport category aircraft. Until more experience is gained with this type of operation the stopping distance should be measured from the edge of the runway.

A chart showing the relationship of exit speed to radii R_1 and R_2 , and length of transition curve L_1 is given in Fig. 6-33.

ICAO has indicated the relationship between aircraft speed and the radius of curvature of taxiway curves as illustrated in Table 6-22. For high-speed exit taxiways ICAO recommends a minimum radius

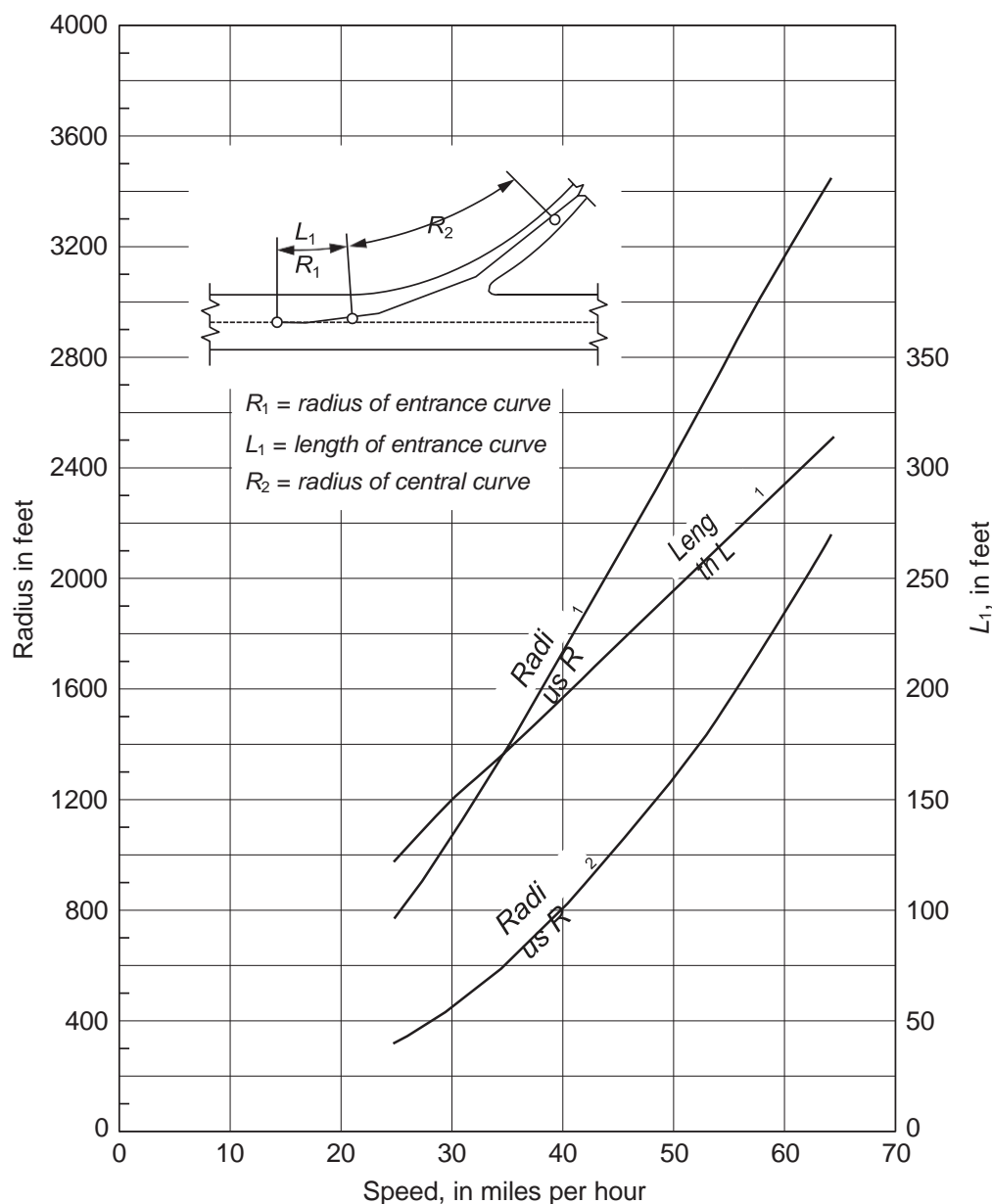


FIGURE 6-33 Radii of curvature and entrance curves for taxiways.

Taxling Speed		Radius of Exit curve	
mph	kph	Feet	Meters
10	16	50	15
20	32	200	60
30	48	450	135
40	64	800	240
50	80	1,250	375
60	96	1,800	540

Source: International Civil Aviation Organization [4].

TABLE 6-22 Radii of Curvature for Transport Category Aircraft

of curvature for the taxiway centerline of 275 m (900 ft) for aerodrome code number 1 and 2 runways and 550 m (1800 ft) for aerodrome code number 3 and 4 runways. This will allow exit speeds under wet conditions of 65 km/h (40 mi/h) for aerodrome code number 1 and 2 runways and 93 km/h (60 mi/h) for aerodrome code number 3 and 4 runways. It also recommends a straight tangent section after the turn-off curve to allow exiting aircraft to come to a full stop clear of the intersecting taxiway when the intersection is 30°. This tangent distance should be 35 m (115 ft) for aerodrome code number 1 and 2 runways and 75 m (250 ft) for aerodrome code number 3 and 4 runways [2, 4].

A configuration for an exit speed of 60 mi/h and a turnoff angle of 30° is shown in Fig. 6-34. The FAA recommends that the taxiway centerline circular curve be preceded by a 1400-ft spiral to smooth the transition from the runway centerline to the taxiway exit circular curve. ICAO recommends the same geometry for both of these high-speed exits. Right-angle or 90° exit taxiways, although not desirable from the standpoint of minimizing runway occupancy, are often

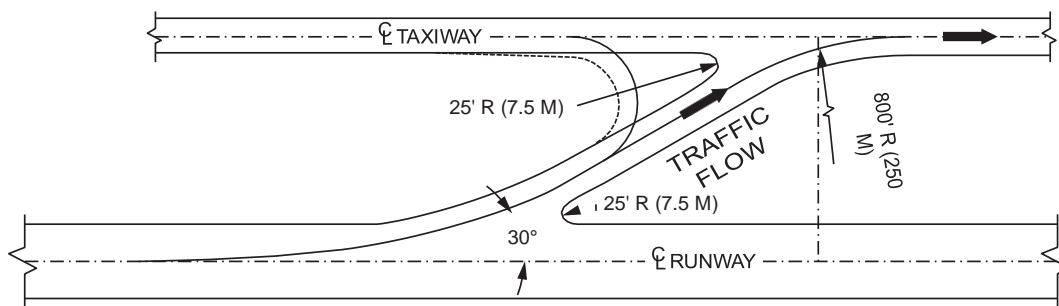


FIGURE 6-34 High-speed exit taxiway.

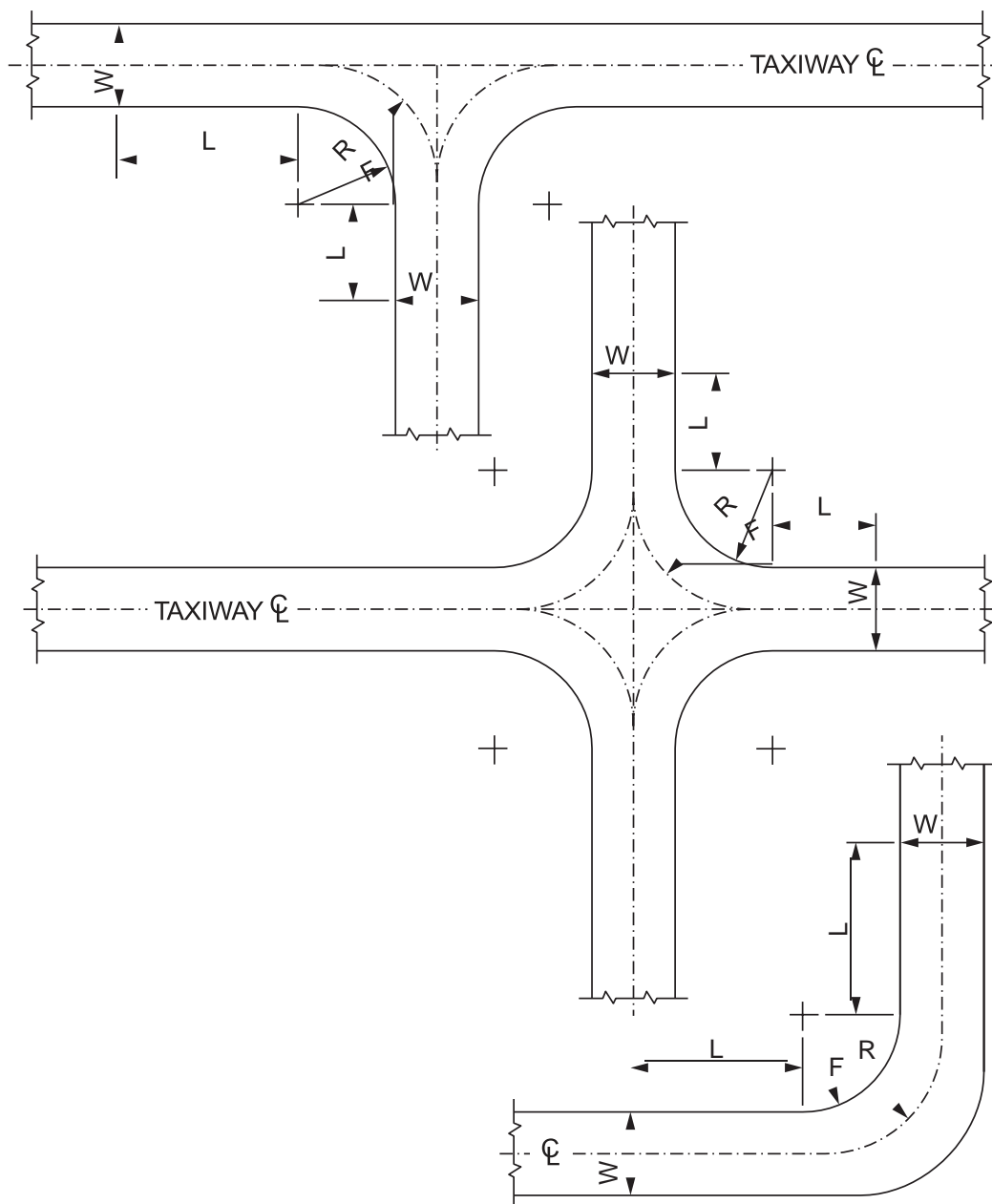


FIGURE 6-35 Common taxiway exit and intersection configurations.

constructed for other reasons. The configurations for a 90° exit and other common taxiway intersection configurations are illustrated in Fig. 6-35. The dimensions labeled in Fig. 6-35 are determined by the aircraft design group of the design aircraft. These dimensional standards are provided in Table 6-23.

Location of Exit Taxiways

The location of exit taxiways depends on the mix of aircraft, the approach and touchdown speeds, the point of touchdown, the exit speed, the rate of deceleration, which in turn depends on the condition of the pavement surface, that is, dry or wet, and the number of exits.

Item	Dim.*	Airplane Design Group					
		I	II	III [†]	IV	V	VI
Radius of taxiway turn [‡]	R	75	75	100	150	150	170
Length of lead-in to fillet	L	50	50	150	250	250	250
Fillet radius for tracking centerline	F	60	55	55	85	85	85
Fillet radius for judgmental oversteering symmetrical widening [§]	F	62.5	57.5	68	105	105	110
Fillet radius for judgmental	F	62.5	57.5	60	97	97	100
Oversteering one side widening [¶]							

*Letters correspond to the dimensions on Fig. 6-35.

[†]Airplanes in airplane design group III with a wheelbase equal to or greater than 60 ft should use a fillet radius of 50 ft.

[‡]Dimensions for taxiway fillet designs relate to the radius of taxiway turn specified.

[§]The center sketch of Fig. 6-35 displays pavement fillets with symmetrical taxiway widening.

[¶]The lower sketch of Fig. 6-35 displays a pavement fillet with taxiway widening on one side.

TABLE 6-23 FAA Taxiway Curvature Dimensional Standards, ft

While the rules for flying transport aircraft are relatively precise, a certain amount of variability among pilots is bound to occur especially in respect to braking force applied on the runway and the distance from runway threshold to touchdown. The rapidity and the manner in which air traffic control can process arrivals is an extremely important factor in establishing the location of exit taxiways. The location of exit taxiways is also influenced by the location of the runways relative to the terminal area.

Several mathematical analyses or models have been developed for optimizing exit locations. While these analyses have been useful

in providing an understanding of the significant parameters affecting location, their usefulness to planners has been limited because of the complexity of the analyses and a lack of knowledge of the inputs required for the application of the models. As a result greater use is made of much more simplified methods.

The landing process can be described as follows. The aircraft crosses the runway threshold and decelerates in the air until the main landing gear touches the surface of the pavement. At this point the nose gear has not made contact with the runway. It may take as long as 3 s to do so. When it does, reverse thrust or wheel brakes or a combination of both are used to reduce the forward speed of the aircraft to exit velocity. Empirical analysis has revealed that the average deceleration of air-carrier aircraft on the runway is about 5 ft/s².

In the simplified procedure, an aircraft is assumed to touch down at 1.3 times the stall speed for a landing weight corresponding to 85 percent of the maximum structural landing weight. In lieu of computing the distance from threshold to touchdown, touchdown distances are assumed as fixed values for certain classes of aircraft. Typically these values range from 500 to 1500 ft from the runway threshold. To these distances are added the distances to decelerate to exit speed. These relationships may be approximated by Eqs. (6-6) and (6-7).

$$D = D_{td} + D_e \quad (6-6)$$

where D = distance from the runway threshold to the exits

D_{td} = distance from the runway threshold to the point where the aircraft touches down

D_e = the distance from the touchdown point to the exit

$$D_e = \frac{V_{td}^2 - V_e^2}{2a} \quad (6-7)$$

where V_{td} = aircraft speed at touchdown

V_e = exit speed of the aircraft

a = deceleration of the aircraft on the runway

Although approach and touchdown speeds vary, they can be approximated for locating exit taxiways. At predominantly air carrier airports air traffic control authorities request general aviation aircraft to increase their speeds above normal to reduce the wide range in speed between air carriers and general aviation. At these airports, the normal approach speeds for general aviation are probably not applicable.

If it is assumed that the distances to touchdown are 1500 ft for air carrier aircraft and 1000 ft for twin-engine general aviation aircraft, a

Type of Aircraft	Touchdown Speed, kn	Exit Speed, mi/h	
		60	15
Small propeller			
GA single engine	60	2,400	1,800
GA twin engine	95	2,800	3,500
Large jet	130	4,800	5,600
Heavy jet	140	6,400	7,100

TABLE 6-24 Approximate Taxiway Exit Location from Threshold, ft

high-speed exit accommodates aircraft exiting the runway at an exit speed of 60 mi/h and a regular exit accommodates aircraft exiting the runway at 15 mi/h, then using approximate touchdown speeds, the approximate exit locations for various types of aircraft may be found as shown in Table 6-24.

These locations are derived using standard sea-level conditions. Altitude and temperature can affect the location of exit taxiways. Altitude increases distance on the order of 3 percent for each 1000 ft above sea level and temperature increases the distance 1.5 percent for each 10°F above 59°F.

During runway capacity studies conducted for the FAA, data were collected on exit utilization at various large airports in the United States [18]. These data, which are tabulated in Table 6-25, indicate the cumulative percentage of each class of aircraft which have exited the runway at exits located at various distances from the arrival threshold. On the basis of these studies, runway exit ranges from the arrival threshold are used in runway capacity studies [5]. These exit ranges are given in Table 6-26. Comparisons between the approximate relationships given in Table 6-24 and the data given in Tables 6-25 and 6-26 indicate that fairly good correspondence results. Variations which occur are due to pilot technique and preference in choosing exits, the wide range of performance characteristics demonstrated by various aircraft in the aircraft approach categories, altitude and temperature considerations, and the amount of runway available for landing. The latter factor is very important because if pilots recognize that the amount of runway available is near the minimum for a particular aircraft they are more likely to touch down closer to the runway threshold and apply larger than normal deceleration and braking to the aircraft.

It is recommended that the point of intersection of the centerlines of taxiway exits and runways, which are up to 7000 ft in length and accommodate aircraft approach category C, D, and E aircraft, should

Dry Runways								
Distance from Threshold to Exit, ft	Regular Exits				High Speed Exits			
	Aircraft Class*				Aircraft Class*			
	A	B	C	D	A	B	C	D
0	0	0	0	0	0	0	0	0
1000	6	0	0	0	13	0	0	0
2000	84	1	0	0	90	1	0	0
3000	100	39	0	0	100	40	0	0
4000		98	8	0		98	26	3
5000		100	49	9		100	76	55
6000			92	71			98	95
7000			100	98			100	100
8000				100				
Wet Runways								
Distance from Threshold to Exit, ft	Aircraft Class*							
	A	B	C	D				
0	0	0	0	0				
1000	4	0	0	0				
2000	60	0	0	0				
3000	96	10	0	0				
4000	100	80	1	0				
5000		100	12	0				
6000			48	10				
7000			88	64				
8000			100	93				
9000				100				

*The aircraft class is the classification of aircraft based upon maximum certified takeoff weight [5].

Source: Federal Aviation Administration [18].

TABLE 6-25 Percentage of Aircraft Exiting at Exits Located at Various Dry Runways

be located about 3000 ft from the arrival threshold and 2000 ft from the stop end of the runway. To accommodate the average mix of aircraft on runways longer than 7000 ft, intermediate exits should be located at intervals of about 1500 ft. At airports where there are extensive

Mix Index*	Exit Range from Arrival Threshold
0–20	2000–4000
21–50	3000–5500
51–80	3500–6500
81–120	5000–7000
121–180	5500–7500

*Mix Index is equal to the percentage of Class C aircraft plus three an aircraft with a maximum certified takeoff weight in excess of class D aircraft, where a class C aircraft is an aircraft with a maximum certified takeoff weight greater than 12,500 lb and up to 300,000 lb and a class D aircraft is an aircraft with a maximum certified takeoff weight in excess of 300,000 lb.

Source: Federal Aviation Administration [5].

TABLE 6-26 Exit Range Appropriate to Runways Serving Aircraft of Different Arrival Mix Indices, ft

operations with aircraft approach category A and B aircraft, an exit located between 1500 and 2000 ft from the landing threshold is recommended.

Planners often find that the runway configuration and the location of the terminal at the airport often preclude placing the exits at locations based on the foregoing analysis. This is nothing to be alarmed about since it is far better to achieve good utilization of the exits than to be too concerned about a few seconds lost in occupancy time.

When locating exits it is important to recognize local conditions such as frequency of wet pavement or gusty winds. It is far better to place the exits several hundred feet farther from the threshold than to have aircraft overshoot the exits a large amount of time. The standard deviation in time required to reach exit speed is on the order of 2 or 3 s. Therefore, if the exits were placed down the runway as much as two standard deviations from the mean, the loss in occupancy time would only be 4 to 6 s. In planning exit locations at specific airports, one needs to consult with the airlines relative to the specific performance characteristics of the aircraft intended for use at the airport.

The total occupancy time of an aircraft can be roughly estimated using the following procedure. The runway is divided into four components, namely, flight from threshold to touchdown of main gear, time required for nose gear to make contact with the pavement after the main gear has made contact, time required to reach exit velocity from the time the nose gear has made contact with the pavement and brakes have been applied, and time required for the aircraft to turn

off on to the taxiway and clear the runway. For the first component it can be assumed that the touchdown speed is 5 to 8 kn less than the speed over the threshold. The rate of deceleration in the air is about 2.5 ft/s^2 . The second component is about 3 s and the third component depends upon exit speed. Time to turnoff from the runway will be on the order of 10 s. Thus the total occupancy time in seconds can be approximated by Eq. (6-8).

$$R_i = \frac{V_{ot} - V_{td}}{2a_1} + 3 + \frac{V_{td} - V_e}{2a_2} + t \quad (6-8)$$

where R_i = runway occupancy time, s

V_{ot} = over the threshold speed, ft/s

V_{td} = touchdown speed, ft/s

V_e = exit speed, ft/s

t = time to turnoff from the runway after exit speed is reached, s

a_1 = average rate of deceleration in the air, ft/s^2

a_2 = average rate of deceleration on the ground, ft/s^2

During the runway capacity studies cited earlier [18] data were also collected on runway occupancy time. These data, which are tabulated in Table 6-27, indicate the total runway occupancy time of each class of aircraft which have exited the runway at exits located at various distances from the arrival threshold. As may be observed in this table, typical runway occupancy times for 60 mi/h high-speed exits are 35 to 45 s. The corresponding time for a 15 mi/h regular exit is 45 to 60 s for air carrier aircraft.

Design of Taxiway Curves and Intersections

The basic design of taxiway curves and intersections for three of the most common types of taxiway intersections have been developed by the FAA [6]. These designs have been taken from this reference and were shown in Fig. 6-35. The dimensions recommended by the FAA for the taxiway width, centerline radius, fillet radius (inner edge radius), and the length of the fillet lead-in are given in Table 6-23. The dimensions given for the fillet radius in this table are related to the taxiway centerline radius.

When an aircraft negotiates a turn with the nose wheel tracking a predetermined curved path, such as a taxiway centerline, the mid-point of the main undercarriage does not follow the same path of the nose gear because of the fairly large distance from the nose gear to the main undercarriage. The relationship between the centerline, which is being tracked by the nose wheel, and position of the main undercarriage are shown in Fig. 6-36. At any point on the curve the distance between the curved path followed by the nose wheel and the mid-point of the undercarriage of main landing gear is referred to as the

Dry Runways								
Distance from Threshold to Exit, ft	Regular Exits				High Speed Exits			
	Aircraft Class*				Aircraft Class*			
	A	B	C	D	A	B	C	D
0	24				19			
1000	24	27			27	24		
2000	34	27			35	24		
3000	44	37	29		43	32	35	35
4000	55	46	38	38		41	35	35
5000	65	56	47	47		49	44	44
6000	76	65	56	56			54	54
7000	76	75	65	65			63	63
8000	76	75	73	73				
9000	76	75	82	82				
10000	76	75	85	85				
11000	76	75	90	90				
Wet Runways								
Distance from Threshold to Exit, ft	Aircraft Class*							
	A	B	C	D				
0	24							
1000	24							
2000	34	27						
3000	44	37	30					
4000	55	47	38					
5000	65	56	47	47				
6000	76	65	56	56				
7000	99	99	65	65				
8000			73	73				
9000			82	82				

*The aircraft class is the classification of aircraft based upon maximum certified takeoff weight [5]

Source: Federal Aviation Administration [8].

TABLE 6-27 Runway Occupancy Time of Aircraft Exiting at Exits Located at Various Distances from the Runway Arrival Threshold, s

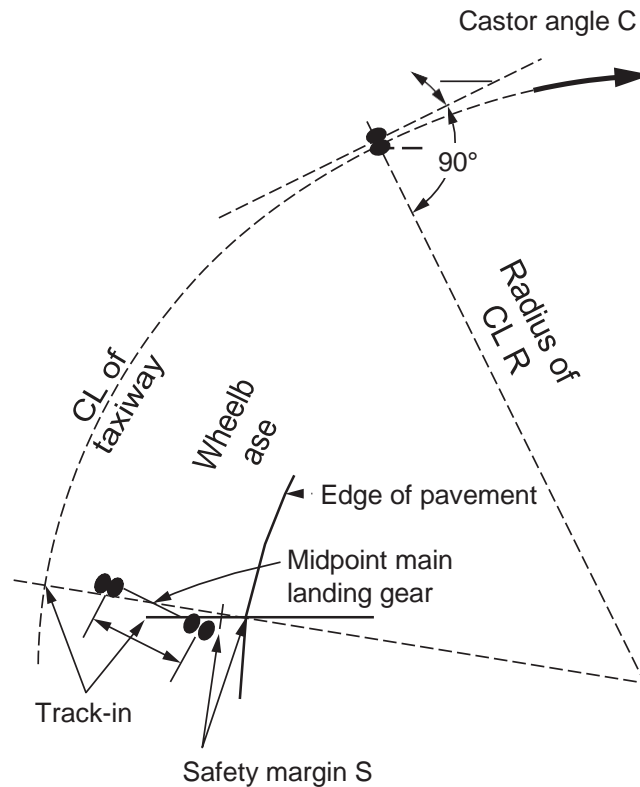


FIGURE 6-36 Path of main gear on curve.

track-in. The track-in varies, increasing progressively during the turning maneuver. It decreases as the nose gear begins to follow the tangent to the curve. Knowing the path of the main gear, the radius of the fillet can be determined by adding an appropriate taxiway edge safety margin S between the outside edge of the tire on the main landing gear closest to the center of the path followed by the nose wheel and the edge of the pavement.

The nose wheel steering angle, the castor angle C is defined as the angle formed by the longitudinal axis of the aircraft and the direction of movement of the nose wheel, or some other reference point or datum point such as the location of the pilot in the cockpit. For preliminary design it is sufficiently accurate to assume that the datum point is the nose wheel.

The size of the fillet depends not only on the wheelbase of the aircraft, the radius of the curve, the width of the taxiway, and the total change in direction, but also on the path that the aircraft follows on the turn. There are three ways in which an aircraft can be maneuvered on a turn. One is to establish the centerline of the taxiway as the path of the nose gear. This is called nose wheel on centerline tracking. Another is to establish the centerline of the taxiway as the path directly beneath the pilot and assume that this path is followed. This is called maintaining cockpit over the centerline tracking. The last is to assume that the nose gear will follow a path offset outward of the

centerline. This is called judgmental oversteering tracking. The latter type of tracking will result in the least amount of taxiway widening but results in greater runway occupancy time and the possibility of pilot error in judging the turn to follow. While there is no agreement on which procedure is desirable, usually maintaining the cockpit over the centerline tracking is preferred.

The principal dimensions of the aircraft related to tracking a curve were given in Fig. 6-36. The geometry of the aircraft tracking the centerline curve from the point of curvature, PC, to the point of tangency, PT, and the various terms used in the equations below to define the movement of the aircraft are given in Fig. 6-37.

The maximum angle formed between the tangent to the centerline and the longitudinal axis of the aircraft will occur at the end of the curve when the nose wheel is at the point of tangency. This angle, called A_{\max} , may be approximated by

$$A_{\max} = \sin^{-1} (d/R) \quad (6-9)$$

where d is the distance from the nose wheel or the pilot cockpit position to the center of the main undercarriage; the wheelbase of the aircraft is often used to approximate this distance and R is the radius the nose wheel or the pilot is tracking on the curve.

The maximum nose wheel steering angle, the castor angle, the angle between the longitudinal axis on the nose gear and the longitudinal axis of the aircraft, B_{\max} is given by

$$B_{\max} = \tan^{-1}(w/d \tan A_{\max}) \quad (6-10)$$

where w is the wheelbase of the aircraft.

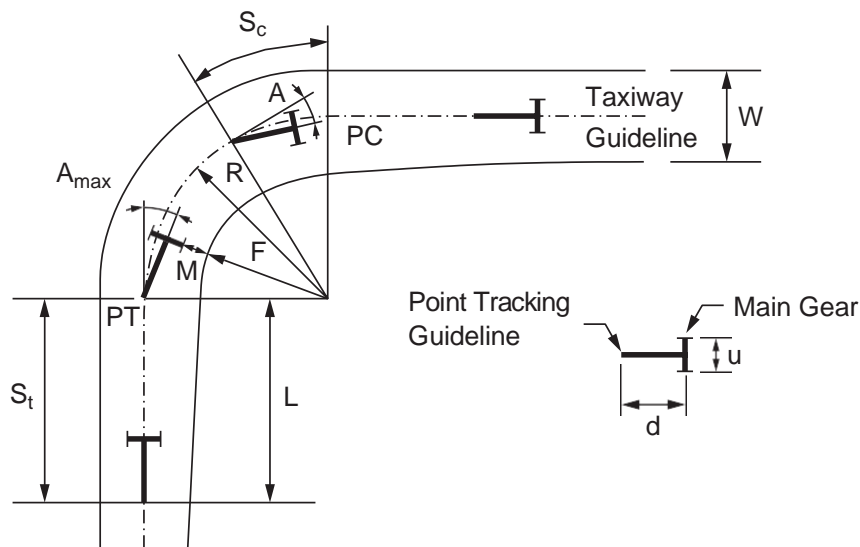


FIGURE 6-37 Taxiway fillet design geometry.

The required fillet radius F is given by

$$F = (R^2 + d^2 - 2Rd \sin A_{\max})^{0.5} - 0.5u - M \quad (6-11)$$

where u is undercarriage width, that is, the distance between the outside tires on the main gear and M is the minimum distance required between the edge of the outside tire and the edge of the pavement, that is, the edge safety margin.

The length of the lead-in to the fillet is given by

$$L = d \frac{(4d \tan 0.5A_{\max})}{W - u - 2M} \ln \quad (6-12)$$

where W is the taxiway width on the tangent and \ln represents the natural logarithm.

These equations may be solved for a given aircraft tracking a curve of radius R to find the necessary lead-in and fillet radius to maintain a minimum edge safety margin between the tire and pavement edge. If the value of the maximum nose wheel steering angle, B_{\max} , exceeds 50° it is recommended that the radius of the centerline curve R which the nose wheel is tracking be increased.

The use of these equations in determining the critical taxiway curve design parameters is shown in Example Problem 6-3.

Example Problem 6-3

Determine the minimum lead-in and the radius of the fillet to maintain the cockpit over the centerline for an aircraft with a 156.1-ft wingspan, a wheelbase of 64.6 ft, undercarriage width of 34.25 ft, and the distance between the main undercarriage and cockpit equal to 72.1 ft. The aircraft is moving between two parallel taxiways through a connecting taxiway which has a centerline perpendicular to the parallel taxiways.

With a wingspan of 156.1 ft this aircraft is in airplane design group IV. From Table 6-23, the taxiway width is 75 ft, the minimum safety margin is 15 ft, the recommended centerline radius is 150 ft, the recommended length of the fillet lead-in is 250 ft and the recommended fillet radius is 85 ft.

To verify that these recommended values are acceptable, we can use Eqs. (6-10) through (6-13). From Eq. (6-10), we have

$$A_{\max} = \sin^{-1}(d/R)$$

or

$$A_{\max} = \sin^{-1}(72.1/150) = 29^\circ$$

From Eq. (6-11),

$$B_{\max} = \tan^{-1}(w/d \tan A_{\max})$$

or

$$B_{\max} = \tan^{-1}[(64.5/72.1) \tan 29^\circ] = 27^\circ$$

Since this is less than 50° the radius is adequate.

From Eq. (6-12), we have

$$F = (R^2 + d^2 - 2Rd \sin A_{\max})^{0.5} - 0.5u - M$$

or

$$F = [150^2 + 72.1^2 - (2)(150)(72.1) \sin 29^\circ]^{0.5} - 0.5(34.25) - 15$$

$$F = 100.1 \text{ ft}$$

From Eq. (6-12),

$$L = \frac{d \ln(4d \tan 0.5A_{\max})}{W - u - 2M} - d$$

$$L = \frac{72.1 \ln \frac{(4)(72.1)(\tan 14.5^\circ)}{75 - 34.25 - (2)(15)}}{72.1} - 72.1$$

or

$$L = 68 \text{ ft}$$

Therefore, both the required fillet radius and the required length of the lead-in to the fillet for this specific aircraft are both well within those recommended for the airplane design group to which this aircraft is assigned.

End-Around Taxiways

In an effort to reduce the number of times aircraft must cross a runway when traveling around an airfield, the FAA has allowed for the design of taxiways that traverse beyond runway thresholds. These taxiways, known as end-around taxiways, are designed to both reduce the risk of runway incursions and increase the overall efficiency of operations on the airfield. For safety considerations, primarily concerned with the transient presence of aircraft immediately off the end of the runway, the FAA has established specific design standards for end-around taxiways. An example of end-around taxiway configuration is shown in Fig. 6-38.

End-around taxiways must remain outside of the runway safety area, and outside of any ILS critical areas. In addition, the tail height of the critical design aircraft at the airport must not exceed any critical Part 77 or TERPS surfaces, when on the end-around taxiway. Furthermore, the location of the end-around taxiway should provide for any aircraft departing on the runway to clear any object on the taxiway by at least 35 ft vertically and 200 ft horizontally from the runway centerline.

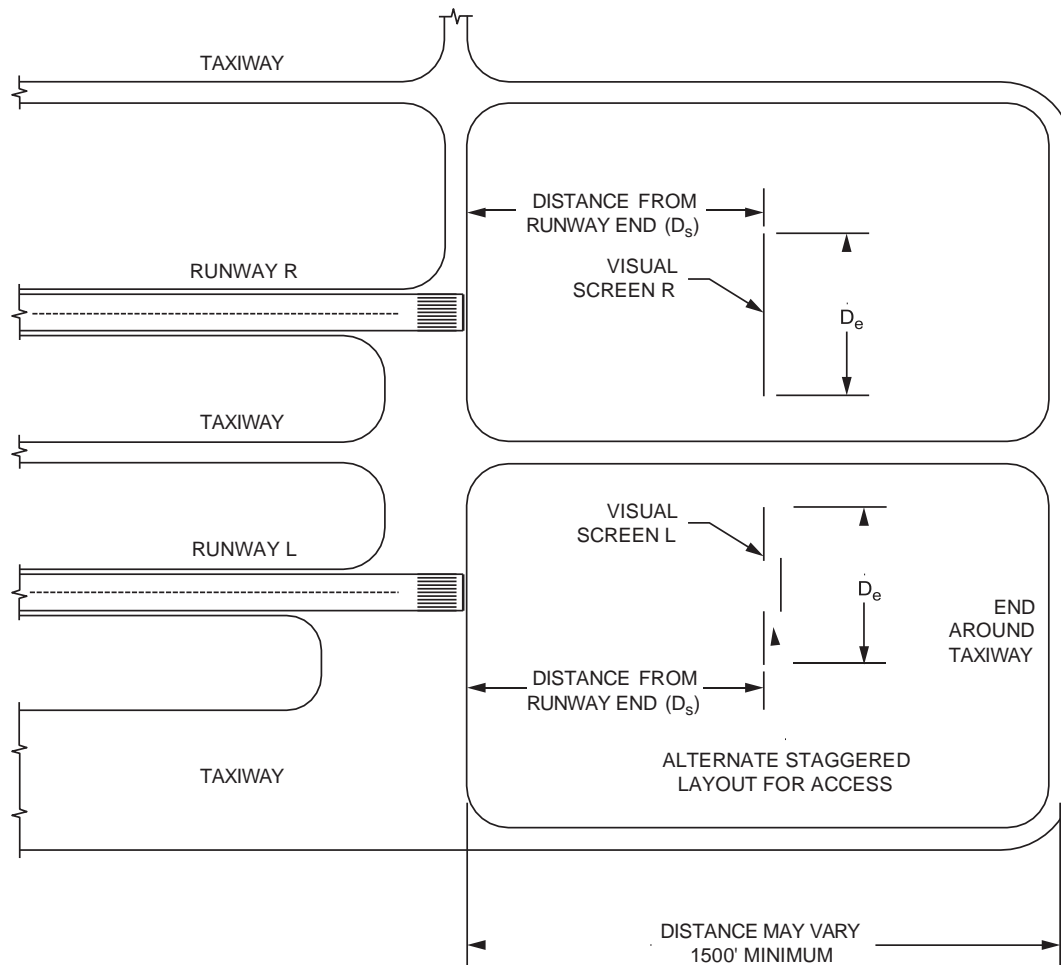


FIGURE 6-38 End-around taxiway.

A diagonal stripe screen is required to be placed between the end of the runway and the end-around taxiway, to provide pilots on the runway visual clarity that any aircraft on the end-around taxiway is not on the runway. An illustration of this screen is found in Fig. 6-39.

Aprons

Holding Aprons

Holding aprons, holding pads, run-up pads, or holding bays as they are sometimes called, are placed adjacent to the ends of runways. The areas are used as storage areas for aircraft prior to takeoff. They are

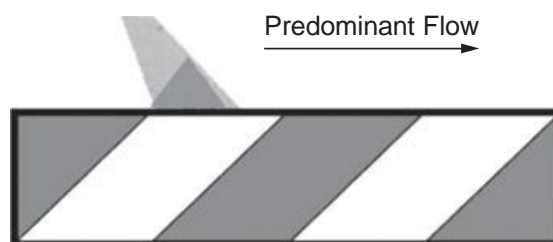


FIGURE 6-39 End-around taxiway screen.

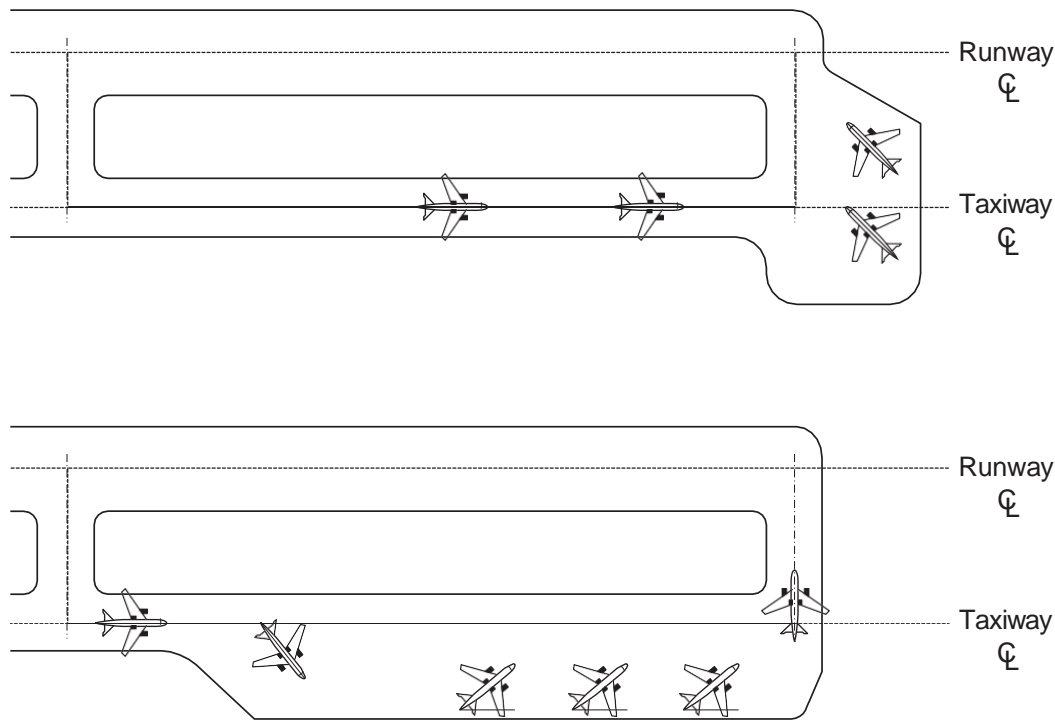


FIGURE 6-40 Typical holding pad configurations.

designed so that one aircraft can bypass another whenever this is necessary. For piston-engine aircraft the holding apron is an area where the aircraft instrument and engine operation can be checked prior to takeoff. The holding apron also provides for a trailing aircraft to bypass a leading aircraft in case the takeoff clearance of the latter must be delayed for one reason or another, or if it experiences some malfunction. There are many configurations of holding aprons, two of which are shown in Fig. 6-40. The important design criteria are to provide adequate space for aircraft to maneuver easily onto the runway irrespective of the position of adjacent aircraft on the holding apron and to provide sufficient room for an aircraft to bypass parked aircraft on the holding apron. The recommendations for the minimum separation between aircraft on holding aprons are the same as those specified for the taxiway object-free area.

The design of a typical flow-through bypass holding pad studied is shown in Fig. 6-41. Holding pads must be designed for the largest

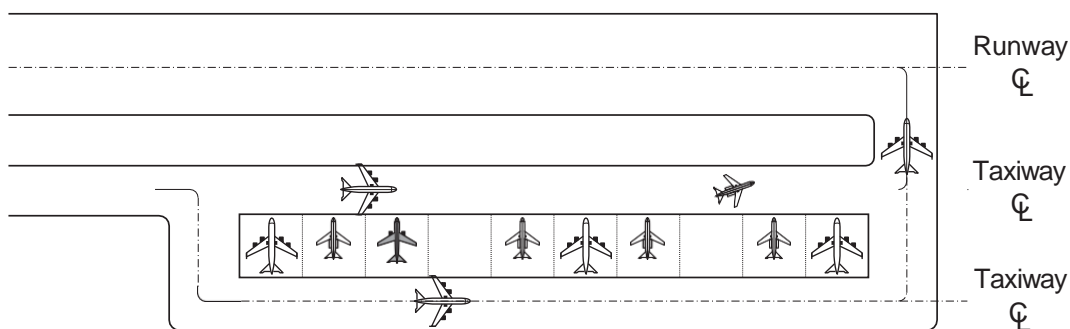


FIGURE 6-41 Flow-through bypass holding pad.

aircraft which will use the pad. The holding pad should be located so that all aircraft using the pad will be located outside both the runway and taxiway object-free area and in a position so as not to interfere with critical ILS signals.

Terminal Aprons and Ramps

Aircraft parking positions, also called aircraft gates or aircraft stands, on the terminal apron or ramp are sized for the geometric properties of a given design aircraft, including wingspan, fuselage length and turning radii, and for the requirements for aircraft access by the vehicles servicing the aircraft at the gates. Both the FAA and ICAO recommend minimum clearances between any part of an aircraft and other aircraft or structures in the apron area as given in Table 6-28.

Example Problem 6-4 illustrates the determination of the terminal apron requirements for aircraft.

Example Problem 6-4

Design a terminal apron with two parallel concourses to accommodate gates for one wide-bodied aircraft and three narrow-bodied aircraft on the face of each of the concourses. The gate design aircraft for the wide-bodied gates is the Boeing 767-200 and the gate design aircraft for the narrow-bodied gates is the McDonnell-Douglas MD-87. Aircraft will park nose-in at each gate and use the gates in a power-in and push-out mode of operation.

The Boeing 767-200 has a fuselage length of 159 ft 2 in and a wingspan of 156 ft 1 in, which places it in airplane design group IV, and the McDonnell-Douglas MD-87 has a fuselage length of 130 ft 5 in and a wingspan of 107 ft 10 in, which places it in airplane design group III.

If the aircraft are arrayed at the concourses as shown in Fig. 6-42, then the size of the terminal apron and the size of each gate position may be determined by referencing the specifications requiring specific separations between aircraft

Airplane Design Group or Aerodrome Code Letter		Minimum Clearance*	
		Feet	Meters
I	A	10	3.0
II	B	10	3.0
III	C	15	4.5
IV	D	25	7.5
V or VI	E	25	7.5

*The FAA recommends the wingtip separation at parking positions to
Source: Federal Aviation Administration [6, 19] and International Civil Aviation Organization [4]

TABLE 6-28 Minimum Clearance between Aircraft and Fixed or Movable Objects at Terminal Apron Parking Positions

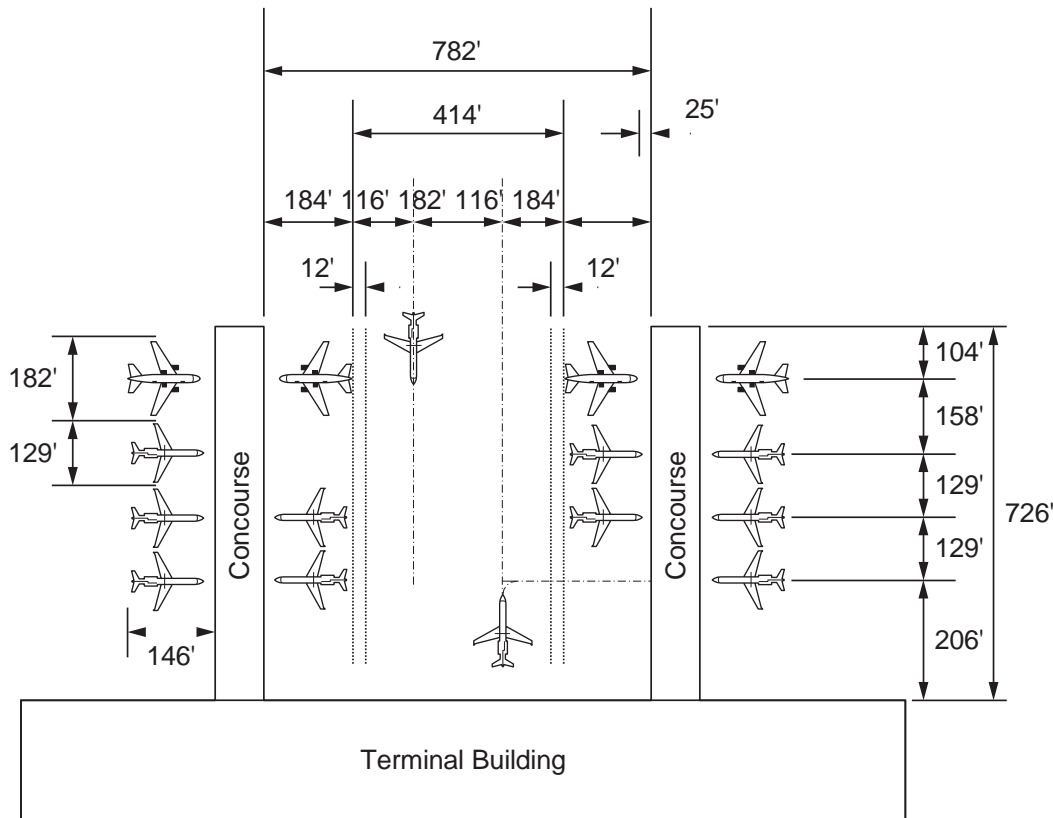


FIGURE 6-42 Terminal apron requirements for Example Problem 6-4.

operating on the taxiways and between aircraft parked at the concourse gates. In this problem the FAA specifications will be used and the design will be based upon the actual dimensions of the aircraft rather than upon the dimensions of the largest aircraft in the airplane design groups to which these aircraft are assigned. Therefore, the relevant separations are contained in the footnotes in Table 6-18.

The Boeing 767-200 is the greater wingspan aircraft and, therefore, the most demanding aircraft for taxiway dimensions. The separation between taxiway centerlines is equal to 1.1 times the wingspan of the most demanding aircraft plus 10 ft. The recommended separation is then equal to $(1.1)(156.1) + 20 = 182$ ft. A ground access vehicle lane will be provided behind each aircraft for the use of aircraft service vehicles. This lane will be 12 ft wide. The distance from the centerline of each taxiway to a fixed or movable object is equal to 0.6 the wingspan plus 10 ft. Therefore, this distance is $(0.6)(156.1) + 10 = 104$ ft. Considering the ground access vehicle lane, the distance from the centerline of each taxiway to the tail of the aircraft is equal to $104 + 12 = 116$ ft.

Table 6-28 indicates that the recommended clearance between the face of each concourse and the nose of the 767-200 aircraft is 25 ft since this aircraft is in airplane design group IV. The length of the 767-200 is 159 ft 2 in and, therefore, the distance from the face of each concourse to the tail of the aircraft is equal to $25 + 156.2 = 182$ ft.

Considering each of these recommended separations, the width of the terminal apron or ramp between the concourses is found to be 782 ft as shown in Fig. 6-42. The clearance between aircraft wingtips or between the aircraft wing and fixed or movable objects is recommended to be 0.1 times the wingspan plus 10 ft. Therefore, the 767-200 requires a clearance between its wingtips and fixed or

movable objects of 0.1 the wingspan plus 10 ft or $(0.1)(156.1) + 10 = 25.6$ ft. This results in a parking position for this aircraft being $156.1 + 25.6 = 182$ ft wide by 182 ft long. The MD-87 requires a clearance of $(0.1)(107.84) + 10 = 20.8$ ft. This results in a parking position for this aircraft being $107.84 + 20.8 = 129$ ft wide. From Table 6-28 this aircraft may be parked as close to 15 ft to the concourse since it is in airplane design group III. This results in the length of a parking position for the MD-87 being $130.42 + 15 = 146$ ft. However, since the parking position for the 767-200 is longer, the actual length provided for the MD-87 is 182 ft.

For aircraft to be moved from their parking positions onto the apron taxilane a separation from the inside edge of the last gate position to the terminal building equal to the fuselage length plus 0.1 times the wingspan plus 10 ft is required. This should allow the aircraft to turn onto the centerline of the taxilane from its gate position and also allow another aircraft to gain access to the gate position before the last aircraft at that gate leaves the apron area. The parking arrangement shown in Table 6-13 shows the MD-87 as the closest aircraft to the terminal building. Therefore, the distance from the centerline of the last gate position to the terminal building is $130.42 + 54 + (0.1)(107.83) + 10 = 206$ ft.

The design must ensure that the clearances provided are adequate based upon the turning capability of the design aircraft. The analyst should verify using the procedures discussed earlier in this chapter that the tracking-in of the aircraft while making a turn from the taxilanes to the parking position will not compromise safe clearances between aircraft. The analyst must also verify for each aircraft that the turning to access gate positions will ensure that no part of the aircraft will compromise these safe clearances.

Using the above dimensions results in a ramp length of 726 ft. The dimensions for all aircraft parking positions or gates are shown in Fig. 6-42.

The terminal apron area for these eight aircraft becomes equal to $(778)(726) = 565,800$ ft², which is equal to about 13 acres or a total ramp area per aircraft of about 1.6 acres. A gate for a 767-200 is $(182)(182) = 33,100$ ft², which is about 0.75 acres. The required gate for an MD-87 is $(129)(146) = 18,800$ ft² which is slightly more than 0.4 acres. Useful rules-of-thumb which allow one to estimate terminal apron gate requirements are a total ramp area of from 1.5 to 2.0 acres per aircraft gate, and an area of from 0.75 to 1.0 acres per gate for a wide-bodied aircraft gate position and an area of about 0.5 acres per gate for a narrow-bodied aircraft gate position.

Terminal Apron Surface Gradients

For fueling, ease of towing and aircraft taxiing, apron slopes or grades should be kept to the minimum consistent with good drainage requirements. Slopes should not in any case exceed 2 percent for utility airports and 1 percent for transport airports. At gates where aircraft are being fueled every effort should be made to keep the apron slope within 0.5 percent.

Control Tower Visibility Requirements

At airports with a permanent air traffic control tower, the runways and taxiways must be located and oriented so that a clear line of sight is maintained to all traffic patterns, the final approaches to all runways, all runway structural pavements, all apron taxiways, and other

operational surfaces controlled by the air traffic control tower. A clear line of sight to all taxiway centerlines is desirable. Operational surfaces not having a clear unobstructed line of sight from the tower are designated as uncontrolled or nonmovement areas. At airports without a permanent air traffic control tower, the runways and taxiways should be located and oriented so that a future tower may be sited in accordance with the continuous visibility requirements. This requirement may be satisfied where adequate control of aircraft exists by other means [6].

A typical air traffic control tower site requires between 1 and 4 acres of land. The site must be large enough to accommodate current and future building needs including employee parking. Tower sites must afford maximum visibility to traffic patterns and clear, unobstructed and direct lines of sight to the runway approaches, the landing area, and all runway and taxiway surfaces. Most towers penetrate the FAR Part 77 surfaces and, therefore, are obstructions to aviation and may be a hazard to air navigation unless an FAA study determines otherwise. The tower must not derogate the signal generated by any existing or planned electronic navigational aid or air traffic control facility.

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CHAPTER 7

Structural Design of Airport Pavements

Introduction

This chapter briefly describes various methods for designing airfield pavements. The term *structural design of airport pavements* as used in this text refers to the determination of the thickness of the components that make up an airfield pavement structure, rather than the design of pavement materials itself.

Airfield pavement is intended to provide a smooth and safe all-weather riding surface that can support the weights of such heavy objects as aircraft on top of the natural ground base. Airfield pavements are typically designed in layers, with each layer designed to a sufficient thickness to be adequate to ensure that the applied loads will not lead to distress or failure to support its imposed loads. The Federal Aviation Administration provides guidance on the design of airfield pavements within its Advisory Circular AC 150/5320-6E, *Airfield Pavement Design and Evaluation*. Originally published in 1975, this advisory circular was completely revised in 2008 to consider new design methods that are based on recently developed computer software models and appropriate for the heaviest of commercial air carrier aircraft. This chapter provides both an account of the historical pavement design and evaluation methods and details the current method of pavement design and evaluation. As with any element of airport planning and design, appropriate Advisory Circulars and software user guides should be studied and referenced prior to performing any airport pavement analysis. These resources may be downloaded from the FAA at <http://www.faa.gov>.

Pavement or pavement structure is a structure consisting of one or more layers of processed materials. A pavement consisting of a mixture of bituminous material and aggregate placed on high-quality granular materials is referred to as *flexible pavement*. When the pavement consists of a slab of portland cement concrete (PCC), it is referred to as *rigid pavement*. Both structures of pavement are

typically found at airports, although often there are preferences to a given type of pavement depending on such factors as the type and frequency of aircraft usage, climatic conditions, and costs of construction and maintenance.

Figure 7-1 illustrates a cross section of a typically layered airfield pavement. As illustrated in Fig. 7-1, airfield pavement, whether flexible or rigid, typically consists of series of layers consisting of a surface course, base course, and one or more subbase courses, resting on the ground, or prepared "subgrade" layer.

The surface course consists of a mixture of bituminous material (generally asphalt) and aggregate ranging in thickness from 2 to 12 in for flexible pavements, and a slab of PCC 8 to 24 in thick for rigid pavements. The principal function of the surface course is to provide for smooth and safe traffic operations, to withstand the effects of applied loads and environmental influences for some prescribed period of operation, and to distribute the applied load to the underlying layers.

The base course may consist of treated or untreated granular material. Like the surface course, it must be adequate to withstand the effects of load and environment and to distribute the applied loads to the underlying layers. Untreated bases consist of crushed or uncrushed aggregates. Treated bases consist of crushed or uncrushed aggregate that has been mixed with a stabilizing material such as cement or bitumen.

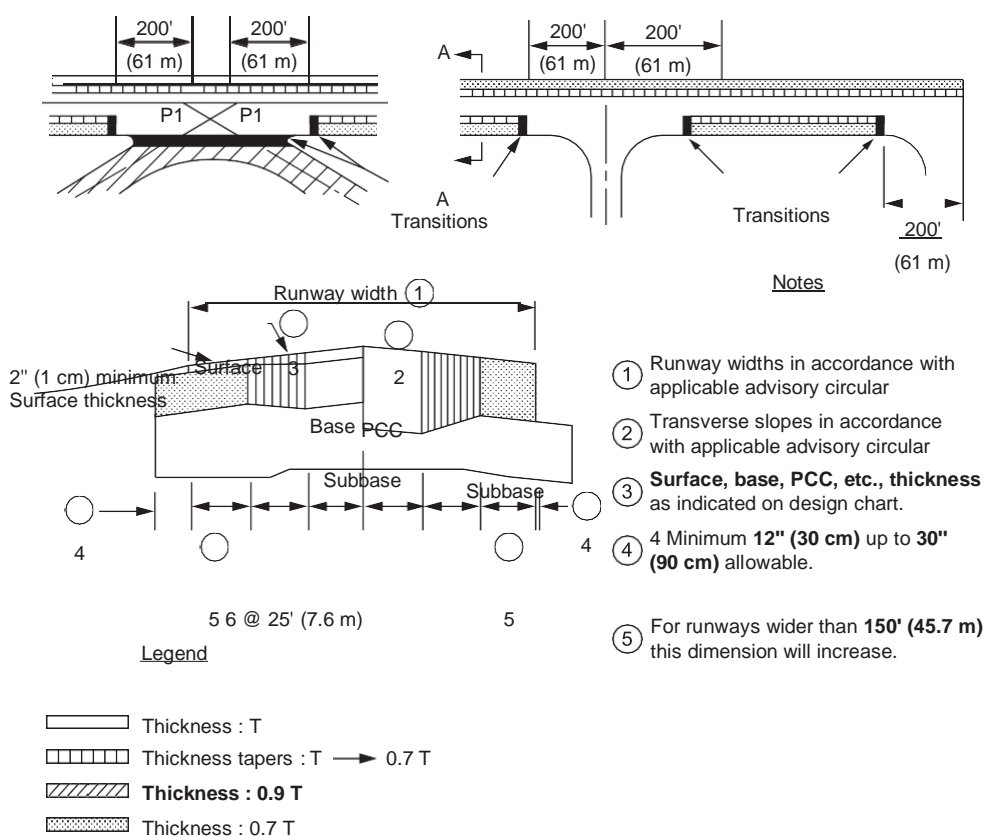


FIGURE 7-1 Typical plan and cross section for airfield pavement.

The subbase course is also composed of treated or untreated material, typically unprocessed pit-run material or material selected from a suitable excavation on the site. The function of the subbase is the same as that of the base. Whether or not a subbase is required, or how many subbase layers are required, is a function of the type of loads on the pavement, as well as the type and quality of soil, or subgrade, on which the pavement will be resting. For most rigid pavements, the surface course rests directly on the subbase.

The design of the thickness of each of the above layers is of primary concern to airport pavement engineers. The two primary factors that contribute to the design thickness of airfield pavement layers are the soil base and the volume and weight of the traffic using the pavement. As such, the first steps in pavement analysis are an investigation of the soil on which the pavement will be placed, and an estimation of the annual traffic volume on the pavement.

Soil Investigation and Evaluation

Accurate identification and evaluation of pavement foundations are essential to the proper design of the pavement structure. The subgrade supports the pavement and the loads placed upon the pavement surface. The function of the pavement is to distribute the loads to the subgrade, and the greater the capability of the subgrade to support the loads, the less the required thickness of the pavement.

Soil investigation consists of a soil survey to determine the arrangement of the different layers of soil in relation to the subgrade elevation, a sampling and testing of the various layers of soil to determine the physical properties of the soil, and a survey to determine the availability and suitability of local materials for use in the construction of the subgrade and pavement. Surveys and sampling are usually accomplished through soil borings to determine the soil of rock profile and its lateral extent. The sampled materials are then tested to determine soil types, gradation or particle sizes, liquid and plastic limits, moisture-density relationships, shrinkage factors, permeability, organic content, and strength properties. In the United States, soil surveys are often conducted using a variety of methods, including referring to U.S. Geological Survey (USGS) geodetic maps, aerial photography, and soil borings. The FAA recommends borings of given spacing and depths for soil surveys as illustrated in Table 7-1.

In the United States, evaluation of sampled soils for the purpose of airfield pavement design is performed according to the U.S. Army Corps of Engineers Unified Soil Classification (USC or "unified") System, as illustrated in Table 7-2. Under the unified system, soils are initially classified as either coarse-grained, fine-grained, or highly organic soils. Coarse-grained soils are those that do not filter through a No. 200 grade sieve. Coarse-grained soils are further divided into gravels and sands, as a function of the percentage of soil that filters

Area	Spacing	Depth
Runways and taxiways	Random across pavement at 200 ft (68 m) intervals	Cut areas—10 ft (3.5 m) below finished grade Fill areas—10 ft (3.5 m) below existing ground
Other areas of pavement	One boring per 10,000 ft ² (930 m ²) of area	Cut areas—10 ft (3.5 m) below finished grade Fill areas—10 ft (3.5 m) below existing ground
Borrow areas	Sufficient tests to clearly define the borrow material	To depth of borrow excavation

Note: For deep fills, boring depths shall be sufficient to determine the extent of consolidation and/or slippage the fill may cause.

TABLE 7-1 Recommended Soil Boring Spacings and Depths

	Major Divisions	Group Symbols	
Coarse-grained soils more than 50% retained on No. 200 sieve	Gravels 50% or more of coarse fraction retained on No. 4 sieve	Clean gravels	GW GP
		Gravels with fines	GM GC
	Sands less than 50% of coarse fraction retained on No. 4 sieve	Clean sands	SW SP
		Sands with fines	SM SC
Fine-grained soils 50% or less retained on No. 200 sieve	Silts and clays liquid limit 50% or less		ML CL OL
	Silts and clays liquid limit greater than 50%		MH CH OH
Highly organic soils			PT

Note: Based on the material passing the 3-in (75-mm) sieve.

TABLE 7-2 Classification of Soils for Airport Pavement Applications

through a No. 4 sieve. Fine grained soils, known also as silts and clays, are subdivided into two groups on the basis of their liquid limits.

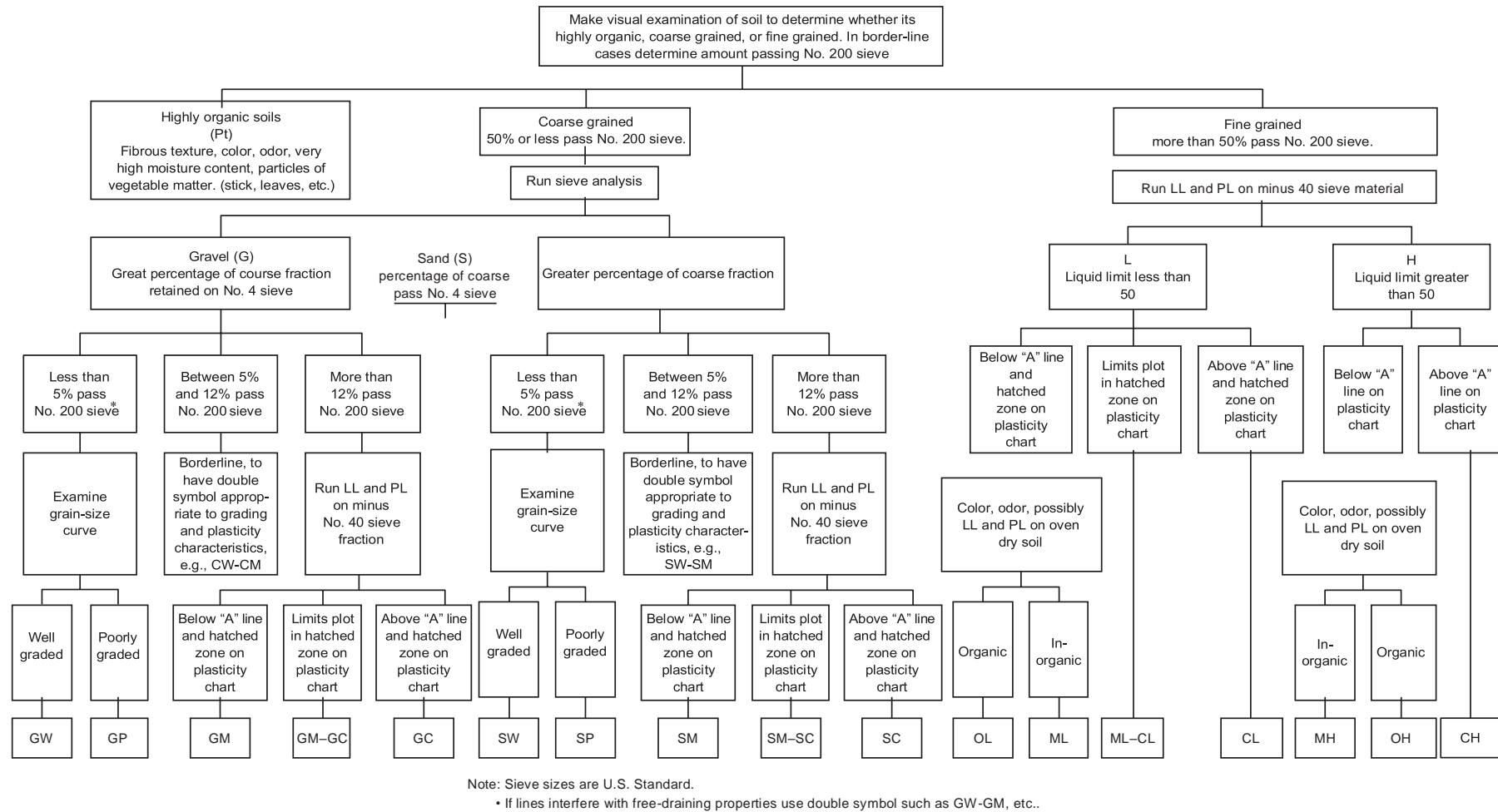
These soils are finally grouped into one of 15 different groupings. These groupings are

- GW: Well-graded gravels and gravel-sand mixtures, little or no fines
- GP: Poorly graded gravels and gravel-sand mixtures, little or no fines
- GM: Silty gravels, gravel-sand-silt mixtures
- GC: Clayey gravels, gravel-sand-clay mixtures
- SW: Well-graded sands and gravelly sands, little or no fines
- SP: Poorly graded sands and gravelly sands, little or no fines
- SM: Silty sands, sand-silt mixtures
- SC: Clayey sands, sand-clay mixtures
- ML: Inorganic silts, very fine sands, rock flour, silty or clayey fine sands
- CL: Inorganic clays of low to medium plasticity, gravelly clays, silty clays, lean clays
- OL: Organic silts and organic silty clays of low plasticity
- MH: Inorganic silts, micaceous or diatomaceous fine sands or silts, plastic silts
- CH: Inorganic clays of high plasticity, fat clays
- OH: Organic clays of medium to high plasticity
- PT: Peat, muck, and other highly organic soils

A flowchart illustrating the procedure for the classification of soils by the unified system is given in Fig. 7-2. The uses of the various soil materials for pavement foundations are described in Table 7-3.

It should be noted that column 11 in Table 7-3 refers to the soil's "field CBR" value, or "California Bearing Ratio," a value of the strength of material used in flexible pavement bases, and column 12 in Table 7-3 refers to the soil's "subgrade modulus" or " k value," a value of the bearing capacity of the soil, estimated using a plate bearing test.

The soil's field CBR value is determined by the CBR method of pavement design, which is applied primarily to flexible pavements. The CBR method of design was developed by the California Division in 1928. The method subsequently was adopted for military airport use by the Corps of Engineers, U.S. Army, shortly after the outbreak of World War II. The outbreak of the war required that a decision be made with little delay concerning a design method. At the time, there were no methods available specifically developed for airport pavements. It was apparent that the time required to develop a completed new method of design would preclude its use in a war emergency.

**FIGURE 7-2** Flowchart for the unified soil classification system.

program. Consequently, it was decided to review all available methods for the design of highway pavements and to select one which could be adopted for airfield use. The criteria for selecting a method were many. Among the more important were (1) simplicity in procedures for testing the subgrade and the pavement components, (2) a record of satisfactory experience, and (3) adaptation to the airport problem in a reasonable time. After several months of investigating of suggested methods, the CBR method was tentatively adopted. Application of the CBR method enables the designer to determine the required thickness of the subbase, base, and surface course by entering a set of design curves using the results of a relatively simple soil test.

The CBR Test

The CBR test expresses an index of the shearing strength of soil. Essentially the test consists of compacting about 10 lb of soil into a 6-in-diameter mold, placing a weight, known as a surcharge, on the surface of the sample, immersing the sample in water for 4 days, and penetrating the soaked sample with the steel piston approximately 2 in in diameter at a specified rate of loading. The resistance of the soil to penetration, expressed as a percentage of the resistance for a standard crushed limestone, is the CBR value for the soil. Thus, a CBR of 50 means that the stress necessary for the surcharge to penetrate the soil sample a specified distance is one-half that required for the surcharge to penetrate the same distance in the standard crushed limestone. The relationship is usually based on a penetration of the piston of 0.1 in with 1000 lb/in² used as the stress required to penetrate the crushed limestone at 0.1 in penetration. As illustrated in Table 7-3, soils range in CBR values from relatively weak fine-grained and highly organic soils with CBR values as low as 3, to wide-grained coarse soils with CBR values as high as 80 (although CBR testing has been found to be somewhat inaccurate for very gravelly soils, and for application a CBR value of no higher of 50 should be applied).

The Plate Bearing Test

The modulus subgrade of reaction, or k value of the subgrade is determined by what is known as a field plate bearing test. This test consists of applying loads by means of a hydraulic jack through a jacking frame on to a steel plate 30 in in diameter on the soil. By loading the plate, a load-versus-deformation curve is obtained. The k value is determined to be the pressure required to produce a unit deflection of the pavement foundation, measured in pounds per cubic inch. k values range from less than 150 (considered "very poor") to more than 300 (considered "very good"). In general, the greater the coarseness of the soil, the higher k value and the less deflection for a given loading can be expected.

Major Divisions		Letter	Name	Value as Base Directly under Wearing Surface	Value as Base Directly under Wearing Surface	Potential Frost Action	Compressibility and Expansion	Drainage Characteristic	Unit Dry Weight (pcf)	CBR	Subgrade Modulus k (pci)
(1)	(2)	(3)	(4)	(5)	(6)	(7)	(8)	(9)	(10)	(11)	(12)
Coarse-gravelly soils	Gravel and gravelly soils	GW	Gravel or sandy gravel, well graded	Excellent	Good	None to very slight	Almost none	Excellent	125–140	60–80	300 or more
		GP	Gravel or sandy gravel, poorly graded	Good	Poor to fair	None to very slight	Almost none	Excellent	120–130	35–60	300 or more
		GU	Gravel or sandy gravel, uniformly graded	Good to excellent	Poor	None to very slight	Almost none	Excellent	115–125	25–50	300 or more
		GM	Silty gravel or silty sandy gravel	Good	Fair to good	Slight to medium	Very slight	Fair to poor	130–145	40–80	300 or more
		GC	Clayey gravel or clayey sandy gravel	Good to excellent	Poor	Slight to medium	Slight	Poor to practically impervious	120–140	20–40	200–300
	Sand and sandy soils	SW	Sand or gravelly sand, well graded	Good	Poor to not suitable	None to very slight	Almost none	Excellent	110–130	20–40	200–300
		SP	Sand or gravelly sand, poorly graded	Fair to good	Not suitable	None to very slight	Almost none	Excellent	105–120	15–25	200–300
		SU	Sand or gravelly sand, Poor uniformly Not suitable graded	Fair to good	Poor	None to very slight	Almost none	Excellent	100–115	10–20	200–300
		SM	Silty sand or silty gravelly sand	Good	Not suitable	Slight to high	Very slight	Fair to poor	120–135	20–40	200–300
		SC	Clayey sand or clayey gravelly sand	Fair to good	Not suitable	Slight to high	Slight to medium	Poor to practically impervious	105–130	10–20	200–300

Fine grained soils	Low compress-ibility LL < 50	ML	Silts, sandy silts, gravelly silts, or diatomaceous soils	Fair to good	Not suitable	Medium to very high	Slight to medium	Fair to poor	100–125	5–15	100–200
		CL	Lean clays, sandy clays, or gravelly clays	Fair to good	Not suitable	Medium to very high	Medium	Practically impervious	100–125	5–15	100–200
		OL	Organic silts or lean organic clays	Poor	Not suitable	Medium to very high	Medium to high	Poor	90–105	4–8	100–200
	High compress ibility LL<50	MH	Micaccous clays or diatomaceous soils	Poor	Not suitable	Medium to very high	High	Fair to poor	80–105	4–8	100–200
		CH	Fat clays	Poor to very poor	Not suitable	Medium	High	Practically impervious	90–110	3–5	50–100
		OH	Fat organic clays	Poor to very poor	Not suitable	Medium	High	Practically impervious	80–105	3–5	50–100
Peat and other fibrous organic soils		Pt	Peat, humus and other	Not suitable	Not suitable	Slight	Very high	Fair to poor			

TABLE 7-3 Characteristics of Soil Related to Airport Pavement Foundations

Young's Modulus (*E* Value)

The most recent accepted method for determining the strength of subgrade by the FAA is based upon the elastic modulus of the subgrade, or *E*, also known as Young's modulus. In general, a structure's *E* value is its measure of "stiffness" or "elasticity." The greater the *E* value, the more stiff the material, and the less the material is susceptible to deformation under a given stress load. A pavement's *E* value may be empirically estimated by evaluating its stress to strain ratio, according to Eq. (7-1)

$$E = \frac{\text{Tensile stress}}{\text{Tensile strain}} = \frac{\sigma}{\epsilon} = \frac{F/A_0}{\Delta L/L_0} = \frac{FL_0}{A_0 \Delta L} \quad (7-1)$$

where *E* = Young's modulus (modulus of elasticity)

F = force applied to the pavement

ΔA_0 = original cross-sectional area through which the force is applied

L = amount by which the shape of the pavement changes

*L*₀ = original shape of the object

Alternatively, the subgrade modulus *E* value, in pounds per square inch, may be found using the following conversion formulas: To find *E* based on CBR value the equation is

$$E = 1500 * \text{CBR} \quad (7-2)$$

To find *E* based on the modulus of subgrade of reaction, *k*, the equation is

$$E = 26k^{1.284} \quad (7-3)$$

This conversation is provided by the FAA in its advisory circular primarily to facilitate the transition from the more traditional pavement design methods to the most current software-based method of pavement design and evaluation.

Effect of Frost on Soil Strength

While there are a variety of soil types, the behavioral properties of any given type are relatively similar regardless of other climatic characteristics, such as the average ambient temperature and amount of precipitation. One factor that does significantly impact the strength of soil, however, is the presence of frost on the surface of or within the soil, either on a seasonal or a permanent basis.

Frost action, if severe, results in nonuniform heave of pavements during the winter because of the formation of ice lenses within the subgrade, known as ice segregation, and in loss of supporting capacity of the subgrade during periods of thaw. Figure 7-3 illustrates the process of ice segregation.

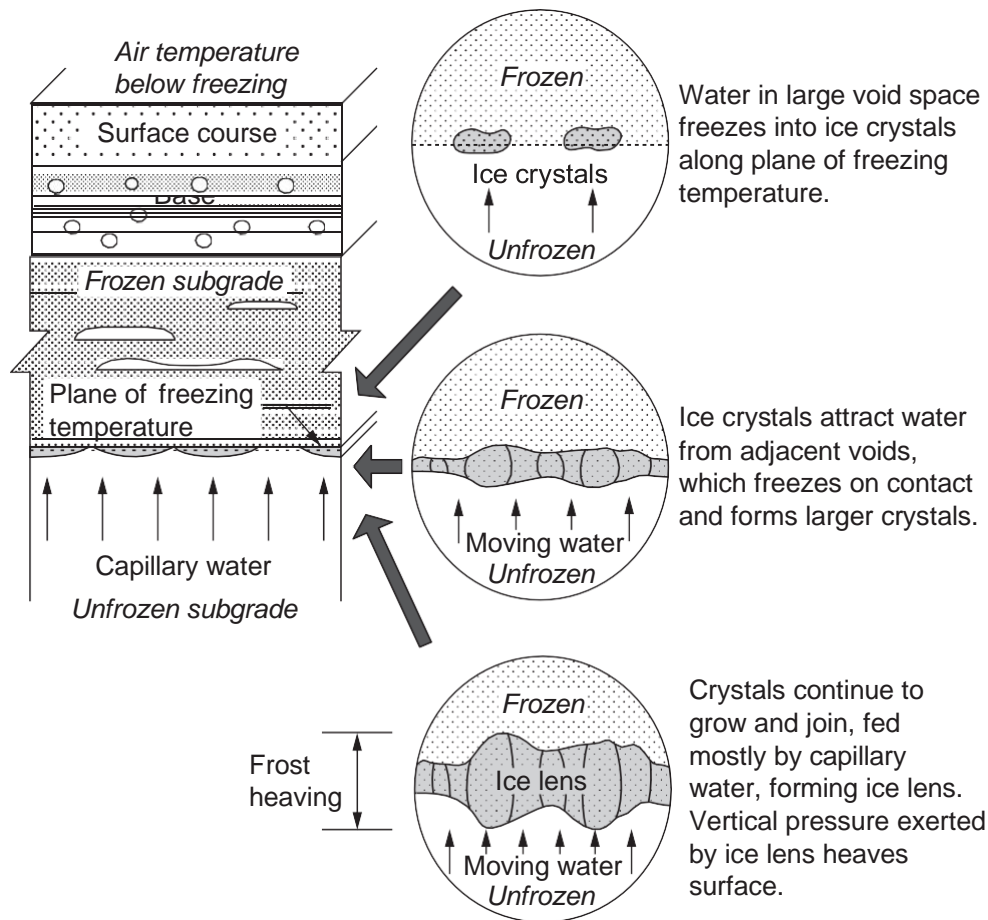


FIGURE 7-3 The process of ice segregation ([http:// www.pavementinteractive.org](http://www.pavementinteractive.org)).

During periods of thaw, the ice lenses begin to melt, and the water which is released cannot drain through the still-frozen soil at greater depths. Thus, lack of drainage results in loss of strength in the subgrade. It is also possible that a reduction in stiffness will occur in subgrade soils during the thaw period, even though ice lenses may not have formed.

Originally developed by the U.S. Army Corps of Engineers, the FAA categorizes soils into four "frost groups." Soils in frost group 1 are least susceptible to frost and associated soil weakening, while soils in frost group 4 are most susceptible. As illustrated in Table 7-4, those soils with larger particle sizes, such as the gravelly soils, are found in frost group 1, while very fine soils are found in frost group 4.

The design of pavements, both flexible and rigid, is modified slightly depending on the propensity of the soil to encounter frost and the depth of the frost, known as frost penetration. These considerations are described in further detail later in this chapter.

Subgrade Stabilization

In addition to frost, factors such as poor drainage, adverse surface drainage, or merely variations in soil depths, contribute to reductions

Frost Group	Kinds of Soils	Percentage Finer than 0.02 mm by Weight	Soil Classification
FG-1	Gravelly soils	3–10	GW, GP, GW-GM, GP-GM
FG-2	Gravelly soils sands	10–20 3–5	GM, GW-GM, GP-GM. SW, SP, SM, SW-SM, SP-SM
FG-3	Gravelly soils Sands, except very fine silty sands Clays, PI above 12	Over 20 Over 15 –	GM, GC SM, SC CL, CH
FG-4	Very fine silty sands All silts Clays, PI = 12 or less Varved clays and other fine grained banded sediments	Over 15 – – –	SM ML, MH CL, CL-ML CL, CH, ML, SM

TABLE 7-4 Frost Design Soil Classification

in the stability of a soil. The FAA allows for the stabilization and treatment of the soils to improve the performance of the constructed pavement. Two types of soil stabilization exist, mechanical stabilization and chemical stabilization. Mechanical stabilization consists of embedding cobble or shot rock sheets within the soil. In some cases, porous concrete or geosynthetics may be used for very soft fine grained soils. Chemical stabilization is achieved by the addition of proper percentages of cement, lime, fly ash, or combinations of these materials to the soil.

FAA Pavement Design Methods

Between 1958 and 2006, the FAA established mandates for aircraft manufacturers to create aircraft, based on their maximum gross take-off weight and landing gear configuration, that produce loads on pavements no greater than 350,000 lb, based on the aircraft at the time that created the heaviest load on airfield pavements, the Douglas DC-8. As aircraft grew in gross weight, landing gear configurations, with

additional wheels and spacings, were created to distribute increased gross weights, resulting in equivalent per wheel loads not exceeding the 350,000 lb maximum.

Equivalent Aircraft Method

Historical airfield pavement design methods recommended by the FAA beginning in 1975 took into account the varying weights and landing gear configurations of the fleet of aircraft that may regularly utilize a given airfield's pavement. This historical method involved determining the number of total annual aircraft departures by each type of aircraft and group them into "equivalent annual departures" of each aircraft in terms of the landing gear configuration of a given design aircraft, that is, the aircraft in the fleet mix that requires the greatest pavement strength. This grouping is based on converting the number of annual departures of all aircraft other than the design aircraft to an equivalent number of annual departures by using the multipliers given in Table 7-5.

The equivalent annual departures of the design aircraft were determined by summing the equivalent annual departures of each aircraft in the group, according to the formula given in Eq. (7-4).

$$\text{Log } R_1 = \text{log } R_2 \times \frac{W_2^{1/2}}{W_1} \quad (7-4)$$

where R_1 = equivalent annual departures by the design aircraft

R_2 = annual number of departures by an aircraft in terms of
design aircraft landing gear configuration

W_1 = wheel load of the design aircraft

W_2 = wheel load of the aircraft being converted

To Convert From	To	Multiply Departures By
Single wheel	Dual wheel	0.8
Single wheel	Dual tandem	0.5
Dual wheel	Dual tandem	0.6
Double dual tandem	Dual tandem	1.0
Dual tandem	Single wheel	2.0
Dual tandem	Dual wheel	1.7
Dual wheel	Single wheel	1.3
Double dual tandem	Dual wheel	1.7

TABLE 7-5 Factors for Converting Annual Departures by Aircraft to Equivalent Annual Departures by Design Aircraft

Because many of the latest generation aircraft require more complex landing gear configurations that are provided in Table 7-5, a special consideration for very heavy aircraft were made by assigning a gross takeoff weight of 300,000 lb and a dual-tandem landing gear configuration to any aircraft with maximum gross takeoff weight greater than 300,000 lb. This rough approximation was, in part, motivation, to develop an entirely new assessment of fleet mix with respect to airfield pavement design and evaluation.

Cumulative Damage Failure Method

The current method of airfield pavement design and evaluation now considers each type of aircraft that uses the pavement explicitly. The “design aircraft” concept has been replaced by design for fatigue failure expressed in terms of a “cumulative damage factor” (CDF). The CDF for a given aircraft is a value between 0 and 1 which expresses the contribution to ultimate pavement failure of the projected number of uses for each aircraft type that use the pavement. Based on Miner’s rule, a traditional theory which estimates the amount of use until failure of a pavement, the CDF for a given fleet of aircraft is determined by Eq. (7-5).

$$\text{CDF} = \sum (n_i/N_i) \quad (7-5)$$

where n_i is the expected number of annual departures of aircraft i and N_i is the number of departures of aircraft i that would lead to pavement failure for each aircraft i in the mix.

When CDF meets or exceeds 1, the cumulative predicted number of operations for each of the aircraft in the mix will lead to failure of a given pavement system. Any value less than 1 represents the fraction of pavement life that has been effectively “used up.” For example, a CDF of 0.75 would indicate that the pavement has used 75 percent of its useful life, and has 25 percent of its life remaining under the predicted traffic usage before fatigue failure.

For both the design of flexible and rigid pavements, the current FAA pavement design method applies a computer software model to estimate the appropriate thickness of designed pavement layers, given the Young’s modulus E value of the subgrade and the expected aircraft fleet mix, such that the CDF of the pavement equals 1 after a 20-year life of the designed pavement.

The FAA approved software, *FAA Rigid and Flexible Iterative Elastic Layered Design* or *FAARFIELD* comes equipped with a library of aircraft, their maximum gross weights, landing gear configuration, and contribution to CDF for the given pavement design. Figure 7-4 illustrates the “aircraft” window of *FAARFIELD* with user inputs of each given aircraft’s estimated departures for the to-be-designed pavement system.

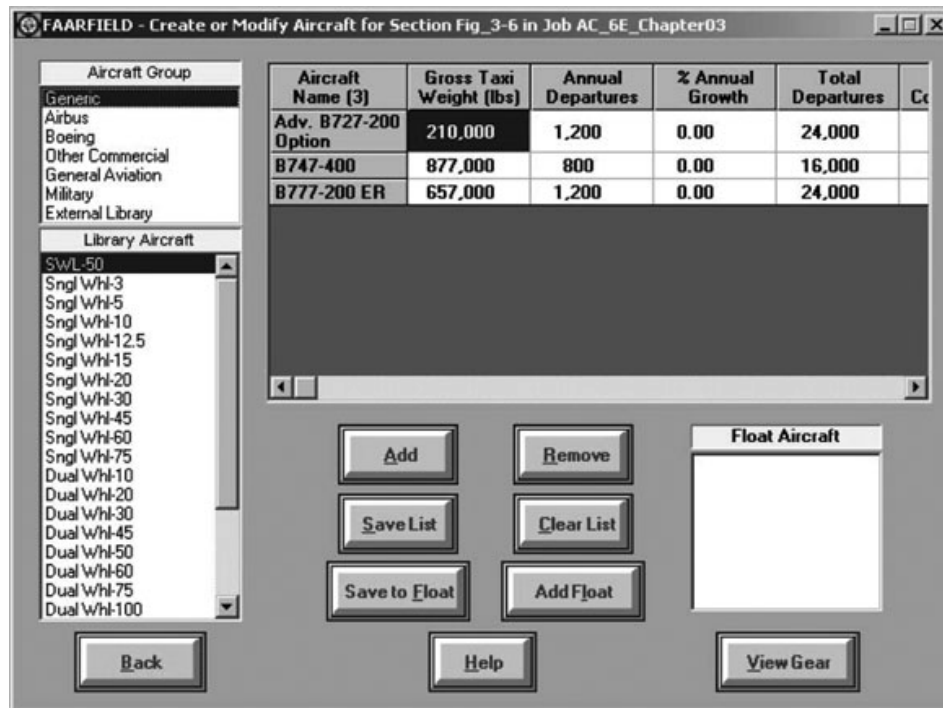


FIGURE 7-4 FAARFIELD aircraft database window.

Design of Flexible Pavements

Flexible pavements consist of a bituminous wearing surface placed upon a base course and, where required by subgrade conditions, a subbase. The entire flexible-pavement structure is ultimately supported by the subgrade. The surface course prevents the penetration of surface water to the base course, provides a smooth, well-bonded surface free of loose particles, resists the shearing stresses caused by aircraft loading, and furnishes a texture of nonskid qualities not causing undue tire wear. The course must also be resistant to fuel spillage and other solvents in areas where maintenance may occur.

The base course is the major structural element of the pavement; it has the function of distributing the wheel loads to the subbase and subgrade. It must be designed to prevent failure in the subgrade, withstand the stresses produced in the base course, resist vertical pressures tending to produce consolidation and deformation of the wearing course, and resist volume changes caused by fluctuations in its moisture content.

Flexible pavement base courses are available in different compositions, or “types,” including:

1. Item P-208—Aggregate Base Course
2. Item P-209—Crushed Aggregate Base Course
3. Item P-211—Lime Rock Base Course
4. Item P-304—Cement Treated Base Course

5. Item P-306—Econocrete Subbase Course
6. Item P-401—Plant Mix Bituminous Pavements
7. Item P-403—HMA Base Course

P-211, P-304, P-306, P-401, and P-403 are considered stabilized based courses.

The function of the subbase, when required, is similar to that of the base course, but since the subbase is further removed from the area of load application, it is subjected to lower stress intensities. Sub-bases are typically required when flexible pavement is to be supported by soils of CBR value less than 20.

Flexible pavement subbase courses are available in different types, including:

1. Item P-154—Subbase Course
2. Item P-210—Caliche Base Course
3. Item P-212—Shell Base Course
4. Item P-213—Sand Clay Base Course
5. Item P-301—Soil Cement Base Course

The subgrade soils are subjected to the lowest loading intensities, and the controlling stresses are usually at the top of the subgrade since stress decreases with depth. However, unusual subgrade conditions, such as layered subgrade materials, can alter the location of controlling stresses.

CBR Method

Prior to 2008, the FAA's standard method for flexible pavement design was known as the CBR method. The CBR method was based on approximation charts that factored in the CBR value of the subgrade and the number and gross weight of equivalent annual departures of the design aircraft. Separate approximation charts were provided by the FAA for different generic aircraft landing gear configurations, and for aircraft greater than 300,000 lb maximum gross weight, specific individual aircraft. Figure 7-5 provides an illustrative example of the CBR method.

The example nomograph found in Fig. 7-5 represents the historical method of estimating the base level thickness of flexible pavement for a Boeing 767. The arrow within the nomograph represents the example for a subgrade with CBR value of 7, a 325,000-lb aircraft gross weight, and 1200 annual equivalent departures, resulting in a required base course of 30 in thickness. The nomograph also provides the necessary thickness for the surface layer, at 4 in thick for critical areas and 3 in thick for noncritical areas, such as pavement shoulders.

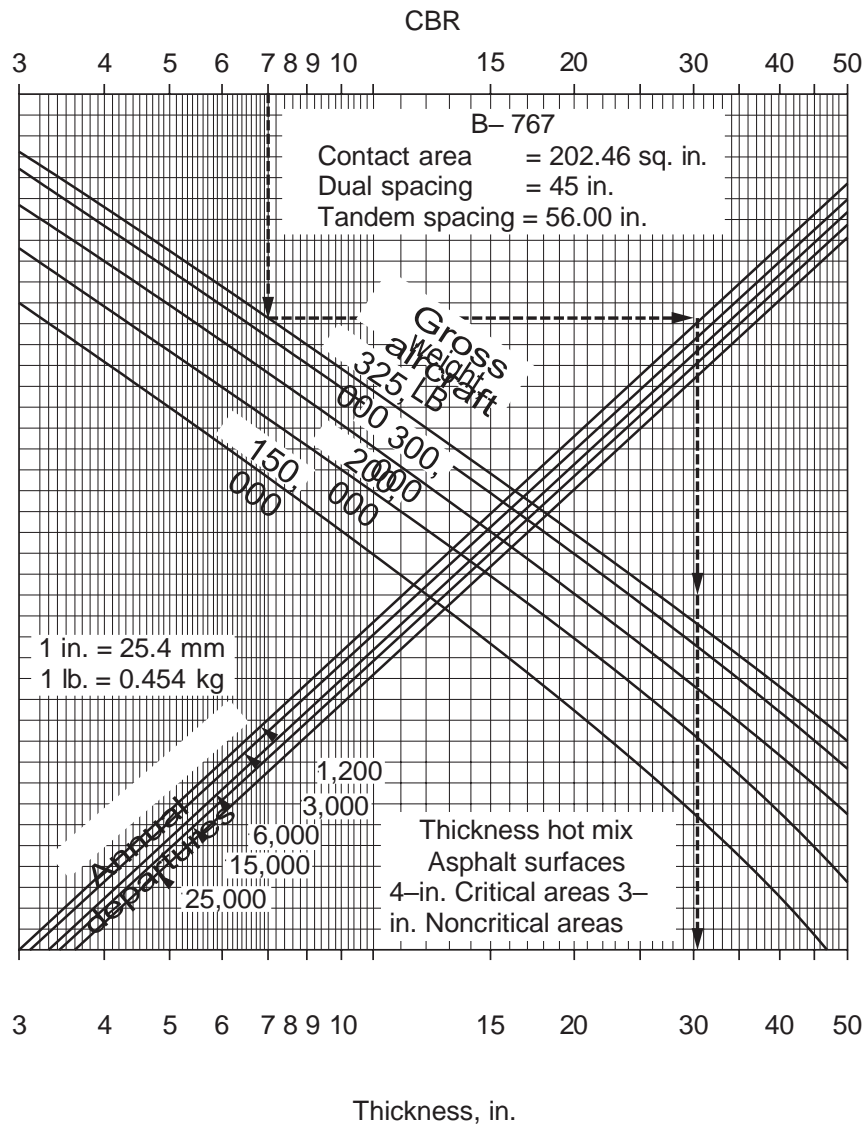


FIGURE 7-5 Example approximation chart, CBR method of flexible pavement design.

Layered Elastic Design

Originally applied in 1995 specifically for the heaviest of aircraft, the FAA adopted the layered elastic design (LED) method of flexible pavement design for all pavements designed to accommodate aircraft greater than 30,000 lb in 2008.

Layered elastic design theory considers the fact that the layers of pavement that support loads are impacted by both vertical and horizontal strains and stress, as illustrated in Fig. 7-6. To accommodate the strain, pavement will deflect with the passing of the load. The magnitude of deflection of a given pavement is a function of its elasticity, E , as measured by Young's modulus. In addition, the ratio of transverse to horizontal deflection of a pavement layer, known as Poisson's ratio, μ , is considered.

The layered elastic design and cumulative damage failure methods of pavement design are applied in the FAA's computer pavement design software, FAARFIELD. FAARFIELD uses a Windows-based

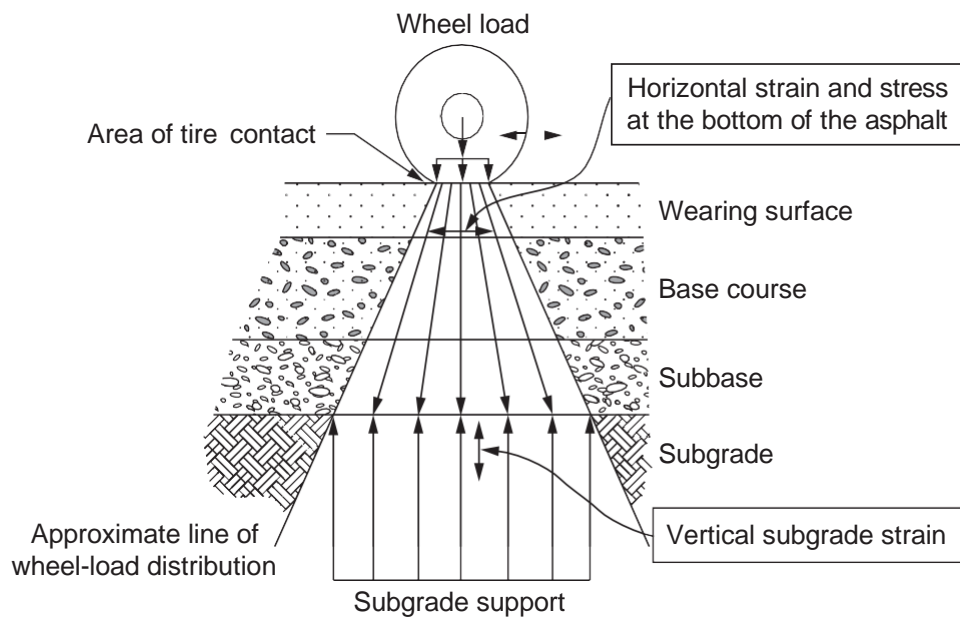


FIGURE 7-6 Visualization of layer elastic design theory.

interface to allow the user to input initial data concerning the subgrade of the area on which the pavement is to be designed, specifically the Young's modulus of the subgrade, as well as the expected fleet mix that will be using the pavement. As illustrated in Fig. 7-7, FAARFIELD provides the recommended thickness of each layer within the flexible pavement structure, using recommended pavement

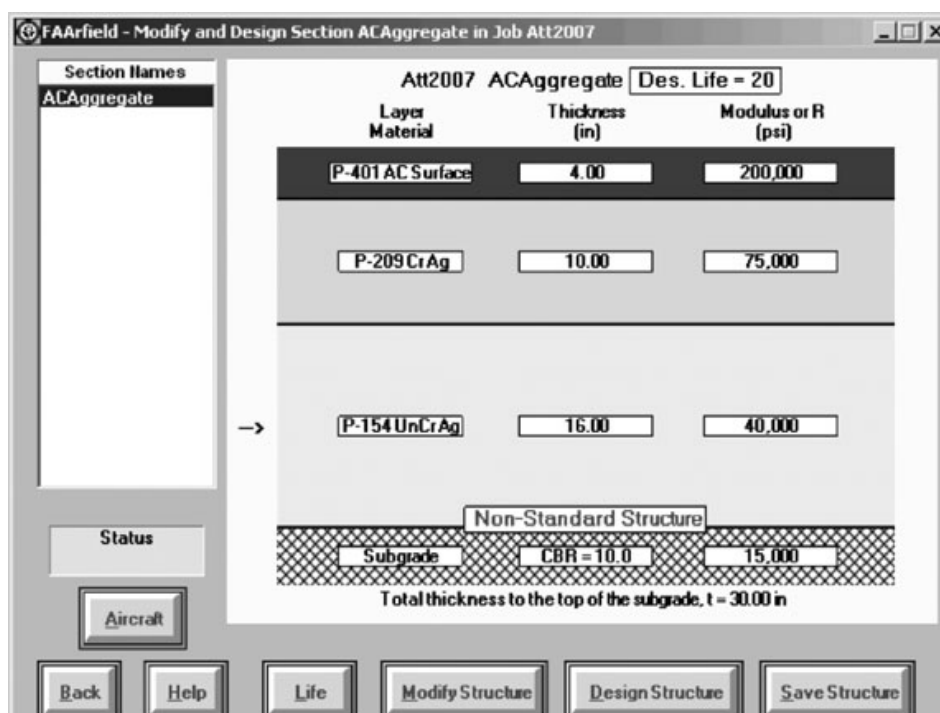


FIGURE 7-7 Example output of FAARFIELD software for flexible pavement using layered elastic design theory.

types (ex. P-401/403 asphalt surfaces), or other user preferred types listed in this section.

Design of Rigid Pavements

Rigid pavements consist of slabs of PCC placed on a subbase that is supported on a compacted subgrade. Like flexible pavements, a properly designed rigid pavement provides a nonskid surface which prevents the infiltration of water into the subgrade, while providing structural support to aircraft which use the pavement.

The subbase under rigid pavements provides uniform stable support for the concrete slabs. As a rule, a minimum thickness of 4 in is required for all subbases under rigid pavements. There are various types of mixtures which are acceptable for rigid pavement subbases including:

- Item P-154—Subbase Course
- Item P-208—Aggregate Base Course
- Item P-209—Crushed Aggregate Base Course
- Item P-211—Lime Rock Base Course
- Item P-301—Soil Cement Base
- Item P-304—Cement Treated Base Course
- Item P-306—Econocrete Subbase Course
- Item P-401—Plant Mix Bituminous Pavements
- Item P-403—HMA Base Course

For rigid pavements accommodating aircraft greater than 100,000lb maximum gross weight a stabilized subbase is required, which include items P-304, P-306, P-401, and P-403.

Westergaard's Analysis

Similar to the CBR method of design for flexible pavements, prior to 2008, rigid pavement design using nomographs and other approximation charts based on theories developed by H. M. Westergaard was the FAA standard. Westergaard's analysis of pavement design was founded in the mid-1920s and focused on the calculations of stresses and deflections in concrete pavements due to applied loading.

Westergaard assumed the pavement slab to be a thin plate resting on a special subgrade which is considered elastic in the vertical direction only. That is, the reaction is proportional to the deflection of the subgrade $p = kz$, where z is deflection and k is a soil constant, referred to as the *modulus of subgrade reaction*. Other assumptions are that the concrete slab is a homogeneous, isotropic elastic solid and that the wheel load of an aircraft is distributed over an elliptical area. Although

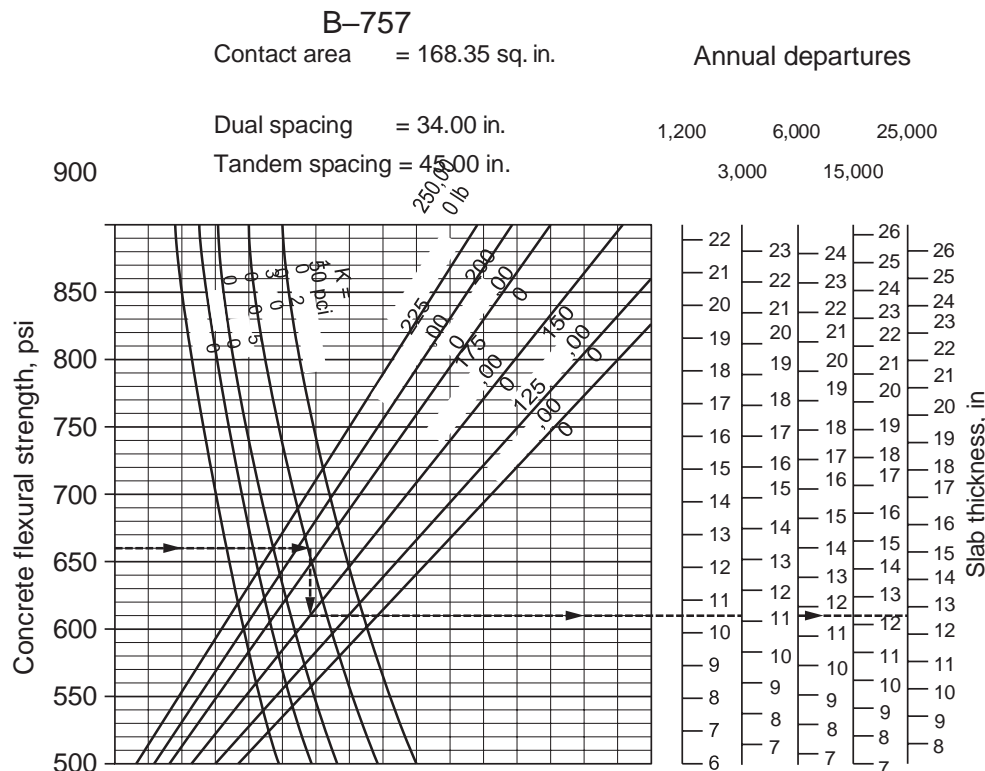


FIGURE 7-8 Example design curve for estimating the slab thickness of rigid pavement using Westergaard's analysis.

these assumptions do not satisfy theory in a strict sense, the results compared reasonably with observations. The Westergaard analysis was used to evaluate stress in a pavement, as well as the deflection of the slab. For airports, Westergaard developed formulas for stresses and deflections in the interior of a slab and at an edge of a slab.

The U.S. Army Corps of Engineers applied Westergaard's formulas toward the creation of approximation charts and design curves. An example is illustrated in Fig. 7-8.

The arrow found in Fig. 7-8 illustrates an example rigid pavement design analysis, considering the use of a PCC mixture of flexural strength of 660 lb/in², a subgrade with k value 100 lb/in³, for a Boeing 757 design aircraft with maximum gross weight of 175,000 lb and 6000 annual equivalent departures, resulting in a design slab thickness of approximately 12 in.

Finite Element Theory

Similar to flexible pavement design, FAARFIELD processes the user defined fleet mix and E value of the subgrade to determine the thickness requirements of the PCC surface. FAARFIELD recommends default subbase layers at 6 in thickness. Multiple subbase layers are recommended for certain subgrades with low E values. Figure 7-9 illustrates an example rigid pavement structure output from FAARFIELD.

FAARFIELD applies finite element theory to estimate the thickness of the PCC surface and any necessary subbase courses. Three-dimensional finite element design theory (3D-FE) is similar to layered

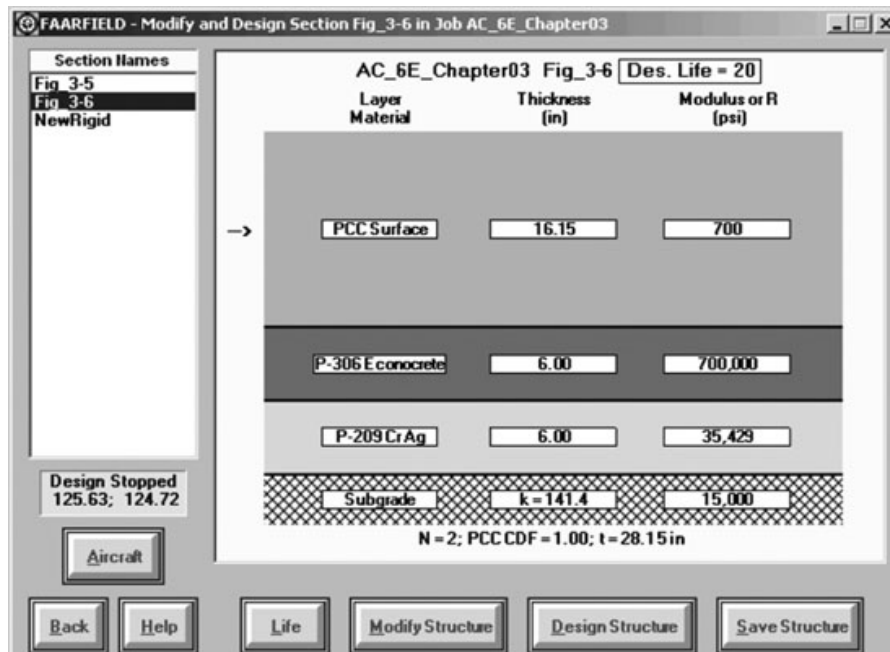


FIGURE 7-9 Example output of FAARFIELD software for rigid pavement design.

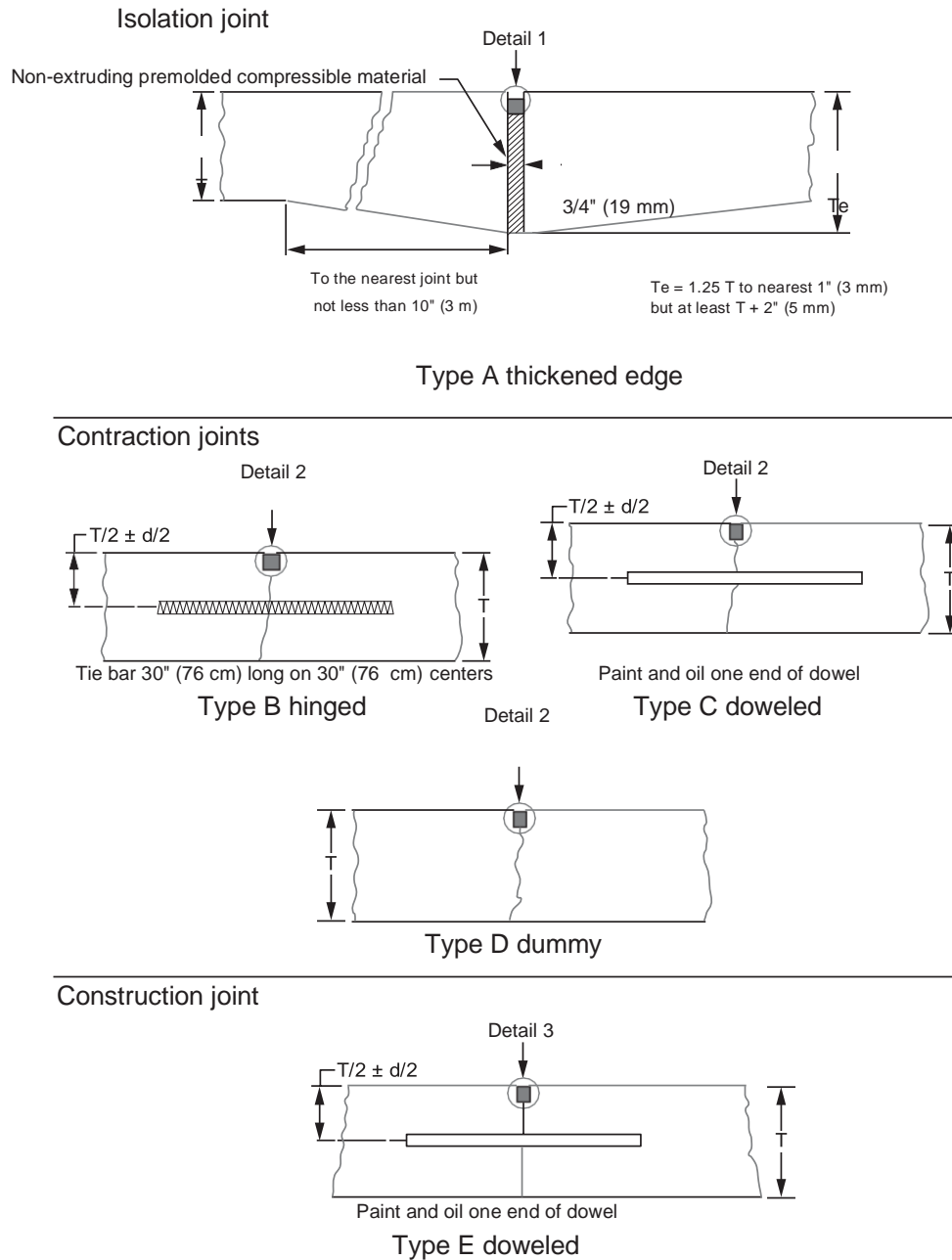
elastic design theory in that it takes into account the Young's modulus of the subgrade and the materials used in the slab and subbase courses, and considers a cumulative damage factor in its analysis. 3D-FE design modeling, however, considers the pavement in discrete sections, rather than a continuous material. This perspective allows for more accurate estimation of stresses and strain on the edges of the rigid pavement slabs, which compared to the transverse stress near the center of the slab, is more critical in rigid pavements.

Joints and Joint Spacing

Slabs of PCC rigid pavement are connected by joints to permit expansion and contraction of the pavement, thereby relieving flexural stresses due to curling and friction and to facilitate construction. There are three types of joints: isolation (type A), contraction (type B, C, or E), and construction (type E) joints. The locations of these joints are illustrated in Fig. 7-10, with details as to their specifications found in Fig. 7-11.

The function of type A *isolation joints* is to isolate adjacent pavement slabs and provides space for the expansion of the pavement, thereby preventing the development of very high compressive stresses which can cause the pavement to buckle.

Contraction joints are provided to relieve the tensile stresses due to temperature, moisture, and friction, thereby controlling cracking. If contraction joints were not installed, random cracking would occur on the surface of the pavement. The spacing between contraction joints is dependent on the thickness of the slab, the character of the



NOTE:

1. Shaded area is joint sealant.
2. Groove must be formed by sawing.

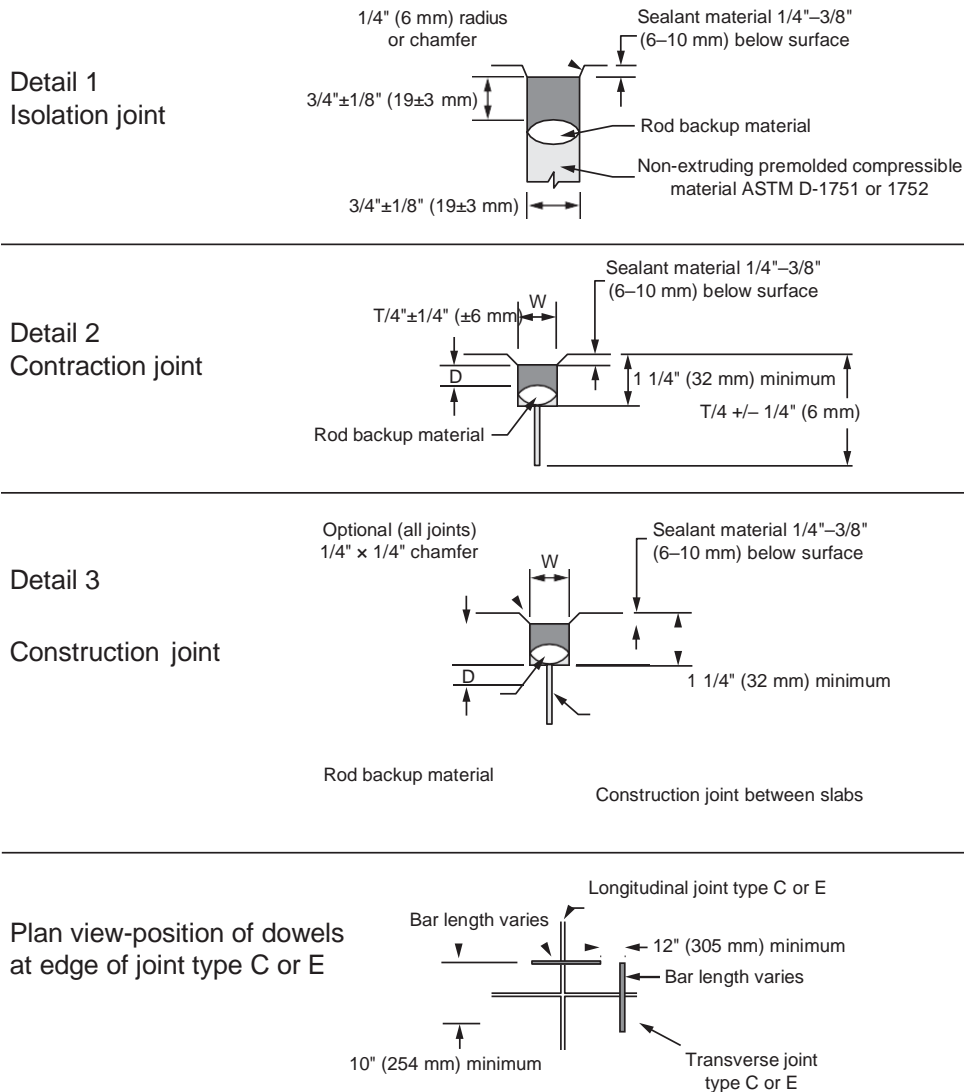
FIGURE 7-10 Location of rigid pavement joint structures.

aggregate, and whether the slab is plain or reinforced. On the basis of experience, it has been found that for plain slabs 8 to 10 in thick, the spacing should be in the range of 15 to 20 ft. For thicker slabs, the spacing can be increased to 25 ft. Contraction joints may be hinged (type B), doweled (type C), or considered a “dummy” joint (type D). The longitudinal and transverse specifications of these joints are in details in Table 7-6.

Construction joints are required to facilitate construction when two abutting slabs are placed at different times.

All joints in concrete pavements are sealed with sealing compound to prevent infiltration of water or foreign material into the joint spaces.

This joint sealant must be capable of withstanding repeated extension and compression as the pavement changes volume.



NOTES:

1. Sealant reservoir sized to provide proper shape factor, W/D . Field poured and preformed sealants require different shape factors for optimum performance.
2. Rod backup material must be compatible with the type of sealant used and sized to provide the desired shape factor.
3. Recess sealant $3/8$ "– $1/2$ " for joints perpendicular to runway grooves.
4. Chamfered edges are recommended when pavements are subject to snow removal equipment of high traffic volumes.

FIGURE 7-11 Rigid pavement joint structure details.

Dowels are load transfer devices which permit joints to open by which prevent differential vertical displacement. Usually dowels are solid, round steel bars, although pipe may also be used. Several different analyses have been proposed for the design of dowels. The spacing of dowels depends on the thickness of the pavement, modulus of subgrade reaction, and the size of the dowel. Table 7-7 contains recommendations for dowel sizes and spacings.

Continuously Reinforced Concrete Pavements

A continuously reinforced concrete pavement (CRCP) is one in which transverse joints have been eliminated (except where the pavement intersects or abuts existing pavements or structures) and the longitudinal reinforcing steel is continuous throughout the length of

the pavement. Other than the design of the embedded

Type	Description	Longitudinal	Transverse
A	Thickened edge isolation joint	Use at intersections where dowels are not suitable and where pavements abut structures. Consider at locations along a pavement edge where future expansion is possible.	Use at pavement feature intersections when the respective longitudinal axis intersects at an angle. Use at free edge of pavements where future expansion, using the same pavement thickness is expected.
B	Hinged contraction joint	For all contraction joints in taxiway slabs <9 in (230 mm) thick. For all other contraction joints in slabs <9 in (230 mm) thick, where the joint is placed 20 ft (6 m) or less from the pavement edge.	Not used.
C	Doweled contraction joint	May be considered for general use. Consider for use in contraction joints in slabs >9 in (230 mm) thick, where the joint is placed 20 ft (6 m) or less from the pavement edge.	May be considered for general use. Use on the last three joints from a free edge, and for three joints on either side of isolation joints.
D	Dummy contraction joint	For all other contraction joints in pavement	For all other contraction joints in pavement.
E	Doweled construction joint	Doweled construction joints excluding isolation joints.	Use for construction joints at all locations separating successive paving operations ("headers").

TABLE 7-6 Pavement Joint Types

steel within the pavement, thickness design of CRCP is identical to other rigid pavement types.

The advantages for placing steel in PCC pavements include:

1. Reducing the number of required joints between slabs, resulting in decreased maintenance costs
2. Prolonging service life when pavement is overloaded
3. Reducing pavement deflection

Thickness of Slab	Diameter	Length	Spacing
6–7 in (150–180 mm)	¾ in* (20 mm)	18 in (460 mm)	12 in (305 mm)
8–12 in (210–305 mm)	1 in* (25 mm)	19 in (480 mm)	12 in (305 mm)
13–16 in (330–405 mm)	1¼ in* (30 mm)	20 in (510 mm)	15 in (380 mm)
17–20 in (430–510 mm)	1½ in* (40 mm)	20 in (510 mm)	18 in (460 mm)
21–24 in (535–610 mm)	2 in* (50 mm)	24 in (610 mm)	18 in (460 mm)

*Dowels noted may be solid bar or high-strength pipe. High-strength pipe must be plugged on each end with a tight-fitting plastic cap or mortar mix.

TABLE 7-7 Dimensions and Spacing of Steel Dowels in Rigid Pavement

The amount of reinforcing steel required to control volume changes is dependent primarily on the slab thickness, concrete tensile strength, and yield strength of the steel. While several procedures have been proposed for estimating the required amount of steel, experience indicates that it should be approximately 0.6 percent of the gross cross-sectional area and that the yield strength should be at least 60,000 lb/in². The minimum amount may be determined by Eq. (7-6).

$$P_s(\%) = \left(\frac{f_t}{f_s} \right) \frac{1}{F} \quad (7-6)$$

where P_s = percentage of embedded steel

f_t = tensile strength of concrete, lb/in²

f_s = allowable working stress in steel, lb/in²

F = coefficient of subgrade friction

The FAA recommends that the cross-sectional area of the reinforcing steel A_s be obtained by Eq. (7-7).

$$A_s = \frac{(3.7)L\sqrt{Lt}}{f_s} \quad (7-7)$$

where

A_s = area of steel per foot of width or length, in²

L = length or width of slab, ft

T = thickness of slab, in

f_s = allowable tensile stress in steel, lb/in²

Note: To determine the area of steel in metric units:

L should be expressed in meters

t should be expressed in millimeters

f_s should be expressed in meganewtons per square meter

The constant 3.7 should be changed to 0.64.

f_s will then be in terms of square centimeters per meter.

The longitudinal embedded steel must also be capable of withstanding the forces generated by the expansion and contraction of the pavement due to temperature changes. Equation (7-8) determines the amount of steel required as a function of temperature.

$$P_s = \frac{50 f_t}{f_s - 195T} \quad (7-8)$$

where

- P_s = embedded steel in percent
- f_t = tensile strength of concrete, 67 percent of the flexural strength is recommended
- f_t = working stress for steel usually taken as 75 percent of specified minimum yield strength
- T = maximum seasonal temperature differential for pavement in degrees Fahrenheit

Longitudinal embedded steel is located at mid-depth or slightly above mid-depth of the slab.

Transverse embedded steel is recommended for CPRP airport pavements to control random longitudinal cracking. Equation (7-9) is used to determine the amount of transverse steel, as a percentage of the total slab area.

$$P_s(\%) = \frac{W_s F}{2 f_s} 100 \quad (7-9)$$

where

- P_s = embedded steel in percent
- W_s = width of slab, in ft
- F = friction factor of subgrade
- f_s = allowable working stress in steel, in lb/in². Yield strength of 0.75 is recommended

Transverse steel is designed in the same way as tie bars.

Design of Overlay Pavements

Overlay pavements are required when existing pavements are no longer serviceable due to either deterioration in structural capabilities or a loss in riding quality. They are also required when pavements must be strengthened to carry greater loads or increased repetitions of existing aircraft beyond those anticipated in the original design. Overlays also provide a solution for increased safety. An example would be to provide improved skid resistance and reduced risk of hydroplaning.

There are several types of overlay pavements. A concrete pavement can be overlaid with additional concrete, a bituminous surfacing, or a

combination of aggregate base course and a bituminous surfacing. Likewise, a flexible type of pavement can be overlaid with concrete, a bituminous surfacing, or the combination of aggregate base course and a bituminous surfacing. The various types of overlay pavements are defined as follows:

1. *Overlay pavement*: The thickness of a rigid or flexible type of pavement placed on an existing pavement
2. *Portland cement concrete overlay*: An overlay pavement constructed of portland cement concrete
3. *Bituminous overlay*: An overlay consisting entirely of a bituminous surfacing
4. *Flexible overlay*: An overlay consisting of a base course and a bituminous surfacing

Figure 7-12 provides an illustration of typical pavement overlay designs.

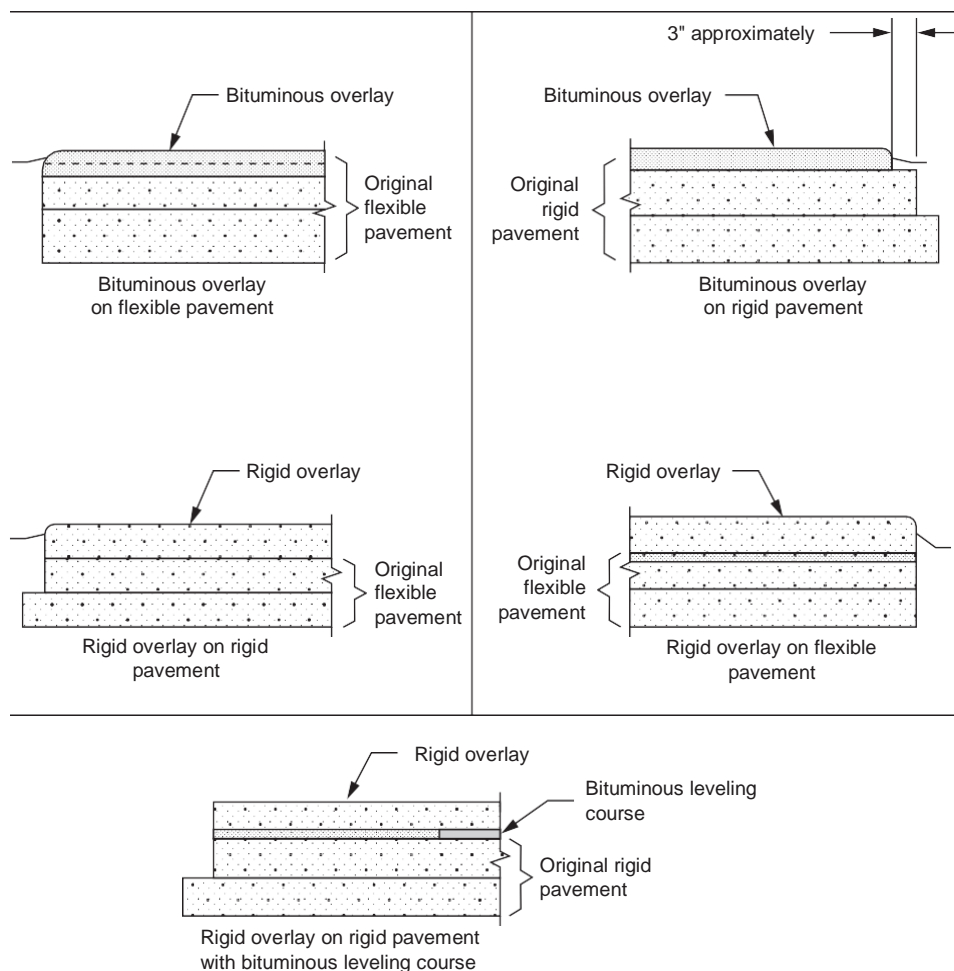


FIGURE 7-12 Typical overlay pavements.

The FAA's FAARFIELD pavement design program includes capabilities for designing airfield pavement overlays. The four types of overlays considered in FAARFIELD are

1. Hot mix asphalt overlay of existing flexible pavement
2. Concrete overlay of existing flexible pavement
3. Hot mix asphalt overlay of existing rigid pavement
4. Concrete overlay of existing rigid pavement

Based on the thickness and condition of the existing pavement layers, FAARFIELD estimates the required thickness of the overlay. Figure 7-13 provides an illustration of FAARFIELD's output for a flexible overlay on an existing flexible pavement.

The design of overlays over an existing rigid pavement is slightly more complex as the condition of the existing rigid pavement plays a significant role in the required thickness of the overlay. The condition of the existing rigid pavement is estimated using a *structural condition index* (SCI), a value which ranges from 0 to 100, in which 100 representing a pavement with no visible structural deficiencies and 0 representing total structural failure. Visible distresses that contribute to a lower SCI include

- Corner breaks
- Longitudinal, transverse, or diagonal cracking

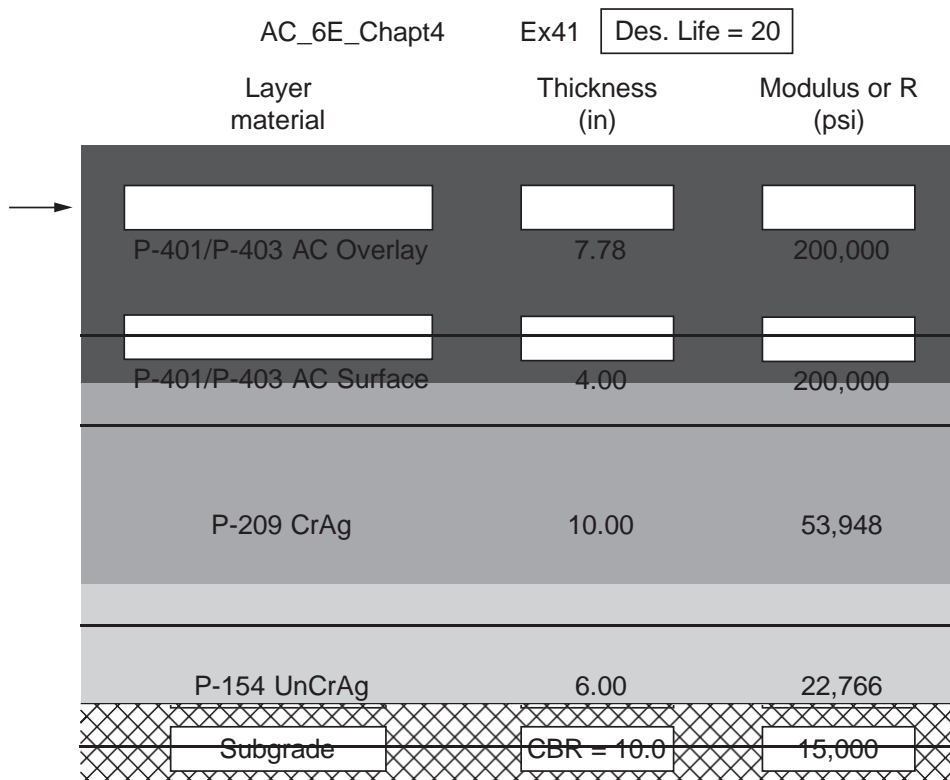


FIGURE 7-13 Design example of flexible overlay on existing flexible pavement.

- Shattered slab
- Shrinkage cracks
- Spalling (cracking, breaking, or chipping of joint/crack edges)

The SCI is estimated using standard pavement structural condition formulas based on empirical analysis that may be found in general concrete structural evaluation references.

In the case when there are no visible or otherwise degradations in structural condition, the FAA calls for the estimation of a *cumulative damage factor used* (CDFU). For aggregate base layers, CDFU may be estimated by Eq. (7-10).

$$CDFU = \frac{L_u}{0.75L_D} \text{ when } L_u < 0.75 L_D \quad (7-10)$$

$$= 1 \text{ when } L_u \geq 0.75L_D$$

where:

L_u = number of years of operation of the existing pavement until overlay

L_D = design life of the existing pavement in years

For rigid pavement bases, CDFU is estimated using the FAARFIELD software, based on the number of years the pavement has been in use to date. Figure 7-14 illustrates the FAARFIELD output for estimating CDFU for rigid pavement bases.

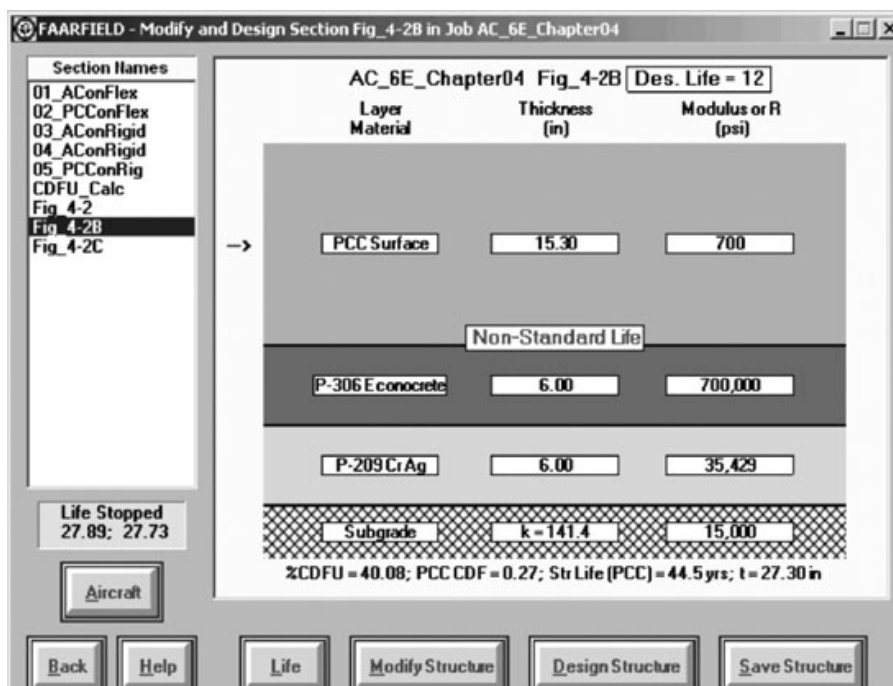


FIGURE 7-14 CDFU estimation for rigid pavement bases using FAARFIELD.

There are several other complexities associated with pavement overlays, particularly with respect to rigid pavements, that are beyond the scope of this text. It is strongly recommended that further study include in-depth review of the FAA Advisory Circular AC 150/5320-6E, "Airfield Pavement Design and Evaluation," as well as familiarization with the FAARFIELD software package.

Pavements for Light Aircraft

Pavements for light aircraft are defined as landing areas intended for personal or other small aircraft engaged in nonscheduled activities, such as recreational, agricultural, or instructional activities, or small aircraft charter operations. Pavements for light aircraft are designed to accommodate aircraft with less than 30,000 lb maximum gross weight. In many cases these aircraft will not exceed 12,500 lb. Figure 7-15 illustrates the composition of light aircraft pavements. Note that, as opposed to pavements for heavier aircraft, light aircraft pavements do not have critical versus noncritical areas and as such the surface thickness of pavement is the same through the paved area.

The FAA FAARFIELD software provides the capability to design pavements for light aircraft, using a similar procedure for typical flexible and rigid pavements. FAARFIELD requires the CBR or modulus E value of the subgrade, the aircraft mix, gross weights, and annual

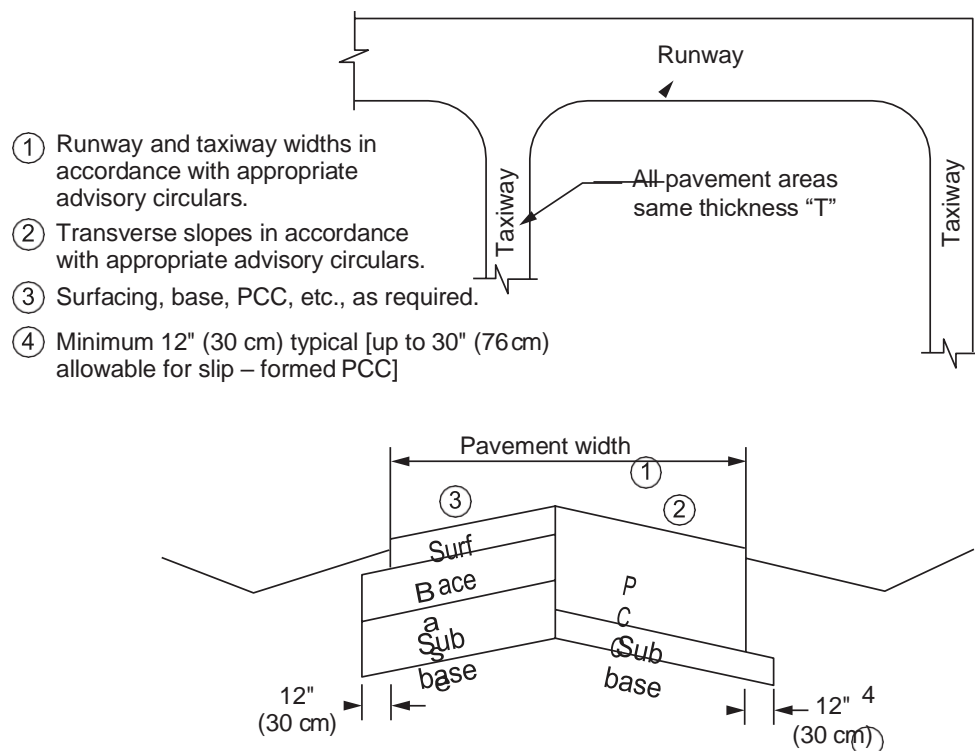


FIGURE 7-15 Typical sections for light aircraft pavements.

departures of all aircraft. For flexible pavements, FAARFIELD will estimate the total thickness of the pavement, including a minimum 2 in surface course. For rigid pavements, FAARFIELD will estimate the slab thickness. In addition FAARFIELD will call for a minimum sub-base thickness of 4 in for aircraft weighing 12,500 lb maximum gross weight or greater.

Other than using flexible or rigid pavement structures, landing facilities for light aircraft may be turf or an aggregate-turf mixture. FAARFIELD also has capabilities for estimating the composition of aggregate-turf mixtures.

Pavement Evaluation and Pavement Management Systems

A pavement management system (PMS) is a mechanism for providing consistent, objective, and systematic procedures for evaluating pavement condition and for determining the priorities and schedules for pavement maintenance and rehabilitation within available resource and budgeting constraints. The pavement management system can also be used to maintain records of pavement condition and to provide specific recommendations for actions which may be required to maintain a pavement network at an acceptable condition while minimizing the cost associated with pavement maintenance and rehabilitation.

A pavement management system evaluates present pavement condition and predicts future condition through the use of a pavement condition indicator. By projecting the rate of deterioration in the pavement condition indicator and adopting some minimum acceptable level for this indicator, a life-cycle cost analysis can be performed for various maintenance and rehabilitation alternatives, and a determination can be made of the optimal time for the application of the most appropriate alternative. The rate of deterioration of a pavement accelerates with time. By implementing a maintenance or rehabilitation strategy to upgrade the pavement condition at the proper time the overall cost of maintenance and rehabilitation can be minimized. As noted by the FAA, the total annual cost to maintain or rehabilitate a pavement in relatively poor condition can be 4 to 5 times that of maintaining or rehabilitating a pavement in relatively good condition.

An effective PMS for use at airports should include the following components:

1. A systematic mechanism for regularly collecting, storing, and retrieving the necessary data associated with pavement use and condition
2. An objective system for evaluating pavement condition at regular intervals

3. Procedures for identifying alternative maintenance and rehabilitation strategies
4. Mechanisms for predicting and evaluating the impact of pavement maintenance and rehabilitation strategies and alternatives on pavement condition, serviceability, and useful service life
5. Procedures for estimating and comparing the costs of various strategies and alternatives
6. Techniques for identifying the optimal strategy or alternative based upon relevant decision criteria

Essential to an effective PMS is the maintenance of a pavement database which should include

1. Information about the pavement structure, including when it was originally constructed, the structural components, the soil conditions, a history of subsequent maintenance and rehabilitation, and the cost of these actions.
2. A record of the airport pavement traffic including the number of aircraft operations by various types of aircraft using the pavement over its life.
3. The ability to regularly track pavement condition, including measures of pavement distress and the causes of distress. A pavement rating system should be developed based upon the quantity, severity, and type of distress affecting the pavement surface condition. This rating system measures pavement surface performance and has implications for structural performance. The periodic collection of condition-rating data is essential to tracking pavement performance

As part of an effective PMS, the evaluation of airport pavements should be a methodical process, which includes a thorough review of construction data and usage history, routine site inspections, and pavement sampling and testing. Types of sampling and testing procedures include: direct sampling, nondestructive testing (NDT), ground penetrating radar, and infrared thermography.

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UNIT – III – Terminal Area Design – SAE1605

CHAPTER 10

Planning and Design of the Terminal Area

Introduction

The terminal area is the major interface between the airfield and the rest of the airport. It includes the facilities for passenger and baggage processing, cargo handling, and airport maintenance, operations, and administration activities. The passenger processing system is discussed at length in this chapter. Baggage processing, cargo handling, and apron requirements are also discussed relative to the terminal system.

The Passenger Terminal System

The passenger terminal system is the major connection between the ground access system and the aircraft. The purpose of this system is to provide the interface between the passenger airport access mode, to process the passenger for origination, termination, or continuation of an air transportation trip, and convey the passenger and baggage to and from the aircraft.

Components of the System

The passenger terminal system is composed of three major components. These components and the activities that occur within them are as follows:

1. The access interface where the passenger transfers from the access mode of travel to the passenger processing component. Circulation, parking, and curbside loading and unloading of passengers are the activities that take place within this component.
2. The processing component where the passenger is processed in preparation for starting, ending, or continuation of an air transportation trip. The primary activities that take place within

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this component are ticketing, baggage check-in, baggage claim, seat assignment, federal inspection services, and security.

1. The flight interface where the passenger transfers from the processing component to the aircraft. The activities that occur here include assembly, conveyance to and from the aircraft, and aircraft loading and unloading.

A number of facilities are provided to perform the functions of the passenger terminal system. These facilities are indicated for each of the components identified above.

The Access Interface

This component consists of the terminal curbs, parking facilities, and connecting roadways that enable originating and terminating passengers, visitors, and baggage to enter and exit the terminal. It includes the following facilities:

1. The enplaning and deplaning curb frontage which provide the public with loading and unloading for vehicular access to and from the terminal building
2. The automobile parking facilities providing short-term and long-term parking spaces for passengers and visitors, and facilities for rental cars, public transit, taxis, and limousine services
3. The vehicular roadways providing access to the terminal curbs, parking spaces, and the public street and highway system
4. The designated pedestrian walkways for crossing roads including tunnels, bridges, and automated devices which provide access between the parking facilities and the terminal building
5. The service roads and fire lanes which provide access to various facilities in the terminal and to other airport facilities, such as air freight, fuel truck stands, and maintenance.

The ground access system at an airport is a complex system of roadways, parking facilities, and terminal access curb fronts. This complexity is illustrated in Fig. 10-1 which shows the various ground access system facilities and directional flows at Greater Pittsburgh International Airport.

The Processing System

The terminal is used to process passengers and baggage for the interface with aircraft and the ground transportation modes. It includes the following facilities:

1. The airline ticket counters and offices used for ticket transactions, baggage check-in, flight information, and administrative personnel and facilities

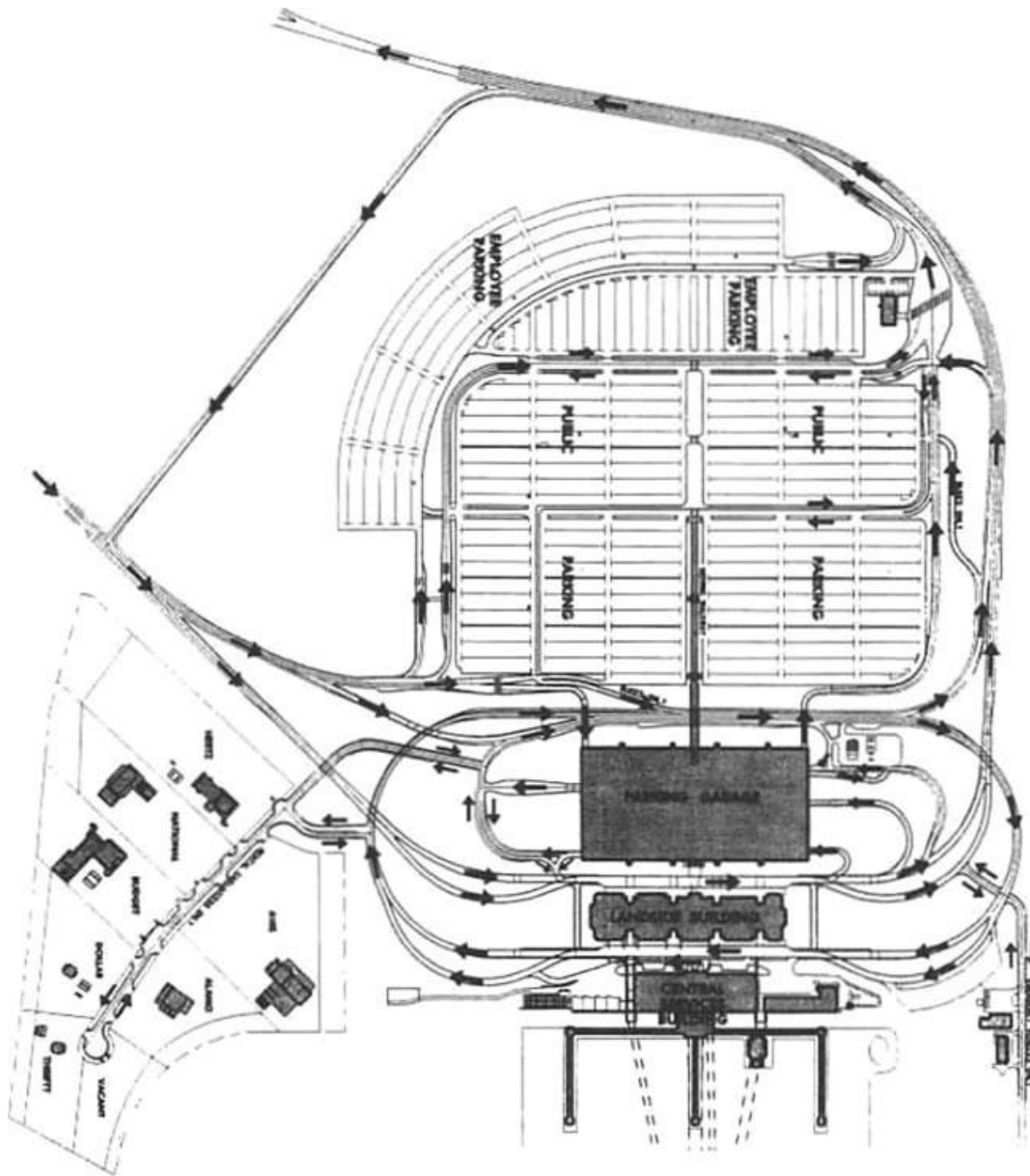


FIGURE 10-1 Ground access system configuration and directional flows for Greater Pittsburgh International Airport (*Tasso Katselas Associates and Michael Baker Jr., Inc. [32]*).

2. The terminal services space which consists of the public and nonpublic areas such as concessions, amenities for passengers and visitors, truck service docks, food preparation areas, and food and miscellaneous storage
3. The lobby for circulation and passenger and visitor waiting
4. Public circulation space for the general circulation of passengers and visitors consisting of such areas as stairways, escalators, elevators, and corridors
5. The outbound baggage space which is a nonpublic area for sorting and processing baggage for departing flights

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6. The intraline and interline baggage space used for processing baggage transferred from one flight to another on the same or different airlines
7. The inbound baggage space which is used for receiving baggage from an arriving flight, and for delivering baggage to be claimed by the arriving passenger
8. Airport administration and service areas used for airport management, operations, and maintenance facilities
9. The federal inspection service facilities which are the areas for processing passengers arriving on international flights, as well as performing agricultural inspections, and security functions

The Flight Interface

The connector joins the terminal to parked aircraft and usually includes the following facilities:

1. The concourse which provides for circulation to the departure lounges and other terminal areas
2. The departure lounge or holdroom which is used for assembling passengers for a flight departure
3. The passenger boarding device used to transport enplaning and deplaning passengers between the aircraft door and the departure lounge or concourse
4. Airline operations space used for airline personnel, equipment, and activities related to the arrival and departure of aircraft
5. Security facilities used for the inspection of passengers and baggage and the control of public access to passenger boarding devices
6. The terminal services area providing amenities to the public and those nonpublic areas required for operations such as building maintenance and utilities

The components of the passenger terminal system together with the specific physical facilities corresponding to them are shown in Fig. 10-2. The relative locations of the various physical facilities in the three level landside building of the midfield terminal complex at Greater Pittsburgh International Airport are shown in Figs. 10-3, 10-4, and 10-5. Figure 10-3 shows the enplaning roadway interface with the departure or check-in level. This level also provides access to the commuter aircraft departure lounge. Figure 10-4 shows the transit level which provides airline baggage makeup space, passenger security processing and access to the automated transit system, the interface between the landside building and the airside building. Figure 10-5 shows the deplaning roadway interface with the arrivals or baggage claim level and contains baggage claim facilities and rental car facilities

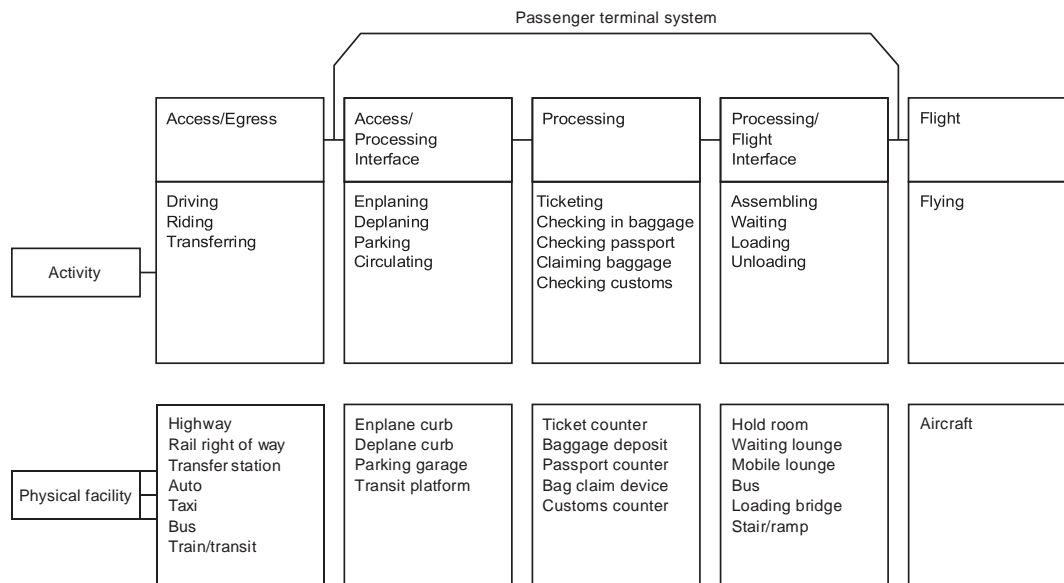


FIGURE 10-2 Components of the passenger terminal system.

as well as a central service building with airport police offices, utility, maintenance, and storage space. The concourse level of the airside building is shown in Fig. 10-6. On this level there is a common core with passenger amenities and four piers providing the departure lounges and boarding devices at the gates providing the interface with aircraft. One of the four piers is for international arrivals and contains the sterile areas for customs and immigration functions required for international passenger processing. The apron level of the airside building is used for airline operations and the lower level provides access to an automated peoplemover transit system.

Design Considerations

In developing criteria for the design of the passenger terminal complex, it is important to realize that there are a number of different factors which enter into a statement of overall design objectives. From these factors general and specific goals are established which set the framework on which design progresses. For example, in designing modifications to the apron and terminal complex at Geneva Intercontinental Airport, the general design objectives included [25]

1. Development and sizing to accomplish the stated mission of the airport within the parameters defined in the master plan
2. Capability to meet the demands for the medium- and long-run time frames
3. Functional, practical, and financial feasibility
4. Maximize the use of existing facilities

Airport Design

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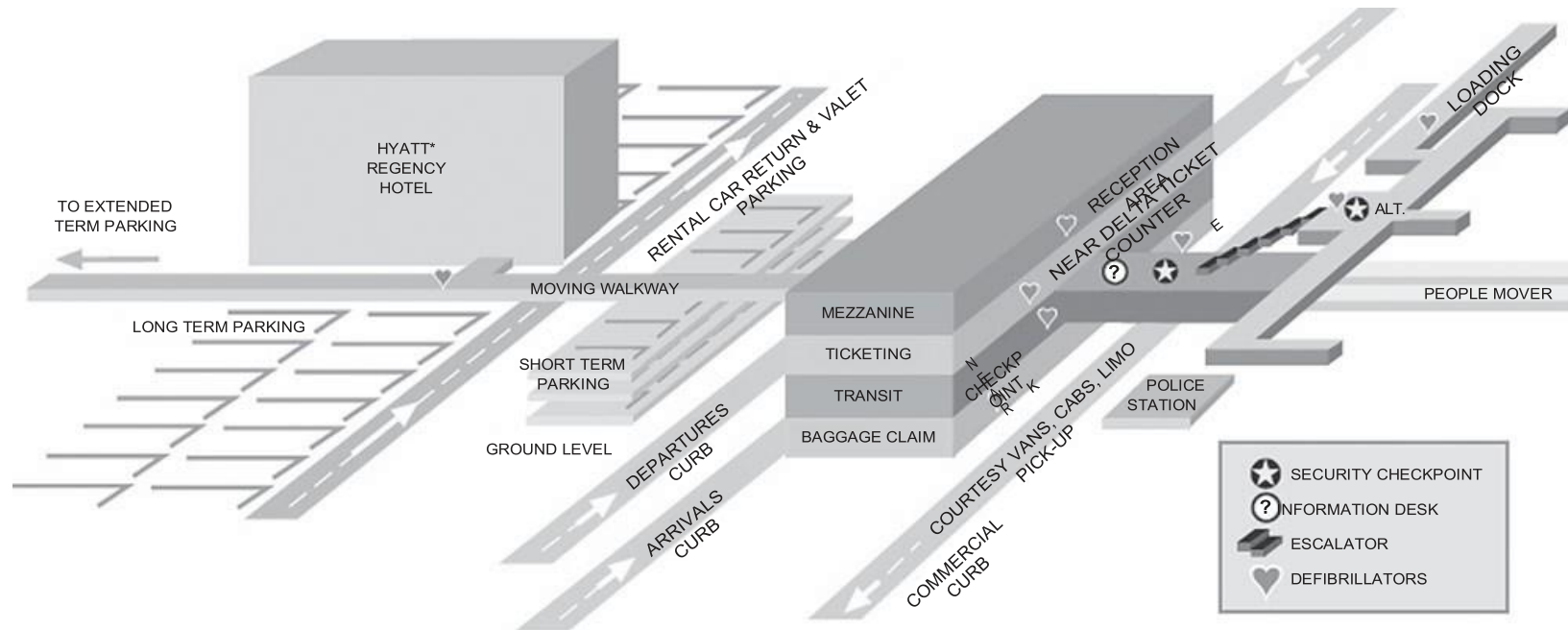


FIGURE 10-3 Landside building enplaning level at Greater Pittsburgh International Airport ([http:// www.pitairport.com](http://www.pitairport.com)).

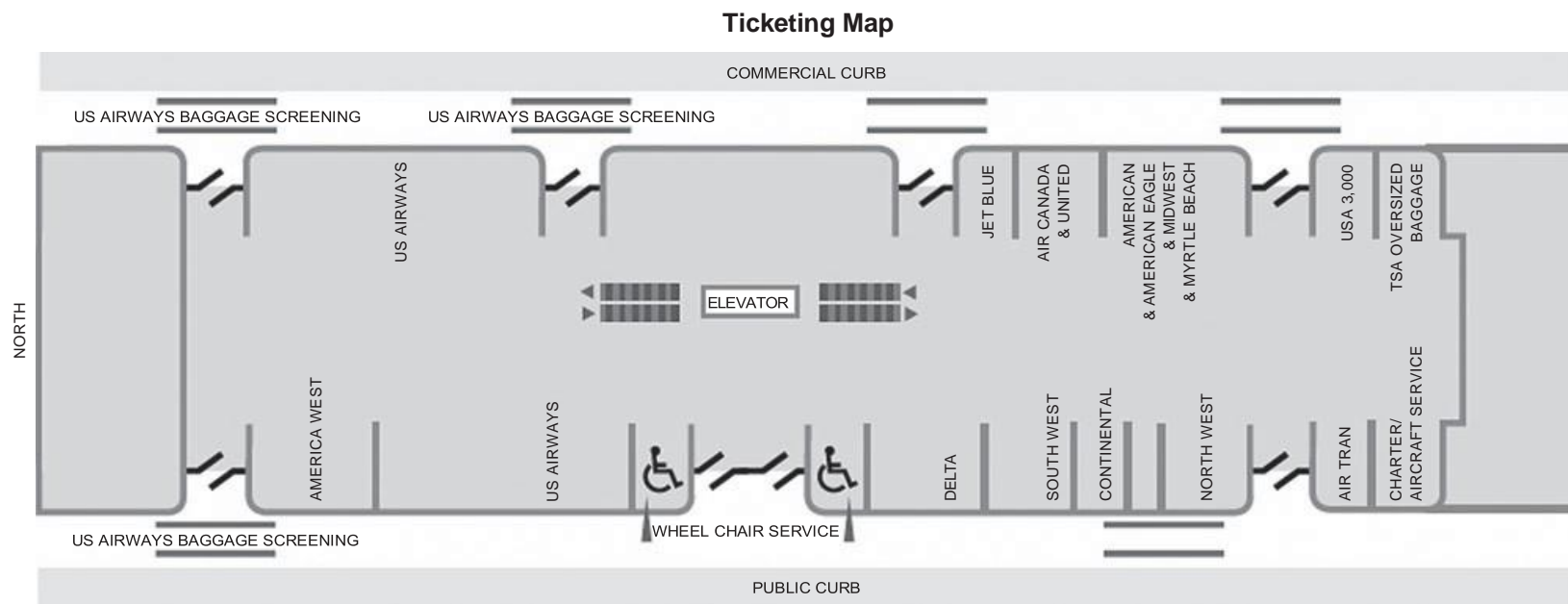


FIGURE 10-4 Landside building transit level at Greater Pittsburgh International Airport ([http:// www.pitairport.com](http://www.pitairport.com)).

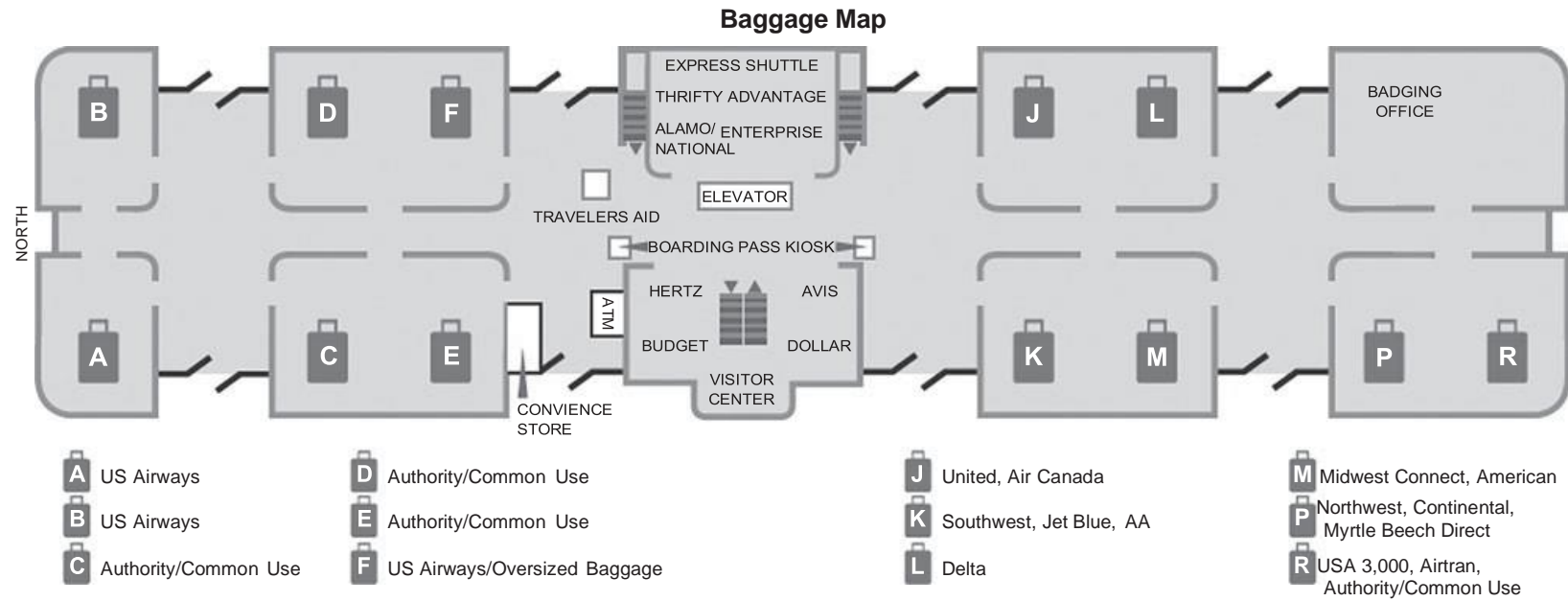


FIGURE 10-5 Landside building deplaning level at Greater Pittsburgh International Airport ([http:// www.pitairport.com](http://www.pitairport.com)).

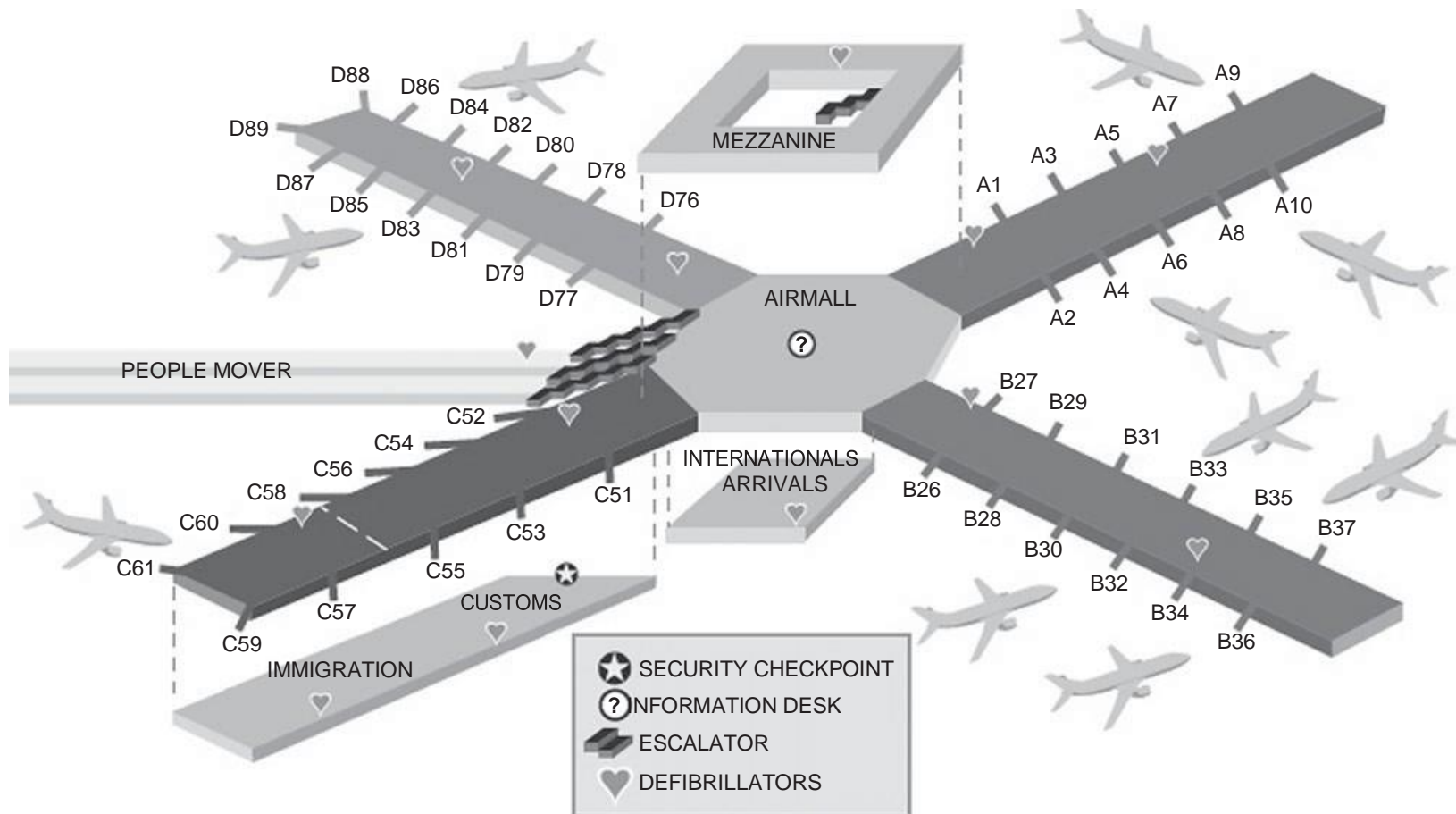


FIGURE 10-6 Airside building concourse level at Greater Pittsburgh International Airport (<http://www.pitairport.com>).

5. Achievement of a balanced flow between access, terminal, and airfield facilities during the peak hour
6. Consideration of environmental sensitivity
7. Maintenance of the flexibility to meet future requirements beyond the current planning horizon
8. Capability to anticipate and implement significant improvements in aviation technology

Specific design objectives were derived from these general objectives which included the needs of the various categories of airport users. These included

1. Passenger objectives
 - a. Responsiveness to the needs of the people relative to convenience, comfort, and personal requirements
 - b. Provision of effective passenger access orientation through concise, comprehensive directional graphics
 - c. Separation of enplaning and deplaning roadways and curb fronts to ensure maximum operational efficiency
 - d. Provision of convenient access to public and employee parking facilities, rental car areas, ancillary facilities, and other on-site facilities
2. Airline objectives
 - a. Accommodation of existing and future aircraft fleets with maximum operational efficiency
 - b. Provision of direct and efficient means of passenger and baggage flow for all passengers, including domestic and international originating, terminating, and transfer passengers
 - c. Provision for economic, efficient, and effective security
 - d. Provision of facilities which will embrace the latest energy conservation measures
3. Airport management objectives
 - a. Maintenance of the existing terminal operation, access system, runway system, and ancillary facilities during all stages of construction
 - b. Provision of facilities which generate maximum revenues from concessionaires and other sources
 - c. Provision of facilities which minimize maintenance and operating expenses
4. Community objectives
 - a. Render a unique and appropriate expression and impression of the community
 - b. Provision of harmony with the existing architectural elements of the total terminal complex
 - c. Coordination with the existing and planned off-airport highway system

The designer should consider the combination of these types of objectives in developing specific design criteria for the passenger terminal complex. These criteria should be used as performance measures for the evaluation of design alternatives. In order to generate performance measures, a detailed design should be provided. The analyst can then proceed to calculate the various performance measures using a number of analytical techniques. Some of these techniques are discussed later in this chapter.

Terminal Demand Parameters

The determination of space requirements at passenger terminals is strongly influenced by the quality of service desired by the various airport users and the community. A review of passenger terminals in relation to passenger volumes at existing airports shows a wide range in the configuration and the amount of area provided per passenger. However, some guidelines for the determination of space requirements can be defined. The purpose of these guidelines is to give general orders of magnitude for values that are subject to change depending on the requirements of specific designs.

The following steps should be followed in determining terminal facility space requirements.

Identify Access Modes and Modal Splits

Vehicle volumes are normally derived from projections of passenger and aircraft forecasts. These volumes critically impact the design of highway access facilities, on-airport roadway and circulation systems, curb frontage requirements for private automobiles, buses, limousines, taxis, and rental cars, and parking. Surveys are normally conducted to determine the access modes of passengers and vehicle occupancy rates [26]. In the absence of such surveys, secondary sources may be investigated to ascertain the access characteristics of passengers in similar airport environments [24, 55, 60].

The most important parameters to be obtained include the typical peak hour volumes of vehicles entering and leaving the airport on the design day, the access facilities used and the duration of use, including parking and curb front. Care should be exercised to include employees and visitors as well as passengers in these access studies, and to correlate the peaking characteristics and access modes of each group of airport traveler.

Identify Passenger Volumes and Types

Passenger volumes can be obtained from forecasts normally done in conjunction with airport planning studies. Two measures of volume are used. The first is annual passenger volume, which is used for preliminary sizing of the terminal building. The second is a more detailed hourly volume. It is customary to use typical peak hour passengers as the hourly design volume for passenger terminal design. This parameter

is a design index and is usually in the range of 0.03 to 0.05 percent of the annual passenger volume but it is significantly affected by the scheduling practices and fleet mix of the airlines.

The identification of passenger types is necessary because different types of passengers place different demands on the various components. Passenger types are usually broadly classified into domestic and international passengers and then further grouped into originating, terminating, connecting or transfer, through, enplaning and deplaning passengers. These various groupings of passengers are made on the basis of the facilities within the terminal which are normally used by each type of passenger. Airports which are used as airline hubs and have a high proportion of connecting passengers require considerably less ground access and landside facilities than airports with a high proportion of originating and terminating passengers. Historical data and forecasts regarding the proportions of the total volumes that are made up by each of the different types of passengers are useful in obtaining estimates of the parameters needed for the design of the various facilities [4, 58].

Identify Access and Passenger Component Demand

This is done by matching the passenger and vehicle types with facilities in the terminal area. The use of tabulations such as the one shown in Table 10-1 is quite helpful. This table shows which passengers are using which facility. By indicating the volume of each type of passenger in the rows corresponding to the facilities, it is possible to generate the total load on each facility. This is done by taking the row sums of the volumes entered.

Facility Classification

The airport terminal facility may be classified by its principal characteristics relative to its functional role. In general, airports are classified as originating-terminating, transfer, or through airports. The facilities required are considerably different in magnitude and configuration for each.

An originating-terminating airport processes a high level of passengers which are beginning or ending the air transportation trip at the airport. At such airports these passengers may be in the order of 70 to 90 percent of the total passengers. These airports can have a relatively long aircraft ground time for long haul international flights but also may have relatively short ground times for domestic operations and operations by low-cost air carriers. In either case, the main flow of passengers is between the aircraft and the ground transportation system and have relatively high requirements for curb frontage, ticketing and baggage claim facilities, and parking. Typical data indicate that the hourly movements of aircraft per gate at such airports can range from on the order of 1.0 to nearly 3.0 operations per hour per gate.

Facility j	Passenger Type i , Arriving			Passenger Type i , Departing			Total Volume V
	Domestic, No Bags, Auto Driver*	Domestic, with Bags, Auto Passenger†	International, with Bags, Auto Passenger	Domestic with Bags, Auto Passenger	Domestic, No Bags, Auto Driver	International, with Bags, Auto Driver	
Curb, arrivals	–	V_{ij}^\ddagger	V_{ij}	–	–	–	
Curb, departures	–	–	–	V_{ij}	–	V_{ij}	
Domestic lobby	–	V_{ij}	–	V_{ij}	V_{ij}	–	
International lobby	–	–	–	–	–	V_{ij}	
Ticketing counter	–	–	–	V_{ij}	–	V_{ij}	
Assembly	–	–	–	V_{ij}	–	V_{ij}	
Baggage check-in	–	–	–	V_{ij}	V_{ij}	V_{ij}	
Security control	–	–	–	V_{ij}	V_{ij}	V_{ij}	
Customs, health	–	–	V_{ij}	–	–	–	
Immigration	–	–	V_{ij}	–	–	V_{ij}	
Baggage claim	–	V_{ij}	V_{ij}	–	–	–	

*Auto driver = passenger driving a car to and from airport.

†Auto passenger = passenger driven to and from airport.

‡ V_{ij} = design volume of passenger type i using facility type j .

TABLE 10-1 Determination of Demand for Various Types of Passenger Facilities

A transfer or connecting airport, on the other hand, has a high percentage of its total passengers connecting between arriving and departing flights. Today many airports in the United States are connecting airports particularly those that are airline hubs. These airports need greater concourse facilities for the processing of connecting passengers and less ground access facility development. Airline ticketing positions and baggage claim facilities are usually less than with originating airports (on a size per passenger basis). However, intraline and interline baggage facilities are usually greater. Care must be exercised in the planning of such airports to locate the gate positions of airlines exchanging passengers in close proximity to each other to minimize central terminal flows and connecting times. Data indicate that such airports demonstrate aircraft activity at the rate of 1.3 to 1.5 aircraft per gate per hour in peak periods.

The through airport combines a high percentage of originating passengers with a low percentage of originating flights. A high percentage of the passengers remain on the aircraft at such points. Aircraft ground times are minimal, averaging between 1.6 and 2.0 hourly movements per gate in peak periods. Departure lounge space, curb frontage, ticketing, security, and baggage facilities are less than at originating airports.

Overall Space Approximations

It is possible to estimate order of magnitude ranges for the overall size of a terminal facility prior to performing more detailed calculations for particular space needs. These estimates allow the planner to broadly define the scope of a project based upon information which summarizes the space provided of other existing facilities.

The FAA has indicated that gross terminal area space requirements of between 0.08 and 0.12 ft² per annual enplaned passenger are reasonable. Another estimate is obtained by applying a ratio of 150 ft² per design hour passenger [43]. Estimates of the level of peak hour passengers, peak hour aircraft operations, and gate positions are also obtained based upon the level of annual enplanements using relationships such as those shown in Fig. 10-7. Others have provided estimating guidelines for total terminal space as shown, for example, in Fig. 10-8 [43].

Approximations of the allocation of space among the various purposes in a terminal building are also useful for preliminary planning. The FAA indicates that approximately 55 percent of terminal space is rentable and 45 percent is non-rentable [49]. An approximate breakdown of these space allocations typically is 35 to 40 percent for airline operations, 15 to 25 percent for concessions and airport administration, 25 to 35 percent for public space, and 10 to 15 percent for utilities, shops, tunnels, and stairways. A final determination of the actual space allocations is obtained through detailed analyses of the performance of the elements of the system as the design process

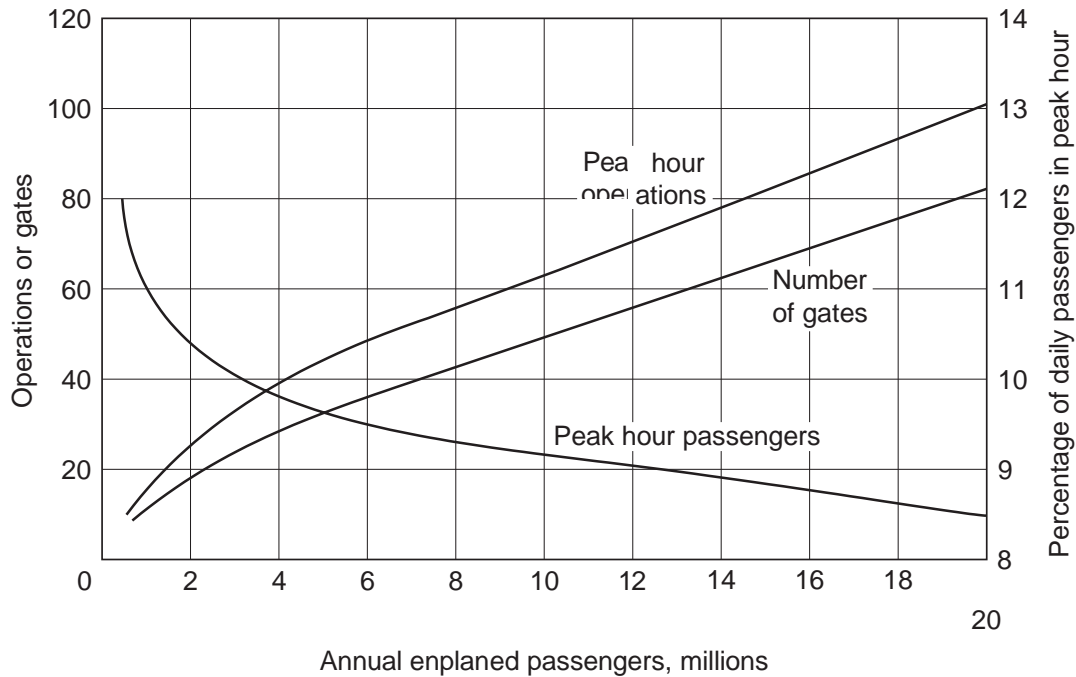


FIGURE 10-7 Estimated peak hour passenger, operations, and gate requirements for intermediate range planning (*Federal Aviation Administration [43]*).

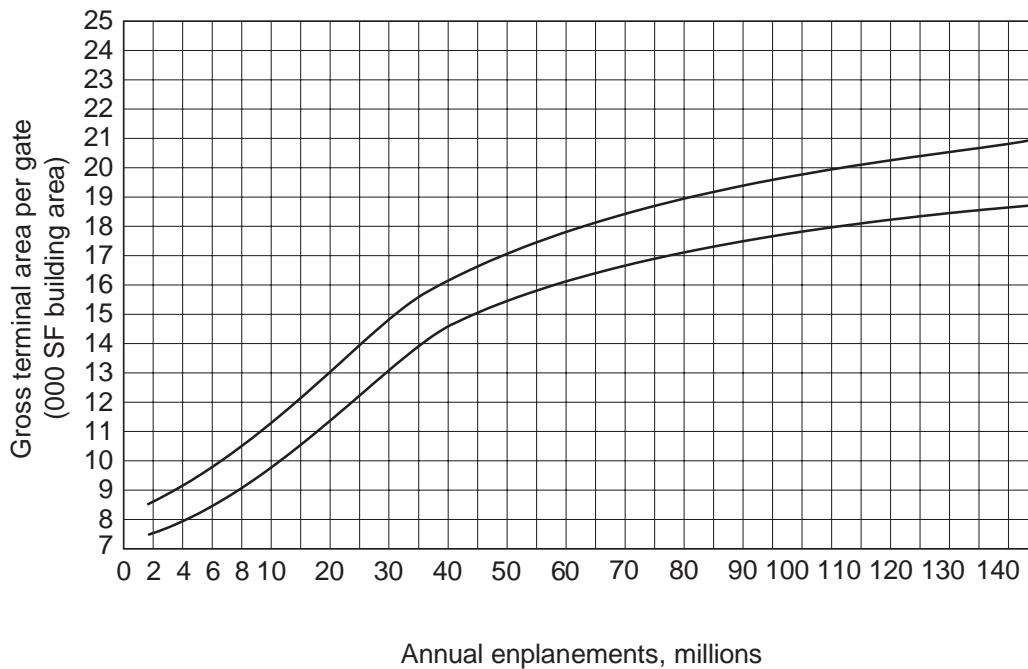


FIGURE 10-8 Gross terminal area estimates for intermediate range planning (*Federal Aviation Administration [49]*).

proceeds from space programming through each of the subsequent phases in the process.

Level of Service Criteria

Considerable research and discussion has taken place in the profession relative to the adoption of level of service standards and associated criteria to evaluate the level of service afforded in the design of

landside processing systems. Although it is relatively simple to develop relationships between aircraft delay on the airside and its economic consequences, such relationships are difficult to either define or develop on the airport landside. Much of the difficulty is related to the fact that the various constituent groups associated with airports view quality of service or level of service from different perspectives [37]. Airlines are concerned with such factors as on-time schedules, the allocation of personnel, airport operating costs, and profitability. Passengers are concerned with the completion of an air transportation trip at a reasonable cost, with minimum delay and maximum convenience, without being subjected to excessive levels of congestion. The airport operator is interested in providing a modern airport facility which meets airline and passenger objectives in harmony with the expectations of the community in which the airport is located. Given the number of possible measures of service quality and the differences in airports throughout the country, it is very difficult to adopt level of service criteria on a broad scale.

Many have examined level of service criteria for airports and attempted to define level of service standards [9, 11, 14, 16, 21, 29, 31, 36, 37, 41, 45, 56]. In general, the level of service measures commonly associated with the airport landside system include measures of congestion within the terminal building and the ground access system, passenger delays and waiting line lengths at the various facilities in the terminal building, passenger walking distances, and total passenger processing time. Most of these parameters can be evaluated in a terminal design with the aid of mathematical modeling. However, the various measures of level of service from the perspective of airport users must be balanced in reaching some acceptable solution to the design problem.

As an illustration of the application of a level of service standard, let us say that an airline may desire to limit the percentage of its passengers which must spend more than some increment of time at an airport check-in facility. One could develop a model which computes the percentage of passengers at the check-in facility for various durations of time when the number of check-in counters operated is varied for some peak hour passenger demand [35]. As an example, the results could be shown graphically as is done in Fig. 10-9. From this illustration, if the criteria were to limit the percentage of passengers spending more than 5 min at this facility to 10 percent, then it would have been necessary for the airline to operate nine check-in counters at the airport during the peak hour.

Such formulations can be examined for the various facilities within the airport landside to obtain quantitative measures of component and system performance. These are discussed later in this chapter.

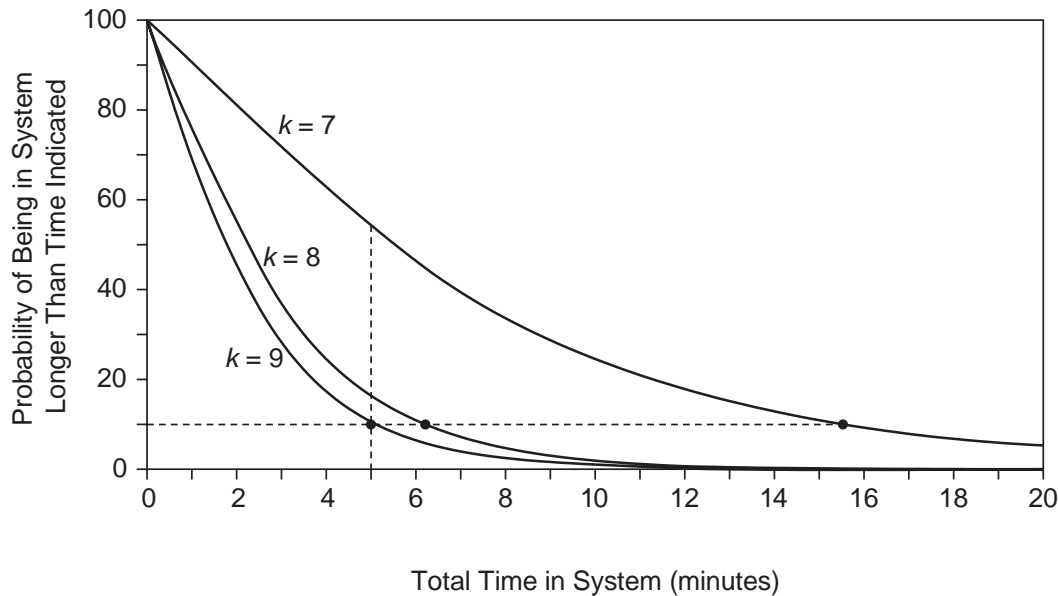


FIGURE 10-9 Impact of the number of check-in counters (k) on passenger time at a check-in facility.

The Terminal Planning Process

The evolution and development of a terminal design is performed in a series of integrated steps. These may be identified as programming, concept development, schematic design, and design development. The terminal facilities are developed in conformity with the planned development of the airside facilities considering the most effective use of the airport site, the potential for physical expansion and operational flexibility, integration with the ground access system, and compatibility with existing and planned land uses near the airport. The planning process explicitly examines physical and operational aspects of the system.

The programming phase defines the objectives and project scope including the rationale for the initiation of the study. It includes a space requirements program, tentative implementation schedules, estimates of the anticipated level of capital investment as well as operating, maintenance, and administrative costs. In concept development, studies are undertaken to identify the overall arrangement of building components, functional relationships, and the characteristics of the terminal building. Schematic design translates the concept and functional relationships into plan drawings which identify the overall size, shape, and location of spaces required for each function. Detailed budget estimates are prepared in schematic design so that comparisons may be made between the space requirements and costs. In design development, the size and character of the entire project is determined and detailed plans of the specific design and allocation of space within the complex are prepared. This phase forms the

basis for the preparation of construction documents, bidding, construction, and final project implementation [50].

In the programming and concept development phases of a terminal design project, the following evaluation criteria are typically used to weigh alternatives:

1. Ability to handle expected demand
2. Compatibility with expected aircraft types
3. Flexibility for growth and response to technology changes
4. Compatibility with the total airport master plan
5. Compatibility with on-airport and adjacent land uses
6. Simplicity of passenger orientation and processing
7. Analyses of aircraft maneuvering routes and potential conflicts on the taxiway system and in the apron area
8. Potential for aircraft, passenger, and vehicle delay
9. Financial and economic feasibility

In the schematic and design development phases, more specific design criteria are examined such as:

1. The processing cost per passenger
2. Walking distance for various types of passengers
3. Passenger delays in processing
4. Occupancy levels and degree of congestion
5. Aircraft maneuvering delays and costs
6. Aircraft fuel consumption in maneuvering on the airport between runways and terminals
7. Construction costs
8. Administrative, operating, and maintenance costs
9. Potential revenue sources and the expected level of revenues from each source

Space Programming

The space programming phase of terminal planning seeks to establish gross size requirements for the terminal facilities without establishing specific locations for the individual components. The nature of the processing components is such, however, that approximate locations are indicated for new and existing terminal facilities due to the sequential nature of the processing system. This section provides guidance concerning the spatial requirements to adequately accommodate the several functions carried out within the various areas of the airport terminal.

The Access Interface System

The curb element is the interface between the terminal building and the ground transportation system. A survey of the airport users will establish the number of passengers using each of the available ground transportation modes such as private automobile, taxi, limousine, courtesy car, public bus, rail, or rapid transit. Ratios may be established for both the passenger and vehicle modal choice for airport access.

Terminal Curb

The length of curb required for loading and unloading of passengers and baggage is determined by the type and volume of ground vehicle traffic anticipated in the peak period on the design day. Airports with relatively low passenger levels may be able to accommodate both enplaning and deplaning passengers from one curb front. Airports with higher passenger levels may find it desirable to physically separate the enplaning from the deplaning passengers, horizontally, if space permits, or vertically if space is limited. There is a tendency at large airports to also separate commercial vehicle traffic from private vehicle traffic.

The determination of the amount of curb space which will be required is related to airport policies relative to the assignment of priorities to the use of curb front and the provision of staging areas for taxis, buses, and other public transport vehicles. The parameters required for a preliminary analysis of curb front needs are the number and types of vehicles at the curb, the vehicle length, and the various occupancy times of different types of vehicles at the curb front for arriving and departing passengers.

Normally, a slot for a private automobile is considered to be about 25 ft, whereas for taxis 20 ft, limousines 30 ft, and transit buses 50 ft are used. Reported dwell times for private automobiles range from 1 to 2 min at the enplaning curb and from 2 to 4 min at the deplaning curb. Taxi dwell times lie closer to the lower range of these values, whereas limousines and buses may be at the curb anywhere from 5 to 15 min. These dwell times are highly influenced by the degree of traffic regulation and enforcement in the vicinity of the curb, and should be verified in specific studies. Normally a wide lane, in the order of 18 to 20 ft, is provided to accommodate direct curb access, maneuvering, and standing vehicles. This usually indicates a minimum of one and preferably two additional lanes in the vicinity of terminal entrances and exits to provide adequate capacity for through traffic. Rules of thumb which may be applied to determine curb front needs indicate that the full length of the curb adjacent to the terminal plus about 30 percent of the maneuvering lane may be considered as the available curb front. Therefore, a 100-ft curb may be considered to provide 130 ft of curb front in 1 h or 7800 foot-minutes of vehicle occupancy. If 120 automobiles per hour demand curb space for an

average dwell time of 2 min, then 6000 foot-minutes of curb front is required, or the peak hour must provide a curb length of 100 ft. Other methods for approximating curb frontage have been reported in the literature [12, 43, 52, 58, 60].

Roadway Elements

The determination of the vehicular demand for the various on and off airport roadways is essential to ensure that adequate service levels are provided airport users. The main components of the highway system providing for access to airports from population and industrial centers is normally within the jurisdiction of federal, state, and local ground transportation agencies. However, coordination in area-wide planning efforts is essential so that the traffic generation potential of airports may be included within the parameters necessary for the proper planning of regional transportation systems. Guidance on the level and peaking characteristics of airport destined traffic may be found in the literature [55, 60].

The provision of adequate feeder facilities from the regional transport network to the airport is largely within the jurisdiction of the airport operator or owner. Vehicle volumes and peaking characteristics are usually determined by correlating modal preference and occupancy rates with flight schedules. Normally roadway facilities are designed for the peak hour traffic on the design day with adequate provision for the splitting and recirculation of traffic within the various areas of the airport property. The main roadway elements which must be considered are the feeder roads into the terminal area, the enplaning and deplaning roadways, and recirculation roadways.

The *Highway Capacity Manual* [33] provides criteria for level of service design and quantitative methods for determining the volumes which can be accommodated by various types of roadway sections. Unfortunately little guidance is available for the level of service design of airport roadways. For preliminary planning, however, it is reasonable to assume that feeder roads on the airport property provide acceptable service when they are designed to accommodate from 1200 to 1600 vehicles per hour per lane. Roadways providing access to the enplaning and deplaning terminal systems provide adequate service when they are designed to accommodate from 900 to 1000 vehicles per hour per lane. Terminal frontage roads and recirculation roads, however, provide adequate service when they are designed to accommodate from 600 to 900 vehicles per hour per lane. It is recommended that for preliminary planning purposes the above ranges of values be used to establish bounds on the sizes of these facilities for a demand-capacity analysis. In schematic design, analysis of the flow characteristics of individual sections of the roadway elements will yield final design parameters.

Parking

Most large airports provide separate parking facilities for passengers and visitors, employees, and rental car storage. In smaller airports these facilities may be combined in one physical location. Passenger and visitor parking are often segregated into short-term, long-term, and remote parking facilities. Those parking facilities most convenient to the terminal are designated as short term and a premium rate is charged for their use. Long-term parking is usually near the main terminal complex, but not as convenient as short term, and rates are usually discounted for long-term users. Remote parking, on the other hand, is usually quite distant from the terminal complex and provisions are normally made for courtesy vehicle transportation between these areas and the main terminal complex. The rates in these facilities are usually the most economical.

Short-term parkers are normally classified as those which park for 3 h or less and these may account for about 80 percent of the parkers at an airport. However, these short-term parkers account for only 15 to 20 percent of the accumulation of vehicles in the parking facility [43]. Preliminary planning estimates of the number of parking spaces required at an airport may be obtained from Fig. 10-10. The range of public parking spaces provided at existing airports varies from 1000 to 3000 per million originating passengers.

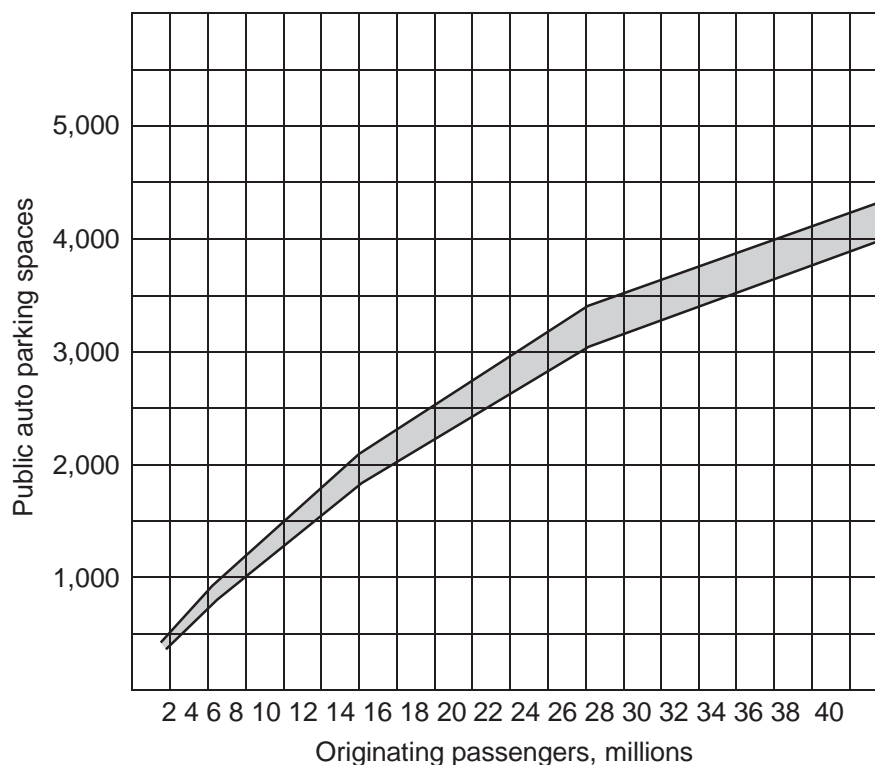


FIGURE 10-10 Public automobile parking space requirements (*Federal Aviation Administration [49]*).

It is recommended, however, that the range for preliminary planning be established between 1000 and 1400 parking spaces per million originating passengers.

More refined estimates of the total amount of parking required and the breakdown of short- and long-term space is obtained from analyses performed in the schematic design phase of the terminal planning process.

Entrances to parking facilities through ticket spitting devices are very common at airports. It has been observed that these devices can process anywhere from 400 to 650 vehicles per hour depending upon the degree of automation used as well as the continuity of the demand flow. It is recommended that the number of entrances be estimated on the basis of 500 vehicles per hour per device in preliminary planning. Parking revenue collection points at parking facility exits process from 150 to 200 vehicles per hour per position.

In parking garages, the capacity of ramps leading from one level to another is important during peak periods when considerable searching for an available space may occur, or vehicles may be directed immediately to a particular level. One-way straight ramps can accommodate about 750 vehicles per hour. However, a reduction in the order of 20 percent should be considered when two-way ramps are utilized. Circular or helical ramps, often used for egress from parking facilities, accommodate about 600 vehicles per hour in one direction.

The precise volume of vehicles which may be accommodated by a particular design will depend to a large extent on the geometric characteristics of the design, continuity of flow, information systems installed, and characteristics of the vehicles and users of the particular facility [22, 40, 60]. Most often, some type of analytical or simulation model is used in the schematic design phase of the project to test a preliminary design.

The Passenger Processing System

The passenger processing system consists of those facilities necessary for the handling of passengers and their baggage prior to and after a flight. It is the element which links the ground access system to the air transportation system. The terminal curbs provide the interface on the ground access side of the system, and the aircraft gates devices provide the interface on the airside of the system. In determining the particular needs of a specific component in this system, knowledge of the types of passengers and the extent of visitors impacting on each component is necessary.

Entryways and Foyers

Entryways and foyers are located along the curb element and serve as weather buffers for passengers entering and leaving the terminal building. The size of an entryway or foyer depends upon its intended

usage. As entrances and exits may be relatively small, sheltered public waiting areas should be provided and sized to meet local needs. Designs must accommodate the physically challenged. These facilities are sized to process both passengers and visitors during the peak hour. Although the time of the enplaning and deplaning peaks may be different, it is likely that the deplaning peak will occur over a shorter time duration than the enplaning peak. It is often useful to subject a preliminary design proposal not only to an average peak hour demand but also a peak 20 or 30 min demand, particularly for the deplaning elements of the system. Preliminary design processing rates for automated doors in the vicinity of the enplaning and deplaning curb front can be taken as from 8 to 10 persons per minute per unit. These values may be reduced by 50 percent if the doors are not automated.

Terminal Lobby Area

The functions of significance to an air traveler performed in a terminal lobby are passenger ticketing, passenger and visitor waiting, and baggage check-in and claiming. Airports with less than 100,000 annual enplanements frequently carry out these functions in a single lobby. More active airports usually have separate lobbies for each function. The size of the lobby space depends on whether ticketing and baggage claim lobbies are separate, if passenger and visitor waiting areas are to be provided, and the density of congestion acceptable. In general, the lobby area should provide for passenger queuing, circulation, and waiting. Waiting lobby areas are designed to seat from 15 to 25 percent of the design hour enplaning passengers and visitors if departure lounges are provided for all gates, and from 60 to 70 percent if they are not provided [43, 50]. Usually about 20 ft² per person is provided for seating and circulation.

Airline Check-In Counter and Ticket Office

The airline check-in counter and ticket office is the area at the airport where the airline and passenger make final ticket transactions and check-in baggage for a flight. This includes the airline check-in counter, airline ticket agent service area, outbound baggage handling device, and support office area for the airline ticket agents. There are three types of ticketing and baggage check-in facilities, the linear, pass through, and island types. Each of these facilities is shown in a typical arrangement in Fig. 10-11.

The check-in transaction takes place at the check-in counter, which is a stand-up desk. To the left and right of the ticket counter position, a low shelf is provided to deposit, check-in, tag, and weigh baggage, if necessary, for the flight. Subsequently, the baggage is passed back by the agent to an outbound baggage conveyance device located near the counter for security screening, sorting, and loading on the aircraft.

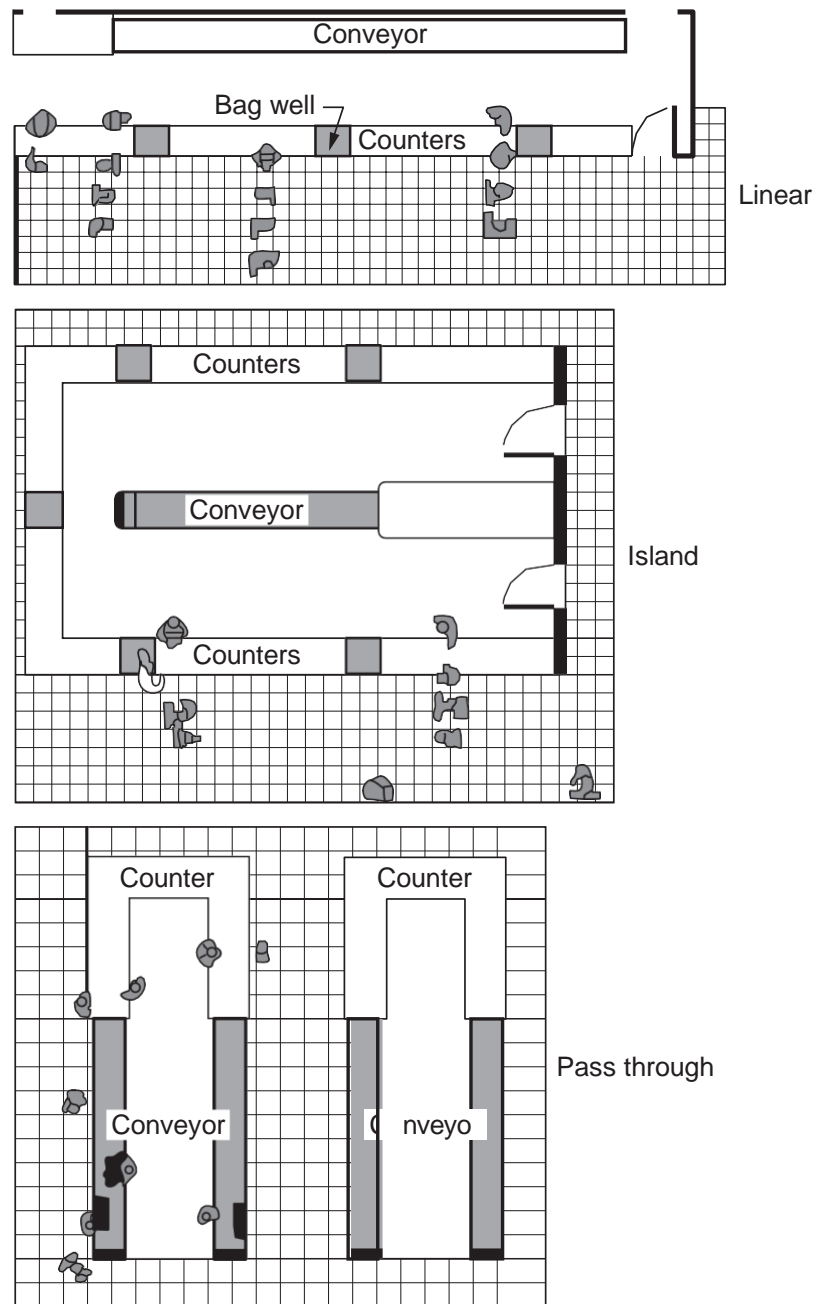


FIGURE 10-11 Typical check-in counter configurations (*Federal Aviation Administration [49]*).

The total number of check-in counter positions required is a function of the level of peak hour originating passengers, the types of facilities provided (i.e., multipurpose, express baggage check-in, and ticketing only) and the queues and delays acceptable to the airlines. In some markets, a considerable number of passengers may be preticketed and a higher percentage of express check-in positions may be warranted either within the terminal building or at the

curb front. This is particularly true at airports serving tourist areas. A reasonable estimate of the number of positions required is gained by assuming the peak loading on the facilities is about 10 percent of the peak hour originating passengers and that a maximum queue length of five passengers per position is a desirable design goal. If this is the case, then 3000 peak hour originating passengers, translates to 300 peak loading passengers, which requires 60 ticketing positions. The sizing of the counter length depends on the mix of position types, but for preliminary estimating purposes a counter length from 10 to 15 ft for two positions is reasonable. Therefore, in this case the total linear footage of counter space would range from 300 to 450 ft. If a queuing depth of 3 ft per passenger is provided, then a minimum queuing depth of 15 ft is required. The counter itself requires a 10-ft depth and a circulation aisle of from 20 to 35 ft is appropriate. Therefore, the area devoted to this function ranges from about 13,500 to 27,000 ft². If single row lobby seating is provided in the area, this increases the area to between 16,500 to 30,000 ft². Approximations gained from other techniques yield similar results [12, 50].

The above calculations are useful for linear type counters where queues form in lines at each position. For corral type queuing similar results are obtained, the principal advantage of this type of queuing being less restriction to circulation in the ticket lobby area. The pass-through and island type counters result in a similar number of positions and counter length but in different geometric arrangements. The circulation and queuing areas may be modified as shown in Fig. 10-11. Normally, in pass-through ticketing and baggage check-in facilities, queuing is along the length of the counter whereas in island types queuing is in lines at the various ticketing positions.

The final determination of the number and mix of check-in positions is made through consultation with the various airlines to be served and through the use of analytical or simulation models [15, 58]. Figure 10-12 shows the terminal lobby area and airline check-in counters at the Pittsburgh International Airport.

The airline ticket office (ATO) support area may be composed of smaller areas for the operations and functions of accounting and safe-keeping of tickets, receipts, and manifests; communications and information display equipment; and personnel areas for rest, personal grooming, and training. On the wall behind the ticket agents are posted information displays for the latest airline information on arriving and departing flights. Typical estimates of ATO space requirements may be obtained by taking the linear footage of counter and multiplying this by a depth of from 20 to 25 ft for this area. This yields a value of from about 6000 to 11,000 ft² for 3000 peak hour originating passengers. As the level of peak hour passengers increases various economies of scale may be gained in the use of such facilities. For example, in moving from 3000 to 6000 peak hour



FIGURE 10-12 Terminal lobby and airline check-in counters at the Pittsburgh International Airport.

originations, the increase in ATO space may be only about 30 percent. Other estimating procedures yield similar results [43, 50]. Again, the final determination of the space requirements is obtained through consultation with the various airlines using the facility.

The proliferation of self-service check-in kiosks at airports, along with the increasing ability for passengers to check-in for flights using the Internet or mobile device has created new challenges for check-in area planning and design. It is clear that there will always be a need for traditional check-in stations staffed by airline personnel. However, the number of staffed stations is becoming smaller compared to the number of installed self-service kiosks.

The implementation of self-service kiosks that have the capability of providing check-in service to more than one air-carrier, known as CUSS—"common use self-service" kiosks are helping to redefine the check-in spacing needs in airport terminals. These CUSS systems may be placed throughout the terminal entry lobby for all passengers to use, regardless of their individual airlines.

Passenger Security Screening

Security screening of passengers is an extremely important function in an airport terminal. The security screening area will include a checkpoint for identification inspection, walk-through metal detectors, and x-ray equipment for carry-on baggage inspection. The location and size of the screening area will be dictated primarily by passenger volume with consideration to issues of queuing, physical search, and

passengers requiring additional processing. The equipment, techniques, and procedures may vary with location and are subject to change at any time. A few years ago, greeters, wellwishers, and visitors could be processed at the security area and proceed to the gate areas, but today only ticketed passengers are permitted in the sterile area. In the United States, the Transportation Security Administration (TSA) has prepared extensive guidelines and space planning and analysis tools [57] to assist planners in the design of the security screening area and these are introduced in Chapter 11.

Departure Lounges

The departure lounge serves as an assembly area for passengers waiting to board a particular flight and as the exit passageway for deplaning passengers. It is generally sized to accommodate the number of boarding passengers expected to be in the lounge 15 min prior to scheduled departure time, assuming this is the point in time when aircraft boarding begins. A conservative estimate of the percentage of passengers in the lounge at this time is 90 percent of the boarding passengers. The space should accommodate space for airline processing and information, passenger queues, seating for enplaning passengers, although all need not be seated, and an exitway for deplaning passengers.

Processing queues should not extend into the corridor to the extent that circulation is impaired. Lounge depths of 25 to 30 ft are considered reasonable for holding boarding passengers. Space for a departure lounge is proportioned on the basis of from 10 to 15 ft² per boarding passenger. Therefore, if a departure lounge is to accommodate 100 boarding passengers for a flight, its area should range from 1000 to 1500 ft². To a greater and greater extent common departure lounge areas are being utilized and these are sized based upon the total peak hour boarding passengers for the gates being served by the common lounge. Since it is likely that boarding for these aircraft will occur at different times in the peak hour, the total space required for separate lounges may be reduced by 20 to 30 percent for common lounges.

The corridor provided for deplaning passengers should be about 10 ft in width. The airline processing area should provide for at least two positions for narrow-bodied aircraft and up to four positions for wide-bodied aircraft to minimize queue lengths extending into the corridor. Processing rates range from one to two passengers per minute at these positions and peak arrival rates range from 10 to 15 percent of the boarding passengers. Therefore, queue depths of about 10 ft are reasonable values for preliminary design for individual departure lounges. For common lounges, the position of the processing area should be such as not to interfere with the circulation of passengers in the vicinity of the entrances and exits from the lounge. Typically these positions are located in the center of a satellite facility or at the end of the corridor

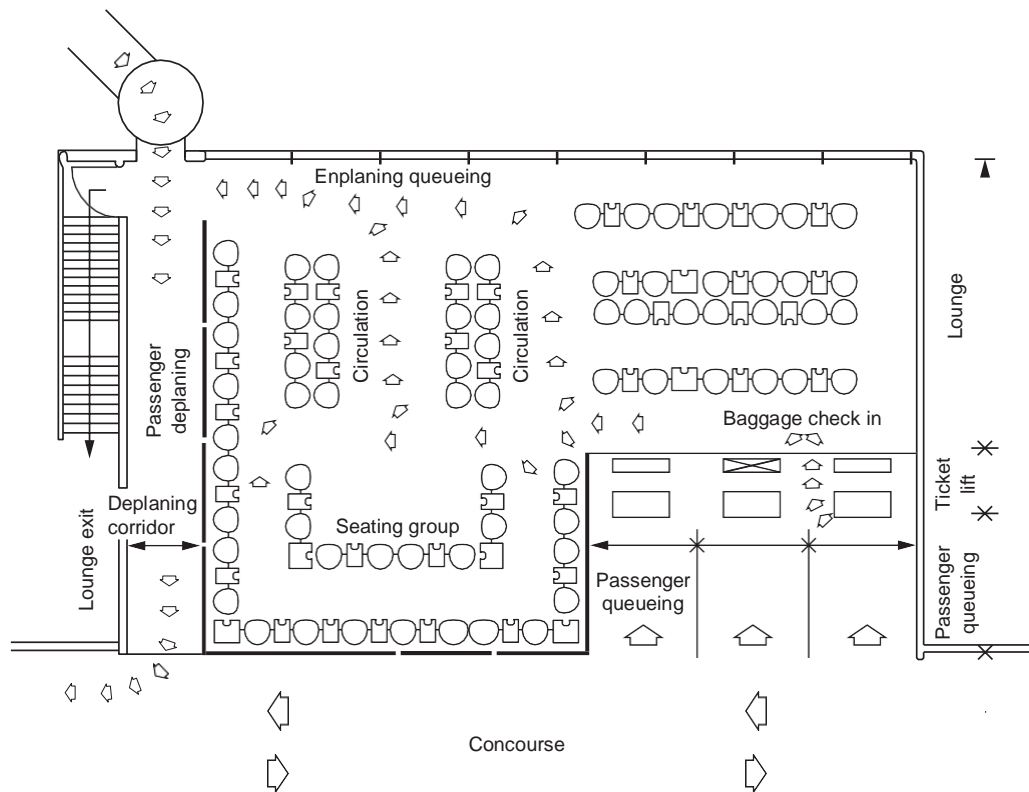


FIGURE 10-13 Departure lounge layout (*Federal Aviation Administration [50]*).

in a pier facility. Figure 10-13 shows the layout for a departure lounge seating about 70 passengers.

Corridors

Corridors provide circulation for passenger and visitor between departure lounges and between departure lounges and the central terminal areas. These should be designed to accommodate physically handicapped persons during the peak periods of high-density flow. Studies have shown that a typical 20-ft-wide corridor will have a capacity ranging from 330 to over 600 persons per minute. For planning purposes, corridor widths should be sized on the basis of about 16.5 passengers per foot of width per minute. The corridor width should be the width required at the most restrictive points, that is, the minimum free-flow width in the vicinity of restaurant entrances, phone booth clusters, or departure lounge check-in points. This standard is based upon a width of 2.5 ft per person and a depth separation of 6 ft between people. The corridor width is adversely impacted by the peaking of deplaning passengers in platoons but the decreased depth separations compensate for the decreased walking rates in these circumstances.

Further guidance on corridor width design is contained in the literature [12, 41, 43, 50].

Baggage Claim Facilities

The baggage claim lobby should be located so checked baggage may be returned to terminating passengers in reasonable proximity to the terminal deplaning curb. At low activity airports, checked baggage may be placed on a shelf for passenger claiming. More active airports have installed mechanical delivery and display equipment similar to that depicted in Fig. 10-14. The number of claim devices required is determined by the number and type of aircraft that will arrive during the peak hour, the time distribution of these arrivals, the number of terminating passengers, the amount of baggage checked on these flights, and the mechanism used to transport baggage from aircraft to

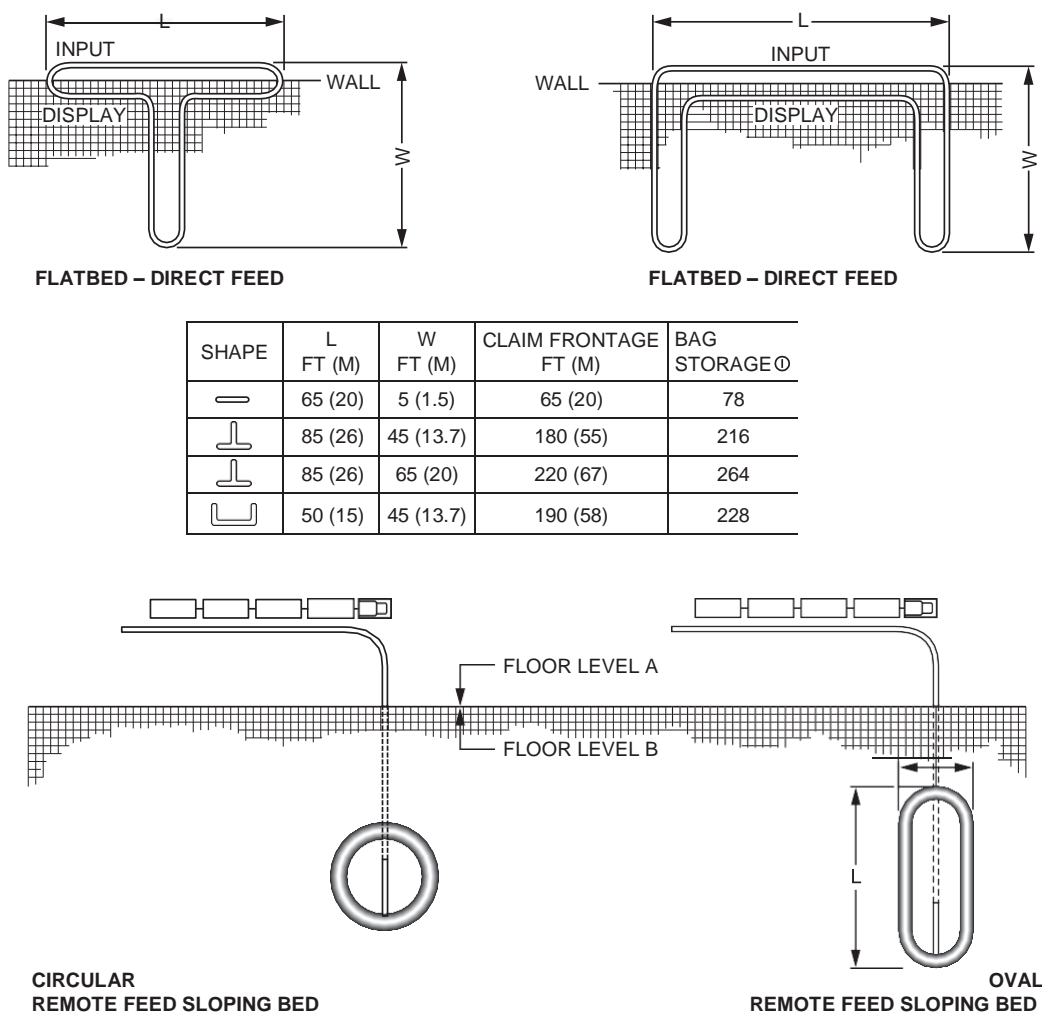


FIGURE 10-14 Mechanized baggage claim devices commonly used at airports (Federal Aviation Administration [43]).

the claim area. In the ideal situation, a baggage claim device should not be shared between flight arrivals at the same time as this leads to considerable congestion in the vicinity of the device and passenger confusion. Greater utilization of the devices is obtained when airlines time the sharing of claim devices for separate flights. Techniques similar to those used to construct ramp charts are useful in scheduling baggage claim devices and for determining the level of congestion in the baggage claim area.

At the present time, except in the most unusual situations, passenger delays in the baggage claim area can be significant due principally to the fact that passengers can travel from aircraft to claim areas much faster than baggage conveyance systems can transport the baggage from aircraft to claim areas. It is therefore essential that claim lobbies be designed to accommodate waiting passengers adequately and provide for rapid claiming of baggage once the baggage is transferred to the claim device. Estimating procedures have been provided by the FAA for sizing claiming devices based upon the equivalent peak 20-min aircraft arrivals. Charts for this purpose are provided in Figs. 10-15 and 10-16. It should be observed that these charts are

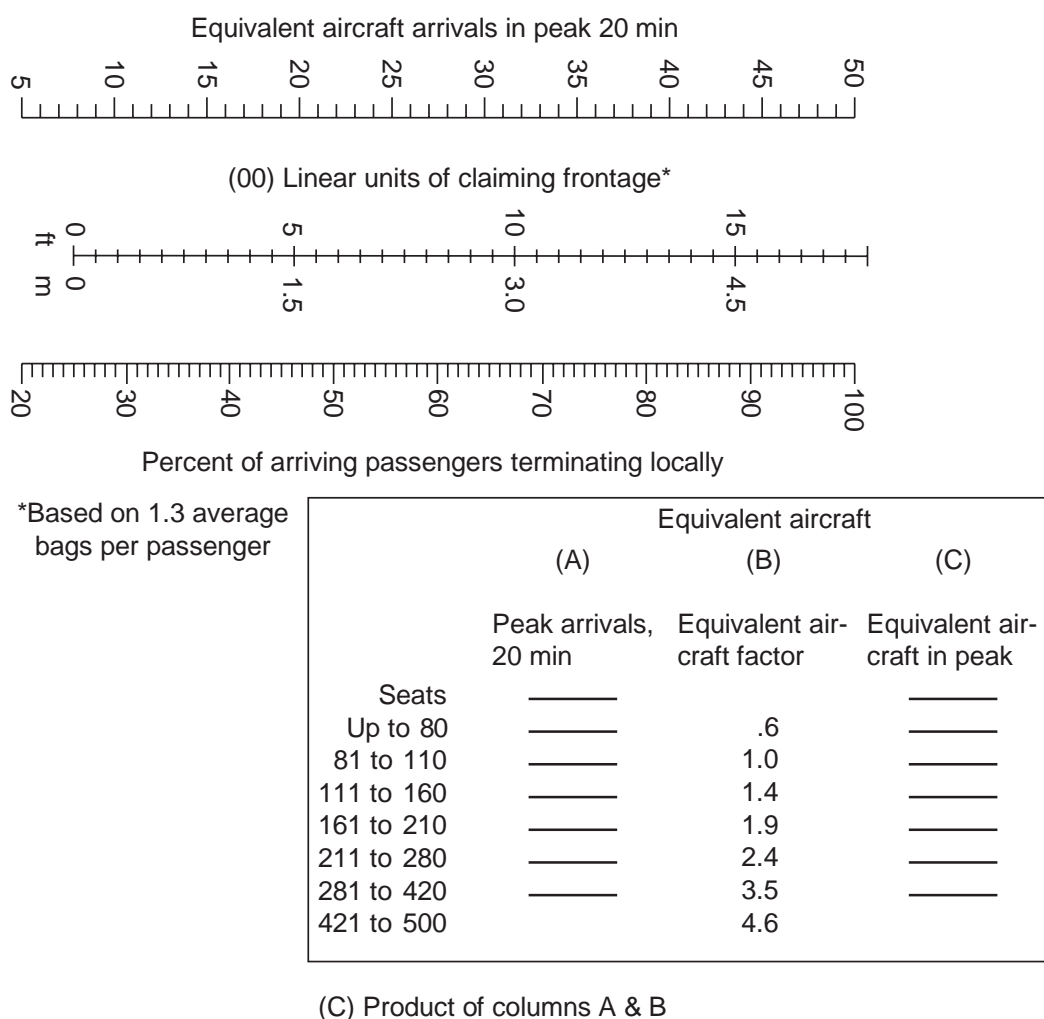
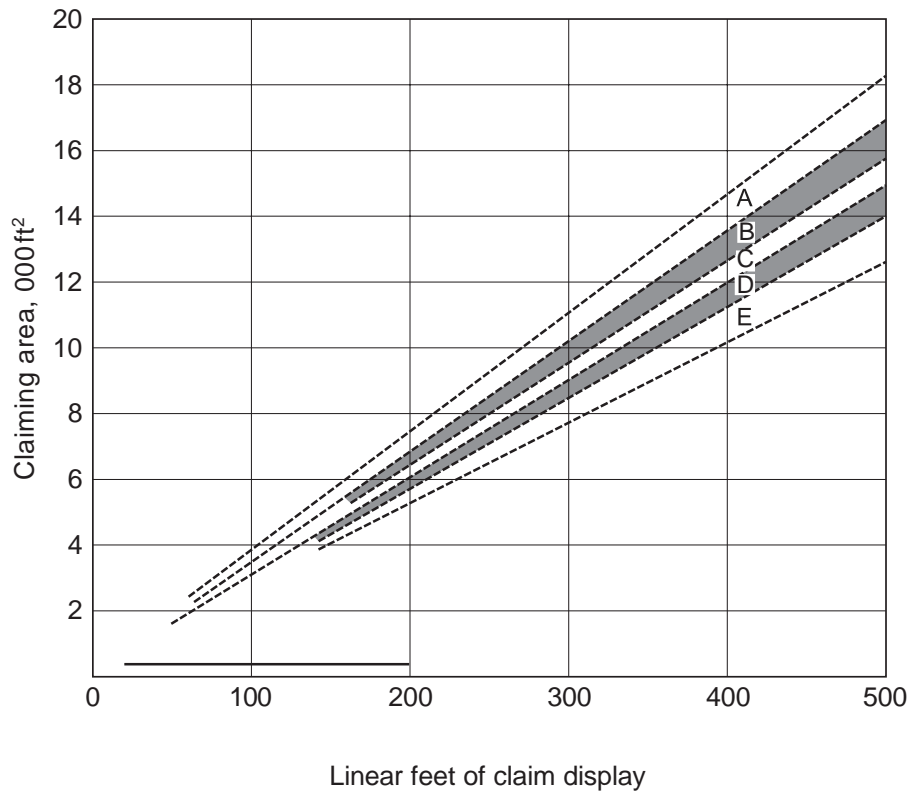


FIGURE 10-15 Estimating nomograph for baggage claim linear claim footage requirements (*Federal Aviation Administration [43]*).



- Areas for optimum configurations of
- A Round – sloping bed/remote feed
 - Tee – flat bed/direct feed
 - B Tee and u-shape alternating @ 75' (flat bed/direct feed)
 - C Oval – flat bed/direct feed
 - Oval – sloping bed/remote feed
 - D Tee and u-shape alternating @ 60' (flat bed/direct feed)
 - E \perp – shape flat bed/direct feed

FIGURE 10-16 Estimating chart for total area of baggage claim facilities (Federal Aviation Administration [43]).

based upon checked baggage at the rate of 1.3 bags per person and adjustments may be required as baggage exceeds this rate.

In preliminary planning studies, great care should be exercised in using the above guidelines. Particular attention should be given to the space provided around a claim device and it is specifically recommended that a clear space of from 13 to 15 ft be provided adjacent to the device for active and waiting claim, as well as claim area circulation. In some instances baggage lobbies have been designed with adequate waiting areas and subsequently other facilities have been moved into these areas, considerably diminishing mobility. It is recommended that an additional 15 to 35 ft of circulation space be provided within the deplaning facility to allow for circulation between the claim devices, rental car positions, and deplaning curbs. If ground transportation facilities are located in these areas,

they should be physically positioned so as not to restrict passenger flow to and from the claim area.

Intraairport Transportation Systems

The use of automated ground transportation systems within the terminal complex at airports is increasing as airports become larger and both the distance and time for passengers to travel through airports have become excessive. In most cases automated ground transportation also provides a clear delineation between the airside and landside functions of an airport and aids in the location of security processing facilities. Moving walkways and automated people mover (APM) systems have become important features in many large terminals.

An Automated People Mover (APM) is an advanced transportation system in which automated driverless vehicles operate on fixed guideways in exclusive rights-of-way. They differ from other forms of transit in that they operate without drivers or station attendants. These systems have been developed and implemented in various sizes and configurations since the early 1970s and today there are over 40 systems operating at airports worldwide.

Most of the early APM systems were implemented to facilitate passenger movement within a terminal or between the central terminal and a satellite building. Figure 10-17 shows an APM vehicle that operates in an underground tunnel that links the central terminal with a mid-field satellite terminal at the Greater Pittsburgh International Airport. More recently, APM systems have been designed to connect airport terminals



FIGURE 10-17 Automated People Mover (APM) Vehicle at the Greater Pittsburgh International Airport.

and link terminals with landside facilities such as parking, car rental services, regional transportation services, hotels, and other related employment and activity centers. Guidance on the planning of such facilities is contained in recent literature [20, 46, 59].

International Facilities

Airports with international operations require space for the inspection of passengers, crew, baggage, aircraft, and cargo. The area required for customs, immigration, agriculture, and public health services may be in a separate facility or in the terminal building itself. These facilities should be designed so that passenger flow between the aircraft and the initial processing station is unimpeded and as short as possible, there is no possibility of contact with domestic passengers or any unauthorized personnel until processing is complete, there is no possible way for an international arrival to bypass processing, and there is a segregated area for in-transit international passengers.

The size of this facility is based on the projection of hourly passengers requiring processing. It is recommended that the appropriate officials and agencies be contacted during the preliminary design phase to determine specific design requirements. Some guidance on the processing rates and sizing of these facilities is found in the literature [43, 58].

Other Areas

Most terminals are developed to accommodate several other activities and the space needs should be determined for each airport based upon local requirements. These activities are identified below.

Airline Activities

The following airline activities may exist at all or some terminal facilities and should be discussed with the airlines which plan on utilizing the facility.

1. Outbound baggage makeup and inbound baggage and conveyance system
2. Cabin services and aircraft maintenance
3. Flight operations and crew ready rooms
4. Storage areas for valuable or outsized baggage
5. Air freight pickup and delivery
6. Passenger reservations and VIP waiting areas
7. Administrative offices
8. Ramp vehicle and cart parking and maintenance

Passenger Amenities

The factors which influence the extent of passenger amenities include the passenger volume, community size, the location and extent of

off-airport services, interests and abilities of potential concessionaires, and rental rates. These generally include

1. Food and beverage services, and newsstands
2. A variety of stores and services
3. Counters for car rental and flight insurance companies
4. Public lockers and public and courtesy telephones
5. Amusement arcades and vending machines
6. Public restrooms

Airport Operations and Services

These facilities and services are normal to most public buildings and include the following:

1. Offices for airport management and staff functions including police, medical and first aid, and building maintenance
2. Building mechanical systems such as heating, ventilation, and air conditioning
3. Communication facilities
4. Electrical equipment
5. Government offices for air traffic control, weather reporting, public health and immigration, and customs
6. Conference and press facilities

Overall Space Requirements

Guidelines have been presented above for the approximate space requirements for the various components in passenger terminal facilities. Once the facilities have been estimated one might compare the space requirements to the approximations given in Table 10-2. The values in this table present overall space requirements which should provide a reasonable level of service and a tolerable occupancy level for the various facilities indicated.

Concept Development

In this phase of the process, the blocks of spaces determined in space programming are allocated in a general way to the terminal complex. There are a number of ways in which the facilities of the passenger terminal system are physically arranged and in which the various passenger processing activities are performed. Centralized passenger processing means all the facilities of the system are housed in one building and used for processing all passengers using the building. Centralized processing facilities offer economies of scale in that many of the common facilities may be used to service a large number of aircraft gate positions. Decentralized processing, on the other hand, means the passenger facilities are arranged in smaller modular units

Component	Space Required In 1000 ft² or 100 m² per 100 Typical Peak Hour Passengers
Ticket lobby	1.0
Baggage claim	1.0
Departure lounge	2.0
Waiting rooms	1.5
Immigration	1.0
Customs	3.0
Amenities	2.0
Airline operations	5.0
Total gross area	
Domestic	25.0
International	30.0

TABLE 10-2 Typical Terminal Building Space Requirements

and repeated in one or more buildings. Each unit is arranged around one or more aircraft gate positions and serves the passengers using those gate positions. There are four basic horizontal distribution concepts, as well as many variations or hybrids which include combinations of these basic concepts. Each can be used with varying degrees of centralization. These concepts are discussed below.

Horizontal Distribution Concepts

The following terminal concepts should be considered in the development of the terminal area plan. Sketches of the various concepts are shown in Fig. 10-18. Many airports have combined one or more terminal types.

Pier or Finger Concept

The pier concept has an interface with aircraft along piers extending from the main terminal area. Aircraft are usually arranged around the axis of the pier in a parallel or nose-in parking alignment. Each pier has a row of aircraft gate positions on both sides, with a passenger concourse along the axis which serves as the departure lounge and circulation space for both enplaning and deplaning passengers. This concept usually allows for the expansion of the pier to provide additional aircraft parking positions without the expansion of the central passenger and baggage processing facility. Access to the terminal area is at the base of the connector or the pier. If two or more piers are employed, the spacing between the two piers must provide for maneuvering of aircraft on one or two apron taxilanes. When each pier serves a large number of gates, and the probability exists that two or more aircraft may frequently be taxiing between two piers and

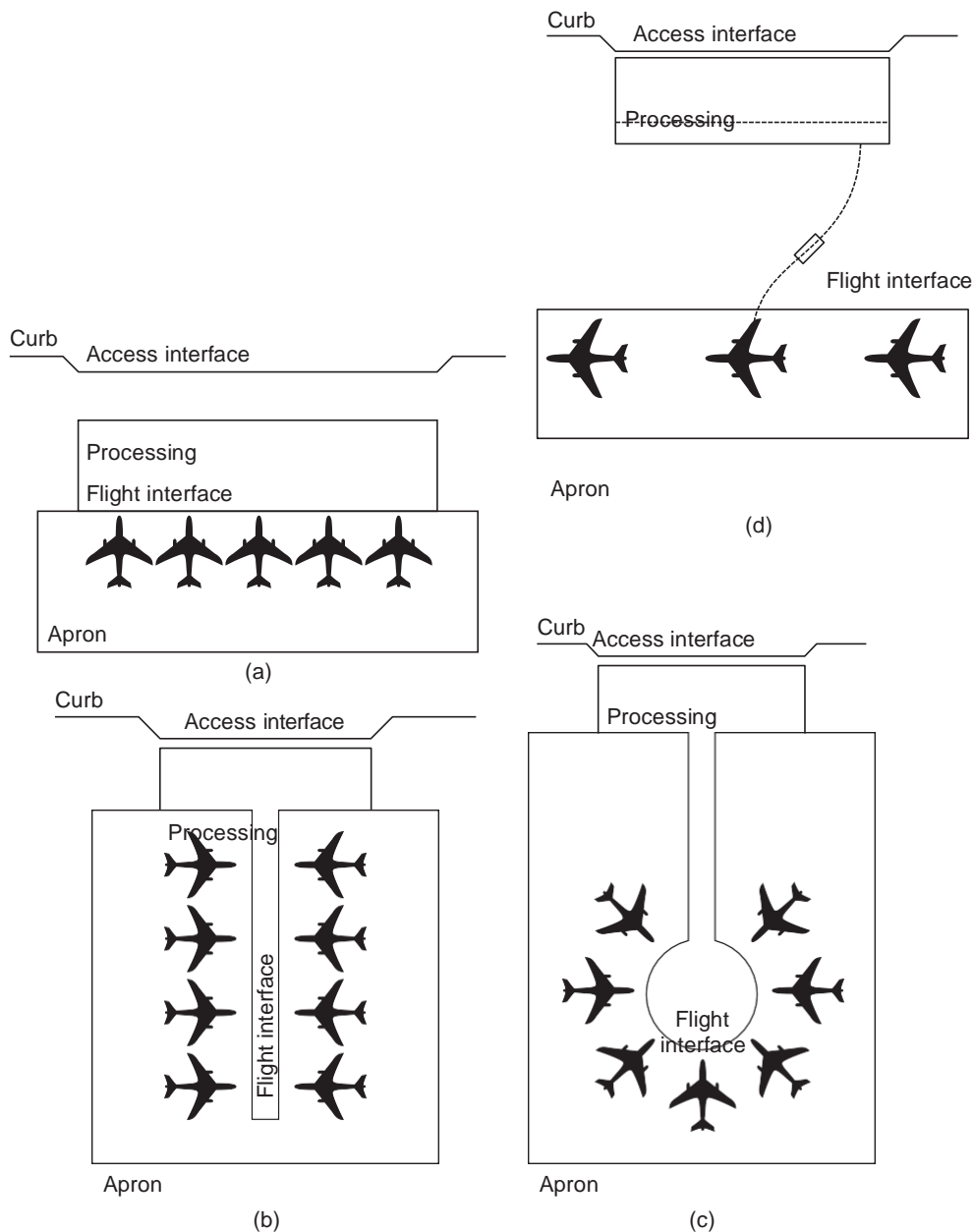


FIGURE 10-18 Horizontal distribution concepts for passenger terminals: (a) linear, (b) pier, (c) satellite, (d) transporter.

will be in conflict with one another, then two taxilanes are advisable. Also, access from this taxiway system by two or more aircraft may require two apron edge taxiways to avoid delays.

The chief advantage of this concept is its ability to be expanded in incremental steps as aircraft or passenger demand warrant. It is also relatively economical in terms of capital and operating cost. Its chief disadvantages are its relatively long walking distance from curbside front to aircraft and the lack of a direct curbside front relationship to aircraft gate positions.

Satellite Concept

The satellite concept consists of a building, surrounded by aircraft, which is separated from the terminal and is usually reached by means of a surface, underground, or above ground connector. The aircraft are normally parked in radial or parallel positions around the satellite. It often affords

the opportunity for simple maneuvering and taxiing patterns for aircraft

but requires more apron area than other concepts. It can have common or separate departure lounges. Since enplaning and deplaning from the aircraft is accomplished from a common and often remote area, mechanical systems may be employed to transport passengers and baggage between the terminal and satellite.

The main advantages of this concept lie in its adaptability to common departure lounge and check-in functions and the ease of aircraft maneuverability around the satellite structure. However, construction cost is relatively high due to the need to provide connecting concourses to the satellite. It lacks flexibility for expansion and passenger walking distances are relatively long.

Linear, Frontal or Gate Arrivals Concept

The simple linear terminal consists of a common waiting and ticketing area with exits leading to the aircraft parking apron. It is adaptable to airports with low airline activity which will usually have an apron providing close-in parking for three to six commercial passenger aircraft. The layout of the simple terminal should take into account the possibility of pier, satellite, or linear additions for terminal expansion. In the gate arrivals or frontal concept, aircraft are parked along the face of the terminal building. Concourses connect the various terminal functions with the aircraft gate positions. This concept offers ease of access and relatively short walking distances if passengers are delivered to a point near gate departure by vehicular circulation systems. Expansion may be accomplished by linear extension of an existing structure or by developing two or more terminal units with connectors.

Both of these concepts provide direct access from curb front to aircraft gate positions and afford a high degree of flexibility for expansion. It does not provide convenient opportunities for the use of common facilities and, as this concept is expanded into separate buildings, it may lead to high operating costs.

Transporter, Open Apron or Mobile Conveyance Concept

Aircraft and aircraft servicing functions in the transporter concept are remotely located from the terminal. The connection to the terminal is provided by vehicular transport for enplaning and deplaning passengers. The characteristics of the transporter concept include flexibility in providing additional aircraft parking positions to accommodate increases in schedules or aircraft size, the capability to maneuver an aircraft in and out of a parking position under its own power, the separation of aircraft servicing activities from the terminal, and reduced walking distances for the passenger.

Concept Combinations and Variations

Combinations of concepts and variations are a result of changing conditions experienced from the initial conception of the airport throughout its life span. An airport may have many types of passenger activity, varying from originating and terminating passengers using the full range of terminal services to passengers using limited services on

commuter or connecting flights. Each requires a concept that differs considerably from the other. In time, the proportion of traffic handled by flights may change, necessitating modification or expansion of the facilities. Growth of aircraft size or a new combination of aircraft types servicing the same airport will affect the type of concept. In the same way, physical limitations of the site may cause a pure conceptual form to be modified by additions or combinations of other concepts.

Combined concepts acquire certain of the advantages and disadvantages of each basic concept. A combination of concept types can be advantageous where more costly modifications would be necessary to maintain the original concept. For example, an airline might be suitably accommodated within an existing transporter concept terminal while an addition is needed for a commuter operation with rapid turn-overs which would be best served by a linear concept extension. In this situation, combined concepts would be desirable. The appearance of concept variations and combinations in a total apron terminal plan may reflect an evolving situation in which altering needs or growth have dictated the use of different concepts. Illustrations of the generation of various horizontal distribution concepts in the concept development phase of Geneva Intercontinental Airport are shown in Fig. 10-19. Applications of several of the concepts to existing airports are shown in Fig. 10-20*a* through 10-20*c*.

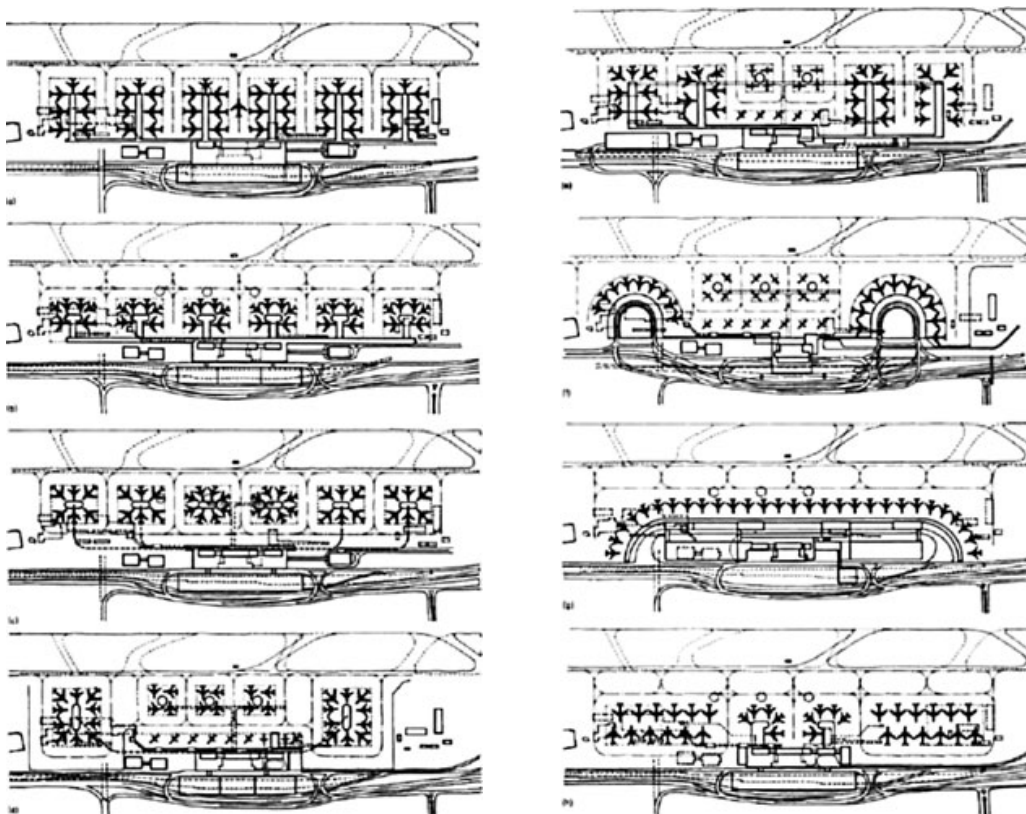


FIGURE 10-19 Conceptual designs for Geneva Intercontinental Airport (*Reynolds, Smith and Hills [25]*).



FIGURE 10-20A Linear, satellite and pier concepts—O'Hare International Airport.



FIGURE 10-20B Pier and satellite concepts—Cleveland Hopkins International Airport.

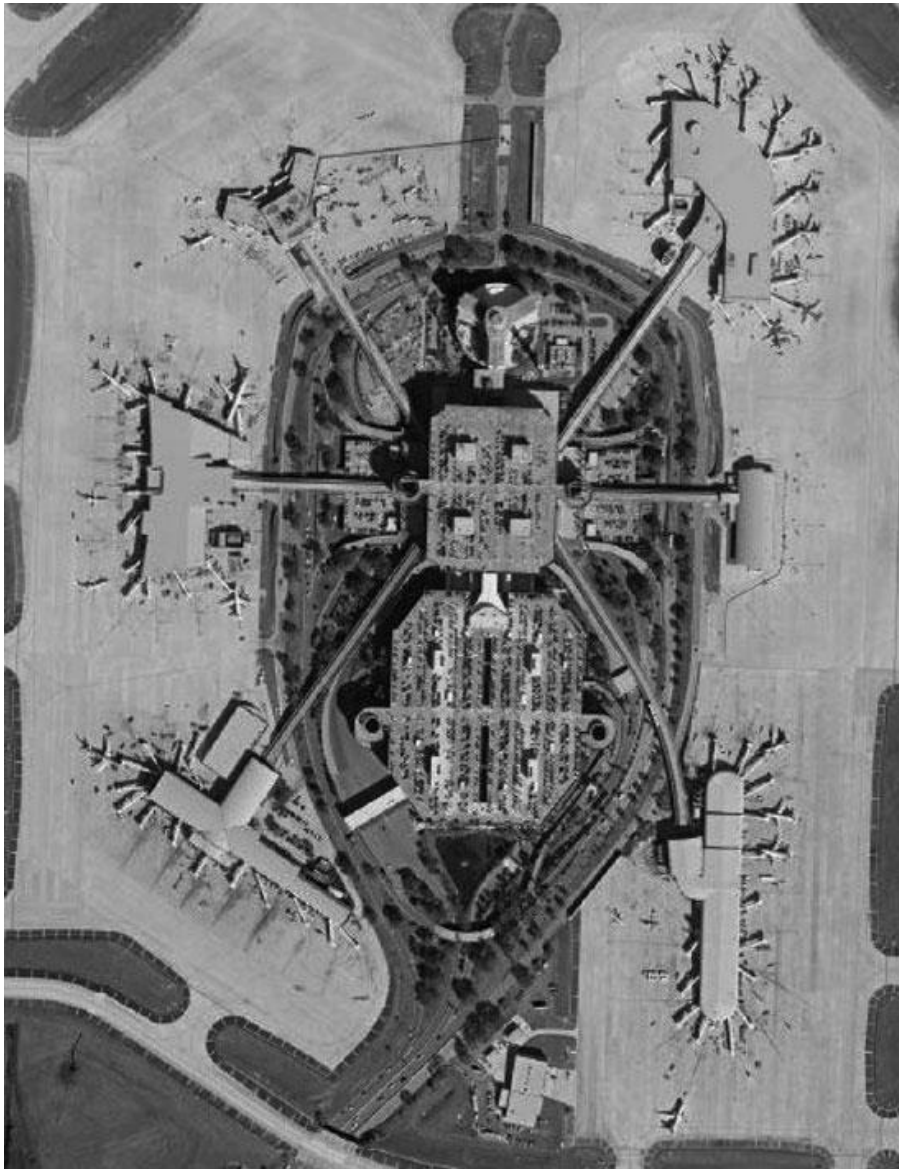


FIGURE 10-20C Satellite airside/landside concept—Tampa International Airport.

Figure 10-20*a* shows the terminal area layout at O'Hare International Airport which shows the linear, satellite and pier concepts. Concourse B is a linear concept, concourse C is a satellite concept connected to the main terminal building by an underground moving sidewalk, and concourses E, F, G, H, K, and L are pier concepts. Figure 10-20*b* shows the terminal area layout at Cleveland Hopkins International Airport which consists of pier and satellite concepts. Figure 10-20*c* shows the terminal area layout at Tampa International Airport which shows five airside satellites connected to the land-side building by automated transit on above ground fixed guideways.

Vertical Distribution Concepts

The basis for distributing the primary processing activities in a passenger terminal among several levels is mainly to separate the flow of arriving and departing passengers. The decision concerning the number of levels a terminal facility should have depends primarily on the volume of passengers and the availability of land for expansion in the immediate vicinity. It may also be influenced by the type of traffic, for example, domestic, international, or commuter passengers being processed, by the terminal area master plan, and by the horizontal processing concept chosen.

With a single level system all processing of passengers and baggage occurs at the level of the apron. Separation between arriving and departing passenger flows is achieved horizontally. Amenities and administrative functions may occur on a second level. With this system, stairs are normally used to load passengers onto aircraft. This system is quite economical and is suitable for relatively low passenger volumes. The single level terminal is shown in Fig. 10-21a.

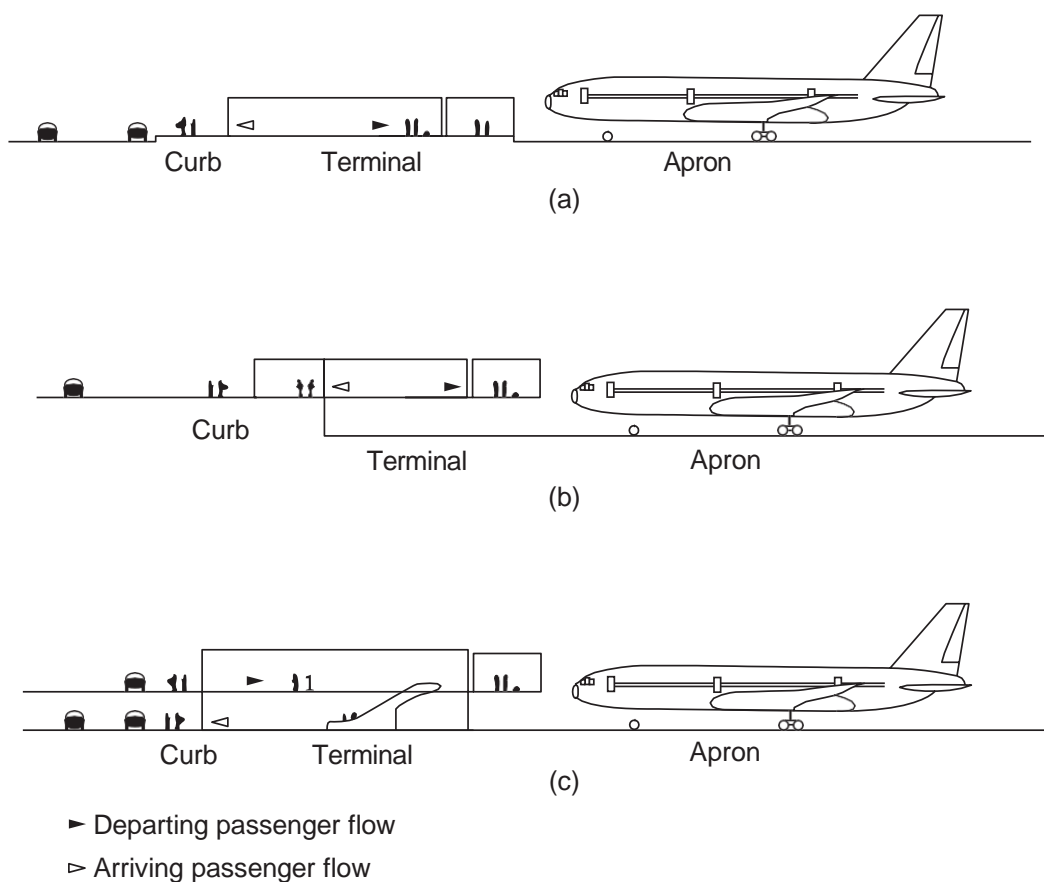


FIGURE 10-21 Vertical distribution concepts: (a) single level, (b) second level loading, (c) two-level system (*Federal Aviation Administration [50]*).

Two level passenger terminal systems may be designed in a number of different ways. In one type, shown in Fig. 10-21*b*, the two levels are used to separate the passenger processing area and the baggage handling areas. Thus, processing activities including baggage claim occur on the upper level, while airline operations and baggage handling activities occur at the lower apron level. The advantage of raising the passenger handling level is that it becomes compatible with aircraft doorsill heights, allowing convenient interface with the aircraft. Vehicular access occurs on the upper level to facilitate the interface with the processing system.

Another articulation of the two-level system separates the arriving and departing passenger streams. In this case departing passenger processing activities occur on the upper level and arriving passenger processing including baggage claim occurs at the apron level. Airline operations and baggage handling also occur at the lower level. Vehicular access and parking occur at both levels, one for arrivals and one for departures, and parking can be surface or structural. An example of this design is shown in Fig. 10-21*c*.

Variations in these basic designs may occur when traffic volumes or the type of traffic so require. For example, for international airport terminals, a third level may be needed for international passengers. Also, at large airports where intraairport transportation systems operate, a special level may be needed to provide for these systems. Figure 10-22*A* shows a multilevel system with structural parking, intraairport transportation, and underground mass transit access. Figure 10-22*B* shows a multilevel system with integrated structural parking. In this variation more direct access to the processing component is attained by providing parking above the processing facility.

Based upon an examination of a large number of airports, it is possible to identify those concepts which are candidates for further consideration. Using the level of annual enplanements and the function nature of the airport, as defined by the relative proportions of originating, terminating, through, or connecting passengers, Fig. 10-23 offers some guidance to the airport planner for the initial identification of appropriate horizontal and vertical distribution concepts. It should be noted, however, that in many instances the constraints of existing terminal facilities, land availability, and the ground access system may restrict the options which are viable alternatives for terminal expansion.

Prior to proceeding with the planning of the airport terminal system, an evaluation of the concepts which have evolved in the conceptual development phase of the project is undertaken, to identify those alternatives which should be brought into the schematic design and design development phase of the project. To do this an overall rating of the various concepts relative to the design criteria is performed.

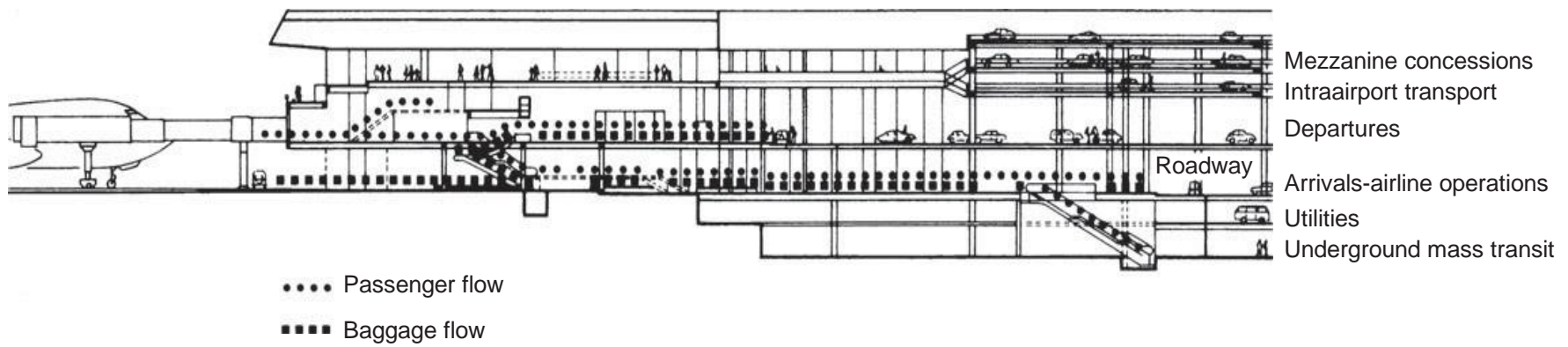


FIGURE 10-22A Multilevel passenger processing system—structural parking adjacent to terminal (*Hamburg Airport Authority*).

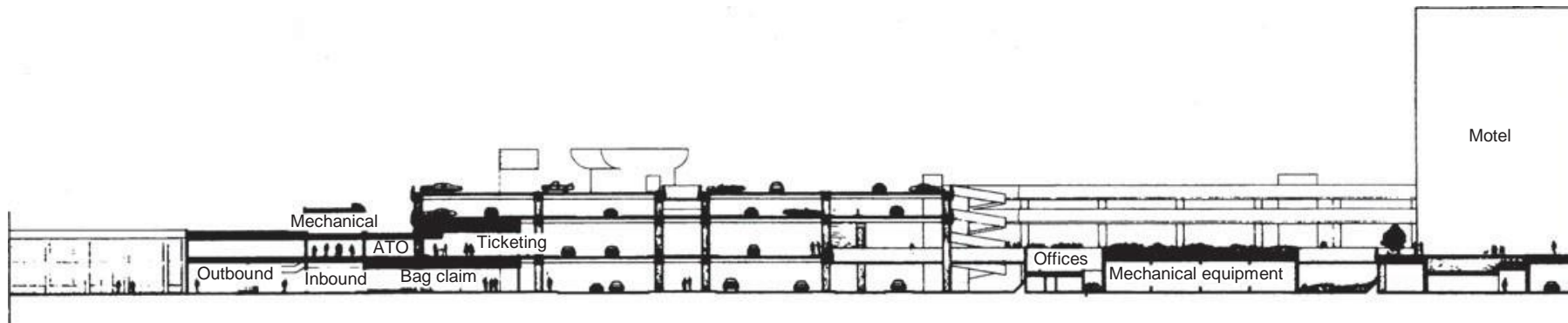


FIGURE 10-22B Multilevel passenger processing system—structural parking above processing area (*Reynolds, Smith and Hills*).

Airport size by enplaned pax/year	Concepts applicable				Physical aspects of concepts							
	Linear	Pier	Satellite	Transporter	Single level curb	Multi level curb	Single level terminal	Multi level terminal	Single level connector	Multi level connector	Apron level boarding	Aircraft level boarding
Feeder under 25,000	X				X		X				X	
Secondary 25,000 to 75,000	X				X		X				X	
75,000 to 200,000	X				X		X		X		X	
200,000 to 500,000	X	X			X		X		X		X	
Primary over 75% pax O/D 500,000 to 1,000,000	X	X	X		X		X		X	X	X	X
Over 25% pax transfer 500,000 to 1,000,000	X	X	X		X		X		X	X	X	X
Over 75% pax O/D 1,000,000 to 3,000,000		X	X	X	X	X		X	X	X	X	X
Over 25% pax transfer 1,000,000 to 3,000,000		X	X		X	X		X	X	X	X	X
Over 75% pax O/D over 3,000,000		X	X	X	X	X		X	X	X	X	X
Over 25% pax transfer over 3,000,000		X	X		X	X		X	X	X		X

FIGURE 10-23 Applicable concepts for airport design (*Federal Aviation Administration [50]*).

The concept evaluation rating factors used for Geneva airport are listed in Table 10-3.

Schematic Design

The schematic design process translates the concept development and overall space requirements into drawings which show the general size, location, and shape of the various elements in the terminal plan. Functional relationships between the components are established and evaluated. The adequacy of the overall space program is evaluated by airport users relative to their specific needs. This phase of the process specifically examines passenger and baggage flow routes through the system and seeks to examine the adequacy of the facility from the point of view of flow levels and flow conflicts.

Passenger convenience: Walking distance from curb to aircraft Walking distance for transfer passengers Walking distance from parking to aircraft Ease of passenger orientation Ease of passenger processing
Operational effectiveness: Efficient taxing routes Ground flow coordination of vehicles and aircraft Apron area maneuverability Apron adaptation to future aircraft Vehicular access flows Direct routes to ancillary facilities
Expansion adaptability: Ancillary facilities, flexible land use Staging adaptability Visual character of increments Gross terminal expandability Expandability of terminal elements
Economic effectiveness: Capital cost Maintenance and operating costs Ratio of revenue- to non-revenue-producing areas

Source: Reynolds, Smith and Hills [25].

TABLE 10-3 Conceptual Development Rating Factors for Evaluation of Terminal Planning Concepts

Modeling techniques are usually employed in this phase of the process to identify passenger processing, travel, and delay times, and the generation of lines at processing facilities. The main purpose for analyzing passenger and baggage handling systems is the determination of the extent and size of the facilities needed to provide a desired level of convenience to the passenger at reasonable cost. In this analysis alternative layouts can be studied to determine which is the most desirable.

Analysis Methods

A number of systems analysis techniques have proven to be useful for the analysis of facilities for passengers and baggage. These include network models, queuing models, and simulation models.

Network Models

These models are particularly useful for representing and analyzing the interrelationships between the various components of a passenger

or baggage processing system. For example, passenger processing can be represented as a network with the nodes representing service facilities and the links representing the travel paths and passenger splits. This type of representation allows estimation of delays to the passenger at various locations within the terminal.

An example of a network that has been applied to the evaluation of arriving passenger delay is the critical path model (CPM) [47]. CPM is used to coordinate the various activities that take place in the system for handling both passengers and baggage. Nodes that represent critical activities, that is, those that take the greatest amount of time, are easily identified and can be analyzed in more detail to determine their effect on the overall performance of the system.

The analysis of the service time and waiting time at each processor in a network model can be obtained through either analytical queuing models or simulation.

Analytical Queuing Models

Queuing theory permits the estimation of delays and queue lengths for service facilities under specified levels of demand. The application of queuing theory yields useful estimates of processing and delay times from which the required sizes of facilities and operating costs may be derived.

Virtually all of the components of the passenger handling system can be modeled as service facilities using queuing models. The diagram in Fig. 10-24 and Example Problem 10-10 showed an example of the application of a deterministic queuing model to the operation of a runway system to determine aircraft delay. A similar type of analysis can be applied to the analysis of passenger processing systems. It is possible to evaluate the impact of adding ticket agents on delays to passengers and on the size of the queues. With this information it is possible to evaluate the feasibility of alternative operating strategies for the ticketing facility.

Diagrams similar to this may be constructed for each of the processors for the passenger and baggage handling system and yield satisfactory results when the demand rate exceeds the service rate.

For the analysis of component delay and queue length when the average demand rate over some period of time is less than the average service rate, queuing theory is used to generate mathematical functions representative of the arrival and service performance of the system. To specify the mathematical formulation of this problem, it is necessary to define the arrival distribution, the service distribution, the number and use of the servers, and the service discipline. Many of the components which service passengers in the airport terminal exhibit a random or Poisson arrival process. The service characteristics are usually exponential, constant, or some general distribution defined by average service times and the variance of average service time. In most cases, there is more than one channel for the performance

of passenger service, and the queuing mechanism is based on first come, first served basis. Extensive research has been carried out in recent years to determine mathematical formulations which adequately represent the processing system [2, 10, 14, 19, 21, 53, 56]. Because of the variability associated with passenger behavior at an airport, it is virtually impossible to obtain precise mathematical formulations for delay at processors. However, reasonable estimates of delay and corresponding queue lengths are possible using simple formulations.

One such formulation [17, 35, 52, 56] is that of a multiple station queuing system with a Poisson arrival distribution and a service time distribution which is characterized by the average service time and the variance of the average service time as shown in Eq. (10-1).

$$W_s = \frac{\sigma^2 + t^2}{2t} \frac{\lambda^k t^{k+1}}{(k-1)!(k-\lambda t) \sum_{n=0}^{k-1} \frac{(\lambda t)^n / n!}{(k-1)!} + \frac{\lambda^k}{(k-1)!} \frac{1}{1 - \lambda t}} \quad (10-1)$$

where W_s = average delay per person

λ = average demand rate

t = average service time for a processor, the reciprocal of the average service rate of a processor, μ

σ = standard deviation of the average service time of a processor

k = number of processors

n = counter in the equation

This equation is valid when the average demand rate on the system of processors λ is less than to total service rate of the processors, $k\mu$; that is, the ratio of the average demand rate to the total service rate ρ is less than 1.

When the number of processors k is equal to 1, this equation reduces to

$$W_s = \frac{\sigma^2 + t^2}{2t} \frac{\lambda}{\mu(\mu - \lambda)} \quad (10-2)$$

or since ρ is equal to $\lambda/k\mu$

$$W_s = \frac{\sigma^2 + t^2}{2t} \frac{\rho}{\mu(1 - \rho)} \quad (10-3)$$

For a single server system which exhibits a Poisson arrival distribution and an exponential service time distribution, Eq. (10-1) reduces to

$$W_s = \frac{\lambda}{\mu(\mu - \lambda)} \quad (10-4)$$

or

$$W_s = \frac{\rho}{\mu(1 - \rho)} \quad (10-5)$$

$$s = \mu - \rho(1 - \mu)$$

For a single server system which exhibits a Poisson arrival distribution and a constant service time distribution, Eq. (10-1) reduces to

$$W_s = \frac{\lambda}{\mu(\mu - \lambda)} \quad (10-6)$$

or

$$W_s = \frac{\rho}{\mu(1 - \rho)} \quad (10-7)$$

When demand is less than capacity, the following expression gives the average line length N over the period being analyzed and consists of those in service and those waiting for service at a processor.

$$N = W_s + \frac{\lambda}{\mu} \quad (10-8)$$

When $\rho > 1$ there is statistical delay plus excess delay which is defined by a deterministic model. For design purposes it is sufficient to estimate delays in such a system in which the statistical delay W_s is computed from the appropriate equation above with $\rho = 0.90$ and the excess delay W_e is added to this from the equation below to compute total processor delay. The rationale for computing the statistical delay with $\rho = 0.9$ is that in reality as demand approached capacity the delay cannot become infinite as airline or airport operating practices will limit delay by utilizing additional servers.

$$W_e = \frac{T - (k/\mu)}{k} \lambda \quad (10-9)$$

where W_e = delay when the demand exceeds the service rate

T = time period during which the demand exceeds the service rate

λ = total demand on the system of processors

k = number of processors

μ = service rate of a processor

It is usually assumed that the time required to reduce the demand to capacity T is about one-half of the time period being analyzed. Typically when demand exceeds capacity, the operator of the facility will increase the number of operating service facilities to alleviate the growth in both waiting time and queue lengths. However, the extent to which this is done is a function of airline operating policies and the availability of additional manpower and physical facilities.

The average line length over the period analyzed N including those in service and those waiting for service; when demand exceeds capacity can be estimated by the following equation.

$$N = W_s + W_e + \frac{\lambda}{\mu} \quad (10-10)$$

Estimates of the waiting times and queue lengths for multiple service channels may be obtained by proportioning the demand equally among processors with the same service characteristics and utilizing single server models. Better estimates may be obtained through the use of multiple channel queuing models whose mathematical formulation was given above. A representation of the passenger delay at baggage claim facilities is given by the relationship [52]

$$W = E(t_1) + \frac{nT}{n+1} - (E(t_2)) \quad (10-11)$$

where $E(t_2)$ = expected value of the time when the first piece of baggage arrives at the claim area

$E(t_1)$ = expected value of the time when passengers arrive at the claim area

n = number of pieces of baggage to be claimed by each passenger

T = length of time from the arrival of the first bag until the arrival of the last bag at the claim device

Others have formulated models which show the buildup and drop-off of passengers and baggage in the claim area [19].

The use of a generalized probability density function called the *Erlang distribution* is recommended as a mechanism for evaluating the service characteristics of the various passenger component processors. By collecting a sample of data relative to a specific component, the constant in the Erlang distribution function can be calculated. This constant determines the particular functional relationships for the server. It is possible that this type of distribution may better describe the queuing characteristics of processors. This distribution has been used successfully in passenger terminal modeling [15].

Great care must be exercised in the application of mathematical models and the interpretation of the results. In most cases, the mathematical representation of the terminal system is best suited for the comparison of alternatives and the identification of those components in the system requiring detailed analyses.

The specification of the service time distribution for use in queuing equations is a function of the distribution of service times demonstrated by the service facility. In general, those processors which exhibit a requirement for small service times, for example, flow through type facilities such as doors, security, and gates are probably best represented by an exponential service time distribution. Those facilities which require a finite service time at a processor such as ticketing, baggage check, seat selection, and rental car checkout, are probably best represented by either a general service time distribution, which is characterized by the average service time and the standard deviation of the average service time, or a constant service time distribution.

In a network model, the average passenger processing time $E(T_p)$ is given by

$$E(T_p) = E(T_w) + E(T_s) + E(T_d) \quad (10-12)$$

where $E(T_w)$ = average passenger delay time throughout the system of processors

$E(T_s)$ = average passenger service time throughout the system of processors

$E(T_d)$ = average passenger travel time through the network of processors

Estimates for the observed range of service time for many of the passenger processing components at an airport are given in Table 10-4.

Example Problem 10-1 presents an illustration of the analysis of the enplaning system at an airport using a network model and the above queuing equations.

Example Problem 10-1 The terminal building layout for enplaning passengers at an airport is given in Fig. 10-24.

Two airlines are servicing the airport, North American Airlines (NA) and Western Pacific Airlines (WP). The enplaning passenger processing facilities and their service rates are given in Table 10-5.

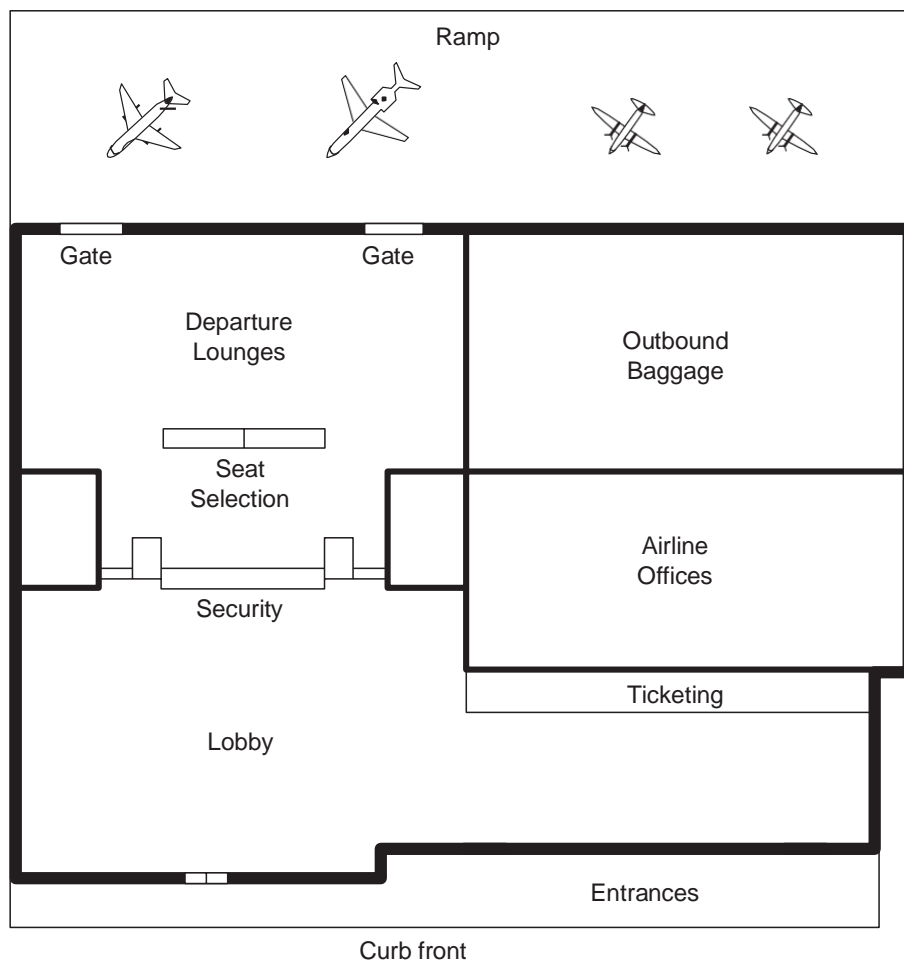


FIGURE 10-24 Layout of enplaning passenger processing system for Example Problem 10-1.

Component Type	Service Rate per Passenger, s	Standard Deviation
Entrance and exit doors		
Automated with baggage	2.0–25	0.5
Automated without baggage	1.0–1.5	0.75
Manual with baggage	3.0–5.0	1.0
Manual without baggage	1.5–3.0	0.75
Stairways	3.0–4.0	1.0
Escalators	1.0–3.0	1.0
Moving sidewalks	1.0–3.0	1.0
Apron doors		
With stairs	4.0–8.0	2.0
Without stairs	3.0–7.0	1.5
Jetway	2.0–6.0	1.0
Ticketing and baggage		
Manual with baggage	180–240	60
Manual without baggage	100–200	30
Baggage only	30–50	10
Information	20–40	10
Automated with baggage	160–220	30
Automated without baggage	90–180	40
Security		
Hand-check baggage	30–60	15
Automated	30–40	10
Seat selection		
Single fights	25–60	20
Multiple fights	35–60	15
Rental car		
Check-in	120–240	60
Checkout	180–300	90
Automated check-in	60–90	20
Baggage claim		
Manual	10–15	8
Automated carousel	5–10	5
Automated racetrack	5–10	5
Automated tee	6–12	5

Sources: Various airport studies.

TABLE 10-4 Observed Service Times for Passenger Processing Facilities at Airports

Type	Airline	Number	Average Service Time per Passenger, s
Entrance doors	All	3	15
Regular ticketing	NC	3	210
	UA	3	180
Express check-in	UA	1	60
Security	All	2	30
Seat selection	NC	1	45
	UA	1	30
Ramp gates	NC	1	20
	UA	1	20

TABLE 10-5 Enplaning Passenger Service Processors for Example Problem 10-1

Demand data collected at the airport indicate that 10 percent of the enplaning passengers proceed directly from curb front to security, 20 percent of Western Pacific Airlines enplaning passengers use the express check-in, there are 0.5 visitors per passenger during the peak hour, and 50 percent of these visitors proceed beyond security.

The peak hour enplaning passenger demand on the design day is expected to be 135 passengers, of which 53 percent use North American Airlines and 47 percent use Western Pacific Airlines. Typically the peak 30 min of the peak hour has 57 percent of the peak hour demand. Assume average walking rates are 1.5 ft/s. Airline operating practices limit the periods when demand exceeds capacity to 15 min.

It is necessary to determine the average enplaning passenger delay and line length at each processor, and the average passenger processing time during the peak 30 min of the peak hour on the design day at this airport.

The link-and-node diagram in Fig. 10-25 shows the relationship between the enplaning passenger processors at the airport, and includes the passenger split between processors, the number of processors, the processing time per passenger, and the distance between processor in feet.

Since the peak 30 min passenger arrival rate is 57 percent of the peak hour arrival rate, the peak 30 min passenger demand into the enplaning passenger system is $0.57(135) = 77$ passengers. Since there are 0.5 visitors per passenger, the peak 30 min visitor demand into the enplaning passenger system is $0.5(77) = 39$ visitors. Combining these demand parameters with the passenger splits given in Fig. 10-25, recognizing that only 50 percent of the visitors proceed beyond security, the processor flow rates in the peak 30 min are given in Fig. 10-26. In this figure the numbers above the lines connecting processors represent the passengers flowing into the processor and the numbers below these lines represent the visitors flowing into each processor when the processor processes visitors.

Assuming that all processors may be modeled by single processor systems in which the demand is split evenly between processors, the demand rate in persons per minute, the service rate in persons per minute, and the ratio of the

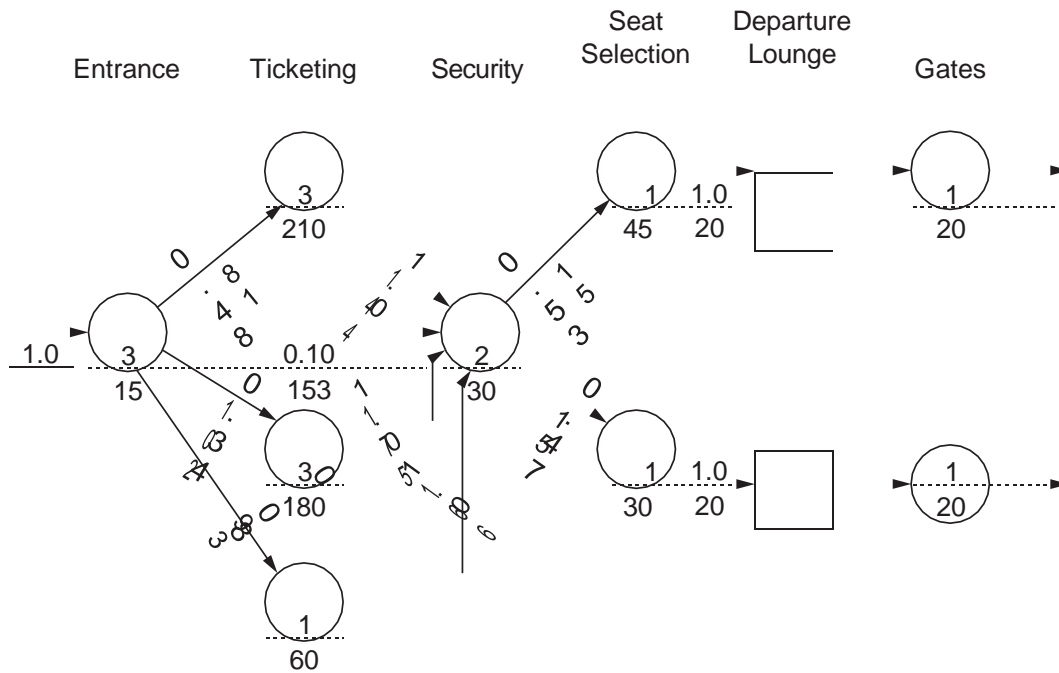


FIGURE 10-25 Link-and-node diagram representing enplaning passenger processing system for Example Problem 10-1.

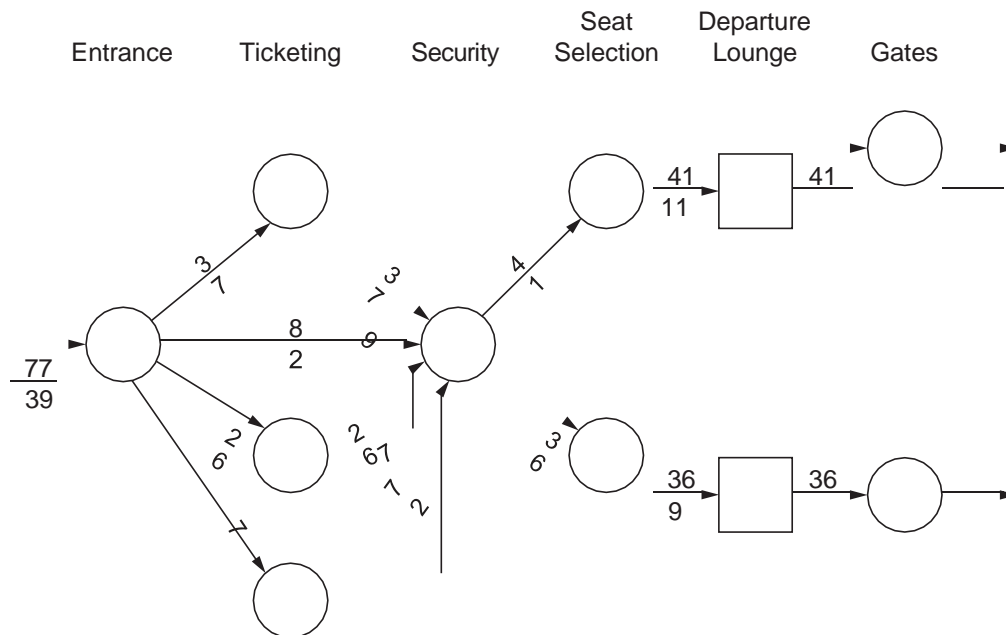


FIGURE 10-26 Link-and-node diagram representing passenger and visitor flows into passenger processing facilities for Example Problem 10-1.

demand to capacity for each processor, the processor utilization, is computed as shown in Table 10-6.

The departure lounges are accumulating processors and are analyzed differently than the queuing processors. Also note that the North American regular ticketing position and the North American seat selection components have a demand which exceeds capacity since the processor utilization is greater than 1.

Based upon an examination airport processor it was found that those processors which are essentially flow-through processors (such as entrance doors, security and ramp gates) behave as Poisson arrivals and exponential service rate

queuing mechanisms, whereas those processors which require a discrete service

Processor	Demand Rate, pers/min	Service Rate pers/min	Processor Utilization
Entrance doors	1.29	4.00	0.32
Regular ticketing			
North American	0.41	0.29	1.41
Western Pacific	0.29	0.33	0.88
Express check-in			
North American	0.23	1.00	0.23
Security	1.62	2.00	0.81
Seat selection			
North American	1.37	1.33	1.03
Western Pacific	1.20	2.00	0.60
Ramp gates			
North American	1.37	3.00	0.46
Western Pacific	1.20	3.00	0.40

TABLE 10-6 Service Processor Characteristics for Example Problem 10-1

function (such as regular ticketing, express check-in, and seat selection) behave as Poisson arrival and constant service rate queuing mechanisms.

Therefore, for the entrance doors, which are Poisson arrivals and exponential services, the average passenger delay is from Eq. (10-4),

$$W = \frac{1.29}{4.00(4.00 - 1.29)} = 0.12 \text{ min}$$

and from Eq. (10-8) the average line length is

$$N = \frac{1.29}{4.00} + \frac{1}{4.00} \cdot 1.29 = 0.1428 \text{ person}$$

For the North American regular ticket counters, which exhibit Poisson arrivals and constant services, since the demand rate is greater than the capacity, we have from Eq. (10-7) with $\rho = 0.9$

$$W = \frac{0.9}{2(0.29)(1 - 0.9)} = 15.52 \text{ min}$$

and from Eq. (10-9)

$$W_e = \frac{15(0.41 - 0.29)}{2(0.29)} = 3.10 \text{ min}$$

Note that the value of k is equal to 1 because the demand was allocated equally among each of the ticketing positions, and the excess delay period T is equal to $0.5(30) = 15 \text{ min}$.

Therefore, $W = 15.52 + 3.10 = 18.62 \text{ min}$ of average delay time per passenger. The average line length is, from Eq. (10-10),

$$N = \frac{0.29}{0.29} + \frac{1}{0.29} \cdot 0.29 = 168.4602 \text{ passengers}$$

Processor	Delay Time		Line Time, min	Length, pers
	Statistical, min	Service Excess, min		
Entrance doors	0.12		0.25	0.48
Regular ticketing				
North American	15.52	3.10	3.45	6.40
Western Pacific	10.98		3.00	4.05
Express check-in				
Western Pacific	0.15		1.00	0.26
Security	2.13		0.50	4.26
Seat selection				
North American	3.38	0.23	0.75	5.80
Western Pacific	0.38		0.50	1.06
Ramp gates				
North American	0.28		0.33	0.84
Western Pacific	0.22		0.33	0.66

TABLE 10-7 Average Passenger Delay Time, Service Time, and Processor Line Lengths in Peak 30 Min for Example Problem 10-1

Computing the delays and queue lengths for each processor results in the values shown in Table 10-7.

From this table it can be seen that considerable delay and line formation exist at the both North American and the Western Pacific regular ticketing facilities, the security check-point, and the North American seat selection facility.

From the passenger splits shown in Fig. 10-25 and the distances between components, the expected values of the passenger delay time, the passenger service time, and the travel time for the average airport passenger in the peak 30 min can be computed.

The expected values are then computed for the delay time, service time, and walking time for the average airport passenger to find the average passenger processing time from Eq. (10-11).

$$E(T_d) = 1.0(0.12) + 0.48(15.52 + 3.10) + 0.34(10.98) + 0.08(0.15) + 1.0(2.13) + 0.53(3.38 + 0.28) + 0.47(0.38 + 0.22)$$

$$E(T_d) = 17.2 \text{ min}$$

$$E(T_s) = 1.0(0.25) + 0.48(3.45) + 0.34(3.00) + 0.08(1.00) + 1.0(0.5) + 0.53(0.75 + 0.33) + 0.47(0.50 + 0.33)$$

$$E(T_s) = 4.5 \text{ min}$$

$$E(D_w) = 0.48(81 + 144) + 0.34(102 + 175) + 0.08(93 + 166) + 0.10(153) + 0.53(55) + 0.47(55)$$

$$E(D_w) = 293 \text{ ft}$$

$$E(T_w) = \frac{293}{1.5(60)} \quad 3.3 \text{ min} = \quad =$$

Time before Scheduled Departure, min	Percentage of Passengers at Airport	Time before Scheduled Departure, min	Percentage of Passengers at Airport
95	0	45	31
90	0	40	40
85	1	35	51
80	2	30	61
75	3	25	71
70	5	20	80
65	8	15	88
60	12	10	94
55	17	5	98
50	23	0	100

TABLE 10-8 Typical Arrival Distribution of Originating Passengers at the Airport for Domestic Flights for Example Problem 10-1

The average passenger processing time for the average airport passenger during the peak 30 min is then

$$E(T_p) = 17.2 + 4.5 + 3.3 = 25 \text{ min}$$

Since departure lounges are only waiting areas, there is no queuing system type delay in these facilities. Generally speaking, these facilities are designed with an area requirement per passenger and visitor in the departure lounge. The area required is based upon the number of passengers and visitors which would be in the departure lounge at the moment the aircraft is allowed to be boarded. Typically the square footage requirement is from 15 to 25 ft² per person in the departure lounge at that point.

Given a typical arrival distribution at the airport, such as that shown in Table 10-8, it is assumed that the total number of passengers and visitors arriving at the airport at the moment the flight is called for boarding is also the number in the departure lounge at that point in time.

In this problem, let us assume that the flight is boarded 15 min before scheduled departure and the airline departure lounge requirements are 20 ft² per passenger.

Therefore, at this time 88 percent of the passengers and their visitors would be in the departure lounge. For Western Pacific Airlines, this means that there are $0.88(0.47)(135)$ passengers and $0.88(0.47)(135)(0.5)(0.5)$ visitors in the departure lounge at this point in time. This totals 56 passengers and 14 visitors. Therefore, the departure lounge size requirements are $20(56 + 14) = 1400 \text{ ft}^2$.

Simulation Models

These models become particularly useful when the analysis of the operation of the passenger handling system is to be performed at a relatively

detailed level or when it is desired to analyze the operation of the system for extended periods of time. They are useful for the analysis of the whole system, as well as of parts of it [28, 34, 39]. When some important inputs to analysis are unobtainable, such as possible future flight schedules, it is possible with the use of computer simulation to analyze the operation of the system under randomly generated inputs.

Simulation is also particularly useful when analysis is to be repeated for varying operating conditions in order to perform sensitivity studies. Computers allow such repeated analysis which would otherwise be prohibitively expensive and very time consuming. Most computer systems have standard simulation packages available which can be adapted to many physical planning problems including airports.

It is important to note that computer simulation is not a substitute for analysis when information on the system is lacking. In order to construct a simulation model nearly as much detailed information about system operation is necessary as for other analytic techniques. The main feature of simulation is the high speed at which computers can perform lengthy calculations. In the analysis of systems operations computer simulation should be used with caution and several runs made so that the statistical reliability of the results may be determined.

Simulation techniques have been studied by the FAA [10, 21, 24] and have been used in many studies to determine facility needs. An outline of the flow of an enplaning passenger through the airport system which can be modeled by the FAA simulation model is given in Fig. 10-27. In the schematic design phase of the planning project at Fort Lauderdale-Hollywood International Airport, simulation was used to determine the parking requirements at the airport and the number of parking toll collection facilities needed at the exit [30]. The inputs into the simulation was the airline flight schedule, the distribution of the total number of parkers relative to the flight schedule, and the historical distribution of parking duration as obtained from the analysis of parking ticket stubs. Alternative flight schedules were used to generate the arrival distribution at the parking facility, and the parking duration distribution, shown in Fig. 10-28, was used to generate the random service times required by the arrivals. As a result of the simulation, it was possible to determine the peaking characteristics of arrivals and departures at the parking garage, and the accumulation of vehicles within the parking facility.

The following steps are recommended in the design of a simulation model for application to airport terminal projects [50]:

1. Define the scope of the simulation in terms of the questions to be answered, the components to be included, and the level of detail required.
2. Specify the required output so that an interpretation of the results will resolve the questions to be addressed.

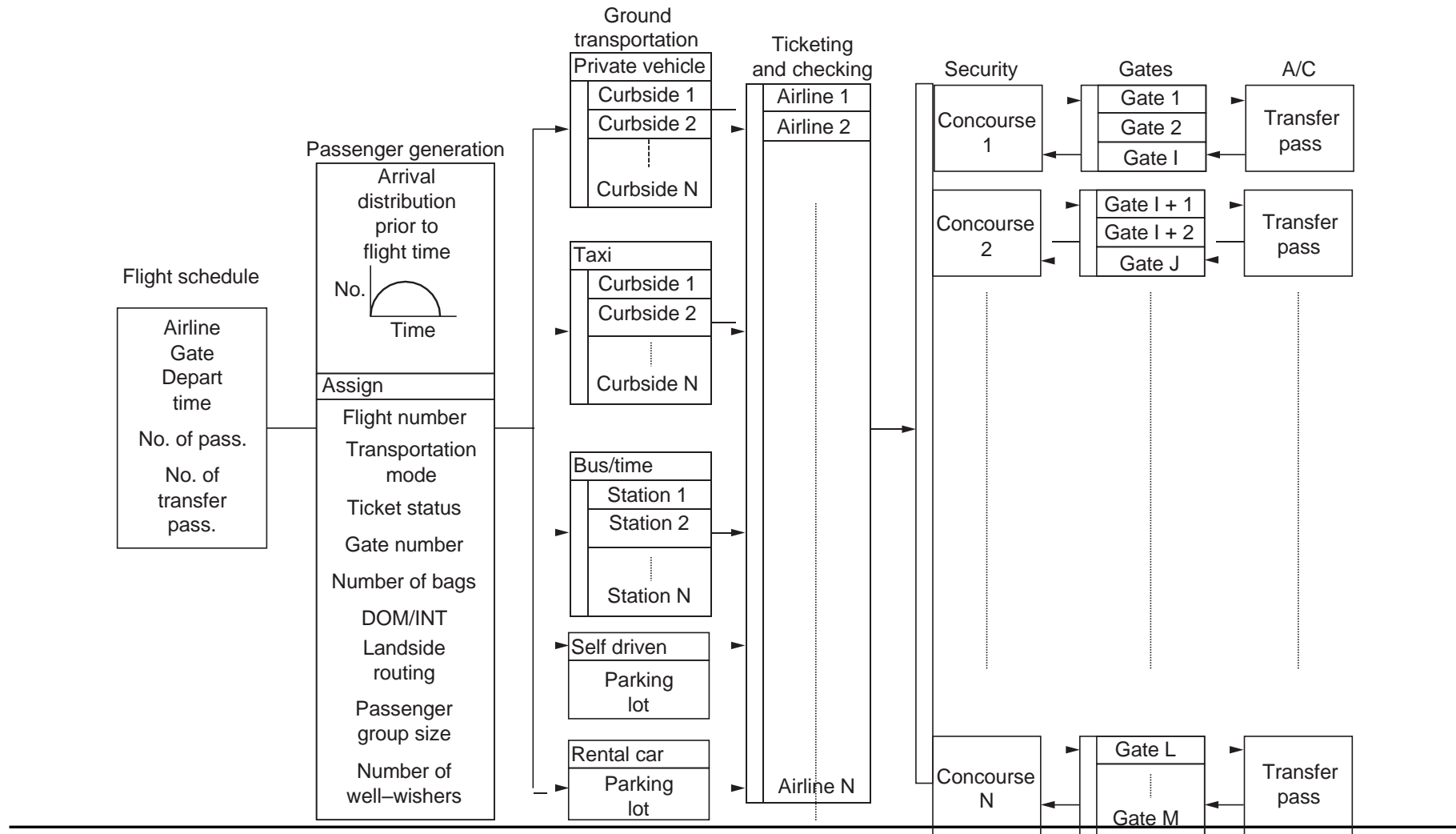


FIGURE 10-27 Flow chart for enplaning passenger simulation model (*Federal Aviation Administration [10]*).

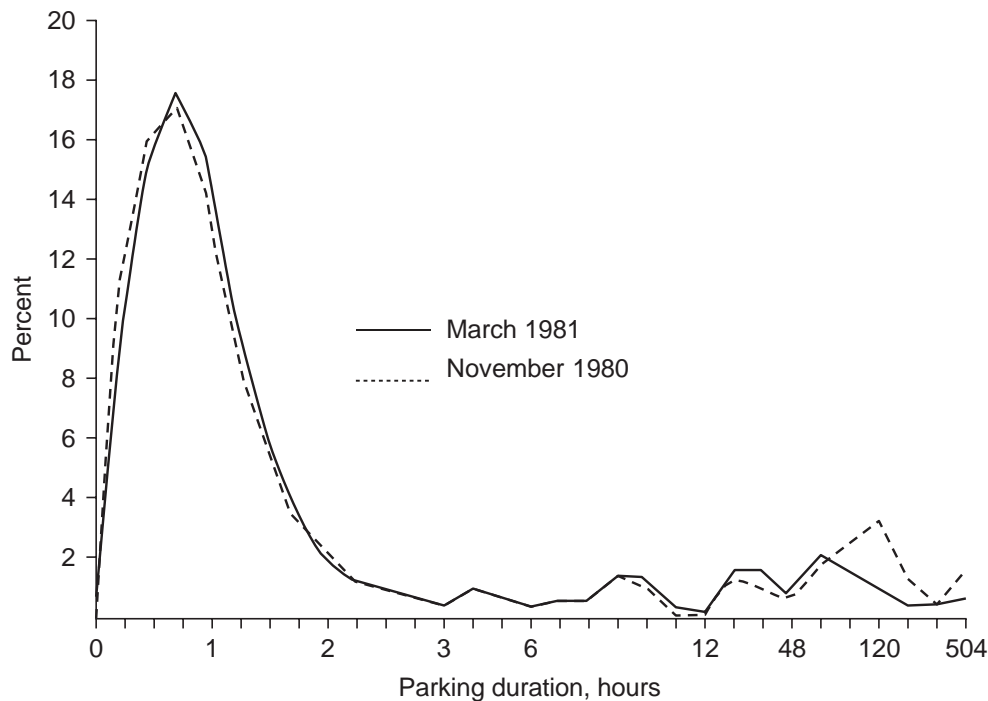


FIGURE 10-28 Parking duration at Fort Lauderdale-Hollywood International Airport (Aviation Planning Associates, Inc. [30]).

3. Structure the model so that the abstract representation of the components in the model and the events and interactions between components are indicative of terminal performance.
4. Define the input data and its variability.
5. Once the model has been developed verify through testing on actual systems.
6. Apply the model and modify the facility design in accordance with the model results.
7. Review the findings and design relative to the degree of variability in the output and through reasonable checks.

Design Development

The final planning phase in terminal projects is called design development. In this phase the size and character of the project is fixed and checked against the findings and recommendations in the prior phases of the project. Acceptance of the project by the airport owner, tenants, and airlines is the final product of this phase. Capital budgeting, operating, maintenance, and administrative costs over the lifetime of the project are determined and a revenue plan is adopted. Agreements are made on rate and charge structures for the airlines, concessionaires, and other tenants. The project moves on toward

implementation through the development of construction documents, bid letting and acceptance, and construction following this phase of planning.

The Apron Gate System

The apron provides the connection between the terminal buildings and the airfield. It includes aircraft parking areas, called *ramps*, and aircraft circulation and taxiing areas for access to these ramps. On the ramp, aircraft parking areas are designated as gates. The discussion in this section is limited to the apron gate area for scheduled commercial aircraft operations. The size of the apron gate area depends on four factors, namely, the number of aircraft gates, the size of the gates, the maneuvering area required for aircraft at gates, and the aircraft parking layout in the gate area. The layout of the apron area is discussed in Chap. 6.

Number of Gates

As in the case with other airport facilities, the number of gates is determined in such a way that a predetermined hourly flow of aircraft can be accommodated. Thus, the number of gates required depends on the number of aircraft to be handled during the design hour and on the amount of time each aircraft occupies a gate. The number of aircraft that need to be handled simultaneously is a function of the traffic volume at the airport. As mentioned earlier, it is customary to use the estimated peak hour volume as the input for estimating the number of gates required at the airport. However, in order to achieve a balanced airport design, this volume should not exceed the capacity of the runways.

The amount of time an aircraft occupies a gate is referred to as the *gate occupancy time*. It depends on the size of aircraft and on the type of operation, that is, a through or turnaround flight. Aircraft parked at a gate are there for passenger and baggage processing and for aircraft servicing and preparation for flight. Larger aircraft normally occupy gates a longer time than small aircraft. This is because large aircraft require more time for aircraft servicing, preflight planning, and refueling. The type of operation also affects gate occupancy time by affecting service requirements. Thus an aircraft on a through flight may require little or no servicing and, consequently, the gate occupancy time can be as low as 20 to 30 min. On the other hand, an aircraft on a turnaround flight will require complete servicing, resulting in gate occupancy times ranging from 40 min to more than 1 h. The table shown in Fig. 10-29 lists the activities that normally take place during a turnaround stop, together with a typical time schedule for these activities.

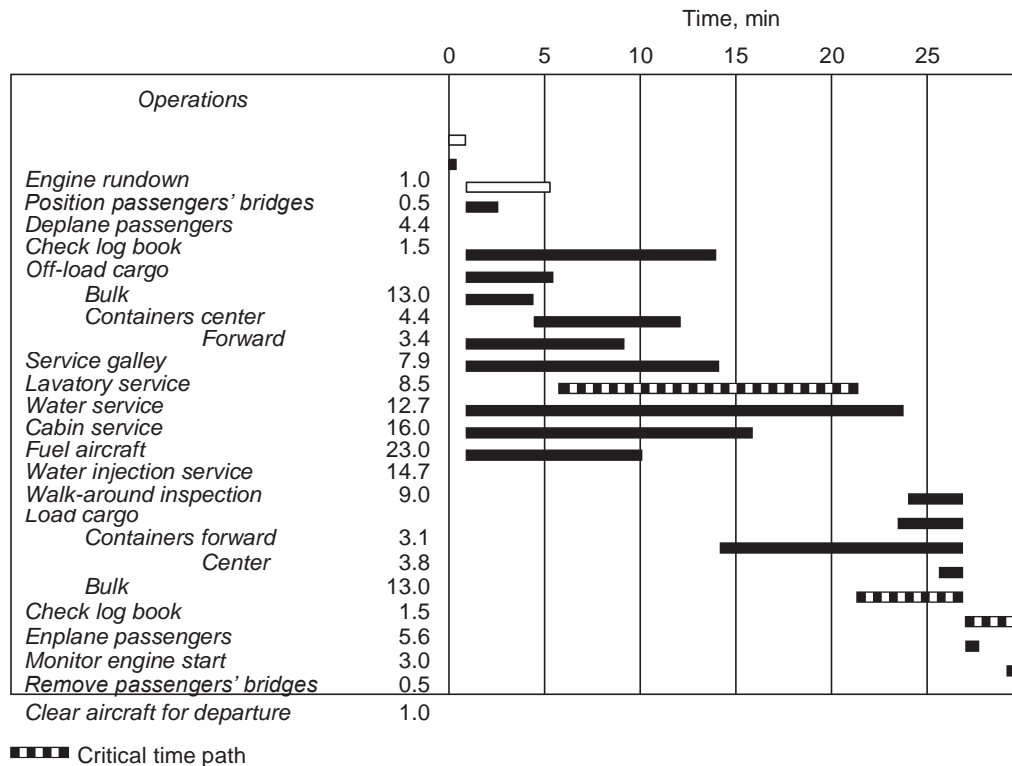


FIGURE 10-29 Typical time schedule of aircraft servicing activities at gate (*Ralph M. Parsons and Federal Aviation Administration [50]*).

If simulation is used a design-day schedule is first forecast and, then based upon the design-day schedule and the practices of the airlines at the airport, a ramp chart based upon the design day is constructed to determine gate requirements.

The average daily gate utilization factor for all gates at an airport usually varies between 0.5 and 0.8. This factor accounts for the fact that it is unlikely that all of the gates available at a terminal building will be used 100 percent of the time. This is caused by the fact that aircraft maneuvering into and out of a gate often blocks other aircraft attempting to move into or out of their gates and by the fact that aircraft schedules often lead to time gaps between the departure of one aircraft and the arrival of another using the same gate. The *gate-use strategy* employed by the airlines at the airport also influences the average gate utilization factor. At airports where gates are used mutually by all airlines, a common gate-use strategy, the gate utilization factor typically varies between 0.6 and 0.8. At airports where groups of gates are used exclusively by different airlines, an exclusive gate-use strategy, the utilization factor drops to about 0.5 or 0.6. The determination of the number of gates needed at an airport should be subjected to the analysis techniques given in Chap. 12 and to the gate-use strategies adopted by the tenant airlines.

An illustration of the use of simulation to generate a design-day schedule from which a ramp chart can be constructed to determine design-day gate requirements is given in Example Problem 10-2.

Example Problem 10-2 The current schedules of airline flights at an airport on a typical day in the peak month are given in Table 10-9. A forecast is made for this airport which indicates that 10 additional flights will be added to the future-year design-day schedule at this airport. It is necessary to perform a simulation for the schedules of these additional flights.

It will be assumed in this problem that the flight schedule of the simulated flights will follow the current day flight schedule distribution.

First it is necessary to obtain a probability distribution for both flight arrival times and gate occupancy times. These are obtained by determining the frequency distribution of each time range for flight arrivals and gate occupancy times and then integrating this frequency distribution to obtain the cumulative probability distribution function.

For simplicity in this problem, the flight arrival time distribution is found by grouping flight arrival times into 1-h increments of time and the gate occupancy

Airline	Flight Number	Time of		Aircraft
		Arrival	Departure	
AE	8/7	7:45 A.M.	9:30 A.M.	727
AE	353	10:30 A.M.	11:15 A.M.	727
AE	319/642	11:30 A.M.	1:00 P.M.	727
AE	421	12:00 P.M.	1:00 P.M.	727
AE	439	1:45 P.M.	2:30 P.M.	727
AE	889	1:45 P.M.	2:30 P.M.	727
AE	852	3:30 P.M.	4:00 P.M.	727
AE	422/660	3:45 P.M.	5:00 P.M.	727
AE	591/544	5:15 P.M.	6:15 P.M.	727
AE	310/390	6:00 P.M.	8:00 P.M.	727
AE	411/428	9:00 P.M.	10:15 P.M.	727
CL	64	7:15 A.M.	7:45 A.M.	737
CL	489	11:15 A.M.	11:45 A.M.	737
CL	41	11:30 A.M.	12:15 P.M.	737
CL	50	1:45 P.M.	2:15 P.M.	737
CL	936	1:45 P.M.	2:15 P.M.	737
CL	81	4:15 P.M.	5:00 P.M.	737
CL	493	8:30 P.M.	9:00 P.M.	737
RX	161	10:15 A.M.	10:45 A.M.	MD8
RX	321/844	4:45 P.M.	5:45 P.M.	MD8

TABLE 10-9 Current Airline Schedule on Typical Day in the Peak Month for Example Problem 10-2

time distribution is found by grouping gate occupancy times into 15 min increments of time.

The flight schedule frequency distribution is then found by counting the number of flights in each 1-h increment of time beginning with 6:00 A.M. and ending with 11:59 P.M. The total number of flights arriving in the periods is shown in Table 10-10.

The cumulative probability distribution function of flight arrival times is then computed by finding the probability that a flight will arrive in a given time period or later beginning with the earliest time period. These values are computed in Table 10-10 and plotted in Fig. 10-30.

By a similar technique, the gate occupancy durations of the flights are grouped as shown in Table 10-11 and the cumulative probability function of these gate occupancy times is plotted in Fig. 10-31.

To simulate a flight arrival and a gate occupancy time for a flight, a table of random digits must be referenced. If this is done, two sets of random numbers, one representing the flight arrival time and one representing the gate occupancy

Time Period		Number of Flights	Arriving in Period or Later	
			Cumulative Number	Cumulative Percentage
From	To			
6:00 A.M.	6:59 A.M.	0	20	1.00
7:00 A.M.	7:59 A.M.	2	20	1.00
8:00 A.M.	8:59 A.M.	0	18	0.90
9:00 A.M.	9:59 A.M.	0	18	0.90
10:00 A.M.	10:59 A.M.	2	18	0.90
11:00 A.M.	11:59 A.M.	3	16	0.80
12:00 P.M.	12:59 P.M.	1	13	0.65
1:00 P.M.	1:59 P.M.	4	12	0.60
2:00 P.M.	2:59 P.M.	0	8	0.40
3:00 P.M.	3:59 P.M.	2	8	0.40
4:00 P.M.	4:59 P.M.	2	6	0.30
5:00 P.M.	5:59 P.M.	1	4	0.20
6:00 P.M.	6:59 P.M.	1	3	0.15
7:00 P.M.	7:59 P.M.	0	2	0.10
8:00 P.M.	8:59 P.M.	1	2	0.10
9:00 P.M.	9:59 P.M.	1	1	0.05
10:00 P.M.	10:59 P.M.	0	0	0.00
11:00 P.M.	11:59 P.M.	0	0	0.00

TABLE 10-10 Aircraft Arrival Distribution for Example Problem 10-2

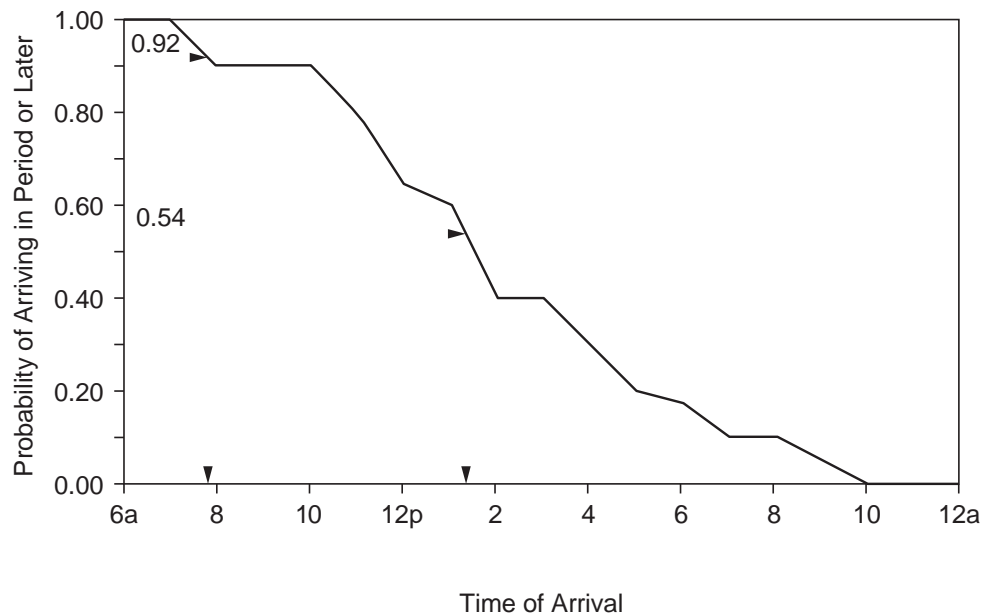


FIGURE 10-30 Flight arrival distribution for Example Problem 10-2.

time, for each of the 10 flights to be simulated are found and these are shown in Table 10-12.

The first simulated flight has an arrival time random number of 0.92 and a gate occupancy time random number of 0.70. Using the flight arrival time

Gate Occupancy Time Period, Min		Number of Flights	Duration Range or More	
From	To		Cumulative Number	Cumulative Percentage
0	14	0	20	1.00
15	29	0	20	1.00
30	44	7	20	1.00
45	59	5	13	0.65
60	74	3	8	0.40
75	89	2	5	0.25
90	104	1	3	0.15
105	119	1	2	0.10
120	134	1	1	0.05
135	149	0	0	0.00
150	164	0	0	0.00
165	179	0	0	0.00
180	194	0	0	0.00

TABLE 10-11 Aircraft Gate Occupancy Time Distribution for Example Problem 10-2

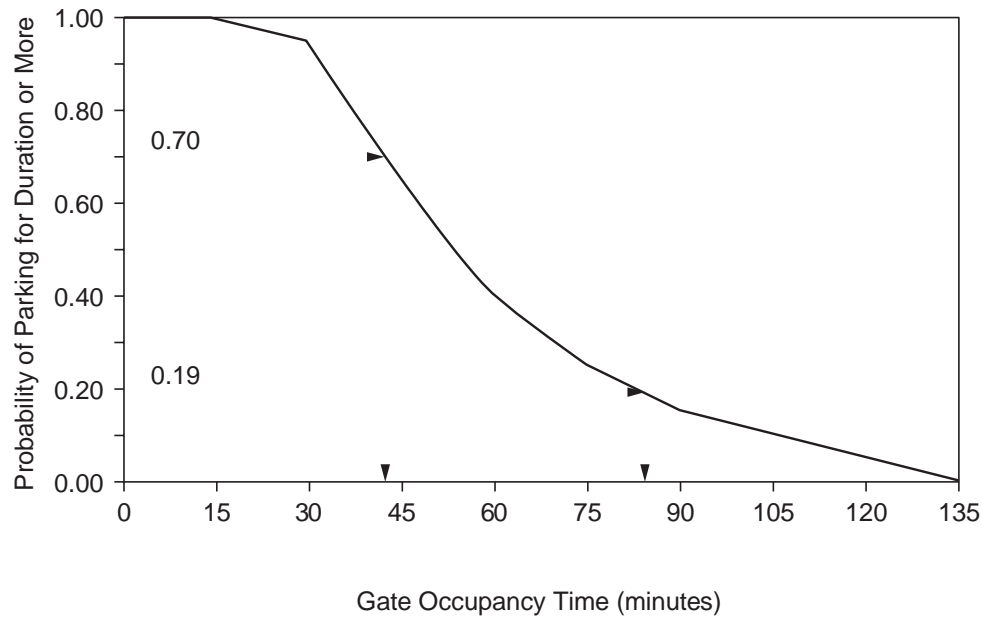


FIGURE 10-31 Flight gate occupancy duration for Example Problem 10-2.

random number of 0.92 with the cumulative probability function of flight arrival times in Fig. 10-30, we find that the simulated arrival time of the first flight is 7:45 A.M. Using the gate occupancy time random number of 0.70 with the cumulative probability function of gate occupancy times in Fig. 10-31, we find that the simulated gate occupancy time of the first flight is 45 min. Both of these numbers are taken to the nearest 15 min time increment for simplicity. Therefore, the first simulated flight has an arrival time of 7:45 A.M. and a flight departure time of 8:30 A.M.

Simulated Flight	Aircraft Arrivals		Gate Occupancy Time		
	Random Number	Arrival Time	Random Number	Duration Time	Departure Time
1	0.92	7.45	0.70	45	8.30
2	0.88	10.15	0.80	45	11.00
3	0.54	1.15	0.19	90	2.45
4	0.18	5.30	0.62	45	4.15
5	0.82	10.45	0.43	60	11.45
6	0.87	10.15	0.86	30	10.45
7	0.75	11.15	0.46	60	12.15
8	0.28	4.15	0.69	45	5.00
9	0.59	1.00	0.87	30	1.30
10	0.60	1.00	0.93	30	1.30

TABLE 10-12 Gate Simulation Results for Example Problem 10-2

This process is continued for each simulated flight. For example, the third simulated flight has an arrival time random number of 0.54 and a gate occupancy time random number of 0.19. Again, using the flight arrival time random number of 0.54 with the cumulative probability function of flight arrival times in Fig. 10-30, we find that the simulated arrival time of the third flight is 1:15 P.M. Using the gate occupancy time random number of 0.19 with the cumulative probability function of gate occupancy times in Fig. 10-31, we find that the simulated gate occupancy time of the third flight is 90 min. Therefore, the third simulated flight has an arrival time of 1:15 P.M. and a flight departure time of 2:45 P.M.

The above simulation process is shown by the dashed lines in Figs. 10-30 and 10-31.

Assuming a simulation was also performed for the airline and the aircraft for each flight, the results of the simulation are tabulated in Table 10-12. The new flight schedule in the design year, with both the current flights and the simulated flights, is shown in Table 10-13. In this table the simulated flights are each given the flight number 9999 for reference purposes.

Ramp Charts

A ramp chart is a graphical representation of the gate occupancy by aircraft throughout the day. Airlines and airports use ramp charts to display the actual gate assignment of aircraft for the flight schedule at the airport.

The ramp chart can also be used to determine the gate requirements at an airport. When it is used for this purpose the ramp charts does not display the actual assignment of aircraft to specific gates for the design-day schedule but only indicates the assignment of aircraft to gates for the determination of the number of gates required at the airport.

There are several factors which influence the gate requirements at an airport. Obviously the flight schedule and the gate occupancy time of aircraft are of paramount importance. However, the scheduling and ramp operating practices of the airlines and the gate-use strategy of the airlines are also important. The scheduling and ramp operating practices give rise to the fact that a gate cannot be used 100 percent of the time as discussed earlier. The gate-use strategy considers whether the gates will be exclusive-use gates, shared-use gates or common gates. Exclusive-use gates are gates which are reserved for the exclusive use of one airline. A shared gate is a gate which is shared by two or three airlines. A common-use gate is a gate which is allocated by the airport based upon the demand for gates and may be used by any airline at the airport.

Gates are sized based upon the geometric properties of the aircraft that will occupy the gates. Therefore, gates may be called *wide-bodied gates* because they are sized to accommodate wide-bodied aircraft. These gates may also be used by narrow-bodied aircraft. Narrow-bodied gates are gates which can only be used by narrow-bodied aircraft. Many airports also have commuter gates which are sized to accommodate commuter aircraft.

Ref No.	Airline	Flight Number	Time of		
			Arrival	Departure	Aircraft
1	AE	8/7	7:45 A.M.	9:30 A.M.	727
2	AE	9999	10:15 A.M.	10:45 A.M.	727
3	AE	9999	10:15 A.M.	11:00 A.M.	727
4	AE	353	10:30 A.M.	11:15 A.M.	727
5	AE	9999	10:45 A.M.	11:45 A.M.	727
6	AE	319/642	11:30 A.M.	1:00 P.M.	727
7	AE	421	12:00 P.M.	1:00 P.M.	727
8	AE	9999	1:00 P.M.	1:30 P.M.	727
9	AE	9999	1:00 P.M.	1:30 P.M.	727
10	AE	439	1:45 P.M.	2:30 P.M.	727
11	AE	889	1:45 P.M.	2:30 P.M.	727
12	AE	852	3:30 P.M.	4:00 P.M.	727
13	AE	422/660	3:45 P.M.	5:00 P.M.	727
14	AE	9999	4:15 P.M.	5:00 P.M.	727
15	AE	591/544	5:15 P.M.	6:15 P.M.	727
16	AE	9999	5:30 P.M.	6:15 P.M.	727
17	AE	310/390	6:00 P.M.	8:00 P.M.	727
18	AE	411/428	9:00 P.M.	10:15 P.M.	727
19	CL	64	7:15 A.M.	7:45 A.M.	737
20	CL	9999	7:45 A.M.	8:30 A.M.	737
21	CL	489	11:15 A.M.	11:45 A.M.	737
22	CL	9999	11:15 A.M.	12:15 P.M.	737
23	CL	41	11:30 A.M.	12:15 P.M.	737
24	CL	9999	1:15 P.M.	2:45 P.M.	737
25	CL	50	1:45 P.M.	2:15 P.M.	737
26	CL	936	1:45 P.M.	2:15 P.M.	737
27	CL	81	4:15 P.M.	5:00 P.M.	737
28	CL	493	8:30 P.M.	9:00 P.M.	737
29	RX	161	10:15 A.M.	10:45 A.M.	MD8
30	RX	321/844	4:45 P.M.	5:45 P.M.	MD8

TABLE 10-13 Simulated Airline Schedule on Typical Day in the Peak Month in the Design Year for Example Problem 10-2

The determination of the number of aircraft gates required at an airport is shown by constructing ramp charts in Example Problem 10-3.

Example Problem 10-3 Let us determine the number of gates required at an airport based upon the design-day schedule shown in Table 10-13. Let us determine the gate requirements under an exclusive gate-use strategy, a shared gate-use strategy, and a common gate-use strategy. Let us assume that the scheduling and operating practices of the airlines require that a minimum time gap of 15 min must be allowed between the departure of a scheduled flight from a gate and the arrival of the next scheduled flight at that gate.

Under an exclusive gate-use strategy each airline will have its own gates which cannot be used by any other airline. To find the number of gates required under this gate-use strategy, a graph is constructed showing time on the X axis and the number of gates on the Y axis. To determine the minimum number of gates required, the flight schedule of each airline must be sorted first by arrival and then departure time. That is, two flights which have the same arrival time are also sorted by the departure time.

In Table 10-13, the flight schedule of each airline has been sorted and a reference number is placed next to each flight for each airline. This reference number will be placed on the ramp chart for illustrative purposes.

To determine the gate required for Alpha Express (AE) Airlines, since the first scheduled flight is scheduled to arrive at 7:45 A.M. and scheduled to depart at 9:30 A.M., a gate is opened on the ramp chart and the block of time from 7:45 A.M. to 9:30 A.M. is filled in on the ramp chart to indicate that the gate is occupied during that period of time. The next scheduled flight by AE Airlines is scheduled to arrive at 10:15 A.M. and depart at 10:45 A.M. Since at the gate just opened up the next scheduled arrival can be placed at this gate 15 min after the departure of the previous aircraft, this flight may be scheduled into that gate and the block of time from 10:15 A.M. to 10:45 A.M. is filled in. The next scheduled flight of AE Airlines is scheduled to arrive at 10:15 A.M. and depart at 11:00 A.M. However, this flight cannot be scheduled into the first gate since that gate is occupied during part of that period of time.

Therefore, a new gate is opened up on the ramp chart to accommodate this aircraft and the block of time from 10:15 A.M. to 11:00 A.M. is filled in at that gate. This process is continued until all of the AE Airline flights have been assigned to gates. As may be seen on the top portion of Fig. 10-32, to accommodate the flights of AE Airlines four gates are required.

The same process is repeated for each airline, and as shown in the middle and bottom portions of Fig. 10-32, this results in three gates being required for Coastal Link (CL) Airlines and one gate being required for Regional Express (RX) Airlines.

Therefore, under an exclusive gate-use strategy, this flight schedule requires eight gates to accommodate the airline schedule at the airport.

For the construction of a shared gate-use strategy ramp chart, let us assume that CL and RX Airlines will share gates at the airport and that AE Airlines will have exclusive-use gates. Therefore, the ramp chart for AE Airlines does not change but the ramp chart for the shared-gate-use airlines must be constructed by first sorting all of the flights of CL and RX Airlines together by arrival time and departure time as shown in Table 10-14. Using the procedure above, Fig. 10-33 is the required ramp chart. It is seen that though AE Airlines requires four gates, RX and CL Airlines now only require at total of three gates. Therefore, the shared-gate-use strategy results in one less gate at the airport to accommodate the design-day flight schedule.

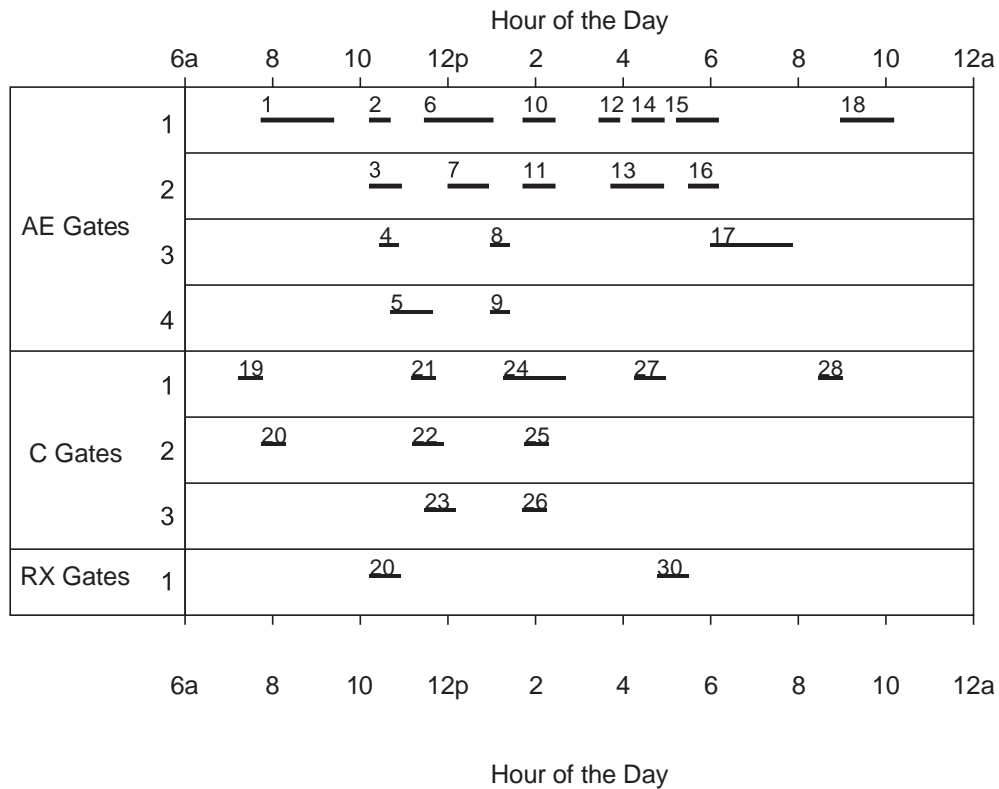


FIGURE 10-32 Ramp chart for exclusive gate use for Example Problem 10-3.

Ref No.	Flight Airline	Number	Time of		
			Arrival	Departure	Aircraft
1	CL	64	7:15 A.M.	7:45 A.M.	737
2	CL	9999	7:45 A.M.	8:30 A.M.	737
3	RX	161	10:15 A.M.	10:45 A.M.	MD8
4	CL	489	11:15 A.M.	11:45 A.M.	737
5	CL	9999	11:15 A.M.	12:15 P.M.	737
6	CL	41	11:30 A.M.	12:15 P.M.	737
7	CL	9999	1:15 P.M.	2:45 P.M.	737
8	CL	50	1:45 P.M.	2:15 P.M.	737
9	CL	936	1:45 P.M.	2:15 P.M.	737
10	CL	81	4:15 P.M.	5:00 P.M.	737
11	RX	321/844	4:45 P.M.	5:45 P.M.	MD8
12	CL	493	8:30 P.M.	9:00 P.M.	737

TABLE 10-14 Simulated Airline Schedule on Typical Day in the Peak Month in the Design Year for CL and RX Airlines for Shared-Gate-Use Strategy for Example Problem 10-3

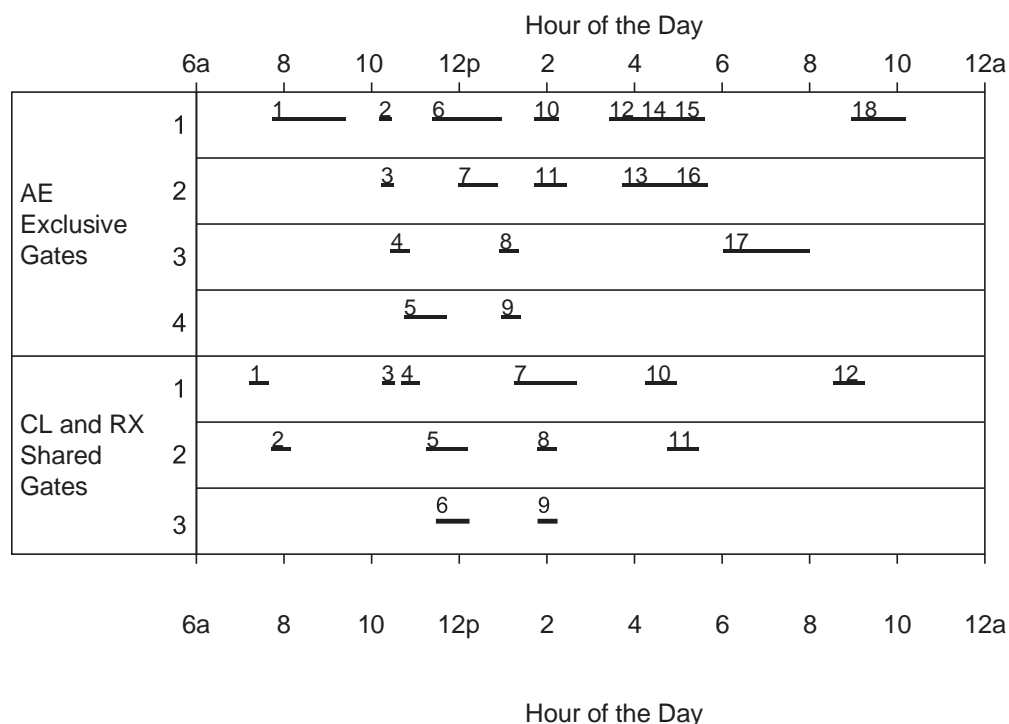


FIGURE 10-33 Ramp chart for shared gate use for Example Problem 10-3.

For the common use-gate strategy, all of the flights of all of the airlines are sorted together by arrival time and departure time and the ramp chart similarly constructed. If this is done, the ramp chart in Fig. 10-34 results which shows that under a common gate-use strategy only five gates are required at the airport to accommodate the design-day schedule.

The peak hour for gate use occurs around 1:00 P.M. and the peak hour gate utilization is defined as the gate time demanded divided by the gate time supplied in the peak hour. The aircraft demand 210 min of gate time in the peak hour and the gate time supplied is 480 min, 420 min, or 300 min for the exclusive, shared and common gate-use strategies, respectively. Therefore, the peak hour gate utilization is 0.44, 0.50, or 0.70 by each of the gate-use strategies.

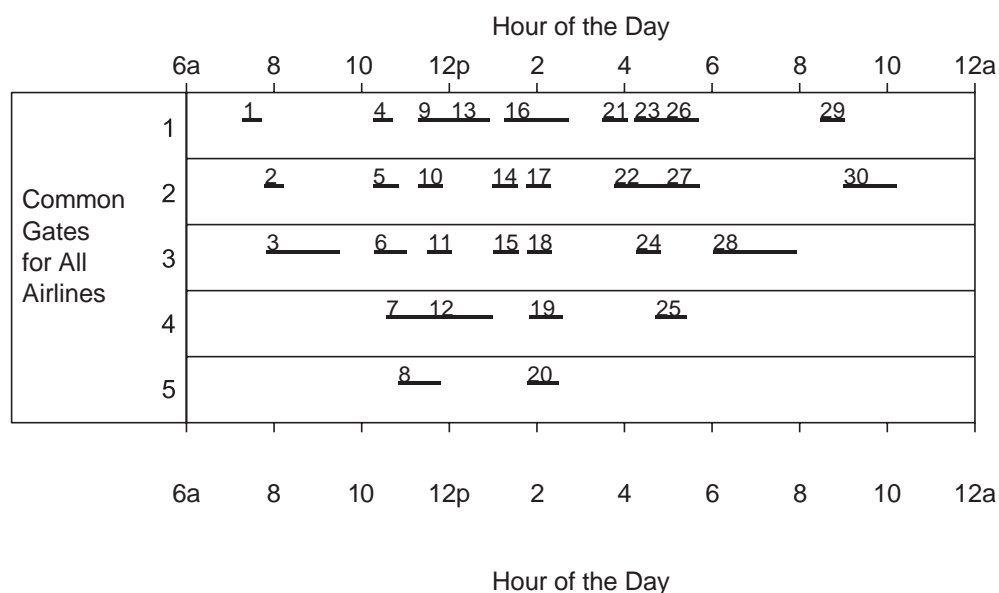


FIGURE 10-34 Ramp chart for common gate use for Example Problem 10-3.

All of the aircraft on the design-day schedule in this are narrow-bodied aircraft. If some are wide-bodied aircraft, the sorting would be done separately for the wide-bodied and narrow-bodied aircraft. To minimize the number of wide-bodied gates, the wide-bodied aircraft would be assigned to the ramp chart first and then the narrow-bodied aircraft would be assigned, recognizing that a narrow-bodied aircraft may use a wide-bodied gate.

Gates at most airports vary within the range of three to five gates per million annual passengers. The total number of gates may have to be modified if not all gates can handle all types of aircraft. This is particularly important at airports where the aircraft mix includes a considerable amount of large and small aircraft. In such situations, and when data are available, it would be preferable to compute gate requirements separately for the different types of aircraft, keeping in mind that the large gates can be used to handle small aircraft, while the reverse is not true. It is also desirable to calculate the gate requirements separately for different types of traffic. For example, at a large international airport separate calculations may be performed for domestic gates, for international gates, and for charter gates.

Gate Size

The size of a gate depends not only on the size of aircraft which it is to accommodate and but also on the type of parking used, that is, nose-in, parallel, or angled parking. The size of the aircraft determines the space required for parking as well as for maneuvering. Furthermore, the size of aircraft determines the extent and size of the servicing equipment that needs to be provided to service the aircraft. The type of parking used at the gates affects the size since the area required to maneuver into and out of a gate varies depending on the way the aircraft is parked.

In view of the large number of factors that affect the size and exact layout of gates, it is desirable to consult the airlines at an early stage in the design process in order to determine the manner in which they plan to maneuver aircraft and the types of servicing facilities they plan to use.

The design of the gates can be worked out with the aid of procedures and dimensions provided by the FAA [6, 7], ICAO [3], and the International Air Transport Association [12]. Included in these references are diagrams which show various dimensions required for different types of aircraft and various parking and maneuvering conditions. Chapter 6 discusses the layout of ramp areas to accommodate aircraft.

While detailed design of aircraft gates requires the aid of charts such as those found in the airplane characteristics manuals published for aircraft, it is usually sufficient for preliminary planning to adopt uniform dimensions between centers of gates and to use these for sizing the apron gate area. The dimensions depend on the type of aircraft. The typical dimensions for the case where aircraft enter a gate under their own power and are pushed out by a tractor are given in Table 10-15.

		Push-Out			Taxi-Out*		
	Aircraft Group	<i>L</i>	<i>W</i>	Area yd ²	<i>L</i>	<i>W</i>	Area yd ²
A	FH-227	103 ft 1 in	115 ft 2 in	1319	148 ft 10 in	140 ft 2 in	2318
	YS-11B	106 ft 3 in	124 ft 11 in	1474	171 ft 0 in	149 ft 11 in	2850
	BAC-111	123 ft 6 in	113 ft 6 in	1557	130 ft 0 in	138 ft 6 in	2001
	DC-9-10	134 ft 5 in	109 ft 5 in	1634	149 ft 2 in	134 ft 5 in	2228
B	DC-9-21,30	149 ft 4 in	113 ft 4 in	1880	149 ft 0 in	138 ft 4 in	2290
	727 (all)	173 ft 2 in	128 ft 0 in	2463	194 ft 0 in	153 ft 0 in	3298
	737 (all)	120 ft 0 in	113 ft 0 in	1507	145 ft 4 in	138 ft 0 in	2228
C	B-707 (all)	172 ft 11 in	165 ft 9 in	3188	258 ft 0 in	190 ft 9 in	5468
	B-720	156 ft 9 in	150 ft 10 in	2627	228 ft 0 in	175 ft 10 in	4454
	DC-8-43, 51	170 ft 9 in	162 ft 5 in	3081	211 ft 10 in	187 ft 5 in	4411
D	DC-8-61, 63	207 ft 5 in	168 ft 5 in	3882	252 ft 4 in	193 ft 5 in	5423
E	L-1011	188 ft 8 in	175 ft 4 in	3676	263 ft 6 in	200 ft 4 in	5865
	DC-10	192 ft 3 in	185 ft 4 in	3959	291 ft 0 in	210 ft 4 in	6801
F	B-747	241 ft 10 in	215 ft 8 in	5795	328 ft 0 in	240 ft 8 in	8771

**L* = perpendicular to face of building; *W* = parallel to face of building.

Source: Federal Aviation Administration [50].

TABLE 10-15 Comparison of Apron Parking Envelope Dimensions for Aircraft Push-out and Taxi-out Gate Use for Nose-In Configuration

Aircraft Parking Type

Aircraft parking type refers to the manner in which the aircraft is positioned with respect to the terminal building and to the manner in which aircraft maneuver in and out of parking positions. It is an important factor affecting the size of the parking positions and consequently the apron gate area. Aircraft can be positioned at various angles with respect to the terminal building line and can maneuver into and out of parking positions either under their own power or with the aid of towing equipment. With aircraft towing it is possible to reduce the size of parking positions. It is advisable in choosing among alternative parking types to consult with the airline in question, as different airlines have different preferences for the available systems. It is also advisable in adopting a parking type to take into consideration the objective of protecting passengers from the adverse elements of noise, jet blast, and weather, and the operating and maintenance costs of needed ground equipment.

The aircraft parking types which have been successfully used at a variety of airports and should be evaluated in any airport planning study include nose-in, angled nose-in, angled nose-out, and parallel. These parking types are shown in Fig. 10-35 and are discussed separately below.

Nose-In Parking

In this configuration the aircraft is parked perpendicular to the building line with the nose as close to the building as permissible. The aircraft maneuvers into the parking position under its own power. In order to leave the gate, the aircraft has to be towed out a sufficient distance to allow it to proceed under its own power. The advantages of this configuration are that it requires the smallest gate area for a given aircraft, causes lower noise levels as there is no powered turning movement near the terminal building, sends no jet blast toward the building, and facilitates passenger loading as the nose is near the building. Its disadvantages include the need for towing equipment

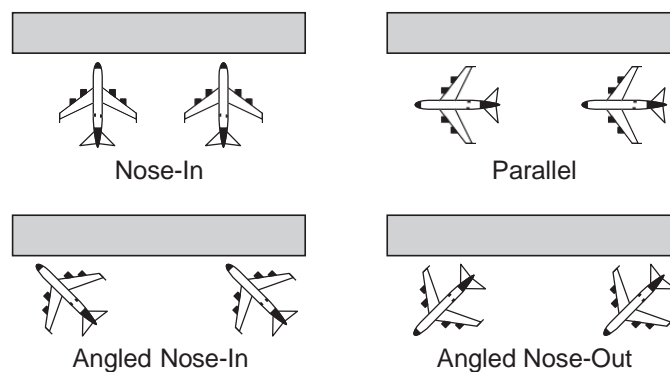


FIGURE 10-35 Aircraft parking types.

and the nose is too far from the building to effectively use the rear doors for passenger loading.

Angled Nose-In Parking

This configuration is similar to the nose-in configuration except that the aircraft is not parked perpendicular to the building. The configuration has the advantage of allowing the aircraft to maneuver in and out of the gate under its own power. However, it requires a larger gate area than the nose-in configuration and causes a higher noise level.

Angled Nose-Out Parking

In this configuration the aircraft is parked with its nose pointing away from the terminal building. Like the angled nose-in configuration, it has the advantage of allowing aircraft to maneuver in and out of gate positions without towing. It does require a larger gate area than the nose-in position, but less than the angled nose-in. A disadvantage of this configuration is that the breakaway jet blast and noise are pointed toward the building when the aircraft starts its taxiing maneuver.

Parallel Parking

This configuration is the easiest to achieve from the aircraft maneuvering standpoint. In this case noise and jet blast are minimized, as there are no sharp turning maneuvers required. It does require, however, a larger gate position area, particularly along the terminal building frontage.

It is evident that no one parking type can be considered ideal. For any planning situation, all the advantages and disadvantages of the different systems have to be evaluated, taking into consideration the preference of the airline that will be using the gates.

Apron Layout

Another factor that affects apron size and installation requirements is the apron layout. This refers to the manner in which the apron is arranged around the terminal building. The apron layout depends directly on the way the aircraft gate positions are grouped around the buildings and on the circulation and taxiing patterns dictated by the relative locations of the terminal buildings and the airfield system.

Aircraft are grouped adjacent to the terminal building in a variety of ways depending on the horizontal terminal concept used. These groupings are referred to as parking systems and are classified as the frontal or linear system, the finger or pier system, the satellite system, and the open apron or transporter system. Each of these were discussed and illustrated earlier.

The choice of aircraft parking system is, naturally, strongly influenced by the horizontal passenger processing concept adopted. For each there are positive and negative attributes that must be weighed

against each other. While the open apron system has the advantage of separating the aircraft and the terminal building from one another, it does require buses or mobile lounges for the conveyance of passengers between them. These vehicles use the apron, and their circulation patterns need careful planning to avoid interference with the flow of aircraft and other service vehicles. While the finger system allows the efficient expansion of gate positions and the efficient use of terminal building space, it may lead to long passenger walking distances if allowed to become excessively long. The frontal system is suitable for the gate arrival processing concept. Other features of these systems were discussed earlier in this chapter.

Apron Circulation

In designing the apron layout it is important to take into account aircraft circulation, particularly the movement of aircraft within the apron gate area and from this area to the taxiways. When the traffic volume is high, it is desirable to provide a taxilane on the periphery of the apron. It is also important to allow sufficient space to permit easy access of aircraft to gates. This is particularly important when pier fingers are used for aircraft parking and the fingers are parallel to each other. Sufficient space must be provided between the fingers to allow aircraft ready access to the gates. The separation between fingers depends on their length and on the size of aircraft to be accommodated. The longer the finger, the more aircraft gates can be accommodated. However, the increase in the number of gates may necessitate the provision of two taxilanes instead of one between the fingers to provide circulation without excessive delay. One taxilane will probably suffice when there are no more than five or six gates on each side of a finger. A large number of gates may require two taxilanes.

Passenger Conveyance to Aircraft

Depending on the passenger processing system used, the type of aircraft parking, and the parking system layout, any of three methods of conveyance can be used between the building and the aircraft. These are walking on the apron, walking through aircraft building connectors such as passenger loading bridges, and by mobile conveyance using any of a variety of apron vehicles.

The first method can be employed with all processing and parking systems. However, as the number of parking positions and the apron size increases, it becomes impractical to use walking for the conveyance of passengers. The economic appeal of this method is overcome by the need to protect the passengers from the elements and from the hazards of walking on the apron.

The second method can be employed for all systems other than where open apron parking is used. A variety of fixed and movable loading systems have been developed for passenger conveyance.

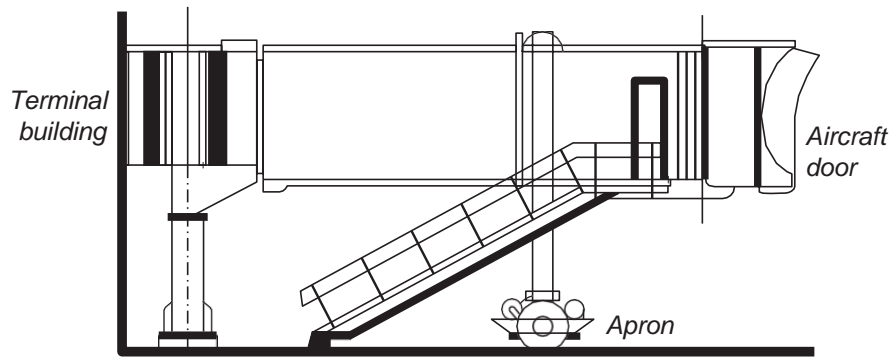


FIGURE 10-36 Typical aircraft loading bridge.

Most common among these are the nose bridges, which are short connectors suitable for use when the aircraft door comes close to the building such as with nose-in parking. Another common system is the telescoping loading bridge. These have the flexibility of extending from the building to reach the aircraft door and of swinging to accommodate different types of aircraft. A typical boarding bridge is shown in Fig. 10-36.

Apron Utility Requirements

Aircraft need to be serviced at their respective gates. Thus certain fixed installations may be required on the apron. Apron congestion is always a problem and, hence, there is a definite trend at larger airports toward replacing mobile servicing equipment with fixed facilities.

Aircraft Fueling

Aircraft are fueled at the apron by fuel trucks, fueling pits, and hydrant systems. At the smaller and even the larger airports the use of fuel trucks is prevalent, but the pattern is changing in favor of the hydrant system at airports requiring large amounts of fuel.

The principal advantage of fuel trucks is their flexibility. Aircraft can be fueled anywhere on the apron, the units can be added or taken away according to need, and the system is relatively economical insofar as airport management and airline operations are concerned. There are, however, disadvantages associated with the use of fuel trucks. Large jet transports require a considerable amount of fuel, from nearly 8000 gal (U.S.) for the McDonnell Douglas MD-88 to almost 50,000 gal (U.S.) for the Boeing 747-100. Two refueler units are normally required, one under each wing. For the large jets, standby units are sometimes required if the fuel requirements are in excess of two units. This means that there are a large number of vehicles on the apron during peak periods, creating a potential hazard of collision with personnel, other vehicles, and aircraft. Since each truck carries a considerable quantity of fuel, it also constitutes a potential fire hazard when moving around on an apron where a

number of other activities are taking place. Trucks are large and awkward and take up valuable space in the operations area. When a truck is empty, it must return to the storage area for refueling before it can be used again. Thus extra trucks must be provided for use during the time when other trucks are being reloaded. When refueling trucks are not in use, parking space must be provided for these vehicles. Modern refuelers are approximately 40 ft in length and weigh as much as 83,000 lb. The capacity of the larger trucks is approximately 8000 gal. For the larger refuelers axle loads are in excess of the legal limits on highways, and, consequently, the airport designer must provide adequate pavement strengths to support these vehicles.

The hydrant system is used at most large airports. In this system, a large fuel storage area, often called a *fuel farm*, is located on the airport property. Fuel is transferred from the fuel storage area to aircraft gate positions through a system of pipes located below the pavement surface. A special valve is mounted in a box in the pavement flush with the pavement surface at each gate position. A special vehicle, a hydrant dispenser, with a hose, meter, filter, and air eliminator is used to connect the fuel supply to the aircraft. One end of the hose has a specially designed valve which is coupled to the valve installed in the pavement. This hose feeds into the meter, filter, and air eliminator, from which another hose, usually on a reel, is led to the fuel intakes on the aircraft.

The principal advantages of the hydrant system are that a continuous supply of fuel is available at the gates, it is safely carried underground, and fuel trucks are eliminated from the apron. The principal disadvantage is that vehicles are not entirely removed from the apron. However, because of their small size, hydrant dispensers reduce possible collision damage to a minimum.

The amounts of fuel required at many airports are so large that, regardless of the type of fueling system used, a central fuel storage area in the vicinity of the landing area is required. If the hydrant system is used, provision must be made for installing pipes from the storage area to the apron.

The location of the hydrant valves at an individual gate will depend upon the location of the fueling connections in the wings of the aircraft occupying the gates. It is desirable that the hose line from the hydrant dispenser to the intakes in the wings not exceed 20 to 30 ft. If a wide variety of aircraft are to be serviced at a gate position, the precise spacing of the hydrant valves should be established in consultation with the airlines. The number of hydrants required per gate position depends not only on the type of aircraft but also on the number of grades of fuel required. Each grade of fuel requires a separate layout.

At a number of airports, hydrant systems are installed by oil companies which contract for fuel with a particular airline or airlines. It is not uncommon to have the hydrant system and fuel trucks operate

simultaneously at the same airport. The trend at the large airports is definitely toward the hydrant system.

Electrical Power

Electrical power is required on the apron for the servicing of aircraft prior to engine starting. External electrical power is also often required for starting the engines. Power requirements vary widely for different aircraft. Consequently, it is necessary to consult with the airlines concerning this matter. Power can be supplied by mobile units or by fixed installations in the pavement. The latter is preferable since it removes the need for a vehicle and to some extent reduces noise which emanates from a motor generator set. For a fixed installation, the most satisfactory technique is to bury conduits under the apron, terminating them at supply points some distance from the hydrant valves but convenient to the aircraft.

Recently, there has been a trend toward fixed ground power and air conditioning systems using terminal power sources. The need for these facilities has grown due to the costs of providing power and conditioned air to aircraft during servicing times at the apron gate by using the power generated by the auxiliary power unit on the aircraft. Considerable operating cost economies have been reported in the use of such systems [27].

Aircraft Grounding Facilities

Grounding facilities will be required on the apron to provide protection of parked aircraft and fuel trucks from static discharge, particularly during fueling operations. The location of the grounding facility will be governed by the location of the hydrant valves. With high fueling rates it is essential that grounding facilities be provided.

Apron Lighting and Marking

Adequate lighting and marking are essential on an apron. Wherever possible, each gate should be floodlighted. Floodlighting removes the need for mobile equipment to use headlights, which experience has shown to cause confusion and glare. A system of elevated lights appears to offer the best method of providing apron illumination. Where pier fingers are utilized, the lights can be attached to the fingers. Lighting should be located so as to provide uniform illumination of the apron area yet not cause glare to the pilot.

When personnel are servicing an aircraft, there is a need for lighting its underside and far side, if the floodlights do not provide the necessary illumination. This can be accomplished by installing flush lights in the pavement. When lights of this type are installed, they should be arranged so as not to confuse the pilot insofar as guidance to the gate position is concerned.

Painted guidelines have proved very desirable as aids to maneuvering aircraft accurately on the apron. The best guide appears to be

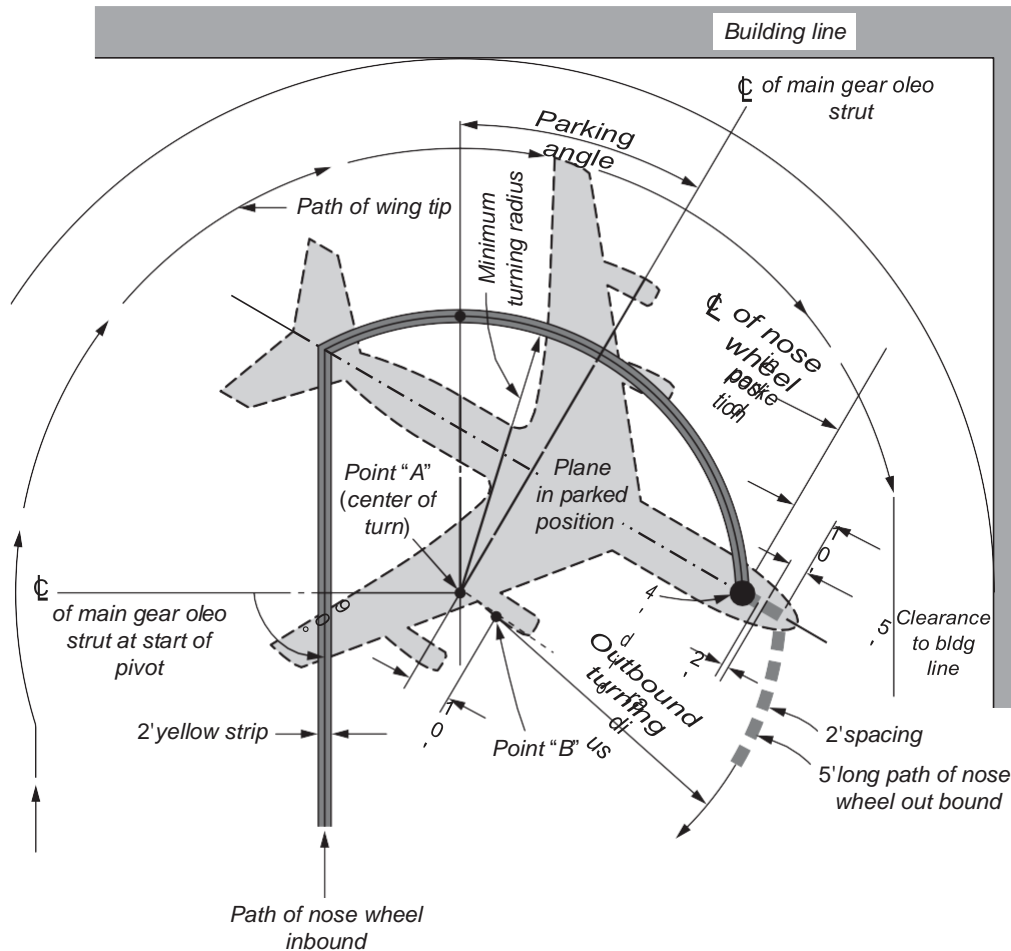


FIGURE 10-37 Typical painted guidelines at gate positions.

a single line, usually the color is yellow, which is followed by the nose gear of the aircraft. A typical layout is shown in Fig. 10-37. It is recognized that a single line will not provide precise guidance for a variety of different aircraft. Usually the guideline is painted for the most critical aircraft using a particular gate position. Smaller aircraft can use the same lines and maneuver without difficulty, especially if personnel on the ground are available to direct the pilot. Because of possible fuel spillage, it is desirable to paint guidelines with special resistant paint in areas where spillage might occur.

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UNIT – IV – Airport Lightning, Marking & Signage – SAE1605

CHAPTER 8

Airport Lighting, Marking, and Signage

Introduction

Visual aids assist the pilot on approach to an airport, as well as navigating around an airfield and are essential elements of airport infrastructure. As such, these facilities require proper planning and precise design.

These facilities may be divided into three categories: lighting, marking, and signage. Lighting is further categorized as either approach lighting or surface lighting. Specific lighting systems described in this chapter include

1. Approach lighting
2. Runway threshold lighting
3. Runway edge lighting
4. Runway centerline and touchdown zone lights
5. Runway approach slope indicators
6. Taxiway edge and centerline lighting

The proper placement of these systems is described in this chapter but no attempt has been made to describe in detail the hardware or its installation. Airfield marking and signage includes

1. Runway and taxiway pavement markings
2. Runway and taxiway guidance sign systems²⁹¹

Airfield lighting, marking, and signage facilities provide the following functions:

1. Ground to air visual information required during landing
2. The visual requirements for takeoff and landing
3. The visual guidance for taxiing

In the United States, the Federal Aviation Administration provides guidance for designing standard airfield lighting, marking, and signage, through published Advisory Circulars. These Advisory Circulars are frequently updated. The standards described in this text are current as of 2007. Current advisory circulars may be found at the FAA's website at <http://www.faa.gov>.

The Requirements for Visual Aids

Since the earliest days of flying, pilots have used ground references for navigation when approaching an airport, just as officers on ships at sea have used landmarks on shore when approaching a harbor. Pilots need visual aids in good weather as well as in bad weather and during the day as well as at night.

In the daytime there is adequate light from the sun, so artificial lighting is not usually required but it is necessary to have adequate contrast in the field of view and to have a suitable pattern of brightness so that the important features of the airport can be identified and oriented with respect to the position of the aircraft in space. These requirements are almost automatically met during the day when the weather is clear.

The runway for conventional aircraft always appears as a long narrow strip with straight sides and is free of obstacles. It can therefore be easily identified from a distance or by flying over the field. Therefore, the perspective view of the runway and other identifying reference landmarks are used by pilots as visual aids for orientation when they are approaching the airport to land. Experience has demonstrated that the horizon, the runway edges, the runway threshold, and the centerline of the runway are the most important elements for pilots to see.

In order to enhance the visual information during the day, the runway is painted with standard marking patterns. The key elements in these patterns are the threshold, the centerline, the edges, plus multiple parallel lines to increase the perspective and to define the plane of the surface.

During the day when visibility is poor and at night, the visual information is reduced by a significant amount over the clear weather daytime scene. It is therefore essential to provide visual aids which will be as meaningful to pilots as possible.

The Airport Beacon

Beacons are lighted to mark an airport. They are designed to produce a narrow horizontal and vertical beam of high-intensity light which is rotated about a vertical axis so as to produce approximately 12 flashes per minute for civil airports and 18 flashes per minute for military airports [28]. The flashes with a clearly visible duration of at least 0.15 s are arranged in a white-green sequence for land airports and a white-yellow sequence for landing areas on water. Military airports use a double white flash followed by a longer green or yellow flash to differentiate them from civil airfields. The beacons are mounted on top of the control tower or similar high structure in the immediate vicinity of the airport.

Obstruction Lighting

Obstructions are identified by fixed, flashing, or rotating red lights or beacons. All structures that constitute a hazard to aircraft in flight or during landing or takeoff are marked by obstruction lights having a horizontally uniform intensity duration and a vertical distribution design to give maximum range at the lower angles (1.5° to 8°) from which a colliding approach would most likely come. The criteria for determining which structures need to be lighted are published by the FAA [18, 19].

The Aircraft Landing Operation

An aircraft approaching a runway in a landing operation may be visualized as a sequence of operations involving a transient body suspended in a three-dimensional grid that is approaching a fixed two-dimensional grid. While in the air, the aircraft can be considered as a point mass in a three-dimensional orthogonal coordinate system in which it may have translation along three coordinate directions and rotation about three axes. If the three coordinate axes are aligned horizontal, vertical, and parallel to the end of the runway, the directions of motion can be described as lateral, vertical, and forward. The rotations are normally called pitch, yaw, and roll, for the horizontal, vertical, and parallel axes, respectively. During a landing operation, pilots must control and coordinate all six degrees of freedom of the aircraft so as to bring the aircraft into coincidence with the desired approach or reference path to the touchdown point on the runway. In order to do this, pilots need translation information regarding the aircraft's alignment, height, and distance, rotation information regarding pitch, yaw, and roll, and information concerning the rate of descent and the rate of closure with the desired path. The glide path, height, time, and distance relationships during a typical landing are shown in Fig. 8-1.

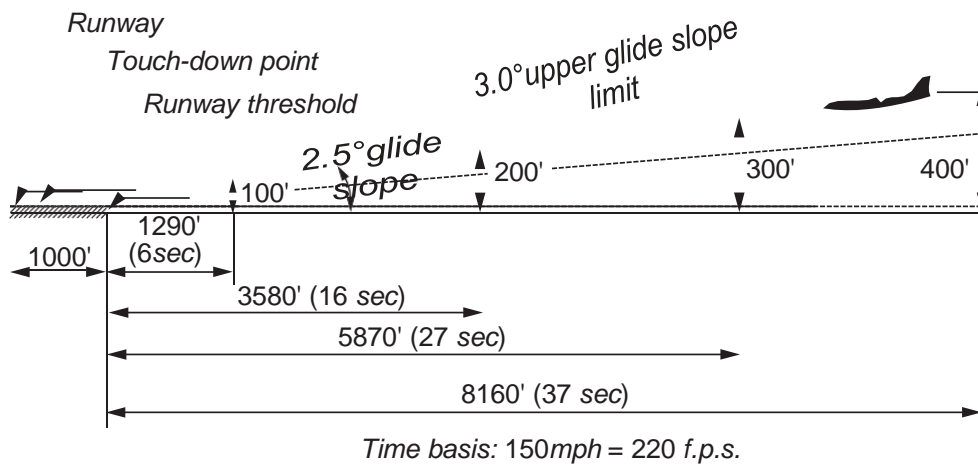


FIGURE 8-1 Glide slope, height, distance, and time relationships.

Alignment Guidance

Pilots must know where their aircraft is with respect to lateral displacement from the centerline of the runway. Most runways are from 75 to 200 ft wide and from 3000 to 12,000 ft long. Thus any runway is a long narrow ribbon when first seen from several thousand feet above. The predominant alignment guidance comes from longitudinal lines that constitute the centerline and edges of the runway. All techniques, such as painting, lighting, or surface treatment that develop contrast and emphasize these linear elements are helpful in providing alignment information.

Height Information

The estimation of the height above ground from visual cues is one of the most difficult judgments for pilots. It is simply not possible to provide good height information from an approach lighting system. Consequently the best source of height information is the instrumentation in the aircraft. However, use of these instruments often requires the availability of precision ground or satellite based navigation technologies. Many airports have no such technologies, and at others only provide lateral approach guidance to certain runways. Consequently two types of ground-based visual aids defining the desired glide path have been developed. These are known as the visual approach slope indicator (VASI) and the precision approach path indicator (PAPI) which are discussed later in this chapter.

Several parameters influence how much a pilot can see on the ground. One of these is the *cockpit cutoff angle*. This is the angle between the longitudinal axis of the fuselage and an inclined plane below which the view of the pilot is blocked by some part of the aircraft, indicated by α in Fig. 8-2. Normally the larger the angle α , the more the pilot can see of the ground. Also important is the *pitch angle*, β ,

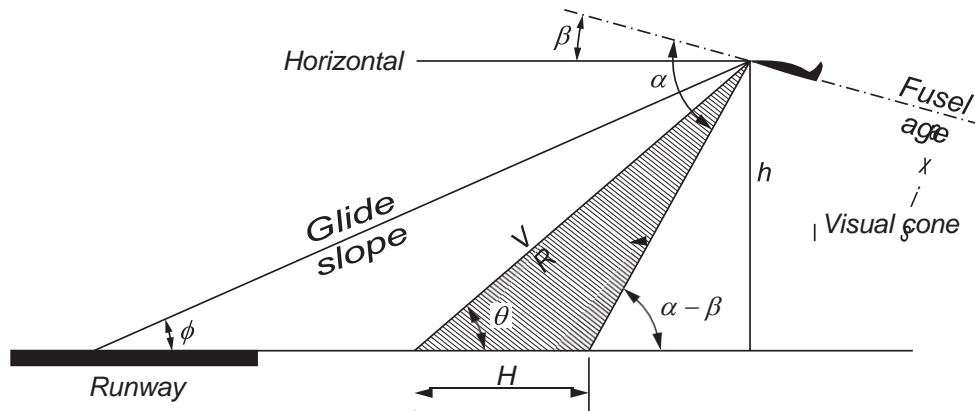


FIGURE 8-2 Visual parameters: ϕ = glide slope angle, α = cockpit cutoff angle, β = pitch angle, VR = visual range, H = horizontal segment of visual range, h = height of glide slope above the runway, and θ = angle formed by VR with the horizontal.

of the fuselage axis during the approach to the runway. Few aircraft approach a runway with the fuselage angle horizontal; they are either pitched up or down. The larger the angle β (in a pitch-up attitude), the larger must be the angle α to have adequate over-the-nose vision. Approach speed has a profound influence on the angle β . As an example, for some aircraft β can be decreased by about 1° with each 5 kn increase in speed above the reference approach speed.

In Fig. 8-2, VR is the visual range or the maximum distance a pilot can see and some height above the runway h . The horizontal segment of the ground that a pilot can see is H . According to Fig. 8-2,

$$H = VR \cos \Theta - h \cot (\alpha - \beta) \quad (8-1)$$

and also

$$\sin \Theta = \frac{h}{VR} \quad (8-2)$$

Note from Eq. (8-1) that for a fixed value of VR the ground segment H increases as the height h of the eyes of a pilot above the ground decreases. Typical values of α range from 11° to 16° and typical values of β are $\pm 0.5^\circ$.

It has been found through experience that 3 s is approximately the minimum reaction time for a pilot to cause the aircraft to react after sighting a visual aid [28]. If a minimum of 3 s is necessary for perception, pilot action, aircraft response, and checking the response, and if the approach speed of the aircraft is 150 mi/h (220 ft/s), then the minimum horizontal segment on the ground should not be less than 660 ft. Using Eq. (8-1) with the glide slope angle, ϕ , of 2.5° and a

value of $\alpha - \beta$ of 12° results in H being about 200 ft when h is 200 ft. However, when h is 100 ft, H is 687 ft. Consequently, lighting systems designed to aid in aircraft approaching to land on a runway have been designed to provide optimal visual guidance when aircraft are at relatively low altitudes on approach, and are angled to be consistent with the downward approach angle of arriving aircraft.

Approach Lighting

Approach lighting systems (ALS) are designed specifically to provide guidance for aircraft approaching a particular runway under nighttime or other low-visibility conditions. While under nighttime conditions it may be possible to view approach lighting systems from several miles away, under other low-visibility conditions, such as fog, even the most intense ALS systems may only be visible from as little as 2500 ft from the runway threshold.

Studies of the visibility in fog [3] have shown that for a visual range of 2000 to 2500 ft it would be desirable to have as much as 200,000 candelas (cd) available in the outermost approach lights where the slant range is relatively long. Under these same conditions the optimum intensity of the approach lights near the threshold should be on the order of 100 to 500 cd. A transition in the intensity of the light that is directed toward the pilot is highly desirable in order to provide the best visibility at the greatest possible range and to avoid glare and the loss of contrast sensitivity and visual acuity at short range.

System Configurations

The configurations which have been adopted are the Calvert system [3] shown in Fig. 8-3 which has been widely used in Europe and other parts of the world, the ICAO category II and category III system shown in Fig. 8-4, and the four system configurations which have been adopted by the FAA in the United States shown in Fig. 8-5. The FAA publishes criteria for the establishment of the approach lighting systems [13] and other navigation facilities at airports [6]. Approach lights are normally mounted on frangible pedestals of varying height to improve the perspective of the pilot in approaching a runway.

The first approach lighting system was known as the Calvert system. In this system, developed by E. S. Calvert in Great Britain in 1949, includes a line of single bulb lights spaced on 100-ft centers along the extended runway centerline and six transverse crossbars of lights of variable length spaced on 500-ft centers, for a total length of 3000 ft. The Calvert system was the first approach lighting system to be certified by ICAO, and is also commonly known as the ICAO category I approach lighting system. An illustration of the Calvert system is found in Fig. 8-3. The Calvert system is still used in developing countries.

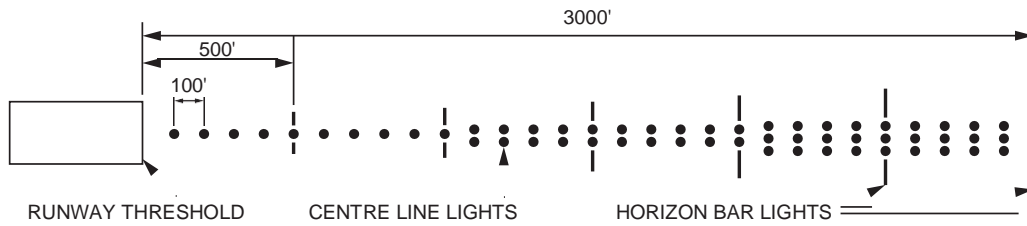


FIGURE 8-3 Calvert approach lighting system.

For operations in very poor visibility, ICAO has certified a modification of the Calvert system, known as the ICAO category II system. The variation calls for a higher lighting intensity to the inner 300 m of the system closest to the runway threshold. The category II and category III system adopted by ICAO shown in Fig. 8-4 consists of two lines of red bars on each side of the runway centerline and a single line of white bars on the runway centerline both at 30 m intervals and both extending out 300 m from the runway threshold. In addition, there are two longer bars of white light at a distance of 150 and 300 m from the runway threshold, and a long threshold bar of green light at the runway threshold. ICAO also recommends that the longer bars of white light also be placed at distances of 450, 500, and 750 m from the runway threshold if the runway centerline lights extend out that distance as shown in Fig. 8-4.

The ALSs currently certified by the FAA for installation in the United States consist of a high-intensity ALS with sequenced flashing lights (ALSF-2), which is required for category II and category III precision approaches, a high-intensity approach lighting system with

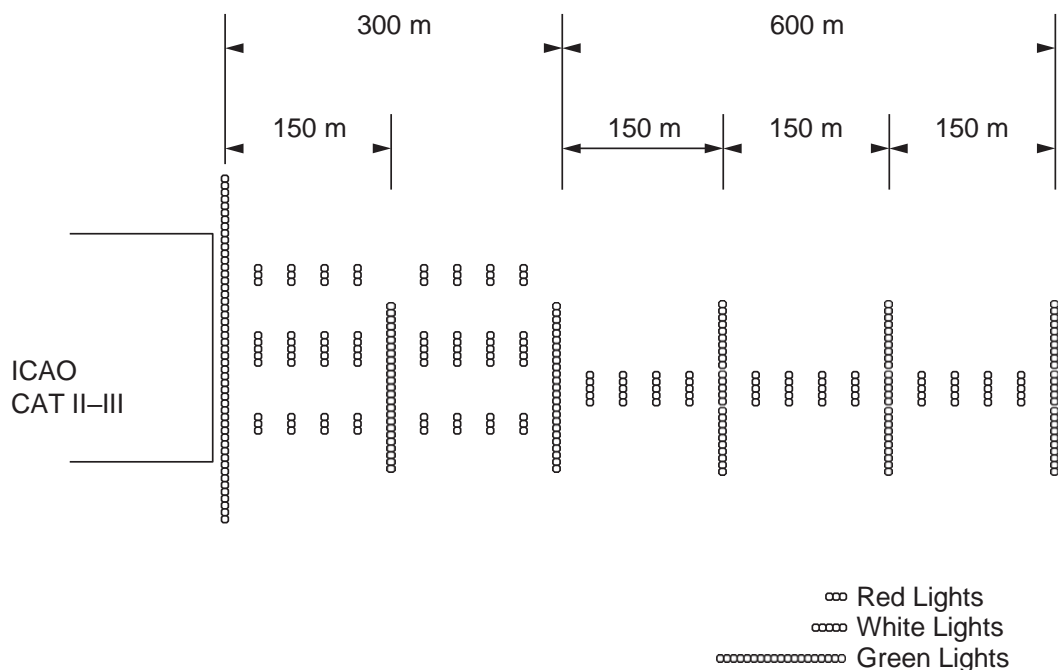


FIGURE 8-4 ICAO CAT II-III approach lighting system.

sequenced flashing lights (ALSF-1), and three medium-intensity ALSs (MALSR, MALS, MALSF).

In each of these systems there is a long transverse crossbar located 1000 ft from the runway threshold to indicate the distance from the runway threshold. In these systems roll guidance is provided by crossbars of white light 14 ft in length, placed at either 100- or 200-ft centers on the extended runway centerline. The 14-ft crossbars consist of closely spaced five-bulb white lights to give the effect of a continuous bar of light.

The high-intensity ALS is 2400 ft long (some are 3000 ft long) with various patterns of light located symmetrically about the extended runway centerline and a series of sequenced high-intensity flashing lights located every 100 ft on the extended runway centerline for the outermost 1400 ft. In the high-intensity ALSs the 14-ft crossbars of five-bulb white light are placed at 100-ft intervals and in the medium-intensity ALSs these crossbars of white light are placed at 200-ft intervals both for a distance of 2400 ft from the runway threshold on the extended runway centerline. The high-intensity ALSs have a long crossbar of green lights at the edge of the runway threshold. The ALSF-2 system, shown in Fig. 8-5*a*, has two additional crossbars consisting of three-bulb white light crossbars which are placed symmetrically about the runway centerline at a distance of 500 ft from the runway threshold and two additional three-bulb red light crossbars are placed symmetrically about the extended runway centerline at 100-ft intervals for the inner 1000 ft to delineate the edges of the runway surface. The ALSF-1 system, shown in Fig. 8-5*b*, has two additional crossbars consisting of five-bulb red light crossbars which are placed symmetrically about the runway centerline at a distance of 100 ft from the runway threshold to delineate the edge of the runway and two additional three-bulb red light crossbars placed symmetrically about the extended runway centerline at 200 ft from the runway threshold.

The MALSR system, shown in Fig. 8-5*c*, is a 2400-ft medium-intensity ALS with runway alignment indicator lights (RAILs). The inner 1000 ft of the MALSR is the MALS portion of the system and the outer 1400 ft is the RAIL portion of the system. The system has sequential flashing lights for the outer 1000 ft of the system. It is recommended for category I precision approaches. The simplified short approach lighting system (SSALR) has the same configuration as the MALSR system.

At smaller airports where precision approaches are not required, a medium ALS with sequential flashers (MALSF) or with sequenced flashers (MALS) is adequate. The system is only 1400 ft long compared to a length of 2400 ft for a precision approach system. It is therefore much more economical, an important factor at small airports. The MALSF, similar to the MALSR shown in Fig. 8-5*d*, is a short approach medium-intensity ALS but the sequenced flashers replace

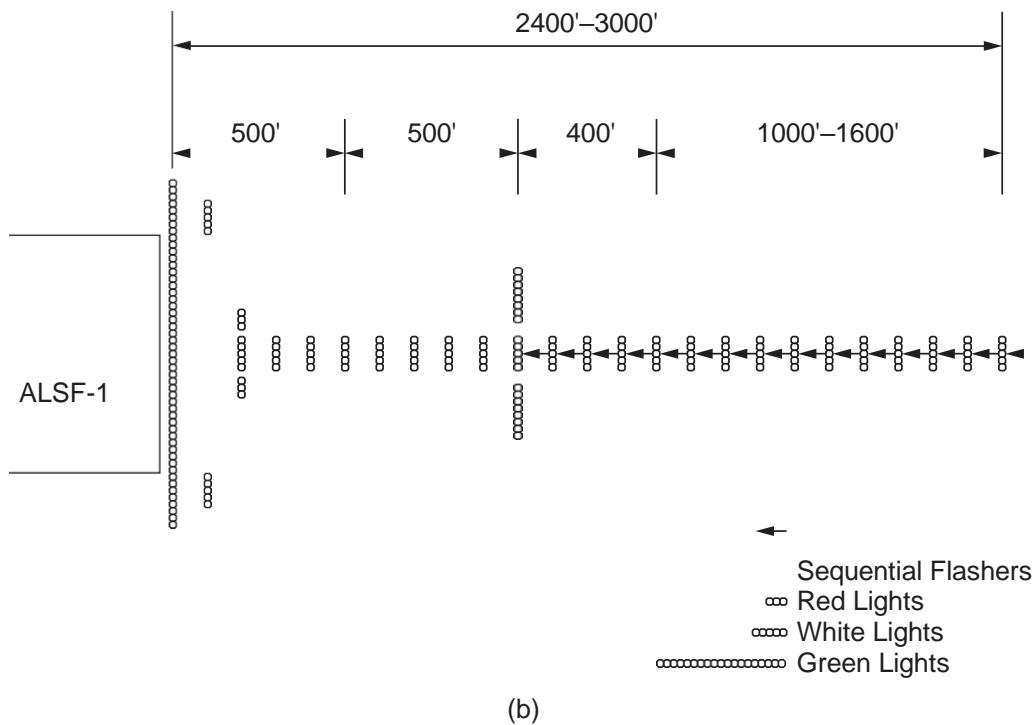
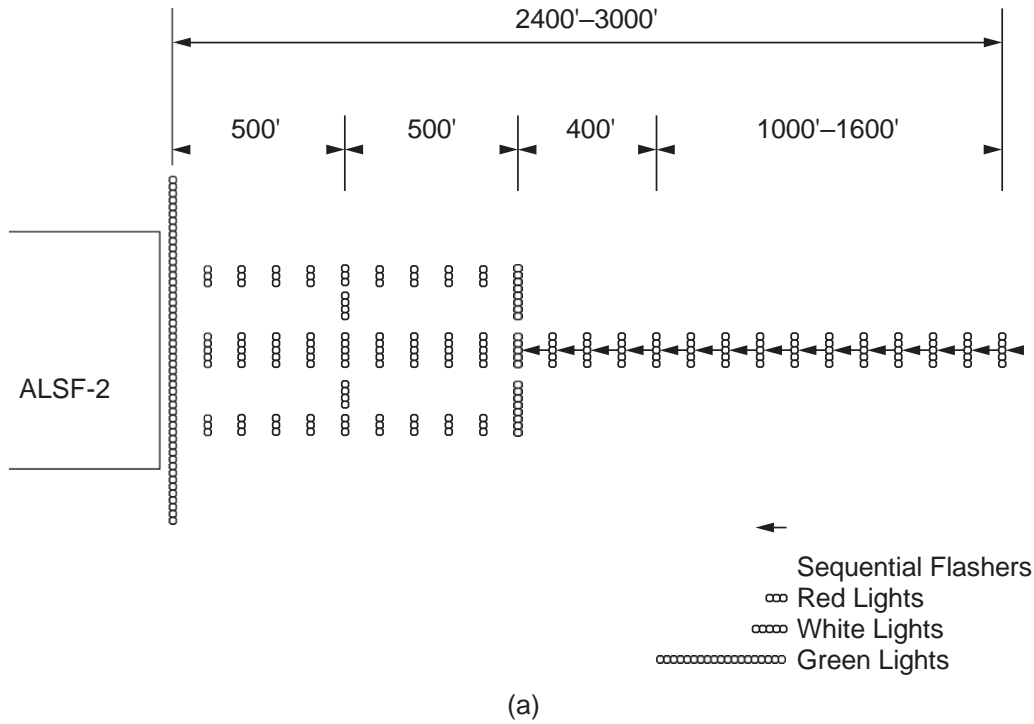
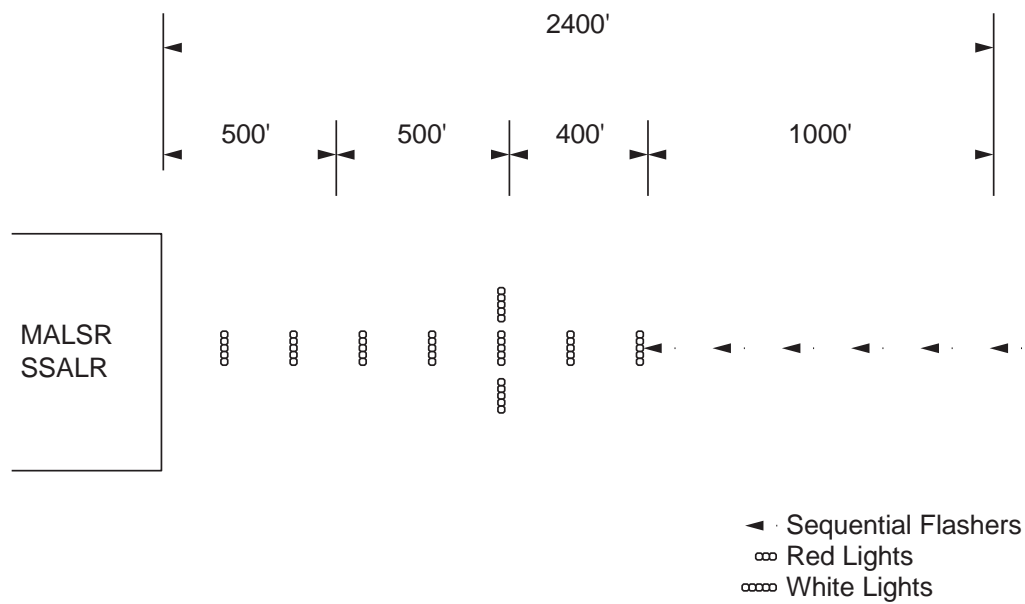
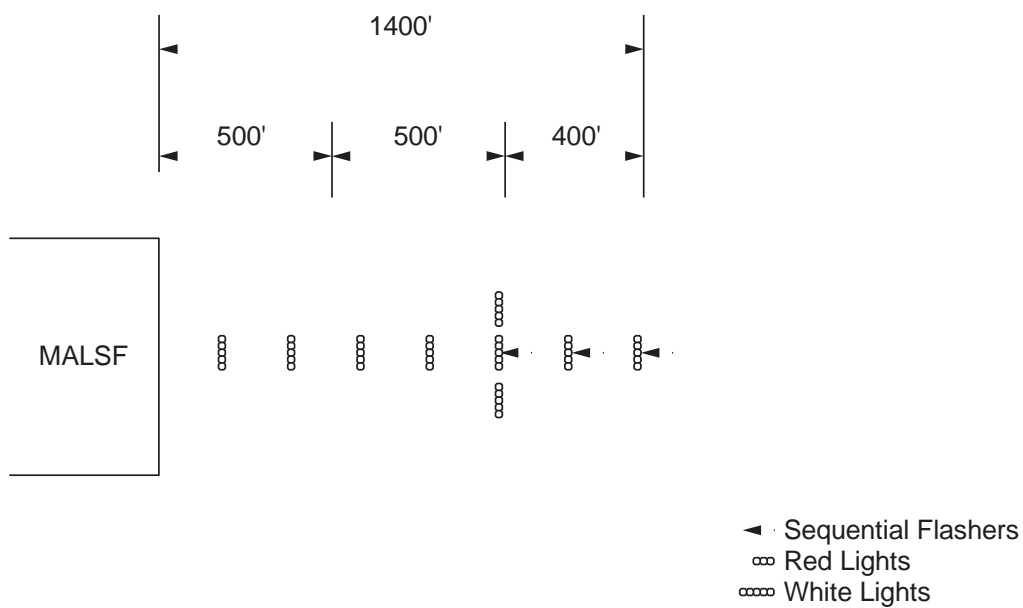


FIGURE 8-5a-d Approach lighting system configurations. (*Continued*)



(c)



(d)

FIGURE 8-5a-d Approach lighting system configurations.

the runway alignment indicator lights and these are only provided in the outermost 400 ft of the 1400-ft system to improve pilot recognition of the runway approach in areas where there are distracting lights in the vicinity of the airport. The MALSR system does not have the runway alignment indicator lights or the sequential flashers.

At international airports in the United States, the 2400-ft ALSs are often extended to a distance of 3000 ft to conform to international specifications.

Sequenced-flashing high-intensity lights are available for airport use and are installed as supplements to the standard approach lighting system at those airports where very low visibilities occur frequently. These lights operate from the stored energy in a capacitor which is discharged through the lamp in approximately 5 ms and may develop as much as 30 million cd of light. They are mounted in the same pedestals as the light bars. The lights are sequence-fired, beginning with the unit farthest from the runway. The complete cycle is repeated every 2 s. This results in a brilliant ball of light continuously moving toward the runway. Since the very bright light can interfere with the eye adaptation of the pilot, condenser discharge lamps are usually omitted in the 1000 ft of the approach lighting system nearest the runway.

Visual Approach Slope Aids

Visual approach slope aids are lighting systems designed to provide a measure of vertical guidance to aircraft approaching a particular runway. The principle of these aids is to provide color-based identification to the pilot indicating their variation from a desired altitude and descent rate while on approach. The two most common visual approach slope aids are the visual approach slope indicator (VASI), and the precision approach path indicator (PAPI).

Visual Approach Slope Indicator

The visual approach slope indicator (VASI) is a system of lights which acts as an aid in defining the desired glide path in relatively good weather conditions. VASI lighting intensities are designed to be visible from 3 to 5 mi during the day and up to 20 mi at night.

There are a number of different VASI configurations depending on the desired visual range, the type of aircraft, and whether large wide bodied aircraft will be using the runway. Each group of lights transverse to the direction of the runway is referred to as a *bar*. The downwind bar is typically located between 125 and 800 ft from the runway threshold, each subsequent bar is located between 500 and 1000 ft from the previous bar. A bar is made up of one, two, or three light units, referred to as *boxes*. The basic VASI-2 system, illustrated in Fig. 8-6, is a two-bar system consisting of four boxes. The bar that is nearest to the runway threshold is referred to as the *downwind bar*, and the bar that is farthest from the runway threshold is referred to as the *upwind bar*. As illustrated in Fig. 8-6, if pilots are on the proper glide path, the downwind bar appears white and the upwind bar appears red; if pilots are too low, both bars appear red; and if they are too high both bars appear white.

In order to accommodate large wide bodied aircraft where the height of the eye of the pilot is much greater than in smaller jets, a third upwind bar is added. For wide bodied aircraft the middle bar becomes

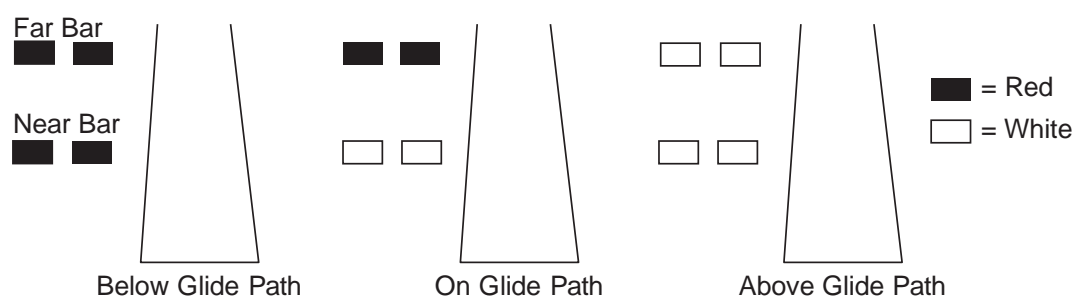


FIGURE 8-6 Two bar VASI system (FAA/AIM).

the downwind bar and the third bar is the upwind bar. In other words, pilots of large wide bodied aircraft ignore the bar closest to the runway threshold and use the other two bars for visual reference. The location of the lights for VASI-6 systems is shown in Fig. 8-7.

The more common systems in use in the United States are the VASI-2, VASI-4, VASI-12, and VASI-16. VASI systems are particularly useful on runways that do not have an instrument landing system or for aircraft not equipped to use an instrument landing system.

Precision Approach Path Indicator

The FAA presently prefers the use of another type of visual approach indicator called the *precision approach path indicator* (PAPI) [20]. This system gives more precise indications to the pilot of the approach path of the aircraft and utilizes only one bar as opposed to the minimum of two required by the VASI system. A schematic diagram of the PAPI system is shown in Fig. 8-8.

The system consists of a unit with four lights on either side of the approach runway. By utilizing the color scheme indicated on Fig. 8-8, the pilot is able to ascertain five approach angles relative to the proper glide slope as compared with three with the VASI system. One of the problems with the VASI system has been the lack of an immediate transition from one color indication to another resulting in shades of colors. The PAPI system resolves this problem by providing an instant transition from one color indication to another as a reaction to the

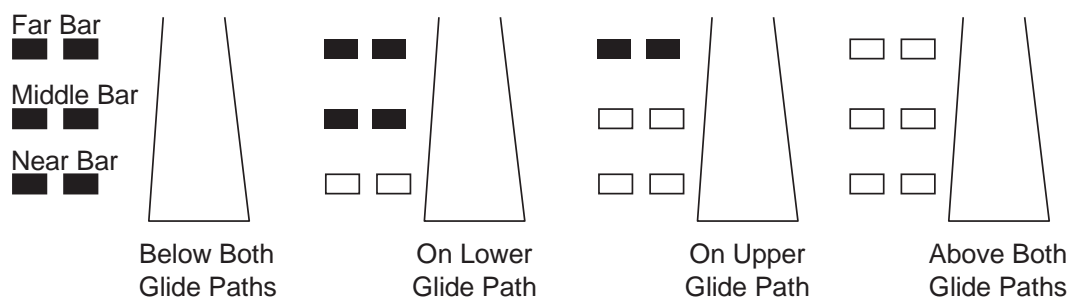


FIGURE 8-7 Three bar VASI-6 system.

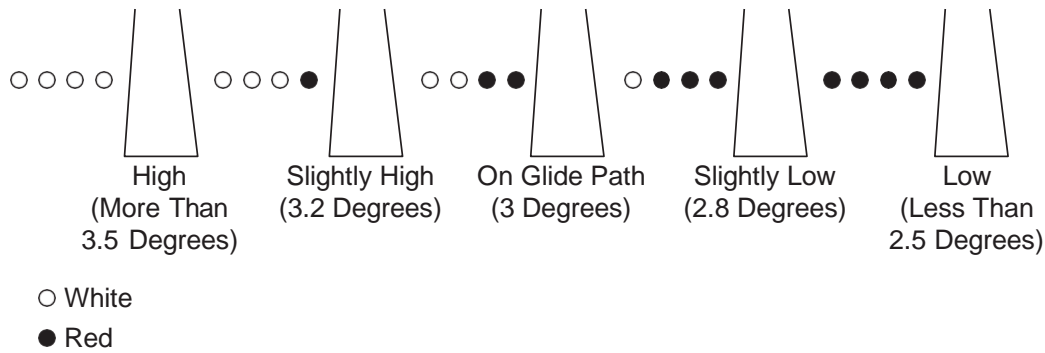


FIGURE 8-8 Precision approach path indicator (PAPI) system.

descent path of the aircraft. An advantage of the system is that it is a one-bar system as opposed to the two-bar VASI system. This results in greater operating and maintenance cost economies, and eliminates the need for the pilot to look at two bars to obtain glide slope indications.

Threshold Lighting

During the final approach for landing, pilots must make a decision to complete the landing or “execute a missed approach.” The identification of the threshold is a major factor in pilot decisions to land or not to land. For this reason, the region near the threshold is given special lighting consideration. The threshold is identified at large airports by a complete line of green lights extending across the entire width of the runway, as shown earlier in Fig. 8-5, and at small airports by four green lights on each side of the threshold. The lights on either side of the runway threshold may be elevated. Threshold lights in the direction of landing are green but in the opposite direction these lights are red to indicate the end of the runway.

Runway Lighting

After crossing the threshold, pilots must complete a touchdown and roll out on the runway. The runway visual aids for this phase of landing are designed to give pilots information on alignment, lateral displacement, roll, and distance. The lights are arranged to form a visual pattern that pilots can easily interpret.

At first, night landings were made by floodlighting the general area. Various types of lighting devices were used, including automobile headlights, arc lights, and search lights. Boundary lights were added to outline the field and to mark hazards such as ditches and fences. Gradually, preferred landing directions were developed, and special lights were used to indicate these directions. Floodlighting was then restricted to the preferred landing directions, and runway edge lights were added along the landing strips. As experience was

developed, the runway edge lights were adopted as visual aids on a runway. This was followed by the use of runway centerline and touchdown zone lights for operations in very poor visibility. FAA Advisory Circular 150/5340-30C provides guidance for the design and installation of runway and taxiway lighting systems. Those planning and designing such systems should refer to the latest changes to this Advisory Circular, commonly found at <http://www.faa.gov>.

Runway Edge Lights

Runway edge lighting systems outline the edge of runways during nighttime and reduced visibility conditions. Runway edge lights are classified by intensity, high intensity (HIRL), medium intensity (MIRL), and low intensity (LIRL). LIRLs are typically installed on visual runways and at rural airports. MIRLs are typically installed on visual runways at larger airports and on nonprecision instrument runways, HIRLs are installed on precision-instrument runways.

Recommended standards for the design and installation of runway edge lighting systems are published by the FAA [21] and are contained in ICAO Annex 14 [1]. These light fixtures are usually elevated units but semiflush lights are permitted. Each unit has a specially designed lens which projects two main light beams down the runway. Elevated runway lights are mounted on frangible fittings and project no more than 30 in above the surface on which they are installed. They are located along the edge of the runway not more than 10 ft from the edge of the full-strength pavement surface. The longitudinal spacing is not more than 200 ft. Runway edge lights are white, except that the last 2000 ft of an instrument runway in the direction of aircraft operations these lights are yellow to indicate a caution zone. A typical layout of low-intensity and medium-intensity runway edge lights is shown in Fig. 8-9, and a typical layout of HIRLs are illustrated in Fig. 8-10. If the runway threshold is displaced, but the area that is displaced is usable for takeoffs and taxiing, the runway edge lights in the displaced area in the direction of aircraft operations are red, as shown in Fig. 8-11.

Runway Centerline and Touchdown Zone Lights

As an aircraft traverses over the approach lights, pilots are looking at relatively bright light sources on the extended runway centerline. Over the runway threshold, pilots continue to look along the centerline, but the principal source of guidance, namely, the runway edge lights, has moved far to each side in their peripheral vision. The result is that the central area appears excessively black, and pilots are virtually flying blind, except for the peripheral reference information, and any reflection of the runway pavement from the aircraft's landing lights. Attempts to eliminate this "black hole" by increasing the intensity of runway edge lights have proven ineffective. In order to reduce

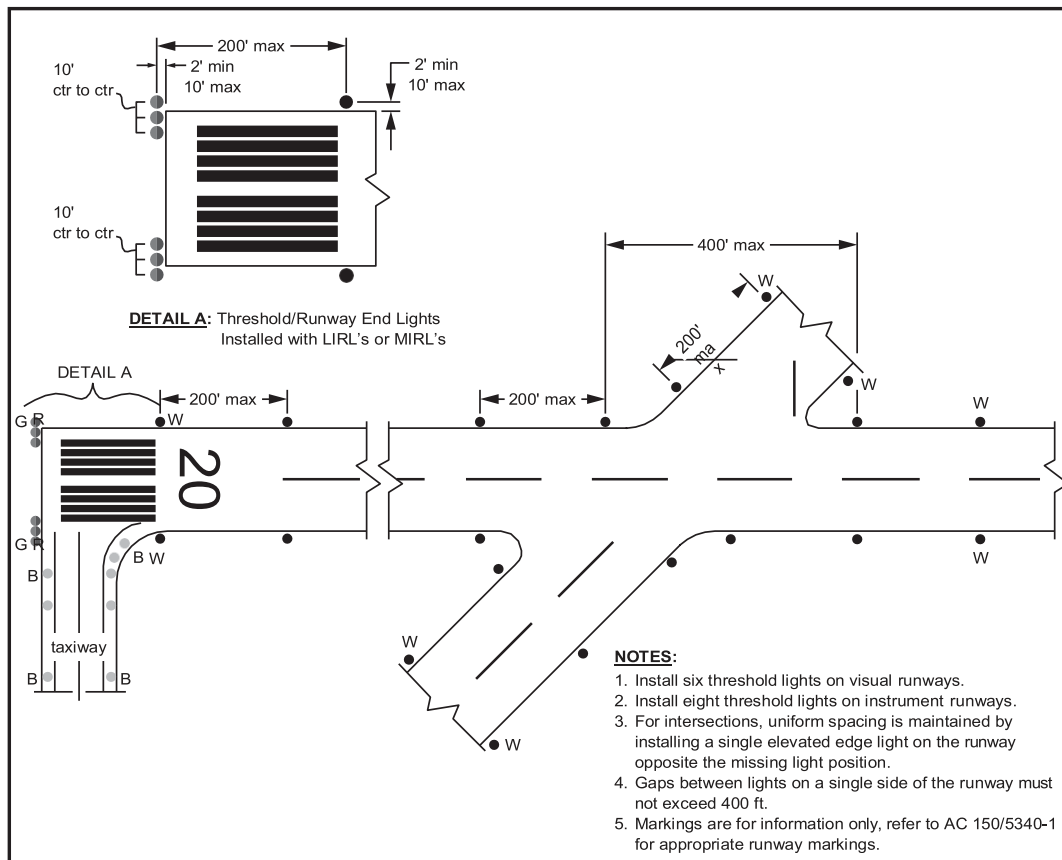


FIGURE 8-9 Low-intensity runway edge lighting specifications (*Federal Aviation Administration*).

the black hole effect and provide adequate guidance during very poor visibility conditions, runway centerline and touchdown zone lights are typically installed in the pavement. An illustration of runway centerline lighting is provided in Fig. 8-12.

These lights are usually installed only at those airports which are equipped for instrument operations. These lights are required for ILS category II and category III runways and for category I runways used for landing operations below 2400 ft runway visual range. Runway centerline lights are required on runways used for takeoff operations below 1600 ft runway visual range. Although not required, runway centerline lights are recommended for category I runways greater than 170 ft in width or when used by aircraft with approach speeds over 140 kn.

When there are displaced thresholds, the centerline lights are extended into the displaced threshold area. If the displaced area is not used for takeoff operations, or if the displaced area is used for takeoff operations and is less than 700 ft in length, the centerline lights are blanked out in the direction of landing. For displaced thresholds greater than 700 ft in length or for displaced areas used for takeoffs, the centerline lights in the displaced area must be capable of being shut off during landing operations.

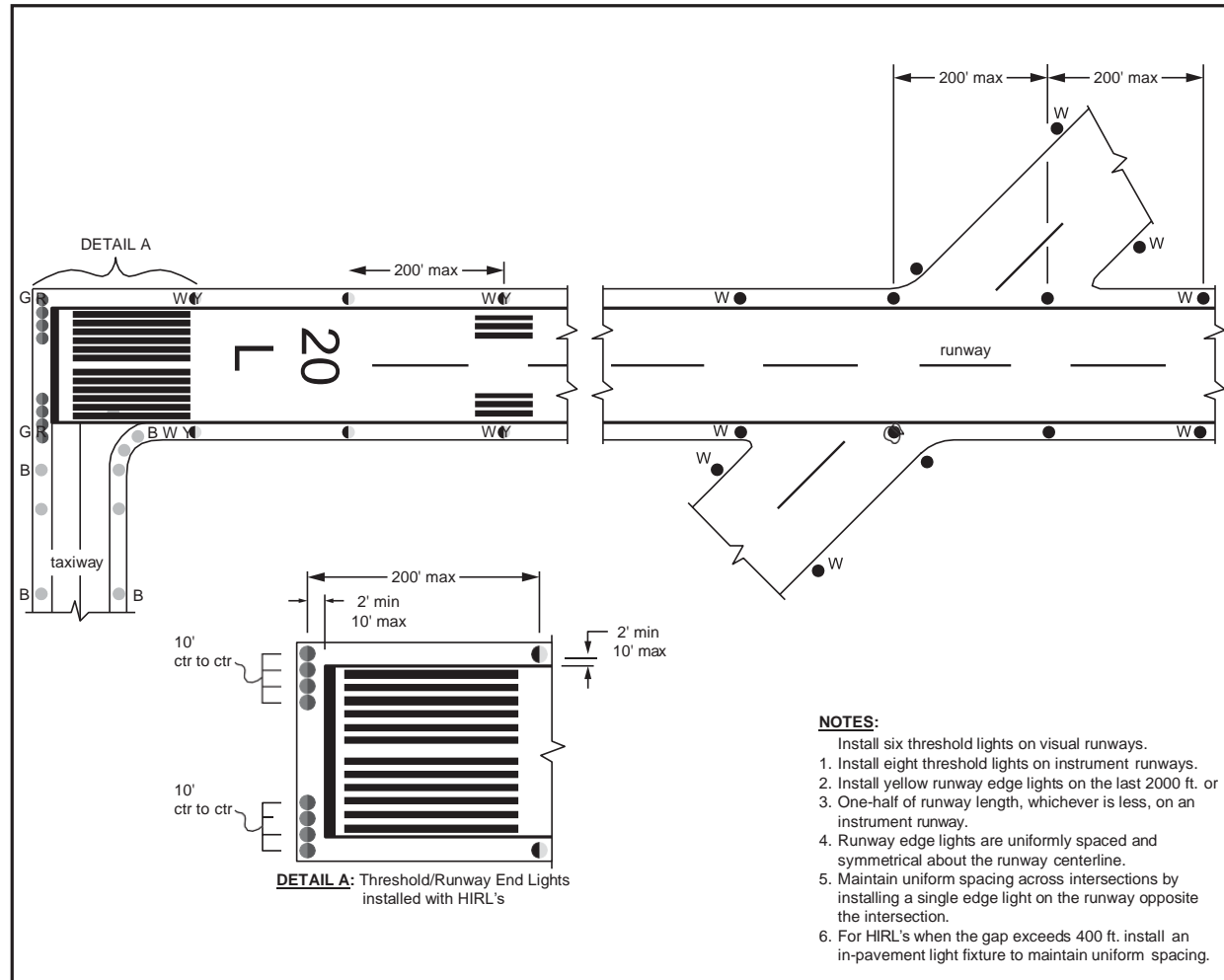


FIGURE 8-10 High-intensity runway edge lighting specifications (*Federal Aviation Administration*).

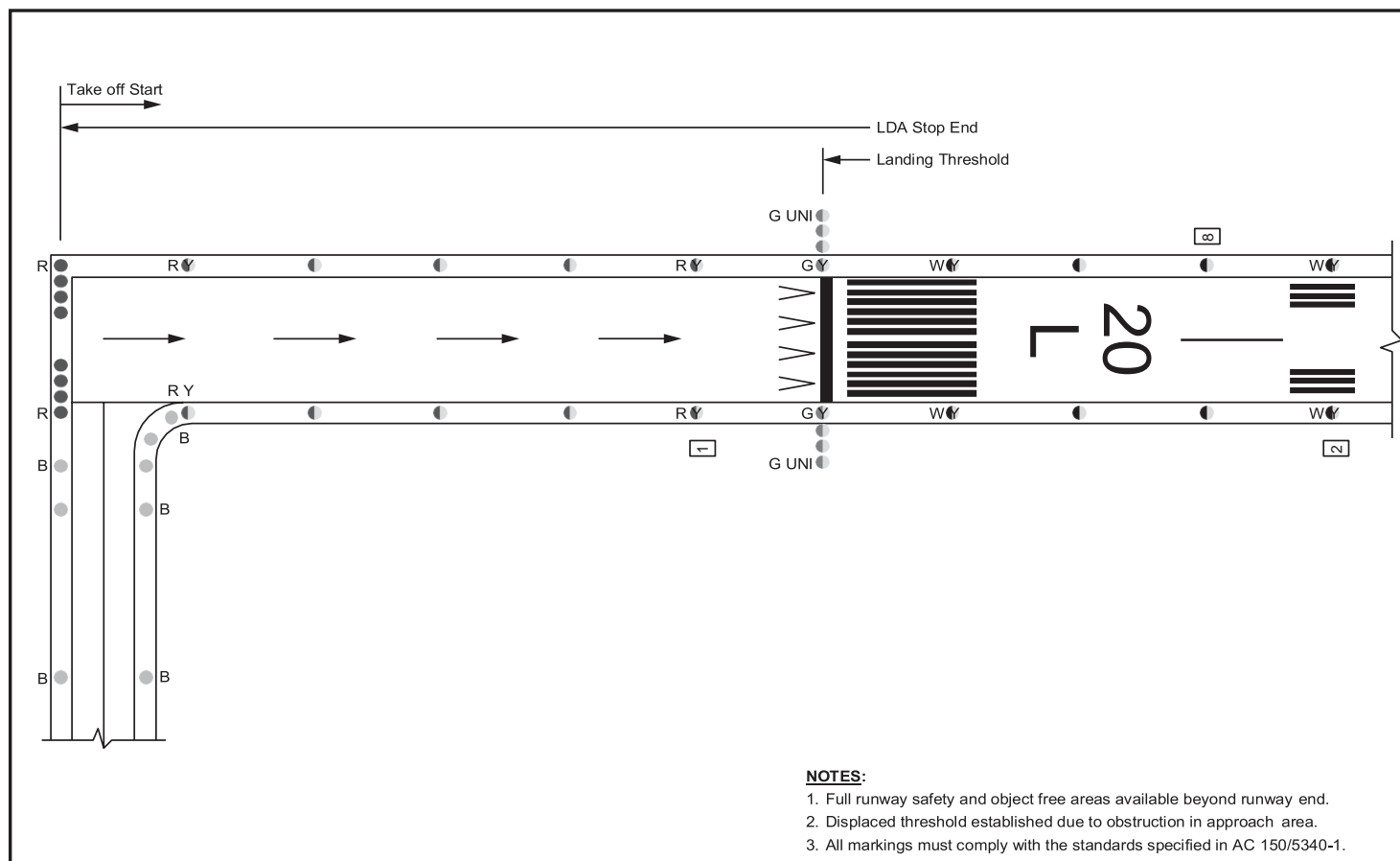


FIGURE 8-11 Runway edge and threshold lighting for a displaced threshold (*Federal Aviation Administration*).

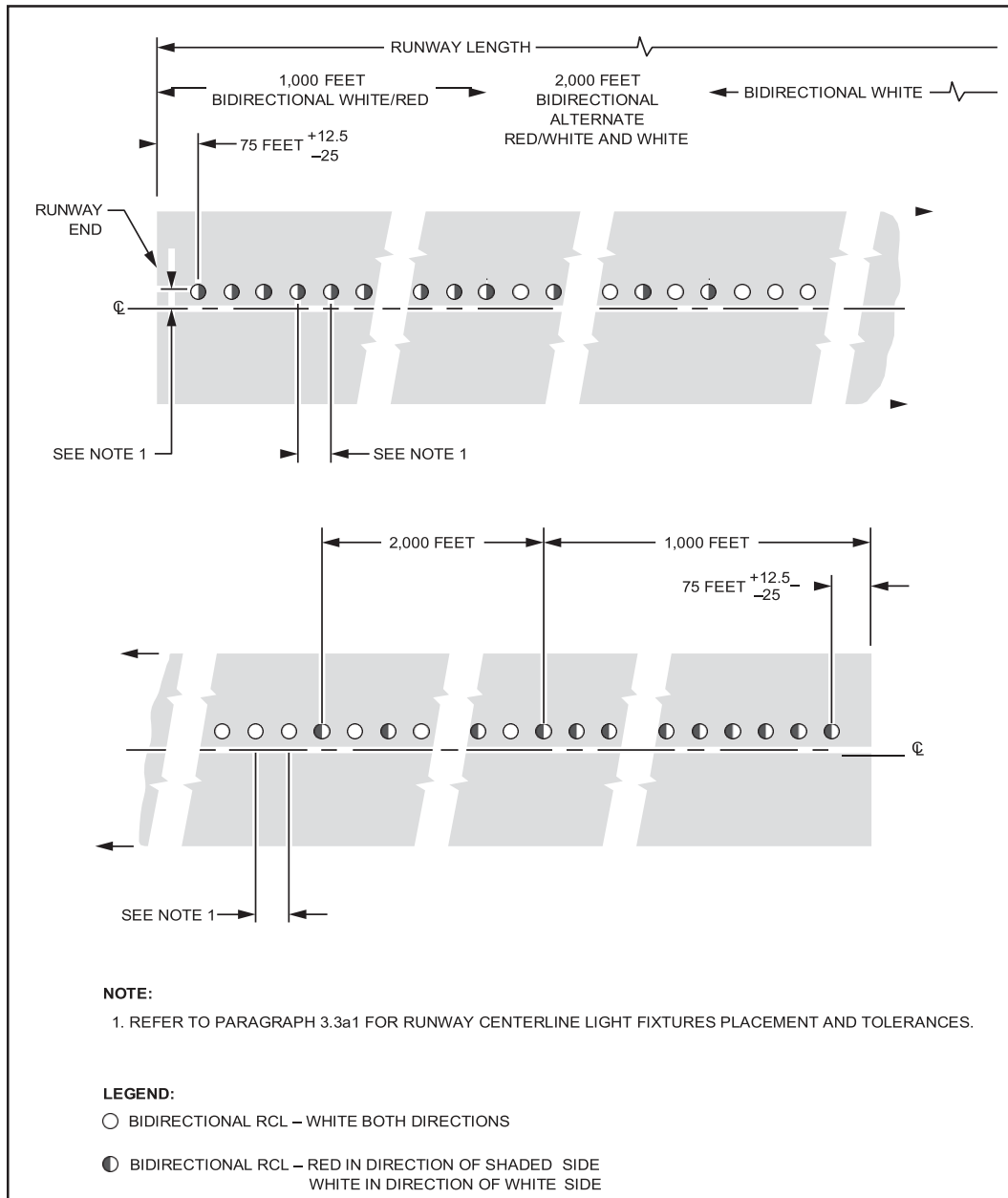


FIGURE 8-12 Runway centerline lighting (*Federal Aviation Administration*).

Runway touchdown zone lights are white, consist of a three-bulb bar on either side of the runway centerline, and extend 3000 ft from the runway threshold or one-half the runway length if the runway is less than 6000 ft long. They are spaced at intervals of 100 ft, with the first light bar 100 ft from the runway threshold, and are located 36 ft on either side of the runway centerline, as shown in Fig. 8-13. The centerline lights are spaced at intervals of 50 ft. They are normally offset a maximum of 2 ft from the centerline to avoid the centerline paint line and the nose gear of the aircraft riding over the light fixtures. These lights are also white, except for the last 3000 ft of runway

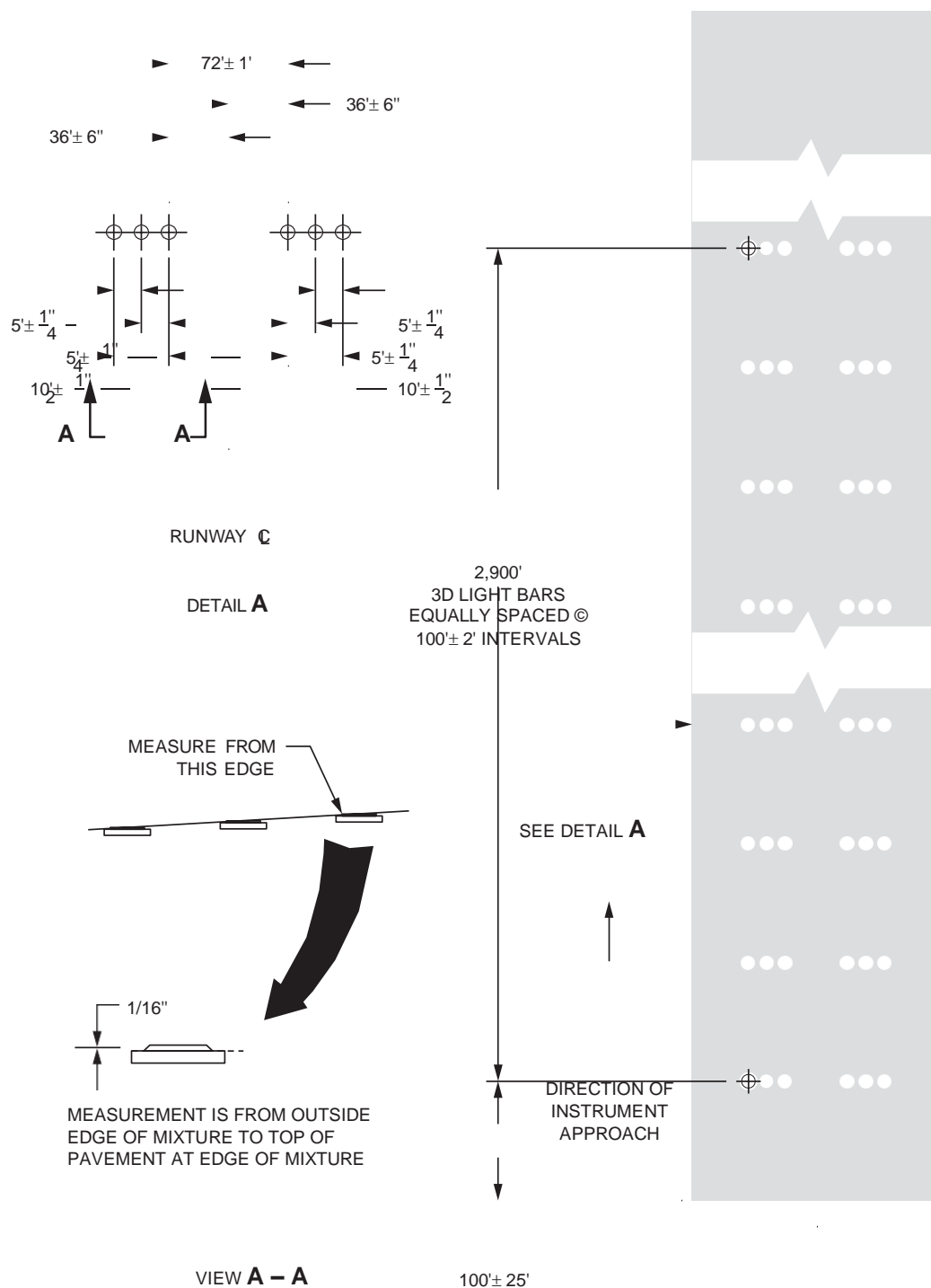


FIGURE 8-13 Runway threshold lighting.

in the direction of aircraft operations, where they are color coded. The last 1000 ft of centerline lights are red, and the next 2000 ft are alternated red and white.

Runway End Identifier Lights

Runway end identifier lights (REIL) are installed at airports where there

are no approach lights to provide pilots with positive visual identification

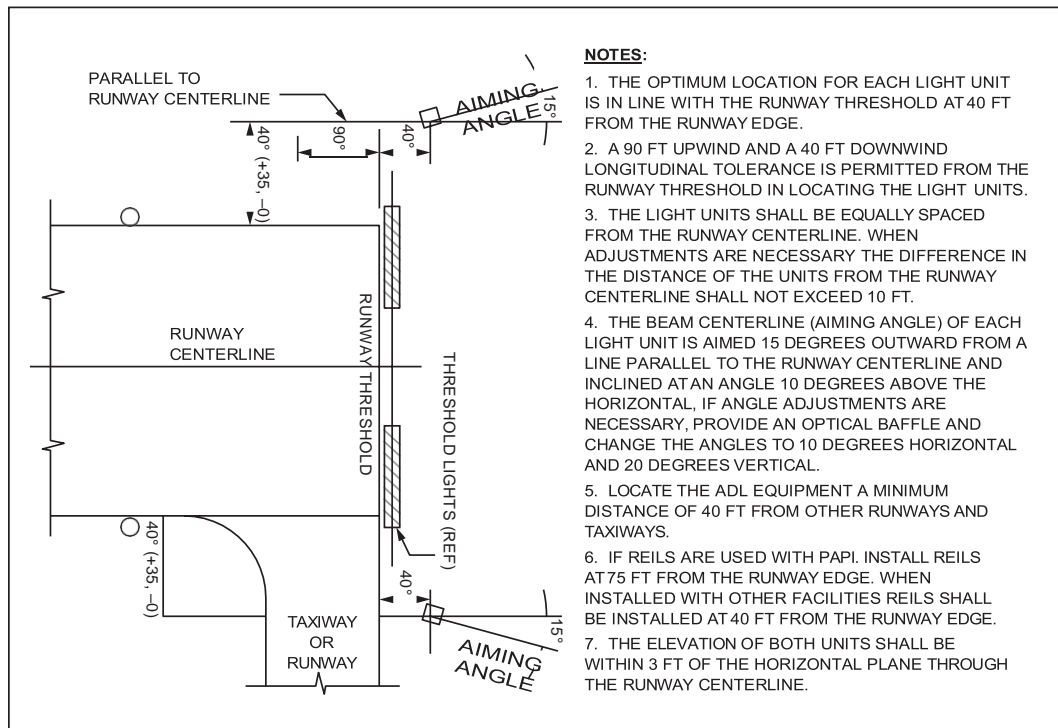


FIGURE 8-14 Typical layout for runway end identifier lights (REILs) (*Federal Aviation Administration*).

of the approach end of the runway. The system consists of a pair of synchronized white flashing lights located on each side of the runway threshold and is intended for use when there is adequate visibility. An illustration and design specifications of REILs may be found in Fig. 8-14.

Taxiway Lighting

Either after a landing or on the way to takeoff, pilots must maneuver the aircraft on the ground on a system of taxiways to and from the terminal and hangar areas. Taxiway lighting systems are provided for taxiing at night and also during the day when visibility is very poor, particularly at commercial service airports.

The following overall guidance should be applied in determining the lighting, marking, and signing visual aid requirements for taxiways:

- In order to avoid confusion with runways, taxiways must be clearly identified.
- Runway exits need to be readily identified. This is particularly true for high-speed runway exits so that pilots can be able to locate these exits 1200 to 1500 ft before the turnoff point.
- Adequate visual guidance along the taxiway must be provided.
- Specific taxiways must be readily identified.

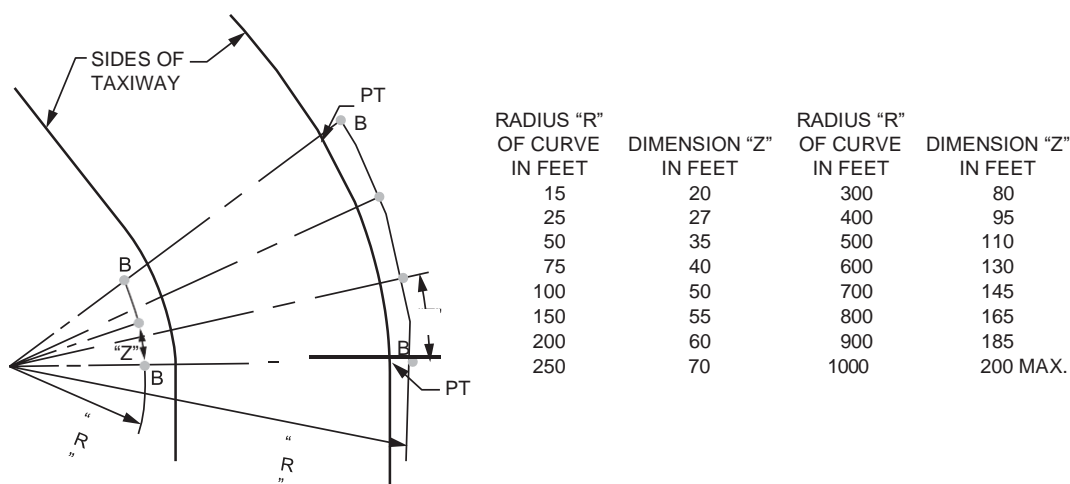
- The intersections between taxiways, the intersections between runways and taxiways, and runway-taxiway crossings need to be clearly marked.
- The complete taxiway route from the runway to the apron and from the apron to the runway should be easily identified.

There are two primary types of lights used for the designation of taxiways. One type delineates the edges of taxiways [21] and the other type delineates the centerline of the taxiway [27]. In addition, there is an increasing use of lighting systems on taxiways, such as runway guard lights (RGLs) and stop bars, to identify intersections with runways, in an effort to reduce accidental incursions on to active runway environments.

Taxiway Edge Lights

Taxiway edge lights are elevated blue colored bidirectional lights usually located at intervals of not more than 200 ft on either side of the taxiway. The exact spacing is influenced by the physical layout of the taxiways. Straight sections of taxiways generally require edge light spacing in 200-ft intervals, or at least three lights equally spaced for taxiway straight line sections less than 200 ft in length.

Closer spacing is required on curves. Light fixtures are located not more than 10 ft from the edge of full strength pavement surfaces. The lights cannot extend more than 30 inches above the pavement surface. The spacing of lights along a curve is shown in Fig. 8-15. Entrance points to runways and exit points from are lighted as shown in Fig. 8-16.



NOTES:

1. For radii not listed, determine "Z" spacing by linear interpolation.
2. "Z" is the arc length.
3. Uniformly space lights on curved edges. Do not exceed the values determined from the above table.
4. On curved edges in excess of 30 degrees arc, do not install less than three lights including those at the points of tangency (PT).

FIGURE 8-15 Typical taxiway lighting on curved sections.

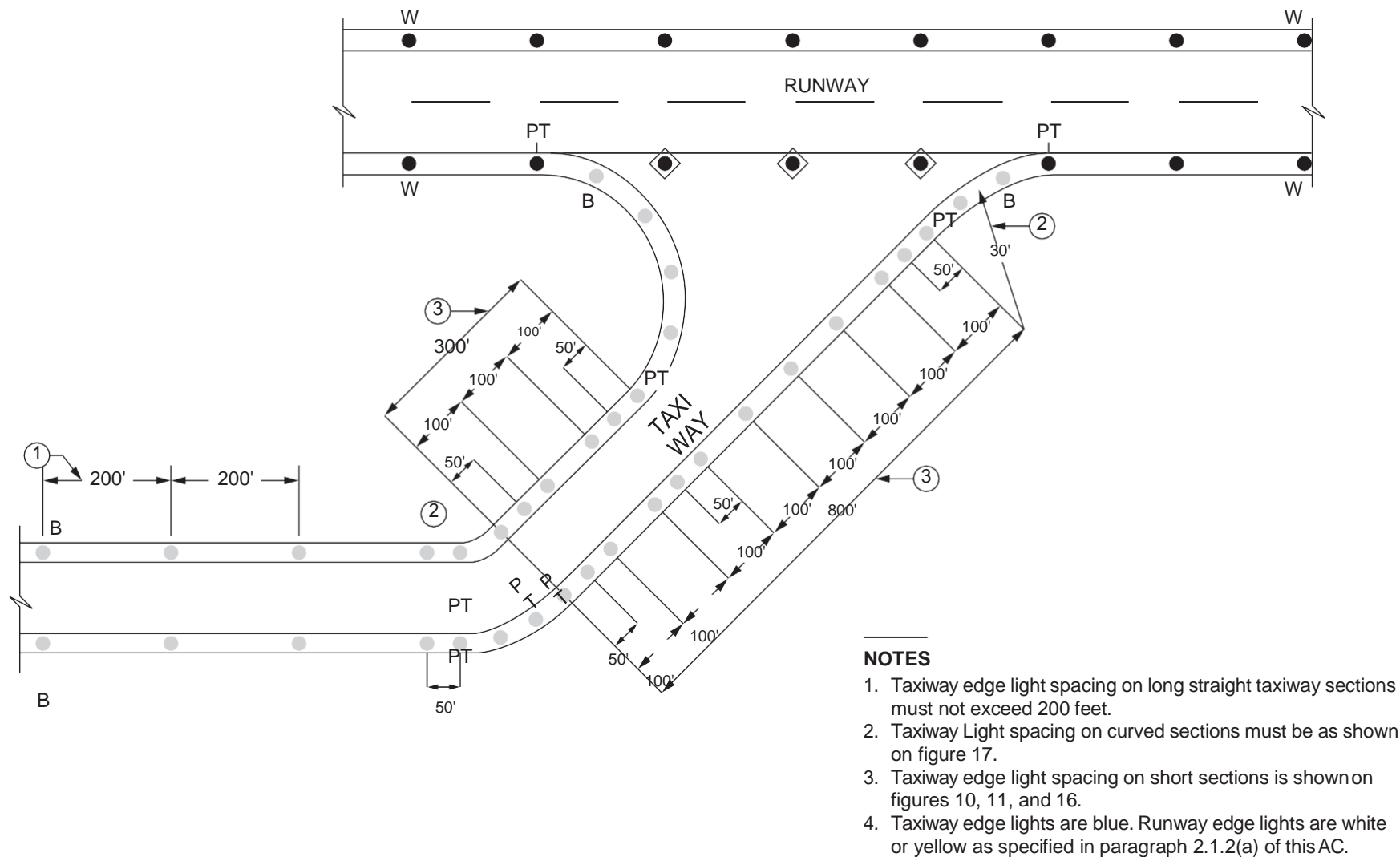


FIGURE 8-16 Taxiway edge lighting configurations on straight line sections, curves, and runway intersection.

	Maximum Longitudinal Spacing	
	1200 ft (365 m) RVR and Above	Below 1200 ft (365 m) RVR
Radius of curved centerlines		
75 ft (23 m) to 399 ft (121 m)	25 ft (7.6 m)	12.5 ft (4 m) 25 ft (7.6 m)
400 ft (122 m) to 1199 ft (364 m)	50 ft (15 m)	25 ft (7.6 m)
≥1200 ft (365 m)	100 ft (30 m)	50 ft (15 m)
Acute-angled exits 50 ft (15 m)		50 ft (15 m)
Straight segments 100 ft (30 m)		50 ft (15 m)

TABLE 8-1 Taxiway Centerline Lighting Spacing

Taxiway centerline lights are in-pavement bidirectional lights placed in equal intervals over taxiway centerline markings. Taxiway centerline lights are green, except in areas where the taxiway intersects with a runway, where the green and yellow lights are placed alternatively. Research and experience have demonstrated that guidance from centerline lights is superior to that from edge lights, particularly in low visibility conditions. The spacing of the lights on curves and tangents is given in Table 8-1 [21].

For normal exits, the centerline lights are terminated at the edge of the runway. At taxiway intersections the lights continue across the intersection. For long-radius high-speed exit taxiways, the taxiway lights are extended onto the runway from a point 200 ft back from the point of curvature (PC) of the taxiway to the point of tangency of the central curve of the taxiway. Within these limits the spacing of lights is 50 ft. These lights are offset 2 ft from the runway centerline lights and are gradually brought into alignment with the centerline of the taxiway.

Where the taxiways intersect with runways and aircraft are required to hold short of the runway, several yellow lights spaced at 5-ft intervals are placed transversely across the taxiway.

Runway Guard Lights

Runway guard lights (RGLs) are in-pavement lights located on taxiways at intersections of runways to alert pilots and operators of airfield ground vehicles that they are about to enter onto an active runway. RGLs are located across the width of the taxiway, approximately 2 ft from the entrance to a runway, spaced at approximately 10-ft intervals, as illustrated in Fig. 8-17. RGLs are unidirectional, colored yellow for aircraft facing the runway.

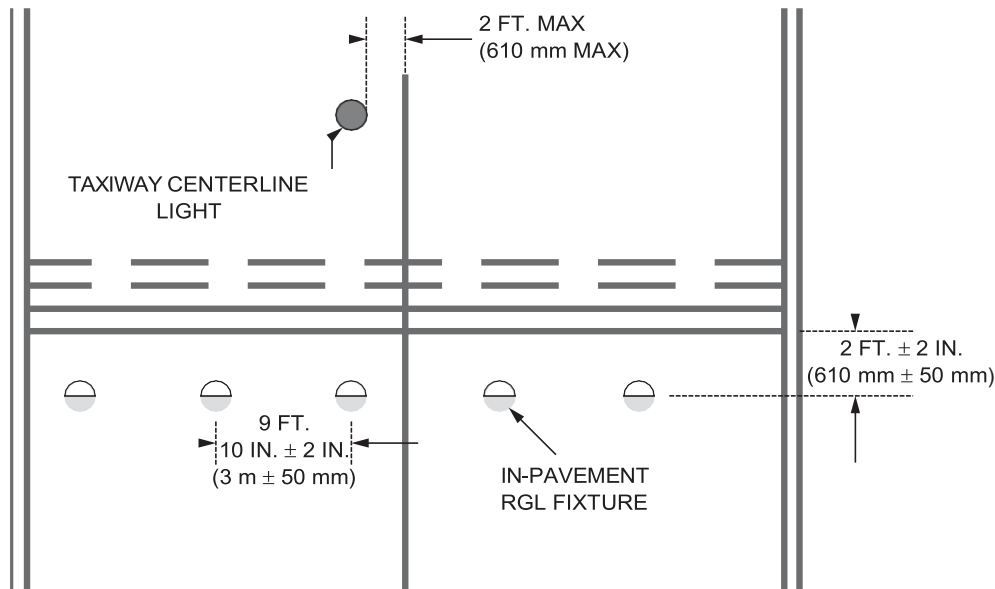


FIGURE 8-17 Runway guard lights.

Runway Stop Bar

Similar to runway guard lights, runway stop bar lights are in-pavement lights on taxiways at intersections with runways. As opposed to RGLs that provide warning to pilots approaching a runway, runway stop bar lights are designed to act as “stop” lights, directing aircraft and vehicles on the taxiway not to enter the runway environment. Runway stop bar lights are activated with red illuminations during periods of runway occupancy or other instances where entrance from the taxiway to the runway is prohibited. In-pavement runway stop bar lighting is typically installed in conjunction with elevated runway guard lights located outside the width of the pavement. An illustration of a typical runway stop bar lighting system is depicted in Fig. 8-18.

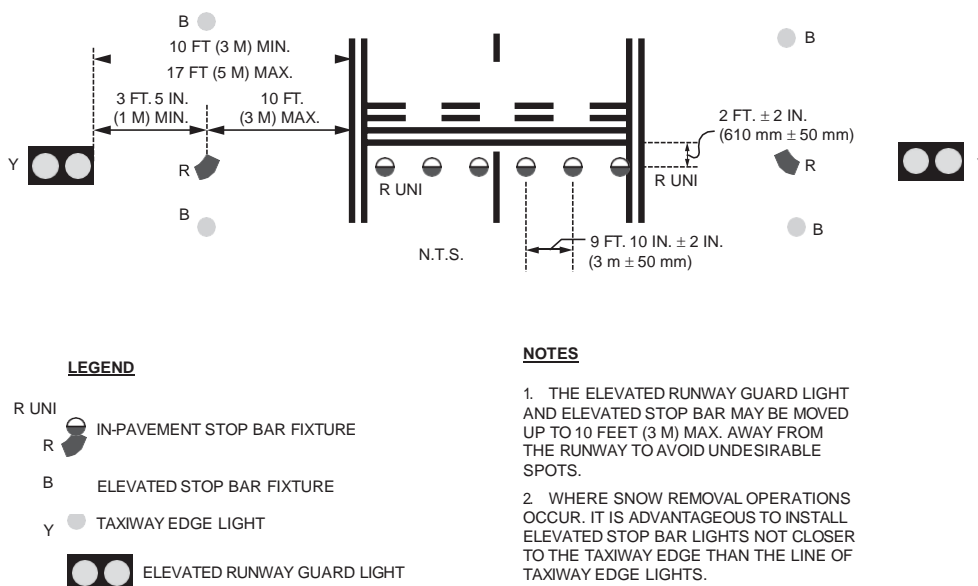


FIGURE 8-18 Runway stop bar lighting.

Runway and Taxiway Marking

In order to aid pilots in guiding the aircraft on runways and taxiways, pavements are marked with lines and numbers. These markings are of benefit primarily during the day and dusk. At night, lights are used to guide pilots in landing and maneuvering at the airport. White is used for all markings on runways and yellow is used on taxiways and aprons. The FAA has developed a comprehensive plan for marking runways and taxiways and they can be found in the FAA Advisory Circular AC 150/5340-1J [17]. Similarly the ICAO recommendations for marking are contained in Annex 14 [2].

Runways

The FAA has grouped runways for marking purposes into three classes: (1) visual, or “basic” runways, (2) nonprecision instrument runways, and (3) precision instrument runways. The visual runway is a runway with no straight-in instrument approach procedure and is intended solely for the operation of aircraft using visual approach procedures. The nonprecision instrument runway is one having an existing instrument approach procedure utilizing air navigation facilities with only horizontal guidance (typically VOR or GPS-based RNAV approaches without vertical guidance) for which a straight-in nonprecision approach procedure has been approved. A precision instrument runway is one having an existing instrument approach procedure utilizing a precision instrument landing system or approved GPS-based RNAV (area navigation) or RNP (required navigation performance) precision approach. Runways that have a published approach based solely on GPS-based technologies are known as GPS runways.

Runway markings include runway designators, centerlines, threshold markings, aiming points, touchdown zone markings, and side stripes. Depending on the length and class of runway and the type of aircraft operations intended for use on the runway, all or some of the above markings are required. Table 8-2 provides the marking requirements for visual, nonprecision, and precision runways.

Figure 8-19 illustrates the required marking for precision runways. Figure 8-20 illustrates the required markings for nonprecision and visual runways (source: FAA AC 150/5340-1J).

Runway Designators

The end of each runway is marked with a number, known as a runway designator, which indicates the approximate magnetic azimuth (clockwise from magnetic north) of the runway in the direction of operations. The marking is given to the nearest 10° with the last digit omitted. Thus a runway in the direction of an azimuth of 163° would be marked as runway 16 and this runway would be in the approximate direction of south-south-east. Therefore, the east end of an east-west runway

Marking Element	Visual Runway	Nonprecision Runway/GPS Nonprecision	Precision Runway/GPS Precision
Designation	X	X	X
Centerline	X	X	X
Threshold marking	X*	X	X
Aiming point	X [†]	X [†]	X
Touchdown zone			X
Side stripes	X [‡]	X [‡]	X

*Only required on runways used, or intended to be used, by international commercial transport.

[†]On runways 4000 ft (1200 m) or longer used by jet aircraft.

[‡]Used when the full pavement width may not be available as a runway.

TABLE 8-2 Required Runway Markings

would be marked 27 (for 270° azimuth) and the west end of an east-west runway would be marked 9 (for a 90° azimuth). If there are two parallel runways in the east-west direction, for example, these runways would be given the designation 9L-27R and 9R-27L to indicate the direction of each runway and their position (L for left and R for right) relative to each other in the direction of aircraft operations. If a third parallel runway existed in this situation it has traditionally been given the designation 9C-27C to indicate its direction and position relative (C for center) to the other runways in the direction of aircraft operations. When there are four parallel runways, one pair is marked with the magnetic azimuth to the nearest 10° while the other pair is marked with the magnetic azimuth to the next nearest 10°. Therefore, if there were four parallel runways in the east-west direction, one pair would be designated as 9L-27R and 9R-27L and the other pair could be designated as either 10L-28R and 10R-28L or 8L-26R and 8R-26L. This type of designation policy is increasingly being applied to three parallel runway configurations, as well. For example, one pair would be designated as 9L-27R and 9R-27L and the third runway may be designated 10-28.

Runway designation markings are white, have a height of 60 ft and a width, depending upon the number or letter used, varying from 5 ft for the numeral 1 to 23 ft for the numeral 7. When more than one number or letter is required to designate the runway the spacing between the designators is normally 15 ft. The sizes of the runway designator markings are proportionally reduced only when necessary due to space limitations on narrow runways and these designation markings should be no closer than 2 ft from the edge of the runway or the runway edge stripes. Specifications for individual runway designators are illustrated in Fig. 8-21.

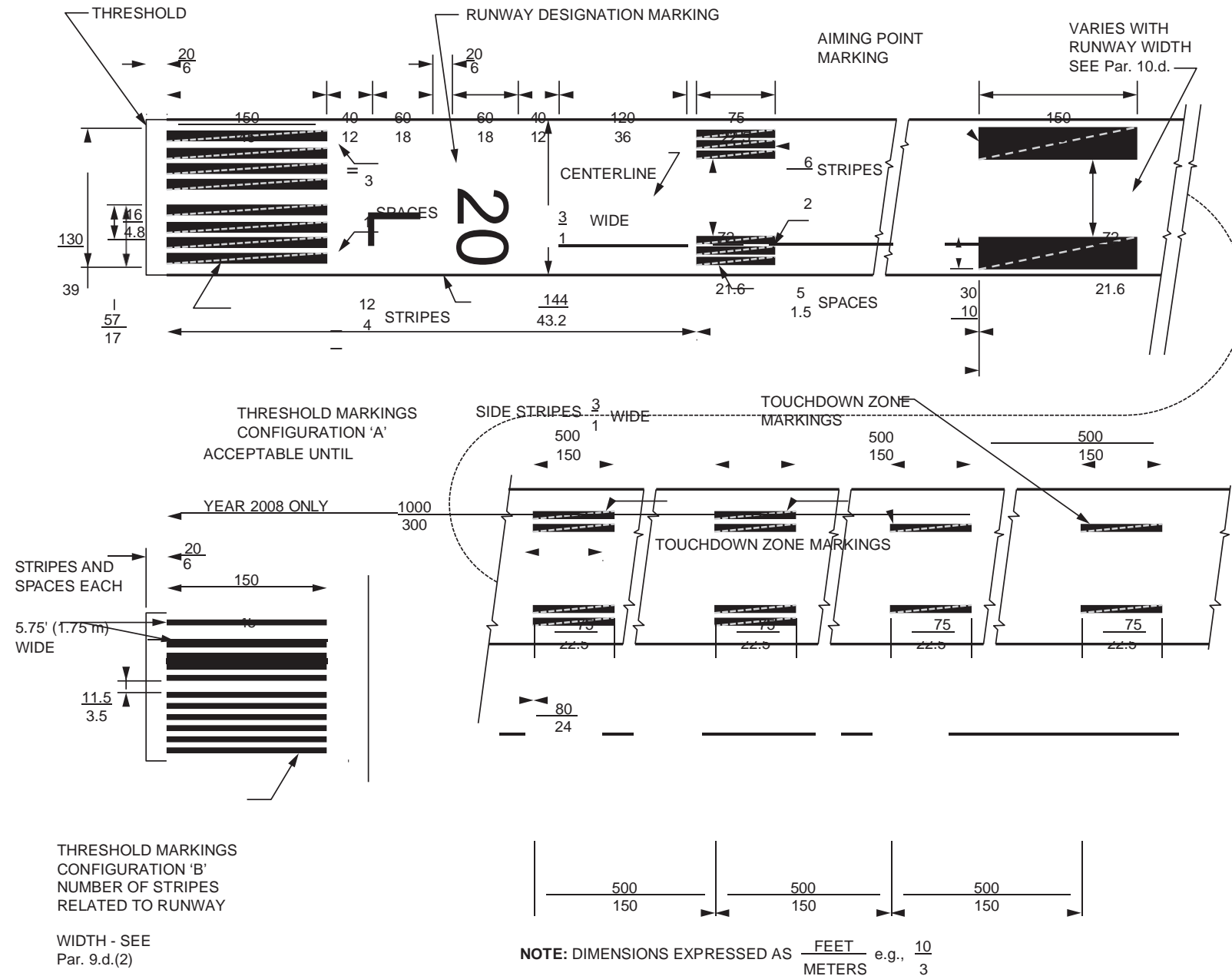
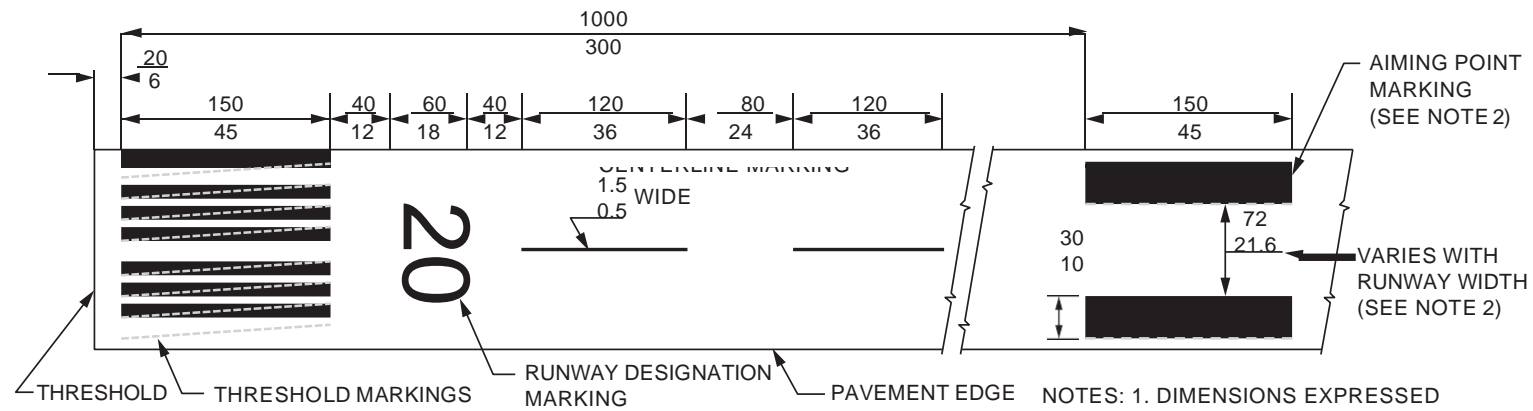


FIGURE 8-19 Precision runway markings.



SEE FIGURE 1 FOR
ALTERNATE

CONFIGURATION

NONPRECISION RUNWAY MARKINGS

2. SEE Par. 10.d. FOR RUNWAY WIDTHS
OTHER THAN 150' (45 m) AND
REQUIREMENTS FOR AIMING
POINT MARKING

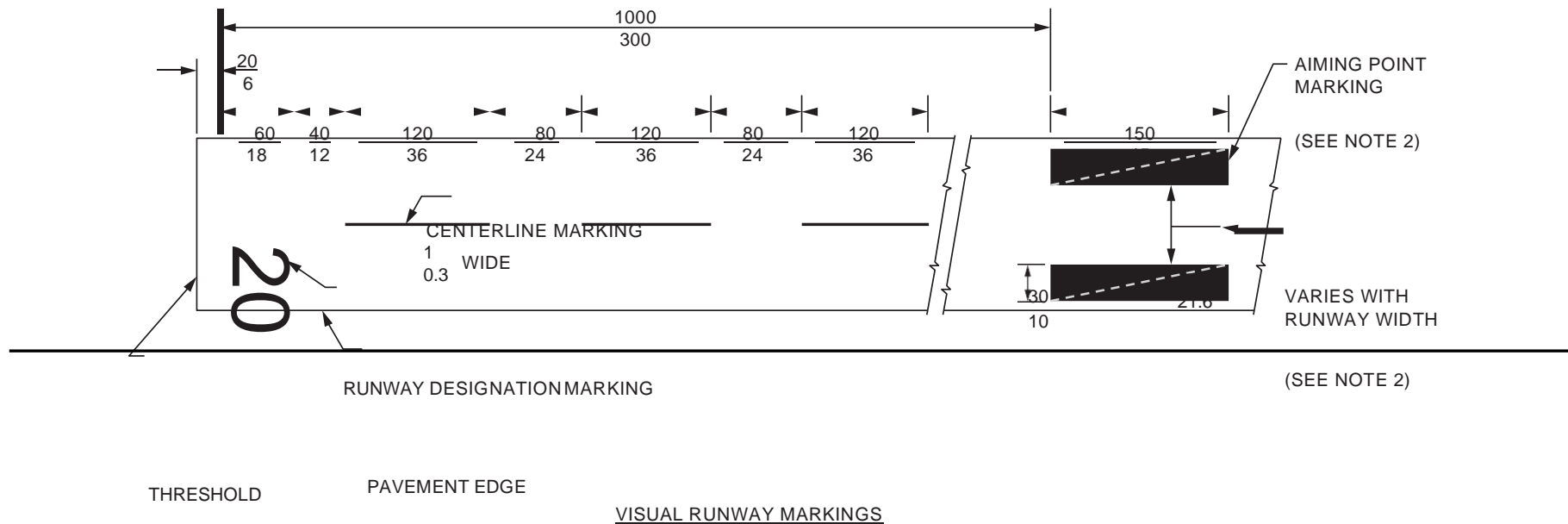
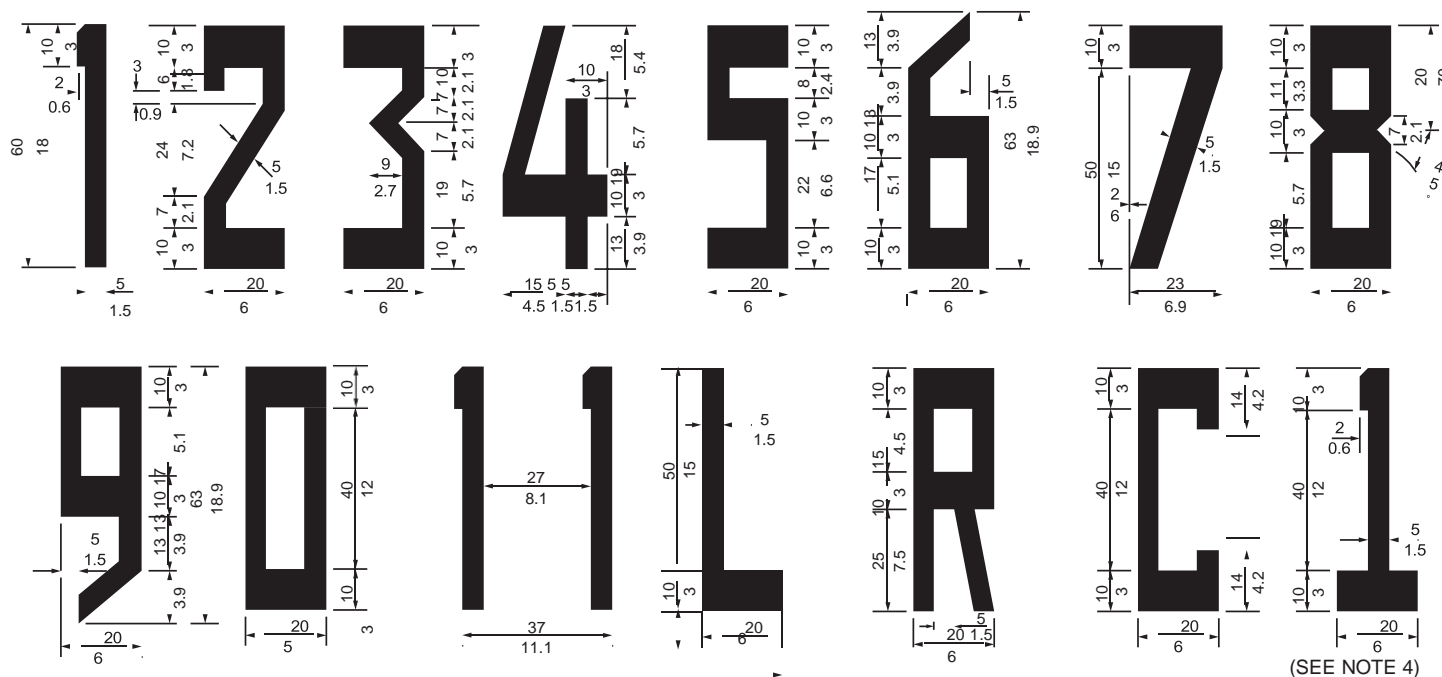


FIGURE 8-20 Nonprecision and visual runway markings.



NOTES:

1. ALL NUMERALS EXCEPT THE NUMBER ELEVEN AS SHOWN ARE HORIZONTALLY SPACED 15 FEET (4.5 METERS) APART.
2. SINGLE DIGITS SHALL NOT BE PRECEDED BY A ZERO.
3. DIMENSIONS ARE EXPRESSED THUS: FEET e.g., 30

METERS 9
4. THE NUMERAL 1, WHEN USED ALONE, CONTAINS A HORIZONTAL BAR TO DIFFERENTIATE IT FROM THE RUNWAY CENTERLINE MARKING.
5. SINGLE DESIGNATIONS ARE CENTERED ON THE RUNWAY PAVEMENT CENTERLINE. FOR DOUBLE DESIGNATIONS, THE CENTER OF THE OUTER EDGES OF THE TWO NUMERALS IS CENTERED ON THE RUNWAY PAVEMENT CENTERLINE.
6. WHERE THE RUNWAY DESIGNATION CONSISTS OF A NUMBER AND A LETTER, THE NUMBER AND LETTER ARE LOCATED ON THE RUNWAY CENTERLINE IN A STACKED ARRANGEMENT AS SHOWN IN FIGURE 1.

FIGURE 8-21 Runway designators.

Runway Width	Number of Stripes
60 ft (18 m)	4
75 ft (23 m)	6
100 ft (30 m)	8
150 ft (45 m)	12
200 ft (60 m)	16

TABLE 8-3 Striping Requirements for Runway Threshold Markings

Runway Threshold Markings

Runway threshold markings identify to the pilot the beginning of the runway that is safe and available for landing. Runway threshold markings begin 20 ft from the runway threshold itself.

Runway threshold markings consist of two series of white stripes, each stripe 150 ft in length and 5.75 ft in width, separated about the centerline of the runway. On each side of the runway centerline, a number of threshold marking stripes are placed, in accordance with the width of the runway, as specified in Table 8-3. Table 8-3 specifies the total number of runway threshold stripes required. For example, for a 100-ft runway, eight stripes are required, in two groups of four are placed about the centerline. Stripes within each set are separated by 5.75 ft. Each set of stripes is separated by 11.5 ft about the runway centerline.

The above specifications for runway threshold markings were adapted by the FAA from ICAO international standards and made mandatory for United States civil use airports in 2008.

Centerline Markings

Runway centerline markings are white, located on the centerline of the runway, and consist of a line of uniformly spaced stripes and gaps. The stripes are 120 ft long and the gaps are 80 ft long. Adjustments to the lengths of stripes and gaps, where necessary to accommodate runway length, are made near the runway midpoint. The minimum width of stripes is 12 in for visual runways, 18 in for nonprecision instrument runways, and 36 in for precision instrument runways. The purpose of the runway centerline markings is to indicate to the pilot the center of the runway and to provide alignment guidance on landing and takeoff.

Aiming Points

Aiming points are placed on runways of at least 4000 ft in length to provide enhanced visual guidance for landing aircraft. Aiming point markings consist of two bold stripes, 150 ft long, 30 ft wide, spaced

Runway Length	Markings on Each End
7990 ft (2436 m) or greater	Full set of markings
6990 ft (2130 m) to 7989 ft (2435 m)	Less one pair of markings
5990 ft (1826 m) to 6989 ft (2129 m)	Less two pairs of markings
4990 ft (1521 m) to 5989 ft (1825 m)	Less three pairs of markings

TABLE 8-4 Touchdown Zone Marking Requirements

72 ft apart symmetrically about the runway centerline, and beginning 1020 ft from the threshold.

Touchdown Zone Markings

Runway touchdown zone markings are white and consist of groups of one, two, and three rectangular bars symmetrically arranged in pairs about the runway centerline. These markings begin 500 ft from the runway threshold. The bars are 75 ft long, 6 ft wide, with 5 ft spaces between the bars, and are longitudinally spaced at distances of 500 ft along the runway. The inner stripes are placed 36 ft on either side of the runway centerline. For runways less than 150 ft in width, the width and spacing of stripes may be proportionally reduced. Where touchdown zone markings are installed on both runway ends on shorter runways, those pairs of markings which would extend to within 900 ft of the runway midpoint are eliminated. In addition, sets of touchdown zone markings are eliminated for shorter runways, as specified in Table 8-4.

Side Stripes

Runway side stripes consist of continuous white lines along each side of the runway to provide contrast with the surrounding terrain or to delineate the edges of the full strength pavement. The maximum distance between the outer edges of these markings is 200 ft and these markings have a minimum width of 3 ft for precision instrument runways and are at least as wide as the width of the centerline stripes on other runways.

Displaced Threshold Markings

At some airports it is desirable or necessary to “displace” the runway threshold on a permanent basis. A displaced threshold is one which has been moved a certain distance from the end of the runway. Most often this is necessary to clear obstructions in the flight path on landing. The displacement reduces the length of the runway available for landings, but takeoffs can use the entire length of the runway. The FAA

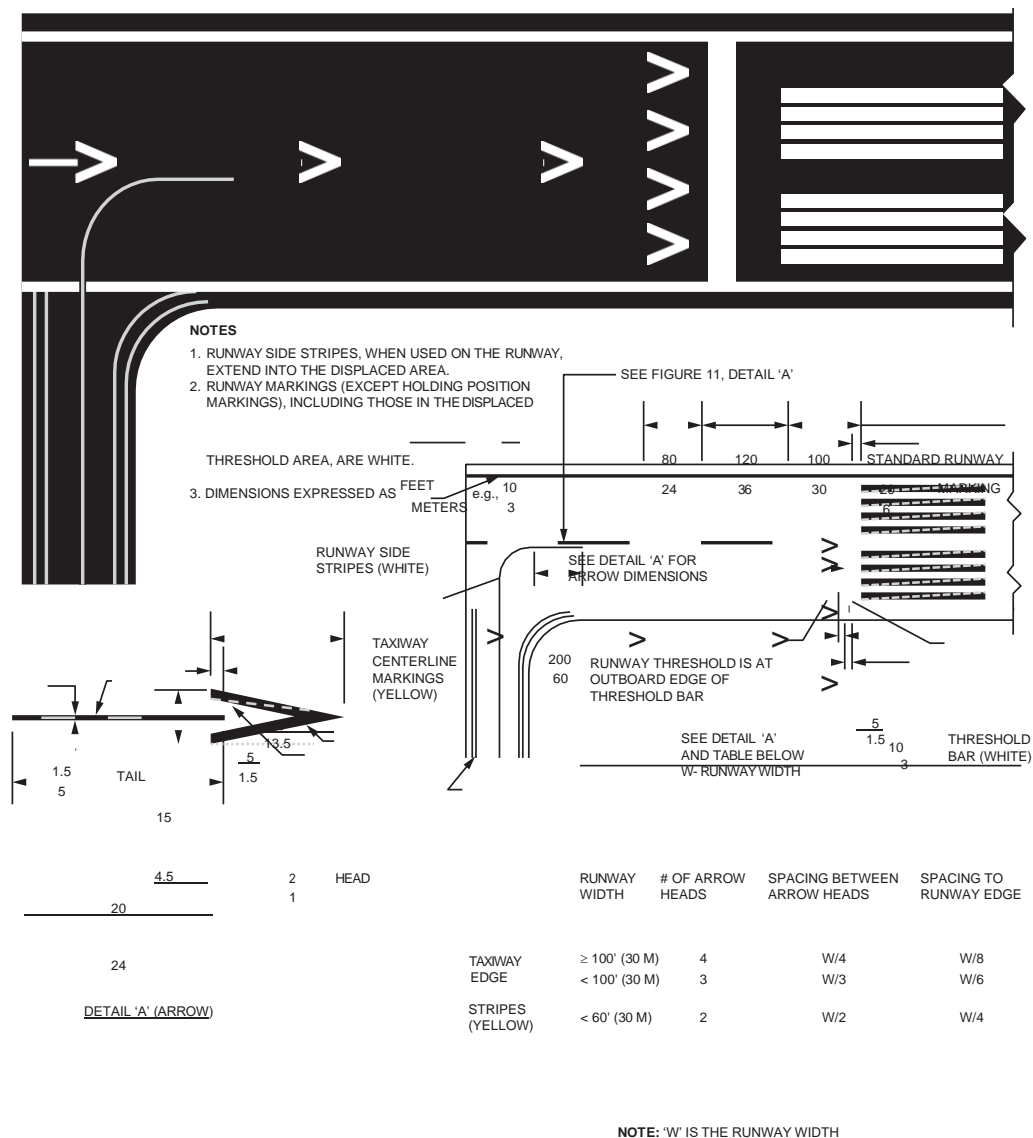


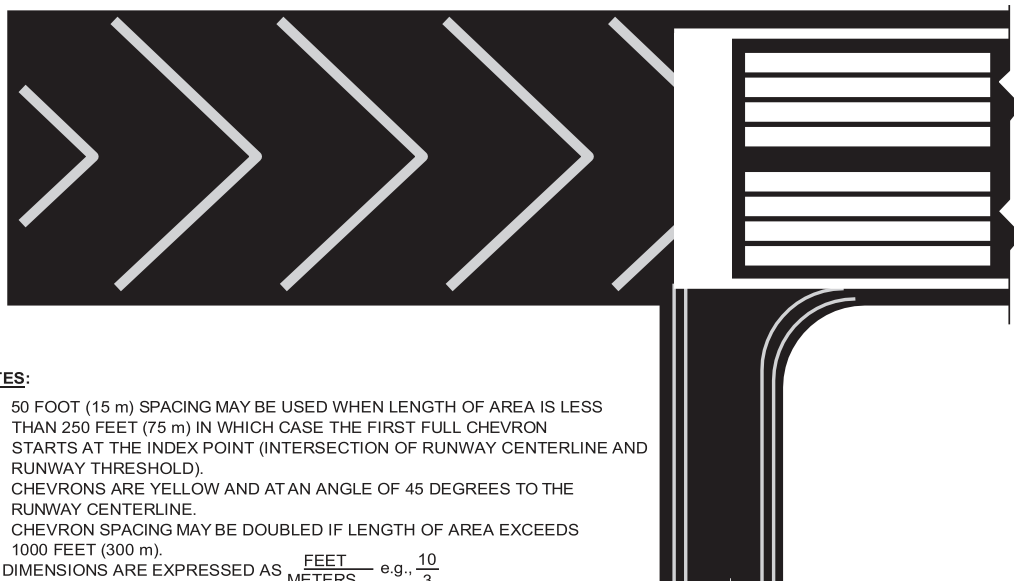
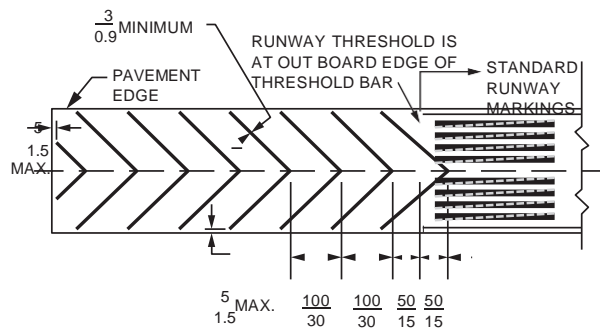
FIGURE 8-22 Displaced threshold markings.

requires that displaced thresholds be marked as shown in Fig. 8-22. These markings consist of arrows and arrow heads to identify the displaced threshold and a threshold bar to identify the beginning of the runway threshold itself. Displaced threshold arrows are 120 ft in length, separated longitudinally by 80 ft for the length of the displaced threshold. Arrow heads are 45 ft in length, placed 5 ft from the threshold bar. The threshold bar is 5 ft in width and extends the width of the runway at the threshold.

Blast Pad Markings

In order to prevent erosion of the soil, many airports provide a paved *blast pad* 150 to 200 ft in length adjacent to the runway end. Similarly, some airport runways have a *stopway* which is only designed to support aircraft during rare aborted takeoffs or landing overruns and is not designed as a full strength pavement. Since these paved areas are not designed to support aircraft and yet may have the appearance of being

so designed, markings are required to indicate this. The markings for blast pads and stopways are shown in Fig. 8-23. Likewise the area adjacent to the edge of the runway may have a paved shoulder not capable

**NOTES:**

1. 50 FOOT (15 m) SPACING MAY BE USED WHEN LENGTH OF AREA IS LESS THAN 250 FEET (75 m) IN WHICH CASE THE FIRST FULL CHEVRON STARTS AT THE INDEX POINT (INTERSECTION OF RUNWAY CENTERLINE AND RUNWAY THRESHOLD).
2. CHEVRONS ARE YELLOW AND AT AN ANGLE OF 45 DEGREES TO THE RUNWAY CENTERLINE.
3. CHEVRON SPACING MAY BE DOUBLED IF LENGTH OF AREA EXCEEDS 1000 FEET (300 m).
4. DIMENSIONS ARE EXPRESSED AS $\frac{\text{FEET}}{\text{METERS}}$ e.g., $\frac{10}{3}$

FIGURE 8-23 Blast pad markings.

of supporting aircraft. These areas are marked with a 3-ft-wide stripe, as shown in Fig. 8-24. Yellow color is used for these types of markings.

Taxiway Markings

Taxiway markings consist of centerline markings, holding position markings, and often edge markings. Taxiways are marked as shown in Fig. 8-25.

Centerline and Edge Markings

The centerline of the taxiway is marked with a single continuous 6-in yellow line. On taxiway curves, the taxiway centerline marking continues from the straight portion of the taxiway at a constant distance from the outside edge of the curve. At taxiway intersections which are designed for aircraft to travel straight through the intersection, the centerline markings continue straight through the intersection. At the intersection of a taxiway with a runway end, the centerline stripe of the taxiway terminates at the edge of the runway.

At the intersection between a taxiway and a runway, where the taxiway serves as an exit from the runway, the taxiway marking is usually

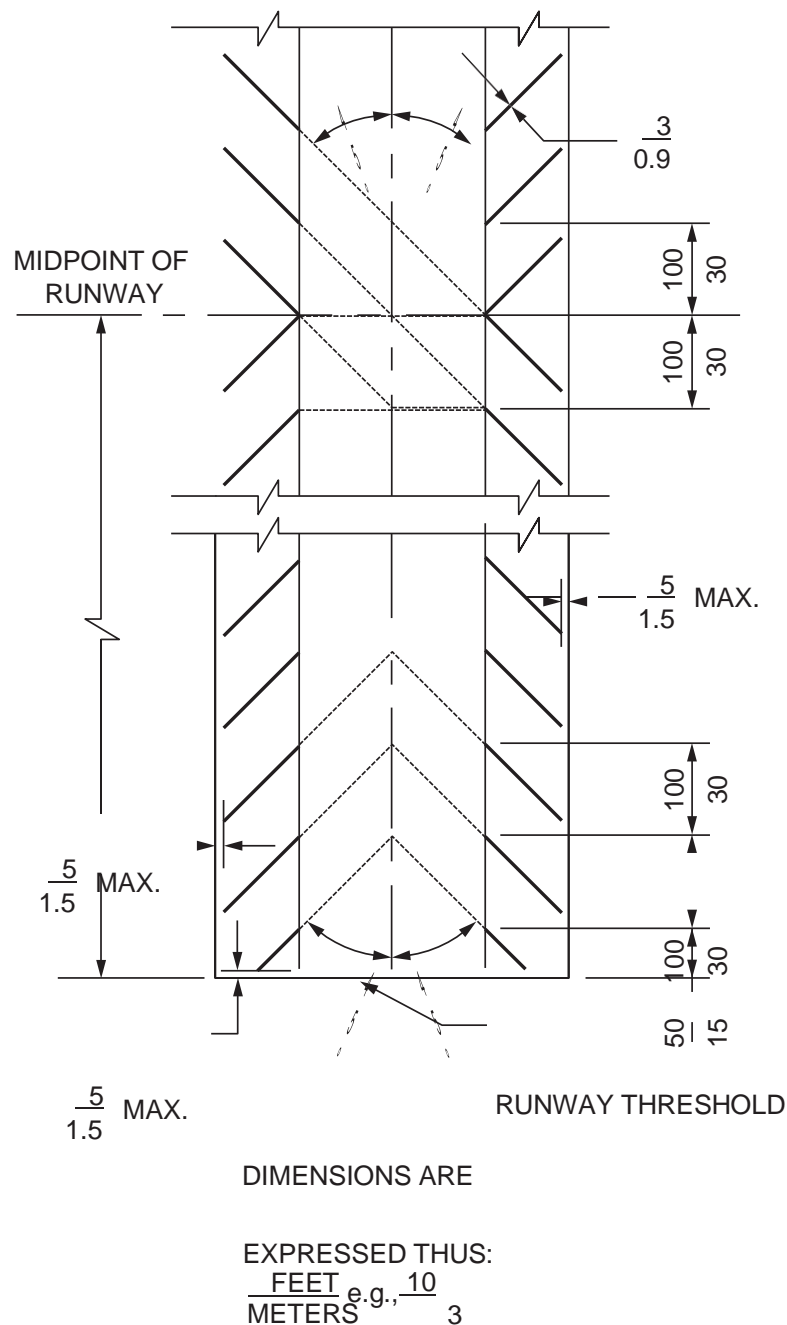


FIGURE 8-24 Runway shoulder markings.

extended on to the runway in the vicinity of the runway centerline marking. The taxiway centerline marking is extended parallel to the runway centerline marking a distance of 200 ft beyond the point of tangency. The taxiway curve radius should be large enough to provide a clearance to the taxiway edge and the runway edge of at least one-half the width of the taxiway. For a taxiway crossing a runway, the taxiway centerline marking may continue across the runway but it must be interrupted for the runway markings.

When the edge of the full strength pavement of the taxiway is not readily apparent, or when a taxiway must be outlined when it is

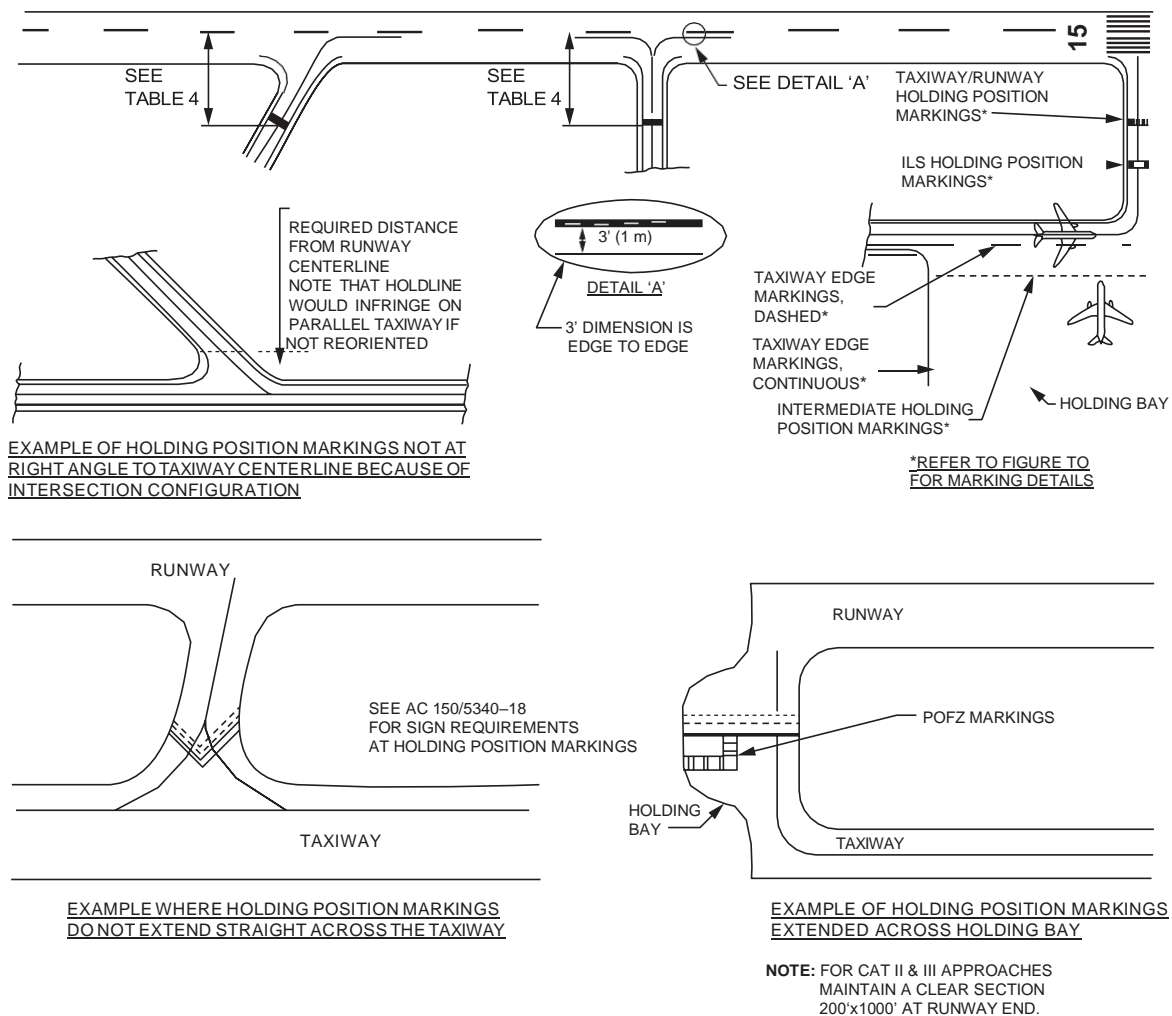


FIGURE 8-25 Taxiway markings.

established on a large paved area such as an apron, the edge of the taxiway is marked with two continuous 6-in wide yellow stripes that are 6 in apart.

Taxiway Hold Markings

For taxiway intersections where there is an operational need to hold aircraft, a dashed yellow holding line is placed perpendicular to and across the centerline of both taxiways.

When a taxiway intersects a runway or a taxiway enters an instrument landing system critical area, a holding line is placed across the taxiway. The holding line for a taxiway intersecting a runway consists of two solid lines of yellow stripes and two broken lines of yellow stripes placed perpendicular to the centerline of the taxiway and across the width of the taxiway. The solid lines are always placed on the side where the aircraft is to hold. The holding line for an instrument landing system critical area consists of two solid lines placed

Aircraft Approach Category and (Airplane Design Group)	Visual and Nonprecision Instrument, ft (m)	Precision Instrument, ft (m)
A and B (I and II) small airplanes only	125 (38)	175 (53)
A and B (I, II, and III)	200 (60)	250 (75)
A and B (IV)	250 (75)	250 (75)
C and D (I through IV)	250 (75)	250 (75)
C and D (V)	250 (75)	280 (85)
C and D (VI)	250 (75)	280 (85)

Source: AC 150/5340-18D

Distances shown above are for planning purposes only. “Hold position markings” must be placed in order to restrict the largest aircraft (tail or body) expected to use the runway from penetrating the obstacle-free zone.

For aircraft approach categories A and B, airplane design group III, this distance is increased 1 ft for each 100 ft above 5100 ft above sea level. For airplane design group IV, precision instrument runways, this distance is increased 1 ft for each 100 ft above sea level.

For aircraft approach category C, airport design group IV, precision instrument runways. This distance is increased 1 ft for each 100 ft above sea level. For airplane design group V, this distance is increased 1 ft for each 100 ft above sea level.

For aircraft approach category D, this distance is increased 1 ft for each 100 ft above sea level.

TABLE 8-5 Location Distances for Holding Position Markings

perpendicular to the taxiway centerline and across the width of the taxiway joined with three sets of two solid lines symmetrical about and parallel to the taxiway centerline. These holding lines are located the minimum distance from the centerline of the runway as indicated in Table 8-5 and illustrated in Fig. 8-26.

Taxiway Shoulders

In some areas on the airfield, the edges of taxiways may not be well-defined due to their adjacency to other paved areas such as aprons and holding bays. In these areas, it is prudent to mark the edges of taxiways with shoulder markings. Taxiway shoulder markings are yellow in color, and are often painted on top of a green background. The shoulder markings consist of 3-ft-long yellow stripes placed perpendicular to the taxiway edge stripes, as illustrated in Fig. 8-27. On straight sections of the taxiway, the marks are placed at a maximum spacing of 100 ft. On curves, the marks are placed on a maximum of 50 ft apart between the curve tangents.

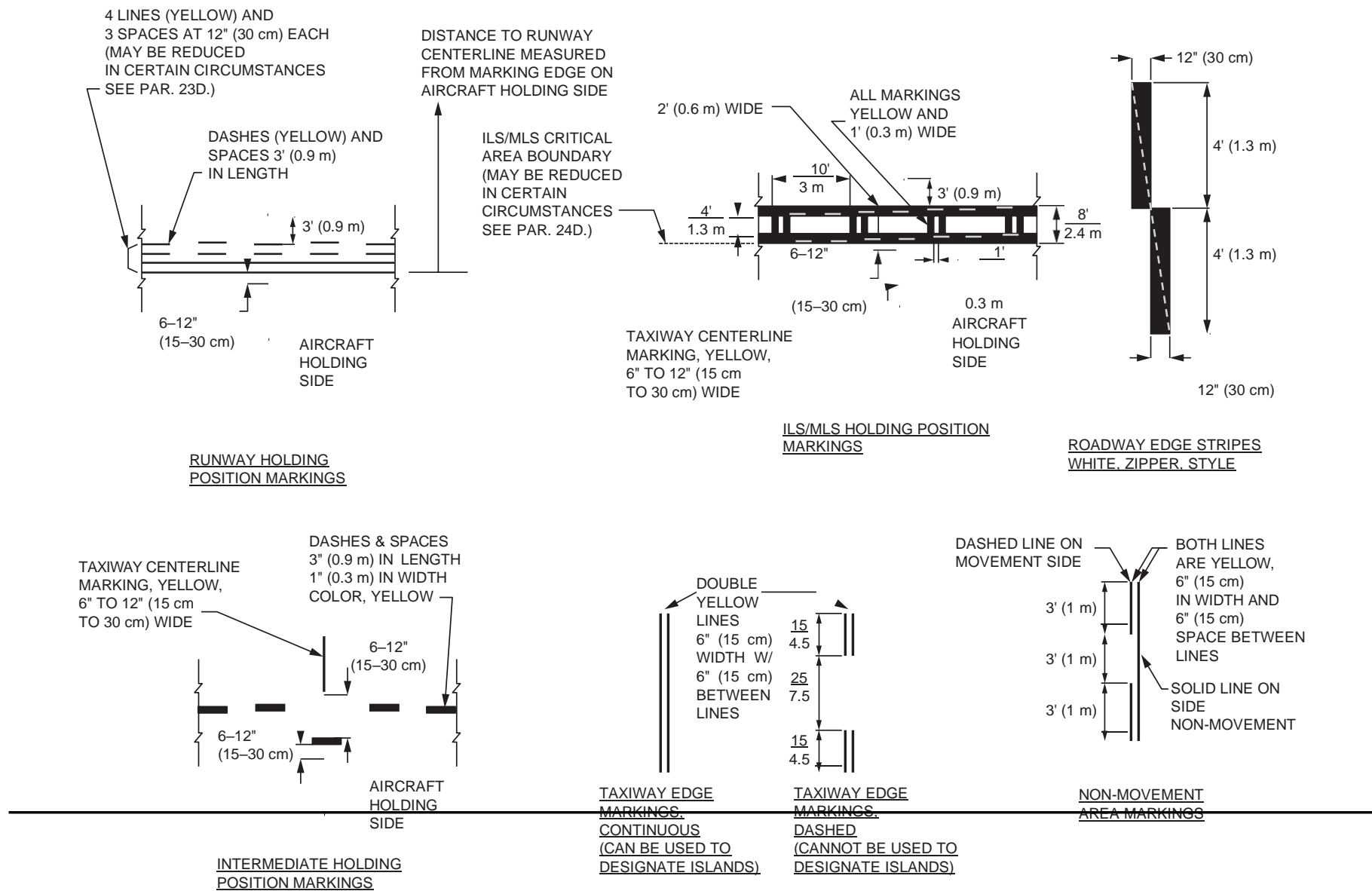


FIGURE 8-26 Taxiway hold short and edge markings.

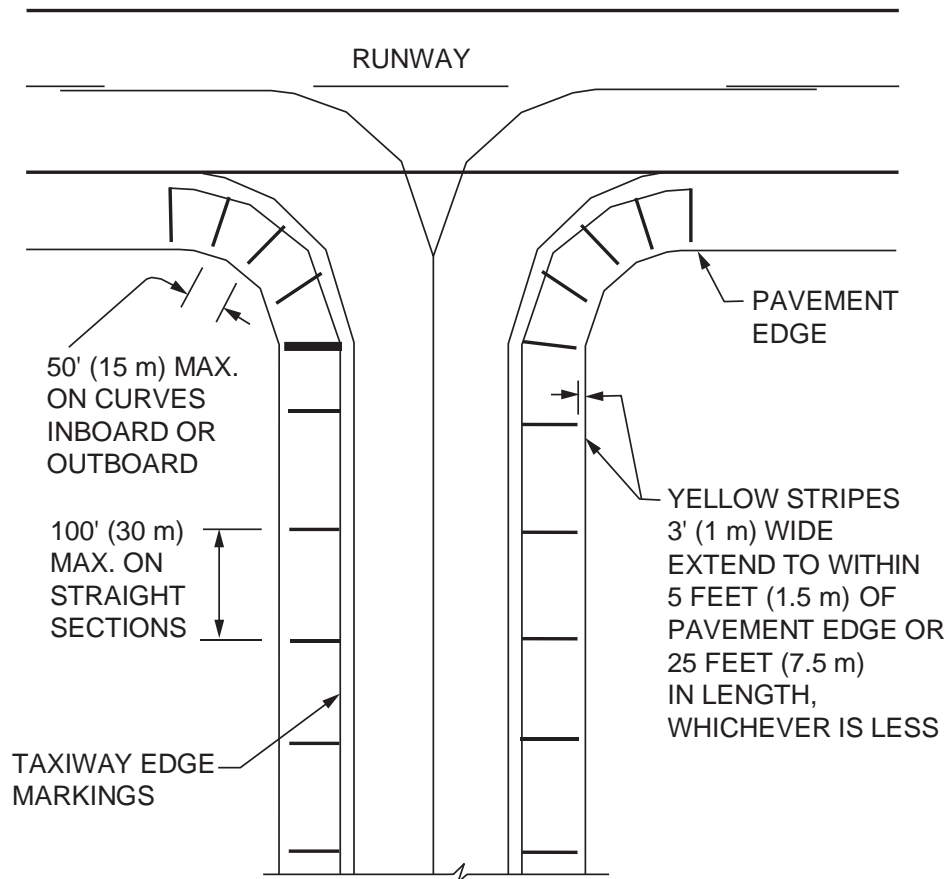


FIGURE 8-27 Taxiway shoulder markings.

Enhanced Taxiway Markings

Beginning in 2008, all airports serving commercial air carriers are required to mark certain critical areas of the airfield with enhanced taxiway markings. These markings are designed to provide additional guidance and warning to pilots of runway intersections. Enhanced markings consist primarily of yellow-painted lines, using paint mixtures with imbedded glass beads to enhance visibility. In addition, yellow markings must be marked on top of a darkened black background.

Taxiway centerlines are enhanced for 150 ft from the runway hold-short markings. The centerline enhancements include dashed yellow lines 9 ft in length, separated longitudinally by 3 ft. These yellow lines are placed 6 in from each end of the existing centerline. An example of enhanced marking is illustrated in Fig. 8-28.

Closed Runway and Taxiway Markings

When runways or taxiways are permanently or temporarily closed to aircraft, yellow crosses are placed on these trafficways. For permanently

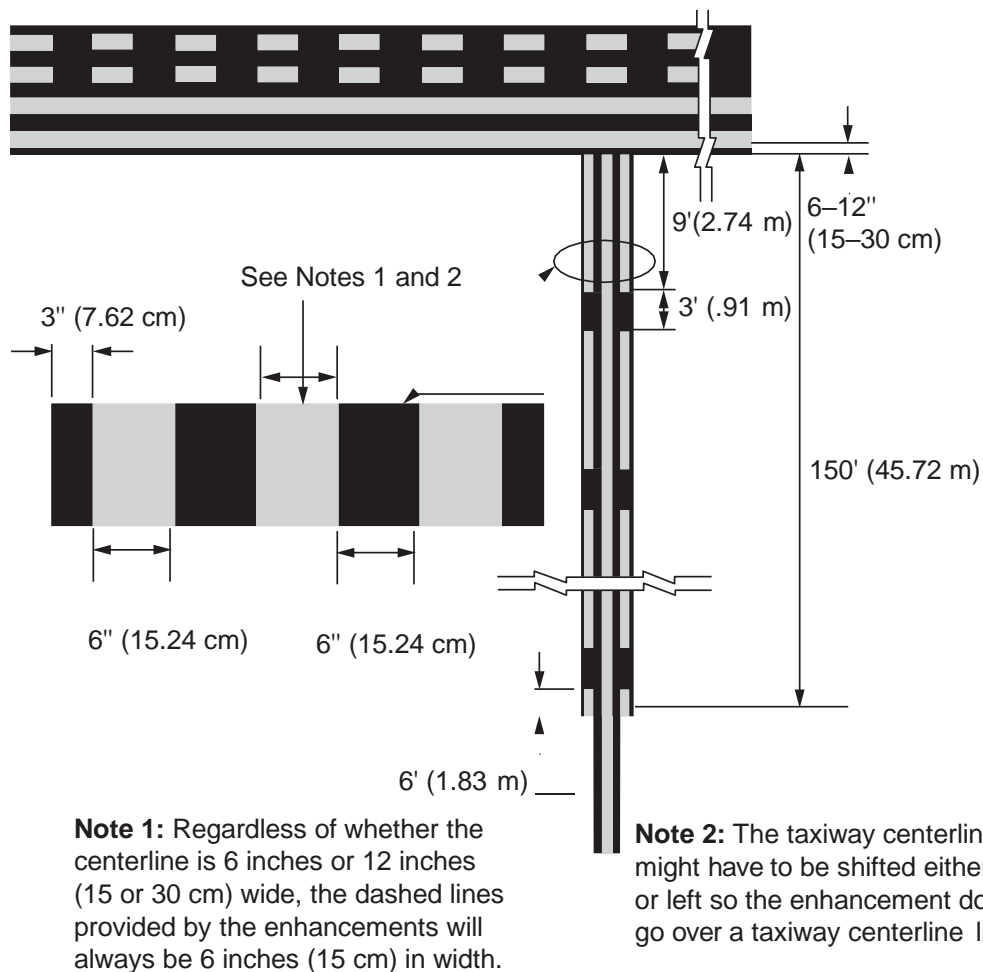


FIGURE 8-28 Example of enhanced taxiway markings.

closed runways, the threshold, runway designation, and touchdown markings are obliterated and crosses are placed at each end and at 1000 ft intervals. For temporarily closed runways, the runway markings are not obliterated, the crosses are usually of a temporary type and are only placed at the runway ends. For permanently closed taxiways, a cross is placed on the closed taxiway at each entrance to the taxiway. For temporarily closed taxiways barricades with orange and white markings are normally erected at the entrances.

Airfield Signage

In addition to markings, signage is placed on the airfield to guide and direct pilots and ground vehicle operators to points on the airport. In addition some signage exists to provide the pilots with information regarding their position on the airfield, the distance remaining on a runway, the location of key facilities at the airport, and often informative signage ranging from voluntary procedures to mitigate noise impacts to warnings about nearby security sensitive areas. FAA Advisory Circular 150/53040-18D describes the U.S. federal standards for airport sign systems.



FIGURE 8-29 Runway distance remaining sign.

Runway Distance Remaining Signs

Runway distance remaining signs are placed on the side of a runway and provide the pilot with information on how much runway is left during takeoff or landing operations. These signs are placed at 1000 ft intervals along the runway in a descending sequential order. Normally, these signs consist of white numerals on a black background, as illustrated in Fig. 8-29.

The FAA recommends that the signs be configured in one of three ways [25]. The preferred method of configuration, and the most economical, is to place double-faced signs on only one side of the runway. In this configuration it is recommended that the signs be placed on the left side of the most frequently used direction of the runway. The signs may be placed on the right side of the runway when necessary due to required runway-taxiway separations or due to conflicts between intersecting runways or taxiways. An alternative method is to provide a set of single-faced signs on either side of the runway to indicate the distance remaining when the runway is used in both directions. The advantage of this configuration is that the distance remaining is more accurately reflected when the runway length is not an even multiple of 1000 ft. Another alternative uses double-faced signs on both sides of the runway. The advantage of this method is that the runway distance is displayed on both sides of the runway in each direction which is an advantage when a sign on one side needs to be omitted because of a clearance conflict. When the runway distance is not an even multiple of 1000 ft, one half of the excess distance is added to the distance on each sign on each runway end. For example, if the runway length available is 8250 ft, the last sign is located at

Sign Size	Legend, in (cm)	Face, in (cm)	Installed (max.), in (cm)	Distance from Defined Pavement Edge
4	40 (100)	48 (120)	60 (152)	50–75 (15–22.5)
5	25 (64)	30 (76)	42 (107)	20–35 (6–10.5)

TABLE 8-6 Runway Distance Remaining Sign Heights and Location Distances

a distance of 1000 plus 125 ft from the end of the runway. A tolerance of ± 50 ft is allowed for the placement of runway distance remaining signs. These signs should be illuminated anytime the runway edge lights are illuminated. The recommended sizes and placement of these signs is given in Table 8-6.

Taxiway Guidance Sign System

The primary purpose of a taxiway guidance sign system is to aid pilots in taxiing on an airport. At controlled airports, the signs supplement the instructions of the air traffic controllers and aid the pilot in complying with those instructions. The sign system also aids the air traffic controller by simplifying instructions for taxiing clearances, and the routing and holding of aircraft. At locations not served by air traffic control towers, or for aircraft without radio contact, the sign system provides guidance to the pilot to major destinations areas in the airport.

The efficient and safe movement of aircraft on the surface of an airport requires that a well-designed, properly thought-out, and standardized taxiway guidance sign system is provided at the airport. The system must provide the pilot with the ability to readily determine the designation of any taxiway on which the aircraft is located, readily identify routings to a desired destination on the airport property, indicate mandatory aircraft holding positions, and identify the boundaries for aircraft approach areas, instrument landing system critical areas, runway safety areas and obstacle free zones. It is virtually impossible, except for holding position signs, to completely specify the locations and types of signs that are required on a taxiway system at a particular airport due to the wide variation in the types of functional layouts for airports. The ICAO also publishes recommendations relative to surface movement guidance and control systems [2, 15].

Taxiway Designations

Taxiway guidance sign systems are in a large part based on a system of taxiway designators which identify the individual taxiway components. While runway designators are based on the magnetic heading of the runway, taxiway designators are assigned based on an

alphabetic ordering system, independent of the taxiways direction of movement. Taxiways are typically identified in alphabetic order from east to west or north to south (i.e., the northern or easternmost taxiway would be designated "A", the next southern or western taxiway would be designated "B," and so forth). Entrance and exit taxiways perpendicular to main parallel taxiways are designated by the letter of the main parallel taxiway from which they spur, followed by a numeric sequence. For instance, the northernmost entrance taxiway off of taxiway "A" would be designated "A1," and so forth. The letters "I" and "O" are not used as taxiway designators due to their similarity in form to the numbers "1" and "0." In addition the letter "X" is not used as a taxiway designator due to its similarity to a closed runway marking. An example taxiway designation scheme is illustrated in Fig. 8-30.

The taxiway guidance sign system consists of four basic types of signs: *mandatory instruction* signs, which indicate that aircraft should not proceed beyond a point without positive clearance, *location* signs, which indicate the location of an aircraft on the taxiway or runway system and the boundaries of critical airfield surfaces, *direction* signs, which identify the paths available to aircraft at intersections, and *destination* signs, which indicate the direction to a particular destination.

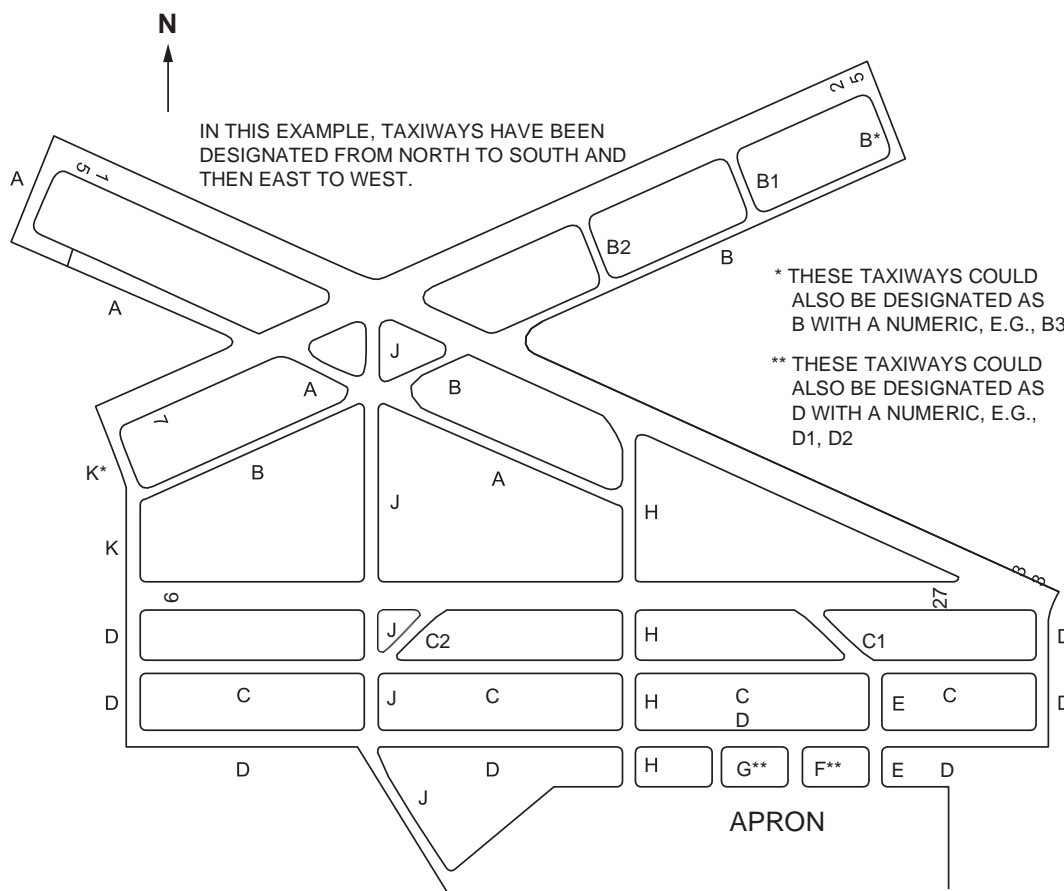


FIGURE 8-30 Example of taxiway designation scheme.

Types of Taxiway Signs

Mandatory Instruction Signs

Mandatory instruction signs denote an entrance to a runway, critical area, or prohibited area. They are used for holding positions signs for runway-taxiway and runway-runway intersections (Fig. 8-31*a*), instrument landing system critical areas (Fig. 8-31*b*), runway approach areas (Fig. 8-31*c*), ILS category II/III critical areas (Fig. 8-31*d*), and to designate areas for which entry is prohibited by aircraft (Fig. 8-31*e*). These signs have white inscriptions on a red background and are installed on the left side of the runway or taxiway. In some cases runway-taxiway intersections require a sign on both sides of the taxiway. This includes situations on taxiways which are at least 150 ft wide, where the painted holding line extends across an adjacent holding bay, where the painted holding line markings do not extend straight across the taxiway, and where the painted holding line markings are located a short distance from an intersection with another taxiway. Generally arrows are not permitted on mandatory instruction signs unless they are necessary at the taxiway-runway-runway intersections to indicate directions to these runways. For runway designation these signs normally contain both designations of the runway and the designation on the left is for the runway to the left and the designation on the right is for the runway to the right.

At controlled airports aircraft and ground vehicles are required to hold at these points unless cleared by air traffic control. At uncontrolled airports, these signs are intended to indicate travel beyond these signs is permitted only after appropriate precautions have been taken.

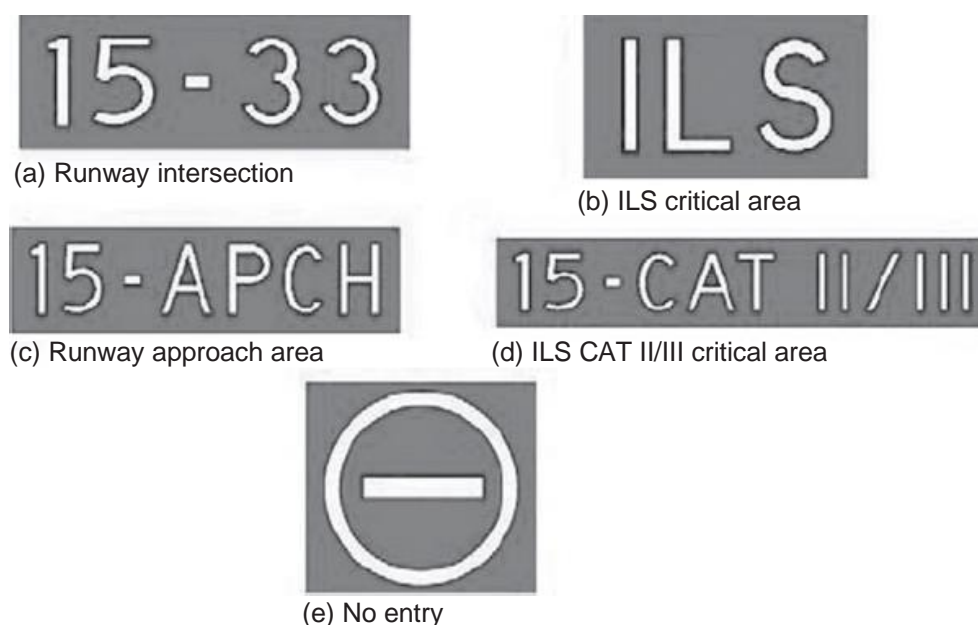


FIGURE 8-31 Mandatory instruction signs.

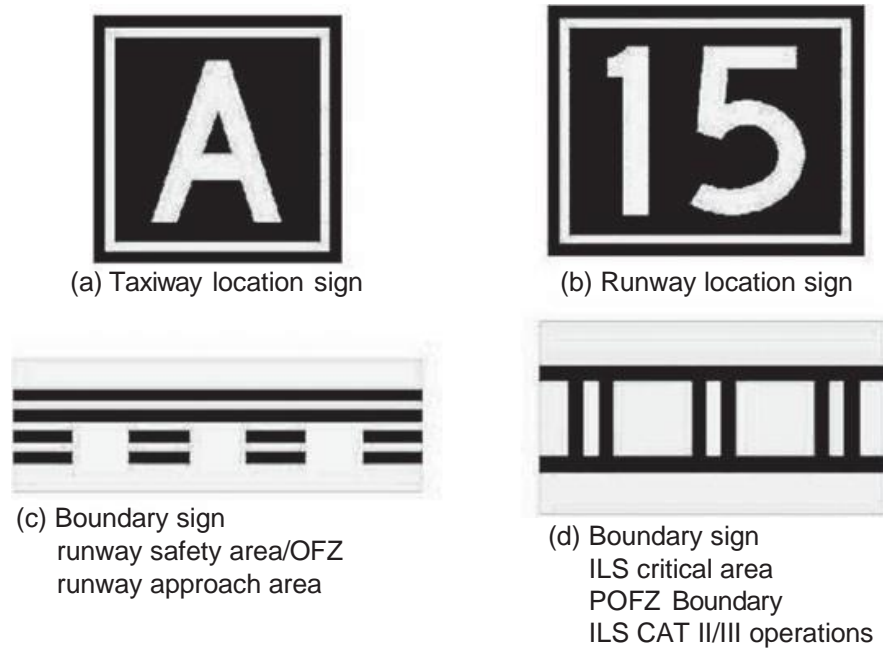


FIGURE 8-32 Location signs.

Location Signs

Location signs are used to identify the taxiway or runway on which an aircraft is located (Fig. 8-32*a* and 8-32*b*). These signs consist of a yellow inscription and border on a black background. Location signs are also used to identify the boundary of the runway safety area or obstacle-free zones (Fig. 8-32*c*), or the instrument landing system critical area (Fig. 8-32*d*) for a pilot exiting a runway. In the latter cases the signs consist of a black inscription and border on a yellow background and the inscription on the sign is the same as relevant holding line marking.

Direction Signs

Direction signs are used to indicate the direction of other taxiways leading out of an intersection. These signs are used as taxiway direction sign (Fig. 8-33) and runway exit sign. The signs have black inscriptions and borders on a yellow background and always contain arrows. The arrows are oriented in the approximate direction of the turn required. These signs should not be located with holding position signs and should not be located between the holding line and the runway.



FIGURE 8-33 Taxiway direction sign.



FIGURE 8-34 Inbound destination sign (to military facility).

Signs used to indicate the direction of taxiways on the opposite side of the runway should be located on the opposite side of the runway. Runway exit signs should be located prior to the exit on the side of the runway on which the aircraft is expected to exit. If the taxiway crosses the runway and the aircraft could be expected to exit on either side, then a runway exit sign should be installed on either side of the runway.

Destination Signs

Destination signs have black inscriptions on a yellow background and always contain arrows. These signs indicate the general direction to a remote location at the airport, such as an inbound destination (Fig. 8-34), and are generally not required where taxiway direction signs are used. Outbound destination signs are used to identify directions to the takeoff runways. These routes normally begin at the entrance to a taxiway from the apron area. More than one runway number may be used, separated by a dot, if the route is common to more than one runway (Fig. 8-35). Inbound destination signs are often used to indicate the general direction to major airport facilities such as passenger terminal aprons, cargo areas, military aprons, or general aviation facilities. These signs should consist of a minimum of three letters to avoid confusion with taxiway guidance signs.

The typical legends found on taxiway destination signs are:

APRON—general parking, servicing, and loading areas

FUEL—areas where aircraft are fueled or serviced

TERM—gate positions at which aircraft are loaded or unloaded

CIVIL—areas set aside for civil aircraft

MIL—areas set aside for military aircraft

PAX—areas set aside for passenger handling

CARGO—areas set aside for cargo handling

INTL—areas set aside for handling international flights FBO—fixed-base operator



FIGURE 8-35 Outbound destination sign (to runways 27 and 33).



FIGURE 8-36 Taxiway ending marker.

Information Signs

Other types of signs may be necessary on the airfield which are not part of the taxiway guidance systems described before. These signs are called information signs and might be used, for example, to indicate a noise abatement procedure to a pilot ready to takeoff on a specific runway. These signs should have black inscriptions on a yellow background. These signs are not required to be lighted.

Taxiway Ending Sign

The sign system does not provide a sign to indicate that a taxiway does not continue beyond an intersection. A frangible, retroreflective sign should be installed on the far side of the intersection if normal visual cues such as marking and lighting are inadequate. This sign is marked with alternating black and yellow diagonal stripes, as illustrated in Fig. 8-36.

The FAA recommends that the following guidelines be applied when designing a taxiway guidance sign system [25]:

1. A holding position sign and taxiway location sign should be installed at the holding position on any taxiway that provides access to a runway.
2. A holding position sign should be installed on any taxiway at the boundary of the instrument landing system critical area or the runway approach area when it is necessary to protect the navigational signal, airspace, or safety area for a runway. This sign should be placed at the entrance to and the exit from such areas.
3. A holding position sign should be installed on any runway where that runway intersects another runway.
4. A sign array consisting of taxiway direction signs should be installed prior to each intersection between taxiways if an aircraft would normally be expected to turn at or hold short of the intersection. The direction sign in the array should include a sign panel, consisting of a taxiway designation and an arrow, for each taxiway that an aircraft would be expected to turn onto or hold short. A taxiway location sign should be included as part of the sign array unless it is determined to be unnecessary. If an aircraft normally would not be expected to turn at

or hold short of the intersection, the sign array is not needed unless the absence of guidance would cause confusion.

5. A runway exit sign identifying the exit taxiway should be installed along each runway for each normally used runway exit.
6. Destination signs may be substituted for direction signs at the intersection between taxiways or for runway exit signs at uncontrolled airports.
7. Standard highway stop signs should be installed on ground vehicle roadways at the intersection of each roadway with a runway or taxiway. For roadway intersections with taxiways, a standard highway yield sign may be used instead of the stop sign.
8. Additional signs should be installed on the airfield where necessary to eliminate confusion or to provide confirmation relative to location.

Signing Conventions

The FAA recommends the following signing conventions [25]:

1. Signs should be placed on the left side of the taxiway as viewed by the pilot of an approaching aircraft. If signs are installed on both sides of the taxiway at the same location, the sign faces should be identical. Signs may be placed on the right side of the taxiway when necessary to meet clearance requirements or where it is impractical to install them on the left side because of terrain features or conflicts with other objects.
2. Some signs may be installed on the back of other signs even though this may result in the sign being on the right side of the taxiway. Signs which may be installed in this manner include
 - a. Runway safety area, obstacle-free zone area, and runway approach area boundary signs may be installed on the back of taxiway-runway intersection and runway approach area holding position signs.
 - b. Instrument landing system critical area boundary signs may be installed on the back of instrument landing system critical area holding position signs.
 - c. Taxiway location signs, when installed on the far side of the intersection, may be installed on the back of direction signs.
 - d. Taxiway location signs may be installed on the back of holding position signs.
 - e. Destination signs may be installed on the back of direction signs on the far side of intersections when the destination referred to is straight ahead.

3. Taxiway location signs installed in conjunction with holding position signs for taxiway-runway intersections should always be installed outboard of the holding position sign.
4. Location signs are normally included as part of a direction sign array located prior to the taxiway intersection. Except for the intersection of two taxiways, the location sign is placed in the array so that the designations for all turns to the left would be located to the left side of the location sign and designations for all turns to the right or straight ahead are located to the right of the location sign.
5. All direction signs have arrows. Arrows on signs should be oriented toward the approximate direction of the turn. Each designation appearing in the array of direction signs should only have one arrow. An exception is when the taxiway intersection comprises only two taxiways and then the direction sign for the taxiway may have two arrows.
6. Destination signs should be located in advance of intersections and should not be collocated with other signs. These may also be installed on the far side of the intersection when the taxiway does not continue and direction signs are provided prior to the intersection.
7. Information signs should not be collocated with mandatory, location, direction, or destination signs.
8. Each designation and its associated arrow included in the array of direction signs or destination signs should be delineated from the other designations in the array by a black vertical border.

Sign Size and Location

Taxiway guidance signs are available in three heights as indicated in Table 8-7. The choice of a particular size sign involves several factors including effectiveness, aircraft clearance, jet blast, and snow removal

Sign Size	Legend, In (cm)	Face, In (cm)	Installed (max.), In (cm)	Perpendicular Distance from Defined Taxiway/Runway Edge to Near Side of Sign, ft (m)
1	12 (30)	18 (46)	30 (76)	10–20 (3–6)
2	15 (38)	24 (61)	36 (91)	20–35 (6–10.5)
3	18 (46)	30 (76)	42 (107)	35–60 (10.5–18)

TABLE 8-7 Taxiway Signage Dimensional Specifications

Aircraft Approach Category and (Airplane Design Group)	Visual and Nonprecision Instrument Runway	Precision Instrument Runway
A and B (I and II) small airplanes only	125 (38)	175 (53)
A and B (I, II, and III)	200 (60)	250 (75)
A and B (IV)	250 (75)	250 (75)
C and D (I through IV)	250 (75)	250 (75)
C and D (V)	250 (75)	280 (85)
C and D (VI)	250 (75)	280 (85)

Perpendicular distance from runway centerline to intersection runway/taxiway centerline is in feet (meters).

TABLE 8-8 Location Distances for Holding Position Markings

operations. Normally, the larger the sign and the closer it is located to the runway or taxiway edge the more effective it is. However, aircraft clearance requirements and jet blast effects require smaller signs when located near the pavement edges, whereas effectiveness requires larger signs when located at further distances. The effects of snow removal operations on the signs should be considered in the choice of sign size and location. The sign used must provide 12 in of clearance between the top of the sign and any part of the most critical aircraft using or expected to use the airport when the wheels of the aircraft are at the defined pavement edge.

The distances shown in Table 8-8 should be used in determining runway holding positions. All signs in an array should be of the same size and at the same height above the ground.

For determining sign locations with respect to intersecting runways, the clearance requirements to other moving aircraft, as given in Table 8-9, should be used. For signs installed at holding positions the signs should be in line with the holding line markings within a tolerance

Airplane Design Group I	Airplane Design Group II	Airplane Design Group III	Airplane Design Group IV	Airplane Design Group V	Airplane Design Group VI
44.5 ft (13.5 m)	65.5 ft (20 m)	93 ft (28.5 m)	129.5 ft (39.5 m)	160 ft (48.5 m)	193 ft (59 m)

TABLE 8-9 Perpendicular Distances for Taxiway Intersection Markings from Centerline of Crossing Taxiway

of 10 ft. Where there is no operational need for taxiway holding line markings the signs may be installed in the area from the taxiway point of tangency to the location where the holding line markings would be installed [25].

Typical locations for taxiway guidance signs are shown in Fig. 8-37. An illustration of the required signs and their placement for a basic airport layout is given in Fig. 8-38 [25].

Sign Operation

Holding positions signs for runways, instrument landing system critical areas, approach areas, and their associated taxiway location signs should be illuminated when the associated runway lights are illuminated. Other taxiway signs should be illuminated when the associated taxiway lights are illuminated.

The installation of retroreflective markings is not mandatory. However, it is quite economical, especially at airports where lights cannot be justified because of the volume or nature of air traffic [12]. The marking is very similar to that used successfully on highways for many years.

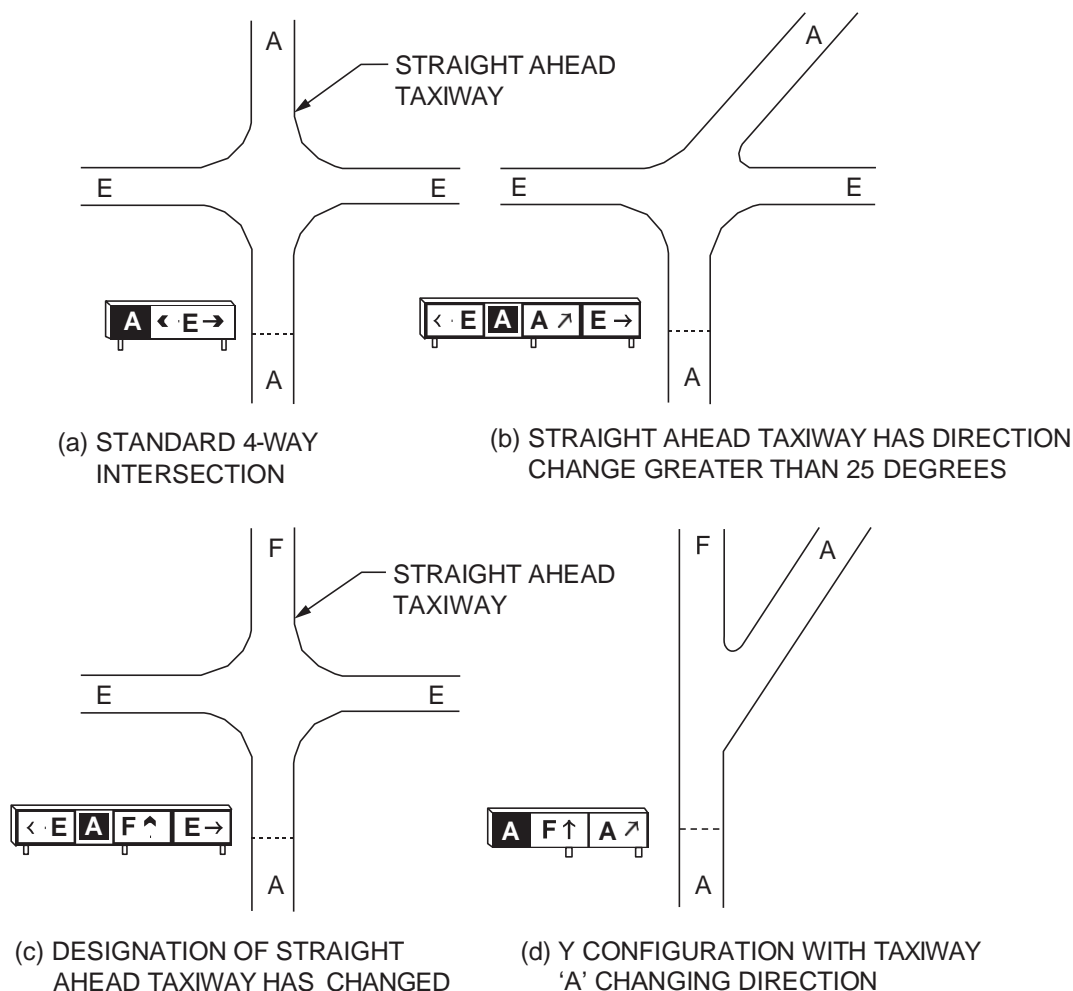


FIGURE 8-37 Signage configuration at taxiway intersections.

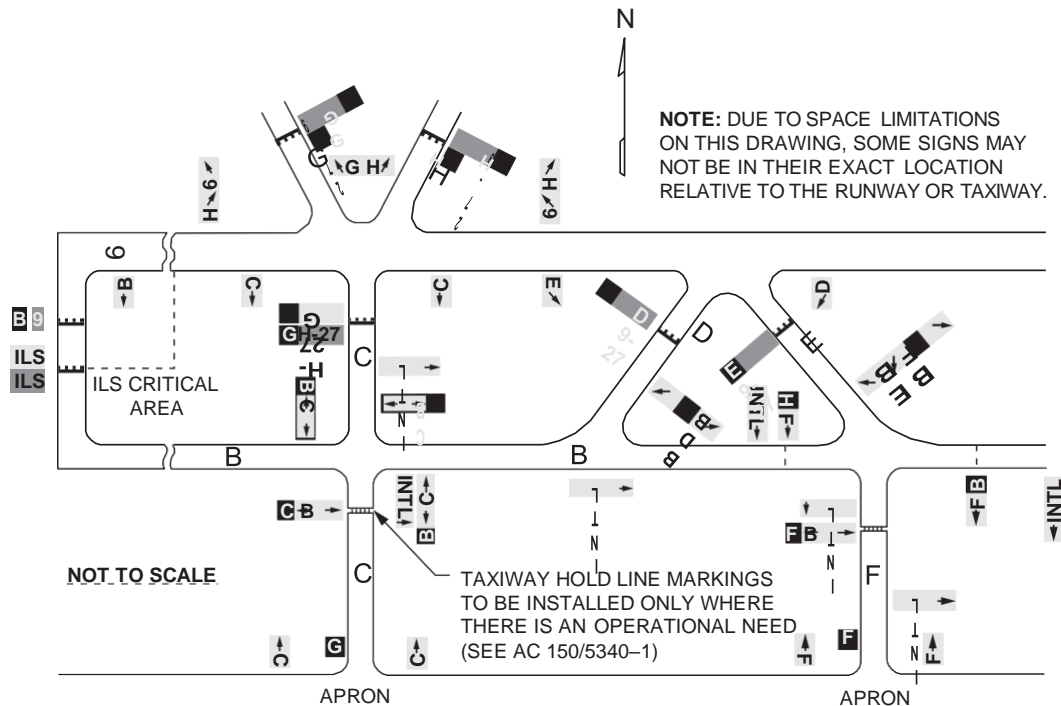


FIGURE 8-38 Typical layout of airfield signage.

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25. *Standards for Airport Sign Systems*, Advisory Circular, AC 150/5340-18D, Federal Aviation Administration, Washington, D.C., 2004.
26. *Taxiway Centerline Lighting System*, Advisory Circular, AC 150/5340-19, Federal Aviation Administration, Washington, D.C., 1968.
27. "The Theory of Visual Judgements in Motion and Its Application to the Design of Landing Aids for Aircraft," E. S. Calvert, *Transactions of the Illuminating Engineering Society*, Vol. 22, No. 10, London, England, 1957.
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CHAPTER 9

Airport Drainage

An adequate drainage system for the removal of surface and subsurface water is vital for the safety of aircraft and for the longevity of the pavements. Improper drainage results in the formation of puddles on the pavement surface, which can be hazardous to aircraft taking off and landing. Poor drainage can also result in the early deterioration of pavements. Flat longitudinal and transverse grades and wide pavement surfaces often pose difficulties in making provision for adequate drainage at airports.

The material in this chapter is principally concerned with estimating the amounts of surface and subsurface runoff and not with the hydraulics of pipes or details of installation. These latter items are adequately covered in texts on hydraulics and literature provided by pipe manufacturers.

The FAA and the Corps of Engineers have developed most of the information on airport drainage in the United States and the material presented in this has been drawn from their work. In 2006, several agencies worked together to combine existing surface drainage topics covered in several manuals into one Unified Facilities Criteria (UFC) document. The resulting manual [1] now serves as the design and analysis standard for surface drainage for the FAA.

Purpose of Drainage

The functions of an airport drainage system are as follows:

1. Interception and diversion of surface and groundwater flow originating from lands adjacent to the airport
2. Removal of surface runoff from the airport
3. Removal of subsurface flow from the airport

In very few cases will the natural drainage on a site be sufficient by itself to satisfy these functions; consequently artificial drainage must be installed.

Design Storm for Surface Runoff

The selection of the severity of the storm which the drainage system should accommodate involves economic consideration. An extremely

severe storm occurring very infrequently would undoubtedly cause some damage if the system were designed for a storm of lesser severity. However, if serious interruptions in traffic are not anticipated, a system designed for the larger storm may not be economically justified. Taking these factors into account, the FAA recommends that for civil airports the drainage system be designed for a storm whose probability of occurrence is once in 5 years [2]. The design should, however, be checked with a storm of lesser frequency (10 to 15 years) to ascertain if serious damage or interruption of traffic would result from such a storm. Drainage for military airfields is based on a 2-year storm frequency [8].

Ordinarily no ponding is permitted on paved surfaces, but in the intervening areas ponding is permitted, provided it will not result in undesirable saturation of the subgrades underneath the pavements.

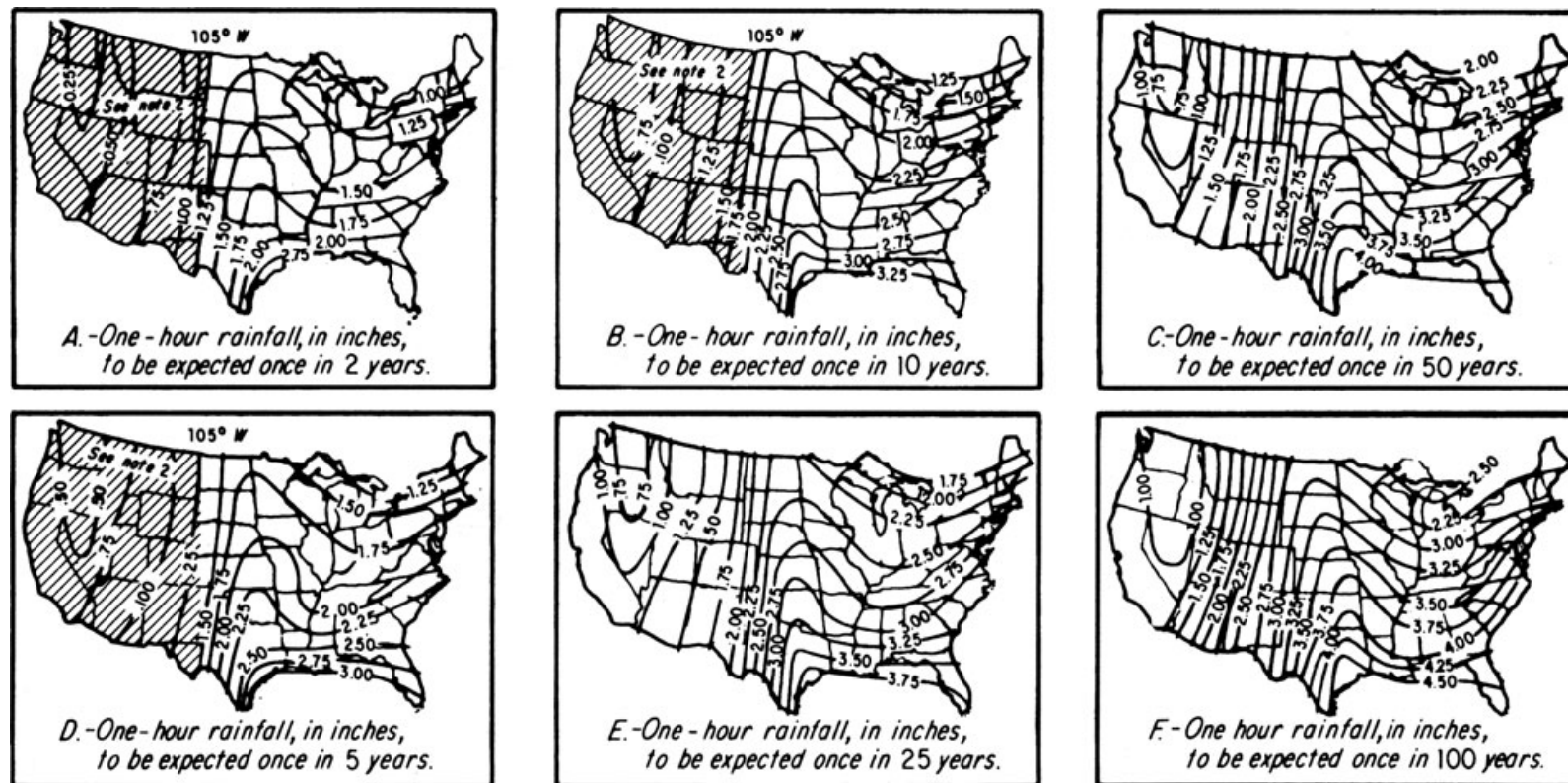
Determining the Intensity-Duration Pattern for the Design Storm

The determination of the amount of rainfall which can be expected at the site of the airport is the first step in the design of a drainage system. Rainfall intensity is expressed in inches per hour for various durations of a particular storm. The expected frequency of occurrence is also an important factor to consider. The severity of storms is related to their frequencies; a storm which is expected to occur once in 100 years will be more severe than one having a frequency of occurrence of once in 5 years.

David L. Yarnell of the U.S. Department of Agriculture conducted extensive investigations concerning rainfall intensities, durations, and frequencies throughout the United States [16]. West of the 105th meridian, where the Yarnell information is not as complete, the National Weather Service has compiled rainfall data which appear in Refs. 12 to 14.

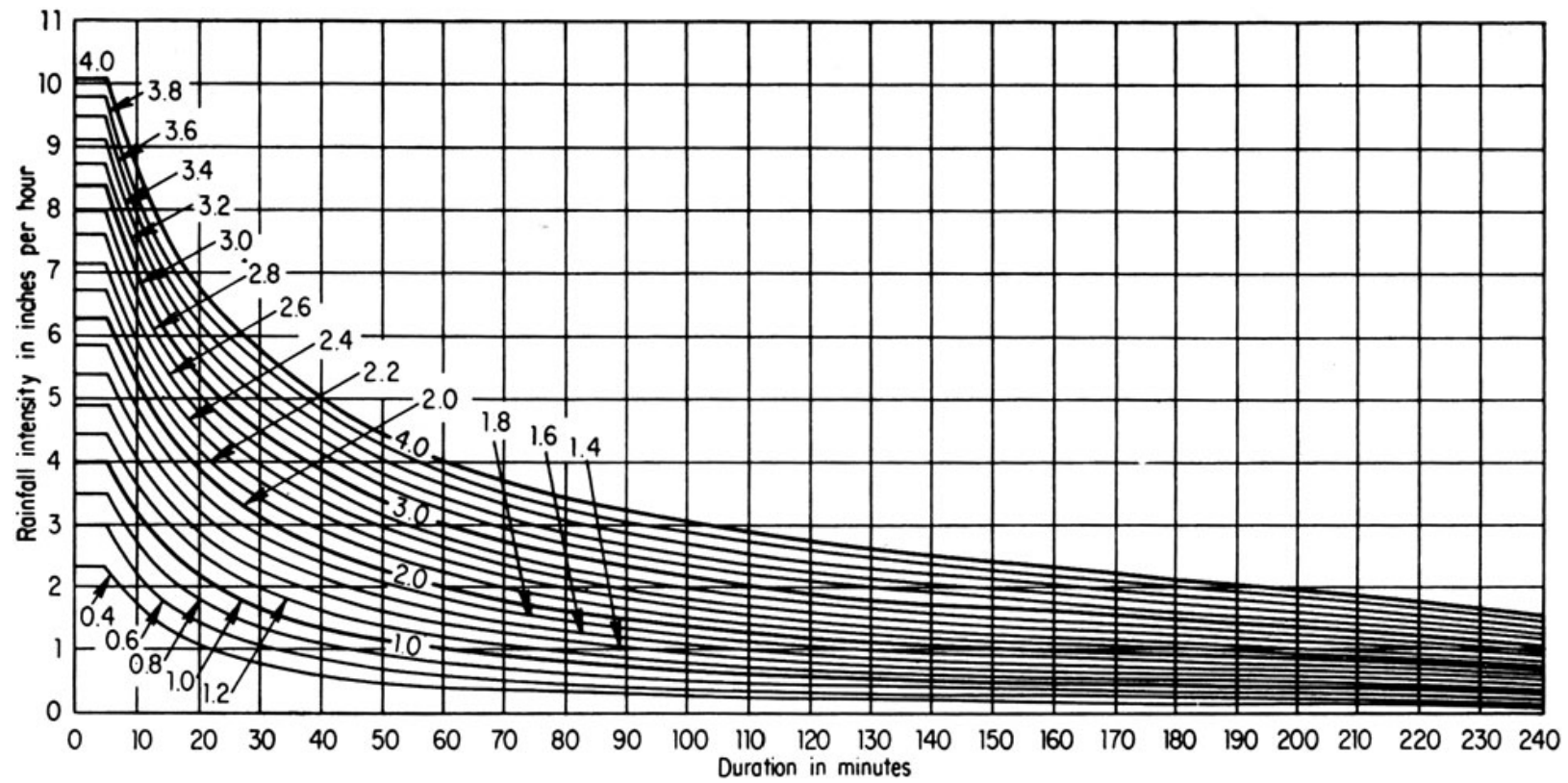
Yarnell developed rainfall intensities for 5-, 10-, 15-, 30-, 60-, and 120-min durations for a storm which can be expected to occur once in 5 years and the intensities for a 1-h duration for storms whose expected frequencies of occurrence are once in 2, 5, 10, 25, 50, and 100 years. The intensities for a duration of 1 h for frequencies of 2, 5, 10, 25, 50, and 100 years are shown in Fig. 9-1.

The 1-h intensity does not by itself portray the intensity-duration pattern of a storm. The Corps of Engineers made extensive studies of rainfall patterns in the United States and found that irrespective of frequency, the intensity-duration patterns of storms were largely governed by their 1-h intensities. That is, two storms of different frequency of occurrence whose 1-h intensities are equal will have similar intensity-duration patterns. This is shown in Fig. 9-2. For example, if the 1-h intensities of storms whose frequencies were 5, 10, or 15 years were all exactly 2.0 in/h, the intensity-duration patterns would be expected to follow the pattern indicated by the curve labeled 2.0.



Notes: 1. Charts shown above were reproduced from Misc. Pub. No. 204, U.S. Department of Agriculture, Rainfall Intensity-Frequency Data by David L. Yarnell.
2. For the Western part of the U.S. West of the 105th Meridian, see One Hour Rainfall Data in Weather Bureau Technical Paper No. 24.

FIGURE 9-1 One-hour rainfall intensities for the United States (*Corps of Engineers*).



Notes:

Curve numbers correspond to one hour values of rainfall or supply indicated by respective curves.

All points on the same curve are assumed to have the same average frequency of occurrence.

FIGURE 9-2 Rainfall intensity-duration curves (*Corps of Engineers*).

If the drainage system is to be designed for a storm whose expected frequency of occurrence is once in 5 years and if detailed data concerning the intensity-duration pattern are nonexistent, the pattern can be approximated from Fig. 9-2, provided the 1-h intensity is known. It goes without saying that if sufficient rainfall data are available at an airport site, the intensity-duration frequency data should be developed from this information rather than from other sources. Rarely, however, does a site have such complete rainfall information.

Determining the Amount of Runoff by the FAA Procedure

The FAA analysis of airport surface drainage revolves about the solution of the *rational method* expression

$$Q = CIA \quad (9-1)$$

where Q = runoff from given drainage basin, ft^3/s

C = ratio of runoff to rainfall

I = rainfall intensity for time of concentration of runoff, in/h

A = drainage area, acres

Examples and charts illustrating the FAA procedure for design have been taken largely from the FAA [2].

Time of Concentration

The *time of concentration* is defined as the time taken by water to reach the drain inlet from the most remote point in the tributary area. The *most remote point* refers to the point from which the time of flow is the greatest. The time of concentration is usually divided into two components: inlet time and time of flow. The *inlet time* is the time required for water to flow overland from the most remote point in the drainage area to the inlet. The *time of flow* is the time taken by the water to flow from the drain inlet through the pipes to the point in the system under consideration. Sometimes the inlet time will be the time of concentration; at other times the time of concentration will be the sum of the inlet time and time of flow.

The time of flow can be computed by the use of well-established hydraulic formulas. The inlet time is obtained largely empirically from the relationship

$$D = kT^2 \quad (9-2)$$

Where D = distance, ft

T = time, min

k = dimensional empirical factor which is dependent on slope, roughness of terrain, extent of vegetative cover, and distance to drain inlet

Inlet times can be estimated from Fig. 9-3.

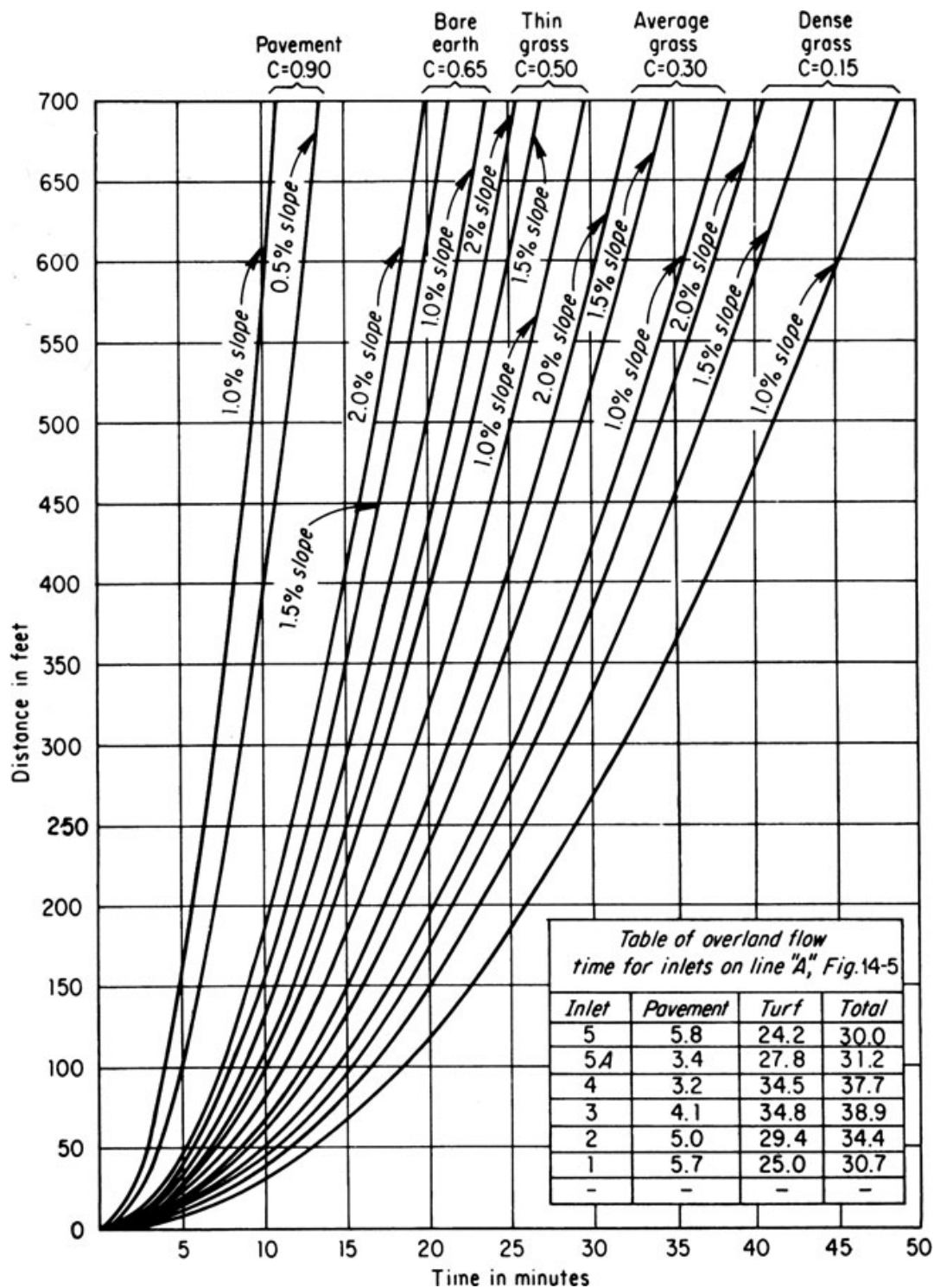


FIGURE 9-3 Inlet time curves (*Federal Aviation Administration [2]*).

Coefficient of Runoff

Application of the rational method requires the exercise of considerable judgment on the part of the engineer. The runoff rate is variable from storm to storm and varies even during a single period of precipitation. The coefficient of runoff depends on antecedent storm conditions, slope and type of surface, and extent of the drainage area. The range of values suggested by the FAA is indicated in Table 9-1.

Types of Surfaces	Factor C
For all watertight roof surfaces	0.75–0.95
For asphalt runway pavements	0.80–0.95
For concrete runway pavements	0.70–0.90
For gravel or macadam pavements	0.35–0.70
For impervious soils (heavy)*	0.40–0.65
For impervious soils with turf*	0.30–0.55
For slightly pervious soils*	0.15–0.40
For slightly pervious soils with turf*	0.10–0.30
For moderately pervious soils*	0.05–0.20
For moderately pervious soils with turf*	0.00–0.10

*For slopes from 1 to 2 percent.

Source: Federal Aviation Administration [2].

TABLE 9-1 Coefficients of Runoff C

For drainage basins consisting of several types of surfaces with different infiltration characteristics, the weighted runoff coefficient should be computed in accordance with

$$C = \frac{A_1 C_1 + A_2 C_2 + A_3 C_3}{A_1 + A_2 + A_3} \quad (9-3)$$

Typical Example—No Ponding

In order that the worst conditions attendant upon the design storm may be used in the design of the pipe system, a separate duration of storm is selected for each subdrainage area tributary to a drain inlet. The duration of the storm is made equal to the sum of the inlet time and time of flow.

Each reach of pipe must be designed to carry the discharge from the inlet at its upstream end plus the contribution from all preceding inlets. For economy of construction, the grade of each reach is determined largely by topography. A minimum mean velocity on the order of 2.5 ft/s should be maintained to provide scouring action so that reduction of the pipe area due to silting will not be a problem.

To clarify the computation of runoff by the FAA method, the following example is presented.

The intensity-duration rainfall pattern for a 5-year-frequency storm at the site of the proposed airport is shown in Fig. 9-4. The layout of the drains on a portion of the airport is shown in Fig. 9-5. Design data for establishing inlet times and coefficients of runoff

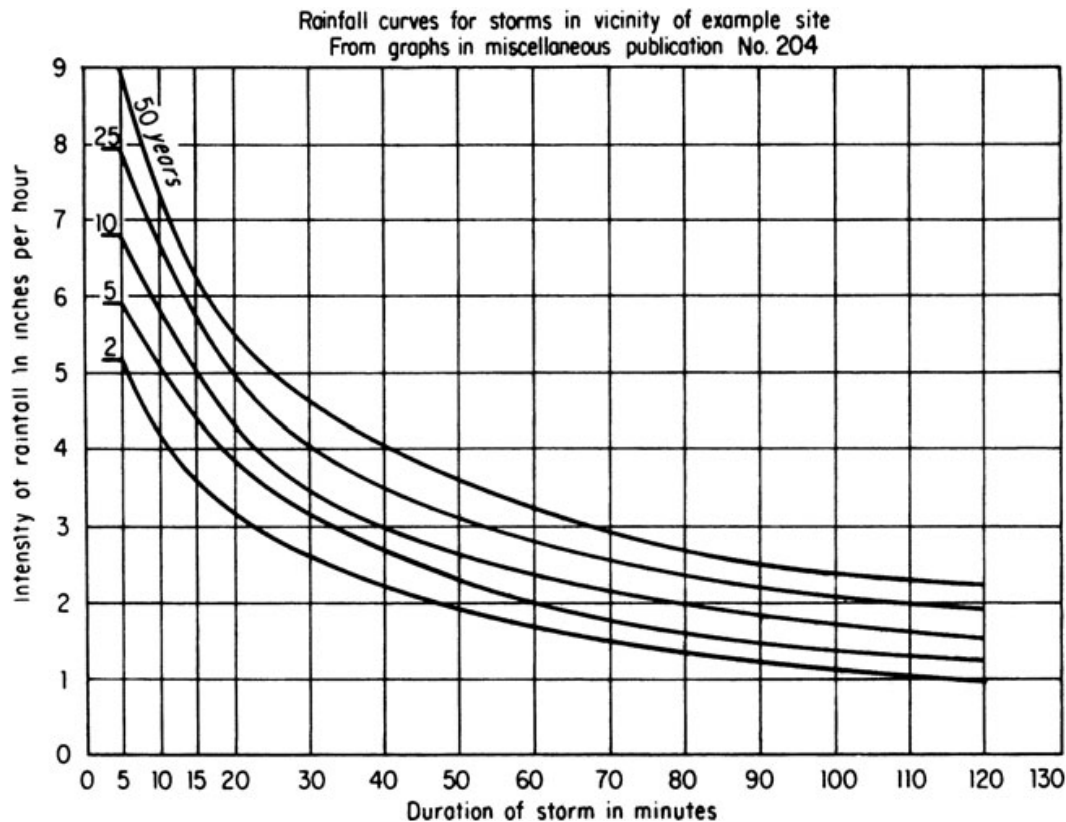


FIGURE 9-4 Intensity-duration rainfall pattern for design storm (*Federal Aviation Administration [2]*).

for drainage areas tributary to drain line *A*, shown in Fig. 9-5, are tabulated in Table 9-2. It is assumed that the coefficients of runoff for pavement and for turf are 0.90 and 0.30, respectively.

From these data, inlet times have been computed by the use of Fig. 9-3 on the basis that the slope of the pavement is 1 percent and the slope of the turfed area is 1.5 percent. The inlet times for this specific problem are shown in Fig. 9-3. The computations for runoff, assuming no ponding, are shown in Table 9-3.

Typical Example—Ponding

In the design of an airfield drainage system, ponding may be used to effect a reduction in the cost of installation. Ponding is simply a means of providing temporary storage of runoff prior to its entry into the underground system. For purposes of design computation, the ponded volume may be assumed to be an inverted pyramid or a truncated pyramid, the height of which is the depth of water above the inlet at any stage. The area of the base of the pyramid is taken as the surface area of the pond. If ponding were permitted, the layout of the drainage system might be as shown in Fig. 9-6. The most remote point to one of the inlets is 950 ft, comprising 100 ft of pavement and 850 ft of turf. The time of concentration is estimated at $4 + 54 = 58$ min. The complete

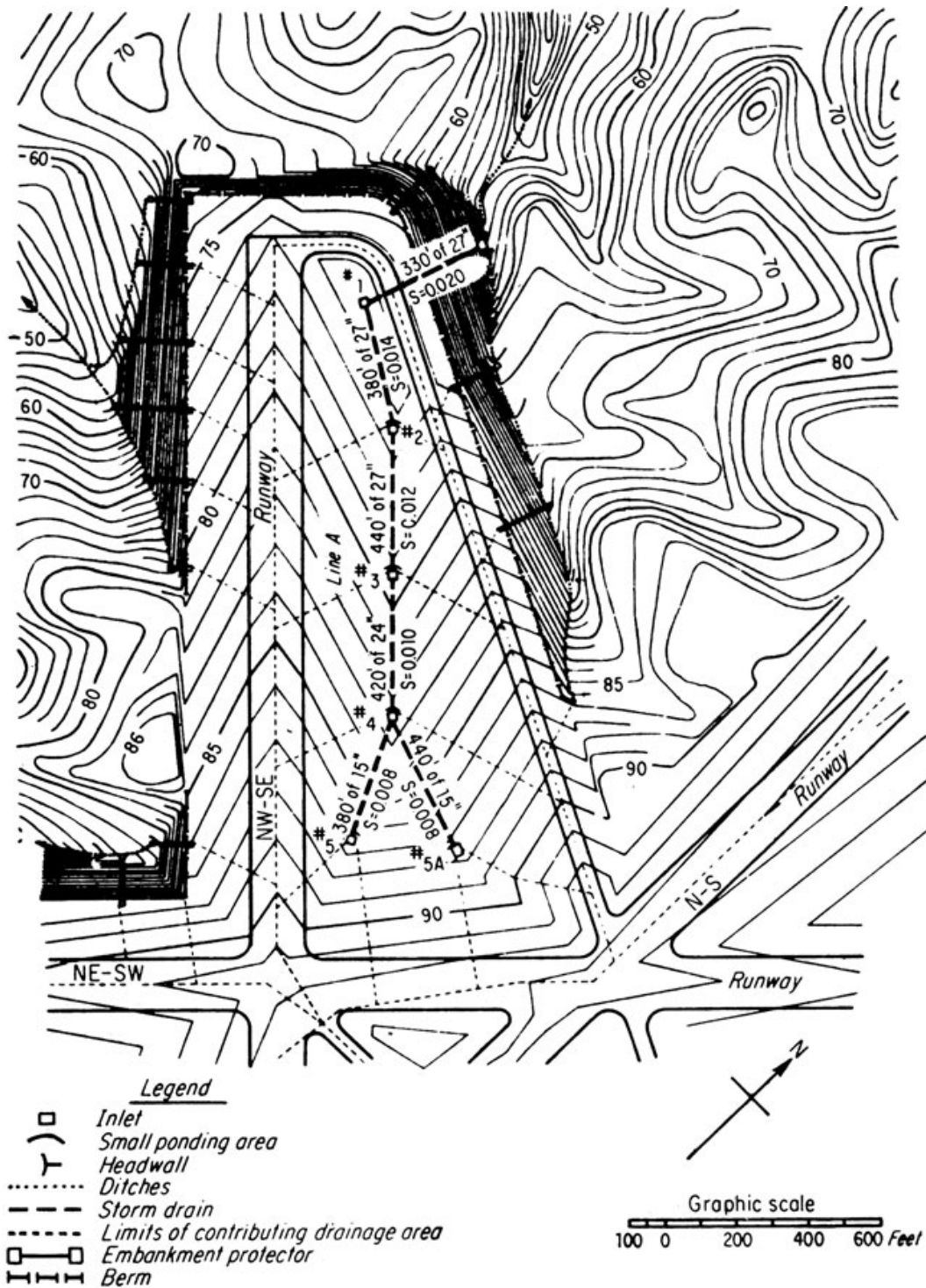


FIGURE 9-5 Portion of airport showing drainage design details (*Federal Aviation Administration [2]*).

drainage area is 31.42 acres, of which 6.44 acres is paved. Assuming that the coefficients of runoff for pavement and turf are 0.90 and 0.30, respectively, the combined C is 0.423. From Fig. 9-4 the rainfall intensities for durations of 5, 10, 15, 20, 30, 60, 90, 120, and 180 min are obtained, and the volumes of runoff are computed as shown in Table 9-4.

	Tributary Area to Inlets, acres				Distance Remote Point to Inlet, ft				
	Pavement	Turf	Both	Subtotal	Pavement	Turf	Total	Line Segment	Length ft
Inlets									
5	1.27	1.05	2.32	2.32	200	340	540	5-4	380
5A	1.02	1.86	2.88	2.88	70	450	520	5A-4	440
4	1.40	7.35	8.75	13.95	60	690	750	4-3	420
3	0.78	6.46	7.24	21.19	100	700	800	3-2	440
2	0.83	4.56	5.39	26.58	150	500	650	2-1	380
1	1.14	3.70	4.84	31.42	190	360	550	1-outlet	330
Outlet	—	—	—	—	—	—	330		
Total	6.44	24.98	31.42						

Weighted Average for <i>C</i>		
To inlet 5:	To inlet 5A:	To inlet 4:
$\frac{1.27}{2.32} (0.90) = 0.49$ $\frac{1.05}{2.32} (0.30) = 0.14$ $C = 0.63$	$\frac{1.02}{2.88} (0.90) = 0.32$ $\frac{1.86}{2.88} (0.30) = 0.19$ $C = 0.51$	$\frac{1.40}{8.75} (0.90) = 0.14$ $\frac{7.35}{8.75} (0.30) = 0.25$ $C = 0.39$
To inlet 3:	To inlet 2:	To inlet 1:
$\frac{0.78}{7.24} (0.90) = 0.10$ $\frac{6.46}{7.24} (0.30) = 0.27$ $C = 0.37$	$\frac{0.83}{5.39} (0.90) = 0.14$ $\frac{4.56}{5.39} (0.30) = 0.25$ $C = 0.39$	$\frac{1.14}{4.84} (0.90) = 0.21$ $\frac{3.70}{4.84} (0.30) = 0.23$ $C = 0.44$

Source: Federal Aviation Administration [2].

TABLE 9-2 Design Data for Line A in Fig. 9-5

Line Segment		Length of Segment, ft	Inlet Time, min	Flow Time, min	Time of Concentration, min	Runoff Coefficient C	Rainfall Intensity I	Tributary Area A, acres	Remarks
Inlet									
5A	5A-4	440	31.2	1.8	31.2	0.51	3.10	2.88	n=0.015
5	5-4	380	30.0	1.6	30.0	0.63	3.15	2.32	
4	4-3	420	37.7	1.1	37.7	0.39	2.80	8.75	See accumulated runoff computed below.
3	3-2	440	38.9	1.0	38.9	0.37	2.70	7.24	Accumulated runoff adjustment negligible.
2	2-1	380	34.4	0.8	39.9	0.39	2.65	5.39	
1	1-outlet	330	30.7	0.5	40.7	0.44	2.60	4.84	
Outlet									
<p style="text-align: center;">Calculation of Example</p> <p>Maximum flow from inlets 5 and 5A will reach inlet 4 in 31.6 and 33.0 min, respectively. All inlet 4 subarea will be contributing to the system only after 37.7 min. Flow from inlets 5 and 5A must be adjusted for 37.7-min time of concentration. Adjusted time of concentration for inlets 5 and 5A (that is, the inlet time for end-of-line structures) equals the time of concentration to inlet 4 less the flow time through the respective pipe segments. For inlet 5, adjusted time of concentration = $37.7 - 1.6 = 36.1$ min. For inlet 5A, adjusted time of concentration = $37.7 - 1.8 = 35.9$ min. By using these adjusted times of concentration, an intensity of rainfall of 2.85 in/h is obtained (slight time difference cannot be read from curves).</p> <p>Applying these data in the formula $Q = CIA$:</p> <p>Adjusted flow from inlet 5 = $0.63 \times 2.85 \times 2.32 = 4.16$</p> <p>Adjusted flow from inlet 5A = $0.51 \times 2.85 \times 2.88 = 4.18$</p> <p>Flow into inlet 4 from inlets 5 and 5A in 37.7 min = 8.34</p> <p>Flow from inlet 4 subarea = <u>9.56</u></p> <p>Accumulated flow entering inlet 4 in 37.7 min = 17.90 ft³/s</p>									

	Runoff Q ft ³ /s	Accumulated Runoff, ft ³ /s	Velocity of Drain, ft/s	Size of Pipe, in	Slope of Pipe, ft/ft	Capacity of Pipe, ft ³ /s	Invert Elevation	Remarks
Inlet								
5A	4.55	4.55	4.0	15	0.008	5.0	81.52 $n=0.015$	5
5	4.60	4.60	4.0	15	0.008	5.0	81.04	
4	9.56	17.90	6.2	24	0.010	20.0	78.00	See accumulated runoff computed below.
3	7.23	25.13	7.4	27	0.012	30.0	73.80	Accumulated runoff adjustment negligible.
2	5.57	30.70	8.0	27	0.014	33.0	68.52	
1	5.54	36.24	9.5	27	0.020	37.5	63.20	
Outlet							56.60	

Source: Federal Aviation Administration [2].

TABLE 9-3 Drainage System Design Data

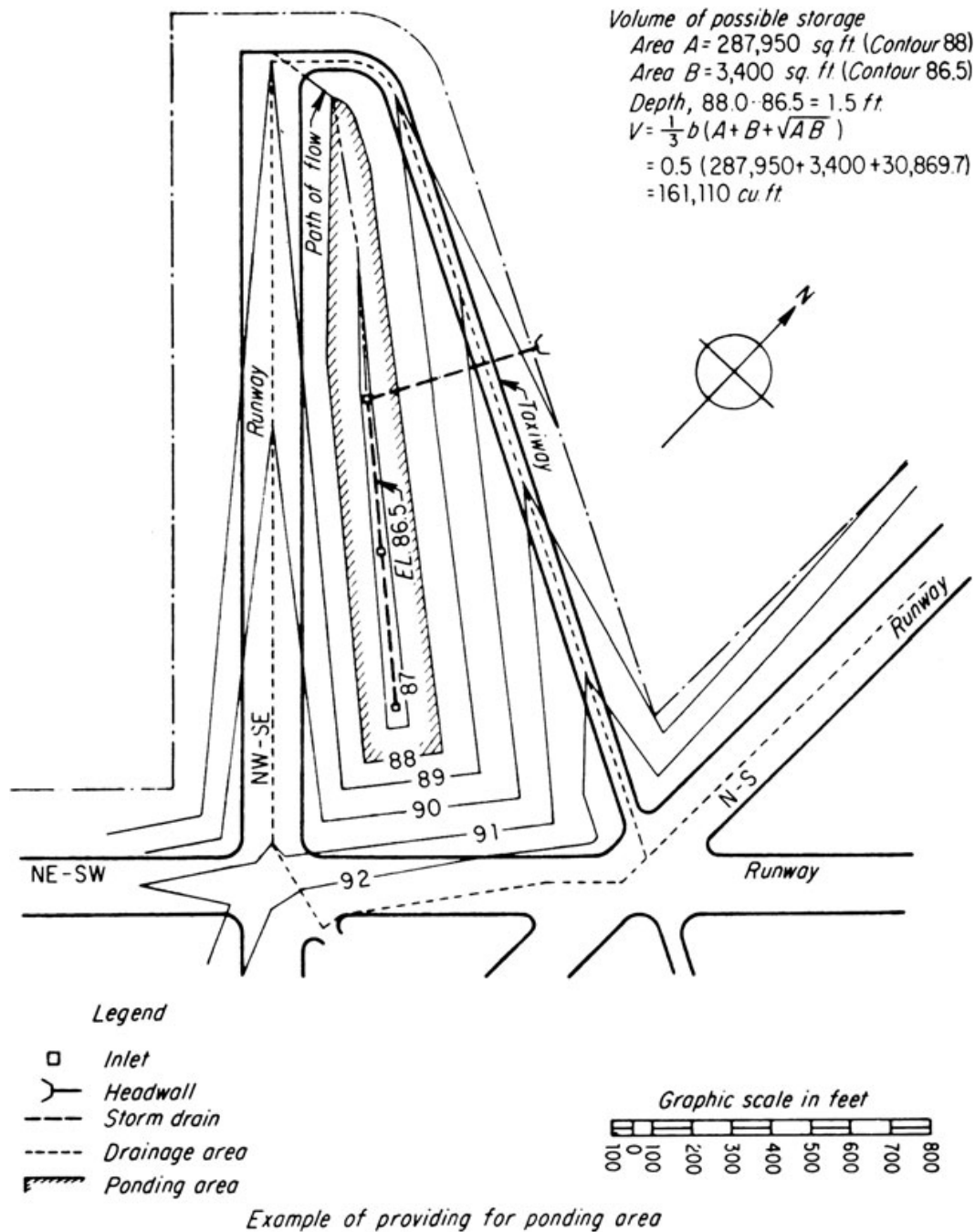


FIGURE 9-6 Layout of drainage for ponding (*Federal Aviation Administration [2]*).

To visualize the effects of ponding, a comparison is made of the discharge capacity of tentative drainage pipes and the cumulative runoff for the design storm frequency. This comparison is best made as a plot of runoff on the ordinate axis and time on the abscissa. An example of such a plot is shown in Fig. 9-7. The discharge capacity for each of four selected pipe sizes is shown as a straight line. These discharge curves were computed for an assigned slope and roughness

Time, min	Intensity* <i>I</i>	$Q = CIA, \text{ft}^3/\text{s}$	Volume $V = CIA t, \text{ft}^3$
5	5.80	77.1	23,100
10	4.96	65.9	39,600
15	4.33	57.5	51,800
20	3.95	52.5	63,000
30	3.18	42.3	76,100
60	2.00	26.6	95,700
90	1.62	21.5	116,300
120	1.26	16.7	120,600
180	0.87	11.6	125,000

*Hourly intensities from Fig. 9-4.

TABLE 9-4 Volume of Runoff—Ponding

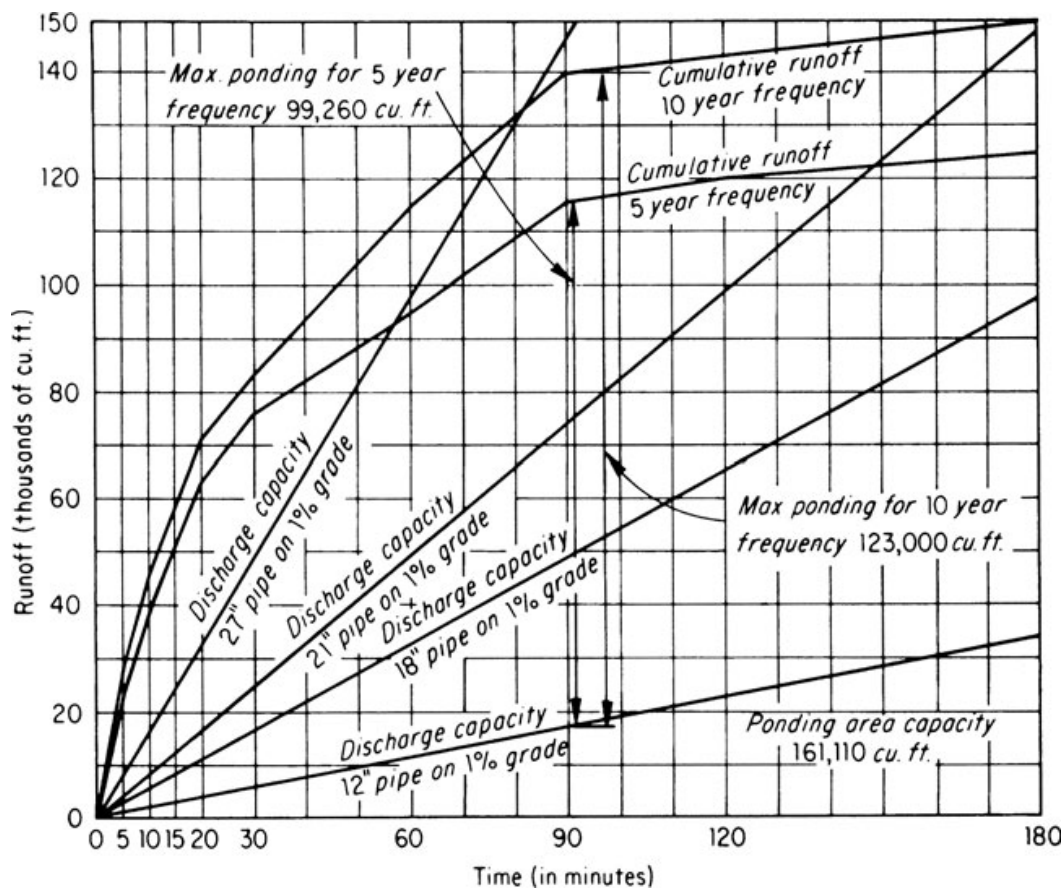


FIGURE 9-7 Cumulative runoff for ponding in Fig. 9-6 (*Federal Aviation Administration [2]*).

coefficient n for each pipe by use of the Manning formula. For this example the pipes were assumed to be concrete with a roughness coefficient n of 0.015 and laid on a 1 percent slope. The discharge in cubic feet per second multiplied by 3600 s is the discharge capacity ordinate in cubic feet at the 60-min abscissa in Fig. 9-7. Each discharge capacity curve must pass through the origin of coordinates, and one point as determined above will define the straight-line relationship.

The significance of the cumulative runoff and discharge capacity curves as plotted in Fig. 9-7 is that the difference in ordinates (cumulative runoff minus discharge capacity) represents the amount of ponding at any instant after the beginning of the storm. The maximum amount of ponding is determined by scaling the largest difference between the cumulative runoff curve and the discharge capacity curve.

It is considered essential that all ponding area edges be kept at least 75 ft from the edges of pavements. In this example, this would mean that the pond should not reach a level above elevation 88.0. The storage capacity below this elevation is 161,000 ft³. If a 12-in-diameter pipe were used, the maximum ponding would amount to 99,260 ft³, considerably less than the available 161,100 ft³. For practical consideration a pipe of lesser diameter is not recommended.

Although not shown in this text, computations were also made for a 10-year-frequency storm. With a 12-in-diameter pipe such a storm would develop a pond of 123,000 ft³, still less than the available capacity of 161,000 ft³.

Determining the Amount of Runoff by the Corps of Engineers Procedure

For determining runoff, the Corps of Engineers uses a relationship for overland flow developed by R. E. Horton [17]. This relationship, as modified by the Corps of Engineers, is as follows:

$$q = \left(\frac{1.483}{n} \right) \left(\tan h^2 \right)^{1/3} \left(\frac{S}{100} \right)^{1/2} t^{1/3} S^{1/4} \quad (9-4)$$

where q = rate of overland flow at lower end of elemental strip of turfed, bare, or paved surface, in/h of ft³/s per acre of drainage area

Q = total discharge from a drainage area, ft³/s; Q equals product of q and drainage area in acres

S = slope of surface or hydraulic gradient, absolute, i.e., 1 percent = 0.01

t = time or duration, min; time from beginning of supply (storm); total time $t = t_c + t_d$

t_c = duration of supply which produces maximum rate of outflow from a drainage area but not in a pipe

t_d = time water flows in pipe
 σ = rate of supply, or rainfall in excess of rate of infiltration,
in/h
 L = effective length of overland or channel flow, ft
 n = retardance coefficient

The term t_c is nothing more than the time of concentration for the drainage area under consideration. The term L , the effective length, represents the length of overland sheet flow from the most remote point in the drainage area to the drain inlet, measured in a direction parallel to the maximum slope, before the runoff has reached a defined channel or ponding basin, plus the length of flow in a channel if one is present. If ponding is permitted, L is measured from the most remote point in the drainage area to the mean edge of the pond.

The term n is referred to as the *retardance coefficient*. Typical coefficients are given in Table 9-5.

When a drainage area is composed of two or three types of surfaces, an average retardance coefficient must be computed. For example, if a drainage area consists of 4 acres of average grass cover and 2 acres of pavement, the average retardance coefficient is equal to

$$\frac{4(0.40) + 2(0.02)}{6} = 0.27$$

Infiltration Rate

Use of the Horton formula requires an estimate of the amount of rainfall which is absorbed in the ground and which therefore does not appear as runoff. This is referred to as *infiltration* and is expressed as a rate in inches per hour. Thus, the intensity of rainfall (in inches per hour) less the infiltration rate is equal to the rate of runoff or the rate of supply σ in the formula for runoff.

Surface	Value of n
Smooth pavements	0.02
Bare packed soil free of stone	0.10
Sparse grass cover, or moderately rough bare surface	0.30
Average grass cover	0.40
Dense grass cover	0.80

Source: Corps of Engineers [8].

TABLE 9-5 Retardance Coefficients

The infiltration rate is dependent largely on the structure of the soil cover, moisture content, and temperature of the air. The infiltration rate is not constant throughout the duration of the storm, but is assumed so in the computations. It is felt that such an assumption is reasonable, especially when the soil is near saturation.

The infiltration rate for paved surfaces is usually assumed to be zero. Infiltration rates for other types of surfaces and soil cover must be estimated from experience. A value of 0.5 in/h has been suggested for turfed areas. Thus, if the rainfall intensity on a turfed area were 2.0 in/h, the rate of supply σ would be 1.5 in/h.

Standard Supply Curves

By use of Eq. (9-4) maximum rates of runoff q for rates of supply σ of 0.8, 1.0, 1.6, and 1.8 in/h are shown in Figs. 9-8 and 9-9. Maximum rates of runoff are also shown for rates of supply of 0.4, 0.6, 1.2, 1.4, 2.0, 2.2, 2.4, 2.6, 2.8, 3.0, 3.2, and 3.4 in/h [8].

Maximum rates of runoff for the curve labeled *supply curve no. 1.0* (Fig. 9-8) were obtained in the following manner. From Fig. 9-2 the intensities of runoff for various durations corresponding to the curve labeled 1.0 are obtained. These intensities are entered as σ in Eq. (9-4), and L is varied to produce the family of curves shown in Fig. 9-8. The curve labeled σ is supply curve no. 1.0, obtained from Fig. 9-2. The dotted line labeled t_c represents the maximum rate of runoff q which would occur from an elemental area with various effective lengths L . For example, the maximum rate of runoff from an area whose effective length L is 60 ft is 2.0 ft³/s. Multiplying this rate by the drainage area yields the maximum total discharge Q .

Figures 9-8 and 9-9 were prepared for $n = 0.40$ and $S = 1$ percent. If these charts are to be used for other cases, the actual effective L for the area under study must be converted in terms of L for $n = 0.40$ and $S = 1$. A conversion chart is shown in Fig. 9-10. For example, if the actual $n = 0.30$ and $S = 2$ percent and the effective length L is 400 ft, then the equivalent effective L for $n = 0.40$ and $S = 1$ percent is 140 ft.

Typical Example—No Ponding

In the Corps of Engineers procedure, a reach of drain pipe is always designed for a storm whose duration is equal to the time of concentration for the drainage area above the pipe. The time of concentration corresponds to the time necessary to produce maximum flow into a particular inlet (which is the same as the time necessary for water to reach an inlet from the most remote point in the area) plus the flow time in the pipe.

To clarify the computation of runoff by the Corps of Engineers procedure, the following example is presented.

Consider the drainage areas shown in Fig. 9-11. The 1-h intensity of the design storm is assumed to be 2.0 in/h. The infiltration rate for the turfed areas is assumed to be 0.5 in/h. The retardance coefficient

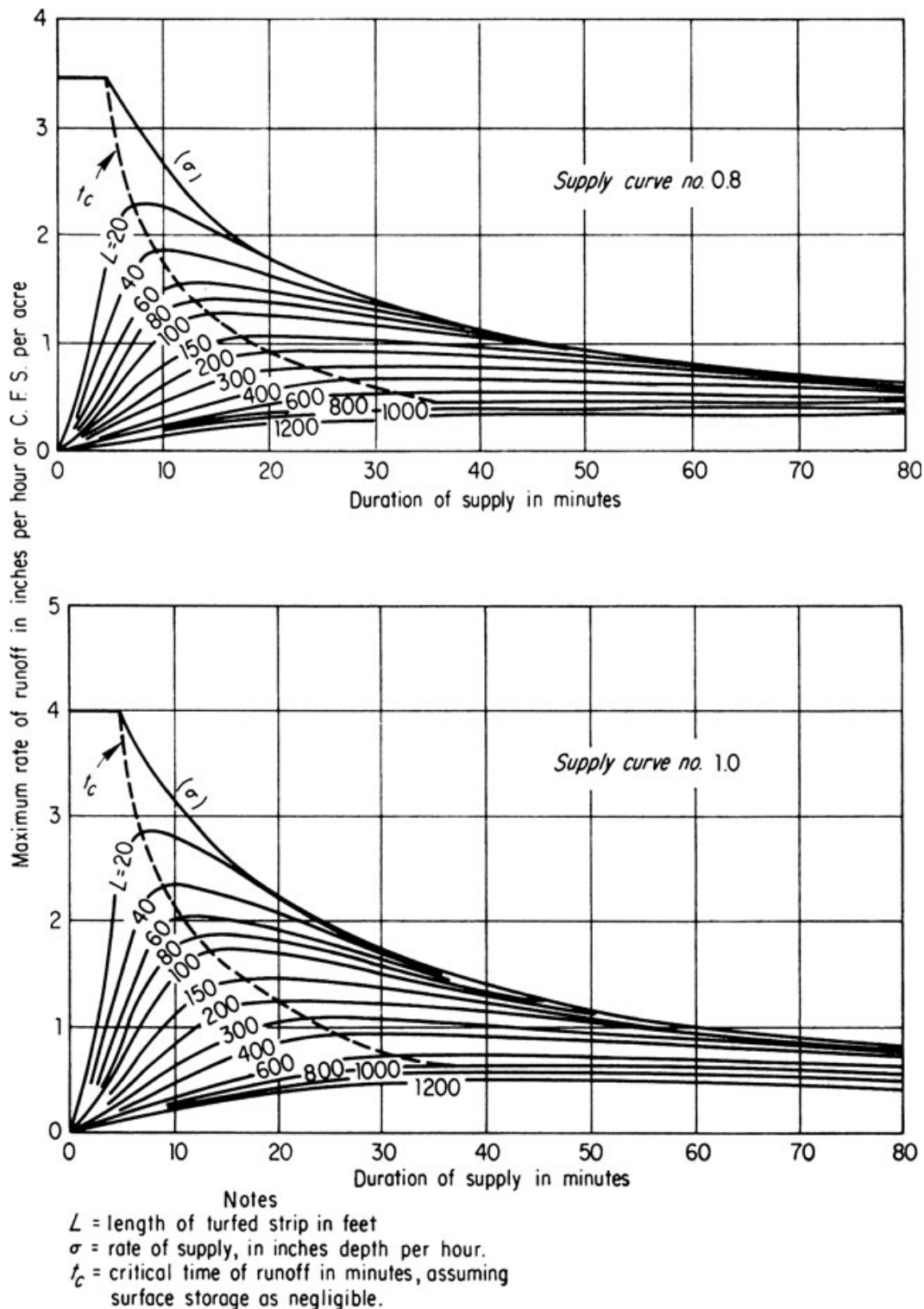


FIGURE 9-8 Standard supply curves, 0.8 and 1.0 in/h (*Corps of Engineers*).

for the pavement is $n = 0.02$, and for the turfed area $n = 0.40$. The drainage areas, retardance coefficients (referred to as *roughness factors*), and actual effective lengths L are shown in Table 9-6. Values of L and S were obtained from a grading plan of the area. The equivalent

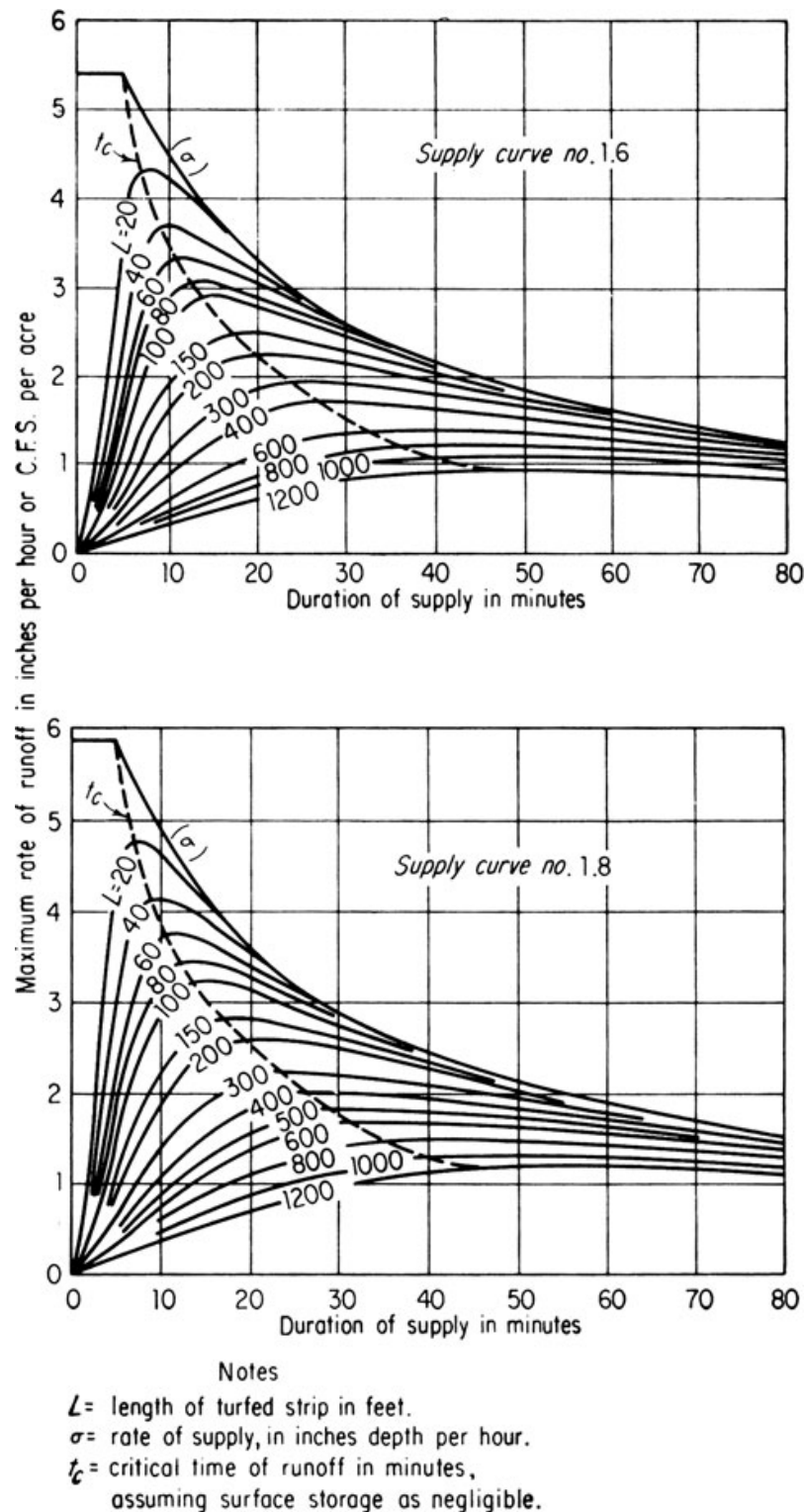


FIGURE 9-9 Standard supply curves, 1.6 and 1.8 in/h (*Corps of Engineers*).

L s are obtained from Fig. 9-10. Column 14, labeled *adopted for selecting diagrams*, designates the nearest whole number which can be identified on the supply curves (Figs. 9-8 and 9-9). The standard supply curve to be used for the example is obtained by weighting the supply curves for the paved and turfed areas. For example, for inlet 4,

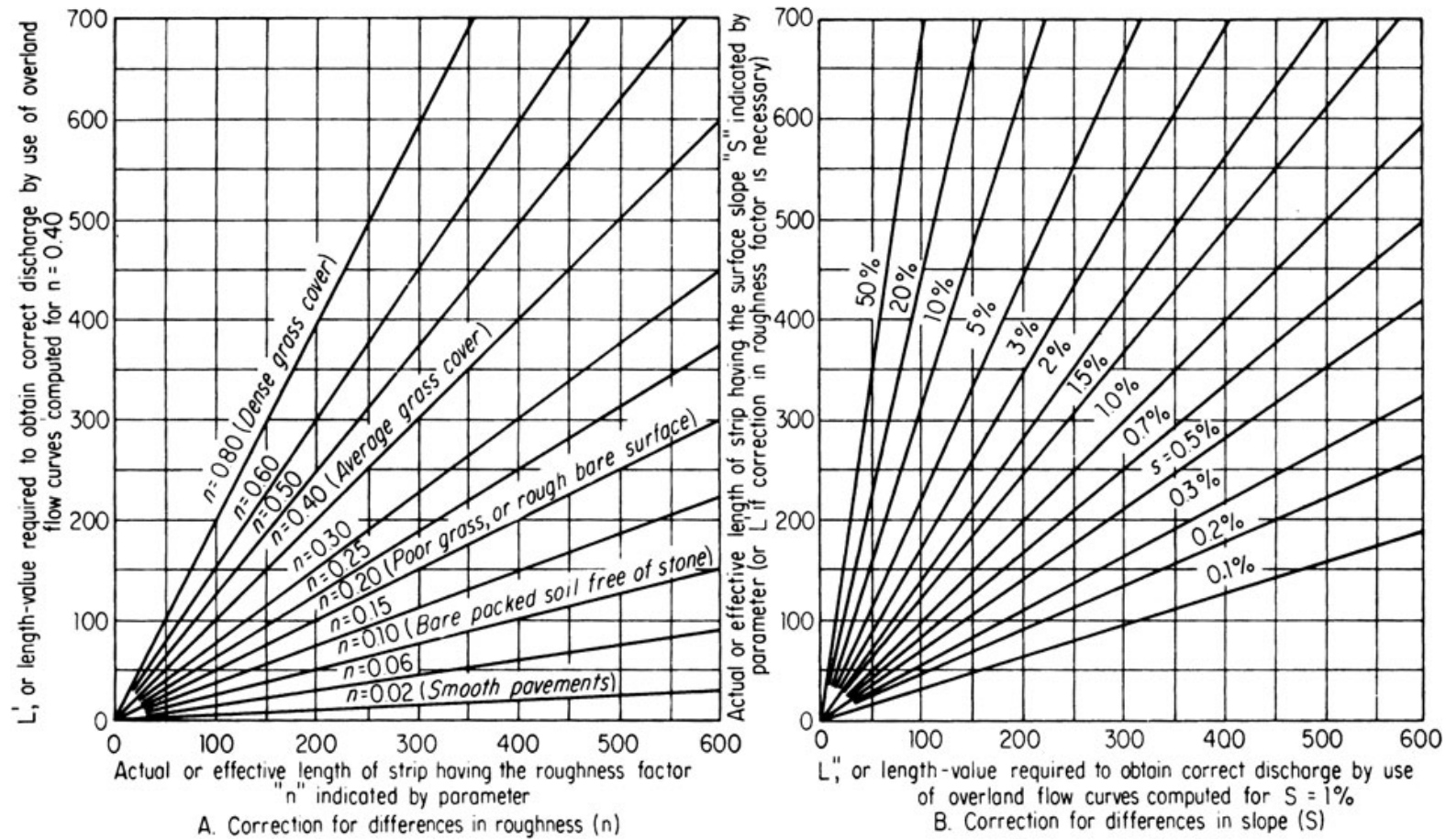


FIGURE 9-10 Modification in L required to compensate for difference in n and S (*Corps of Engineers*).

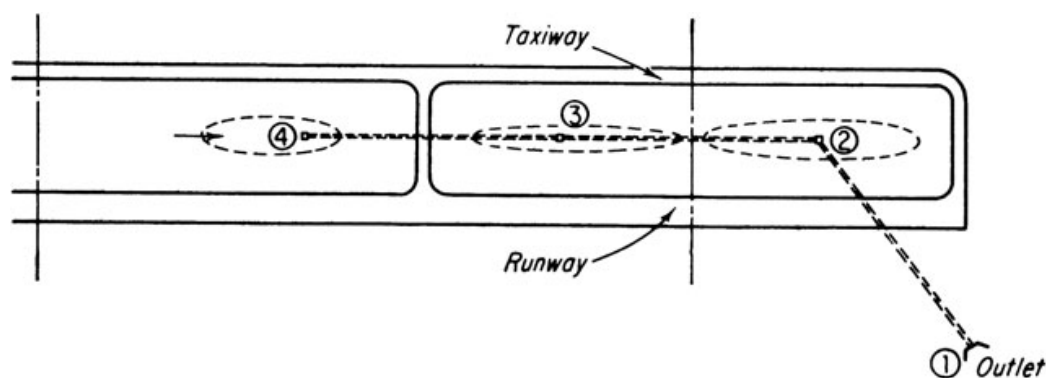


FIGURE 9-11 Portion of airport showing drainage layout (*Corps of Engineers*).

the paved area is 5.97 acres and the supply curve is 2.0 in/h; the turfed area is 26.81 acres and the supply curve is 1.5 in/h. The weighted supply curve is equal to

$$\frac{5.97(2) + 26.81(1.5)}{5.97 + 26.81} = 1.6$$

In columns 20 and 21, the critical inlet time t_c (the time that will produce the maximum discharge) and the corresponding rates of runoff are listed. These values are obtained from Fig. 9-9. In columns 23 and 24, additional rates of runoff for arbitrarily selected times are listed. This is done to facilitate computation for various times of concentration for the several points along a drainage system.

The next step is to compute the volumes of runoff into inlets 4, 3, and 2. The computations are shown in Table 9-7. Obviously the duration of a storm necessary to provide the maximum rate of runoff into inlet 4 is equal to 24 min. The pipe from inlet 4 to inlet 3 is designed for a storm of this duration. At inlet 3 the time of concentration is 24 min plus the flow time in the pipe from inlet 4 to inlet 3 (9.2 min). The pipe from inlet 3 to inlet 2 would be designed for a storm of 33.2-min duration. Enter Fig. 9-9 (supply curves 1.6) with 33 min as the abscissa, and read the rates of runoff for effective lengths L of 280 ft (inlet 3) and 330 ft (inlet 4). Multiply these rates by their respective drainage areas. According to the computations at inlet 3, the area directly tributary to it contributes 62.5 ft³/s, and the area tributary to inlet 4 contributes 59.0 ft³/s. Thus the pipe from inlet 3 to inlet 2 should be designed for a capacity of 59.0 + 62.5 = 121.5 ft³/s. The same process would be repeated for the design of the pipe from inlet 2 to the outlet.

It should be emphasized that the duration of the storm for the analysis of a particular point along the drainage system always corresponds to the time of concentration above this point. Had the inlet

time t_c for the area directly tributary to inlet 3 been larger than the sum of the inlet time for the area tributary to inlet 4 plus the flow time to inlet 3, the former would have established the duration of the storm for the design of the pipe from inlet 3 to inlet 2.

Typical Example—Ponding

If ponding is permissible, the first step is to establish the limits of the ponding area. From a grading and drainage plan, the volumes in the various ponds can be computed. These volumes are then expressed in terms of cubic feet per acre of drainage area, as shown in column 9 of Table 9-8. The actual and equivalent L values are determined in the same manner as for the case of no ponding, with one exception. The actual L is measured to the mean edge of the pond rather than to the drain inlet. The actual and equivalent effective lengths are listed in columns 12 and 13.

The Corps of Engineers has developed charts which yield drain inlet capacities to prevent ponds from exceeding certain specified volumes. Typical charts are shown in Figs. 9-12 and 9-13. The volumes are computed for various supply curves (Fig. 9-2), assuming the slope of the basins forming the drainage areas is 1 percent. The supply curves represent the intensity-duration pattern for storms whose 1-h intensities correspond to the supply curve numbers. The volumes of runoff for a specific supply curve are computed in a manner similar to the procedure used by the FAA. The cumulative volumes of runoff are compared with the various capacities of drain inlets to arrive at the volumes of storage shown in Figs. 9-12 and 9-13. Since the volumes of runoff depend on L and S , charts must be prepared for a wide range of L values. Figures 9-12 and 9-13 show drain inlet capacities for L equal to 100, 200, 300, and 400 ft. Additional charts have been prepared for $L = 0, 40, 600, 800, 1000$, and 1200 ft [8].

The physical significance of the charts may be described by reference to the following example. Suppose that L for a large drainage area is 100 ft and that the runoff pattern corresponds to supply curve 2. Assume that the maximum permissible ponding is 300 ft³/acre of drainage area. From Fig. 9-12 a pipe which has a capacity of 1.0 ft³/s per acre of drainage area would be adequate to prevent the pond from exceeding a volume of 3000 ft³ during any part of the storm. The dashed lines labeled 4 are equal to rates of supply corresponding to a duration of 4 h. Although smaller drain inlets are possible, it is felt that the sizes corresponding to a duration of 4 h are about the minimum from a practical standpoint.

The required drain inlet capacities for the drainage layout in Fig. 9-11 were obtained from Figs. 9-12 and 9-13 and are tabulated in Table 9-8. Note that the time of concentration is not a factor in these computations.

Supply Curve Nos.					Drainage Section Assuming								
For paved areas	2.0												
For bare areas	1.5												
For turfed areas													
	Drainage Area (DA), acres				Permissible Ponding						Length L, ft		
Inlet No.	Paved, n= 0.02	Unpaved		Total	Depth at Inlet, ft	Pond Area, 1000 ft²	Volume 1000 ft³	Volume, ft³/acre DA	Average Roughness Factor n	Average Slope S, %	Actual or Effective Length, ft	EquivalentL for n= 0.40 and	L Adopted for Selecting Diagrams
		Bare	Turf n= 0.40										
1	2	3	4	5	6	7	8	9	10	11	12	13	14
4	5.97		26.81	32.78					0.33	2.0	575	330	330
3	5.69		25.54	31.23					0.33	2.8	575	280	280
2	5.69		25.54	31.23					0.33	2.8	575	280	280

Source: Corps of Engineers [8].

TABLE 9-6 Airfield Drainage—Drain Inlet Capacities

Supply Curve Nos.							Drainage Section				
For paved areas		2.0									
For bare areas											
For turfed areas		1.5									
Point of Design				Critical Runoff Time to Produce Maximum Flow In Underground Drain						Rate of	
Inlet or Junction	Distance, ft		Critical Inlet	t_c , min	Assumed Velocity In Pipe, ft/s	Drain time, min		Approximate t_c (col. 5 + col. 8)	Adopted t_c , min	4	3
	From Main Outlet	From Pre- ceding Inlet				From Pre- ceding Inlet	Accu- mula- tion Total				
1	2	3	4	5	6	7	8	9	10	11	12
4	4805	—	4	24	3.0			24	25	59.0	
3	3155	1650	4	24	3.0	9.2	9.2	33	30	59.0	62.5
2	1505	1650	4	24	3.0	9.2	18.4	42	40	55.8	56.2

Source: Corps of Engineers [8].

TABLE 9-7 Airfield Drainage—Size and Profile of Underground Storm Drains

No Ponding of Runoff										
Standard Supply Curve No. 4 DA					Drain Inlet Capacity			Critical Contribution to System		
Paved Areas	Unpaved Areas		Total	Weighted Supply Curve (col. 18 + col. 5)	t _c , min	q _s , ft ³ /s/acre	Q _s , ft ³ /s (col. 21 4 col. 5)	t _c , min	q _s , ft ³ /s/acre	Q _s , ft ³ /s (col. 24 4 col. 5)
	Bare	Turf								
15	16	17	18	19	20	21	22	23	24	25
11.94		40.22	52.16	1.6	24	1.8	59.0	30	1.8	59.0
								40	1.7	55.8
11.38		38.31	49.69	1.6	23	2.0	62.5	25	2.0	62.5
								30	2.0	62.5
								40	1.8	56.2
11.38		38.31	49.69	1.6	23	2.0	62.5	25	2.0	62.5
								30	2.0	62.5
								40	1.8	56.2

[illegible]

Supply Curve Nos.					Drainage								
For paved areas	2.0												
For bare areas													
For turfed areas	1.5												
Drainage Area (DA), acres					Permlssible Ponding						Length L, ft		
Inlet	Paved, <i>n</i> = 0.02	Unpaved		Total	Depth at Inlet, ft	Pond Area, 1000 ft²	Volume 1000 ft³	Volume, ft³/acre DA	Average Roughness Factor <i>n</i>	Average Slope <i>S</i> , %	Actual or Effective Length, ft	Equivalent <i>L</i> for <i>n</i> = 0.40 and	<i>L</i> Adopted for Selecting Diagrams
		Bare	Turf <i>n</i> = 0.40										
1	2	3	4	5	6	7	8	9	10	11	12	13	14
4	5.97		26.81	32.78	3.0	138	206	6,292	0.33	2.0	525	300	300
3	5.69		25.54	31.23	1.73	145	125	4,016	0.33	2.8	340	200	200
2	5.69		25.54	31.23	2.73	270	368	11,800	0.33	2.8	340	200	200

*Not required when appreciable ponding is permissible.

Source: Corps of Engineers [8].

TABLE 9-8 Airfield Drainage—Drain Inlet Capacities Required to Limit Ponding to Permissible Volumes

Layout of Surface Drainage

A finished grade contour map of the runways, taxiways, and aprons is extremely helpful for the layout of a storm drain system. Several trial drainage layouts may be necessary before the most economical system can be selected. The grades of the storm drain should be such as to maintain a minimum mean velocity on the order of 2.5 ft/s to provide sufficient scouring action to avoid silting. To maintain an adequate cross section for flow at all times, the diameter of the storm drain should not be less than 12 in.

Water from a drainage area is collected into the storm drain by means of inlets. The inlet structure consists of a concrete box, the top of which is covered with a grate made of cast iron, cast steel, or reinforced concrete. The grates must support aircraft wheel loads and

Section East Side of Airfield										
Standard Supply Curve No. & DA					Drain Inlet Capacity			Critical Contribution to System		
Paved Areas	Unpaved Areas		Total	Weighted Supply Curve (col. 18 ÷ col. 5)	t_c , min	q_s , ft ³ /s/acre	Q_s , ft ³ /s (col. 21 × col. 5)	t_c , min	q_s , ft ³ /s/acre	Q_s , ft ³ /s (col. 24 × col. 5)
	Bare	Turf								
15	16	17	18	19	20	21	22	23	24	25
11.94		40.21	51.15	1.6	*	0.52	17.05			
11.38		38.31	49.69	1.6	*	0.52	16.24			
11.38		38.31	49.69	1.6	*	0.52	16.24			

should therefore be designed for contact pressures for the aircraft which will be served by the airport.

On long tangents, drain inlets are usually placed at intervals varying from 200 to 400 ft. The location of the inlets depends on the configuration of the airport and on the grading plan. Normally, if there is a taxiway parallel to the runway, the inlets are placed in a valley between runway and taxiways, as indicated in Fig. 9-11. If there is no parallel taxiway, the drains are placed near the edge of the runway pavement or at the toe of the slope of the graded area. The FAA recommends that the inlets not be closer than 75 ft to the edge of the pavement.

On aprons, inlets are usually placed in the pavement proper. This is the only way a large apron area can be drained. All grates should be securely fastened to the frames so that they will not be jarred loose with the passage of traffic (see Fig. 9-14).

Adequate depths of cover should be provided over the pipes so that the pipes can support traffic. The recommended minimum depths of cover are shown in Table 9-9.

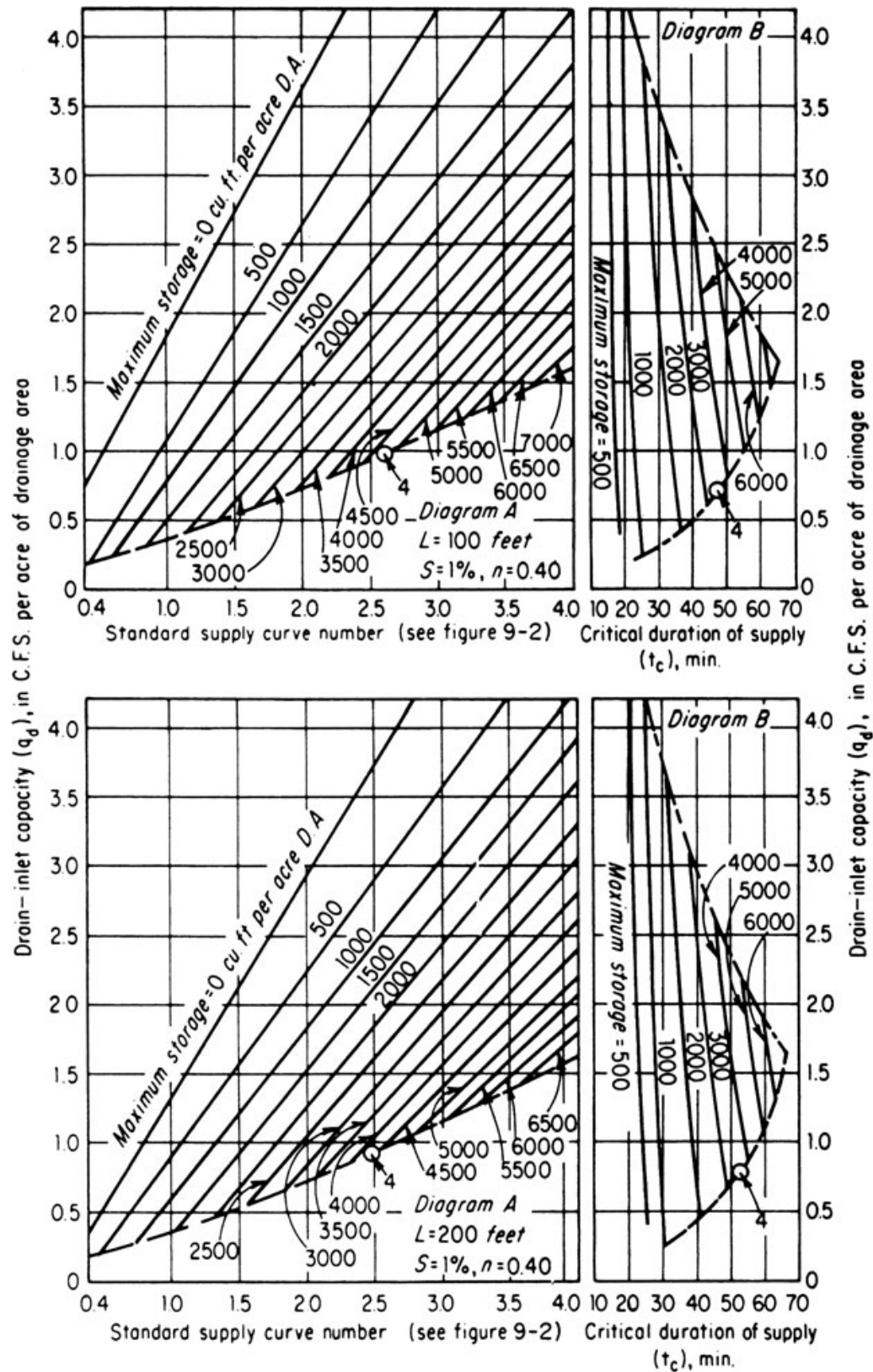


FIGURE 9-12 Drain inlet capacity versus maximum surface storage, $L=100$ ft and $L=200$ ft, C.F.S. = cubic feet per second (*Corps of Engineers*.)

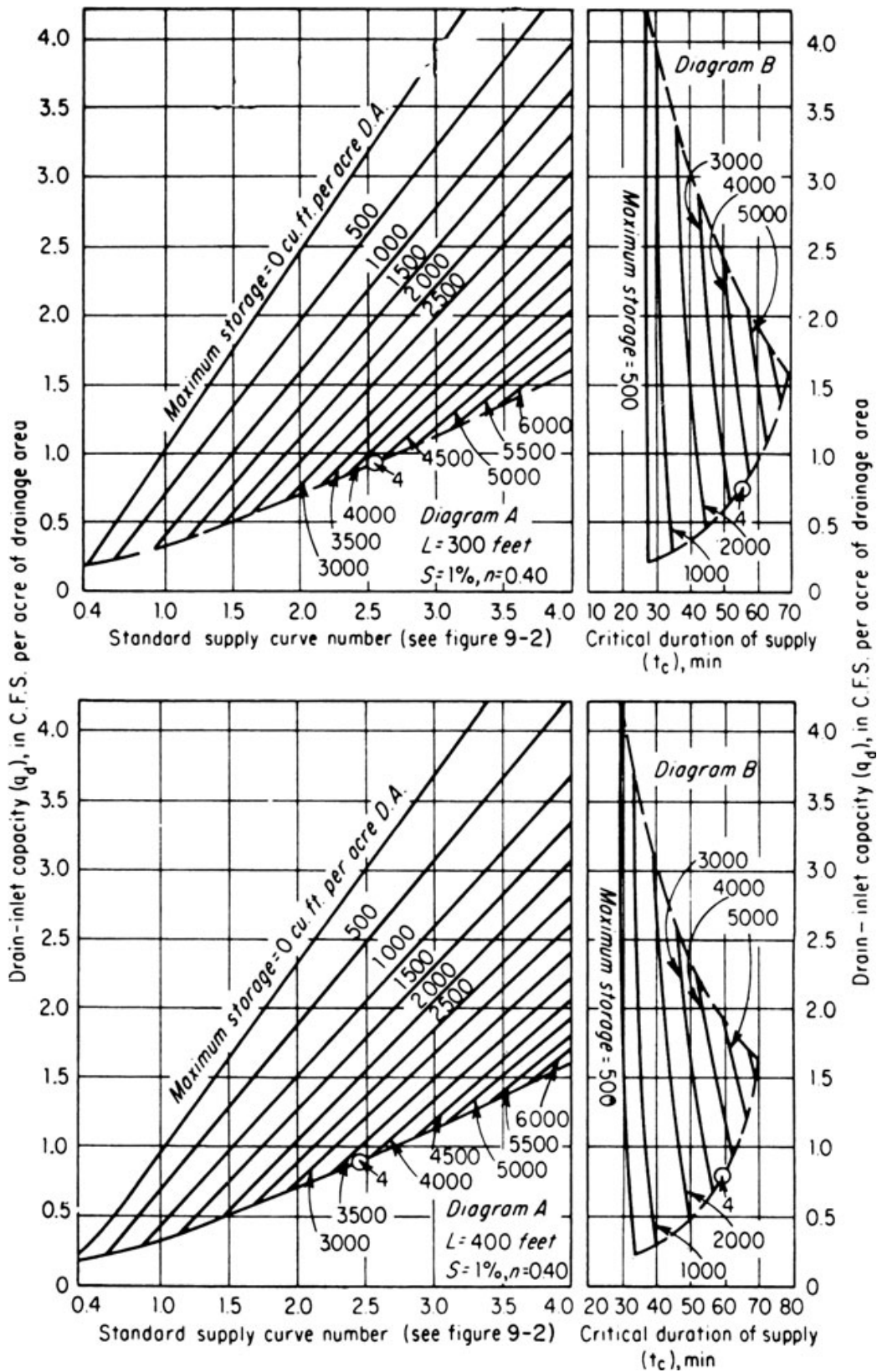


FIGURE 9-13 Drain inlet capacity versus maximum surface storage, $L = 300$ ft and $L = 400$ ft, C.F.S. = cubic feet per second (*Corps of Engineers.*)

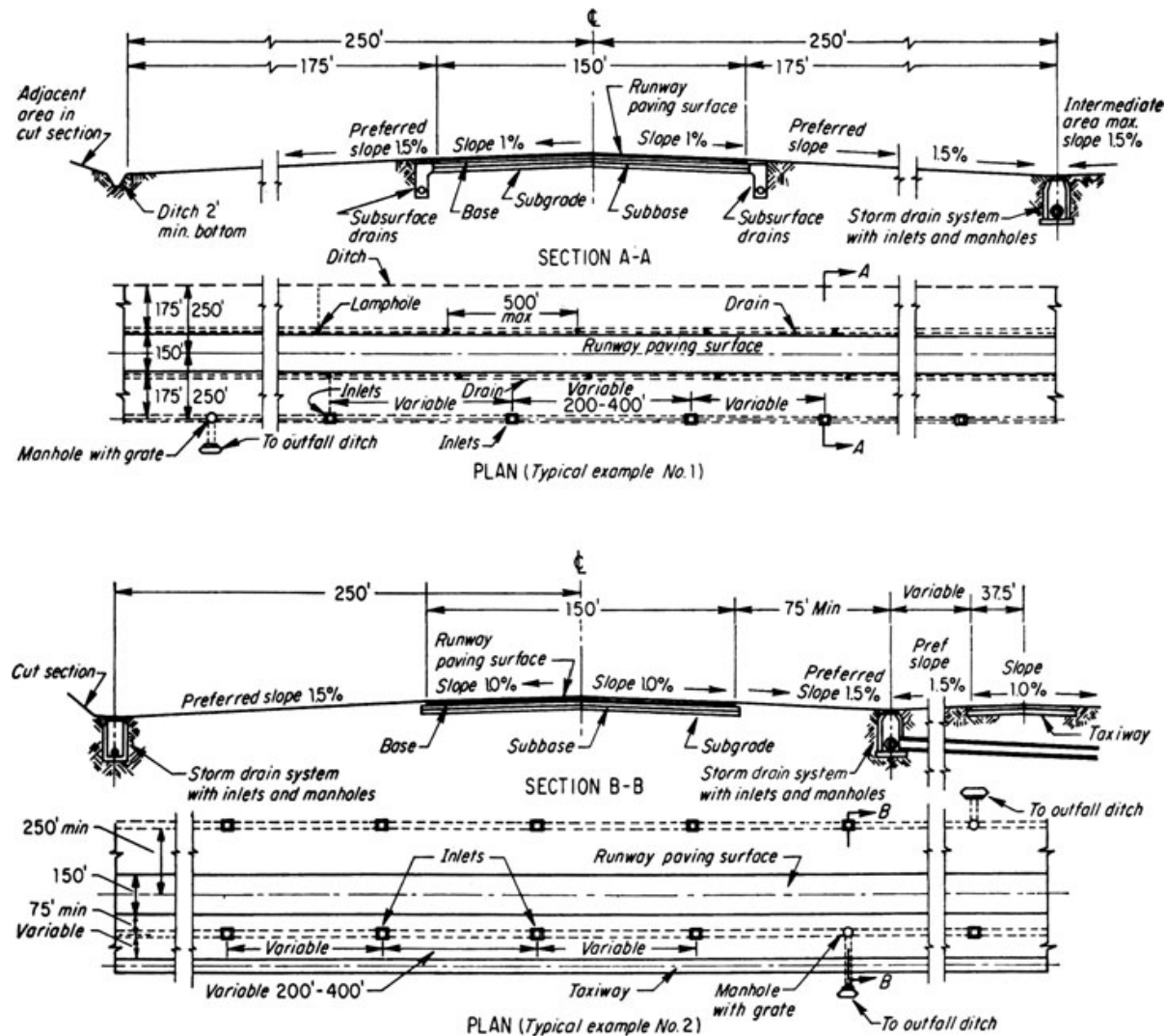


FIGURE 9-14 Recommended pavement drainage sections (*Federal Aviation Administration [2]*).

Flexible Pavement																
Pipe Cover																
Kind of Pipe	Nominal Diameter of Pipe, In								Nominal Diameter of Pipe, In							
	12	18	24	30	36	42	48	60	12	18	24	30	36	42	48	60
	Wheel Load 15,000 lb								Wheel Load 30,000 lb							
Clay sewer pipe	2.5	3.0	3.0	3.5	3.5				3.0	3.5	4.0	4.5	4.5			
Clay culvert pipe	1.5	1.5	1.5	2.0	2.0				2.5	3.0	3.0	3.0	3.0			
Concrete sewer pipe	2.5	3.0	3.0						3.0	3.5	4.0					
Concrete sewer pipe (extra-strength)	1.5	1.5	2.0						2.5	3.0	3.0					
Reinforced-concrete culvert pipe																
Class I								3.0								4.5
Class II	2.5	2.5	2.5	2.5	2.5	2.5	2.5	2.5	3.0	3.0	3.0	3.0	3.0	3.5	3.5	3.5
Class III	2.0	2.0	2.0	2.0	2.0	2.0	2.0	2.0	2.5	2.5	2.5	2.5	2.5	2.5	3.0	3.0
Class IV	1.5	1.5	1.5	1.5	1.5	1.5	1.5	1.5	2.0	2.0	2.0	2.0	2.0	2.0	2.0	2.0
Class V	1.0	1.0	1.0	1.0	1.0	1.0	1.0	1.0	1.5	1.5	1.5	1.5	1.5	1.5	1.5	1.5
Corrugated metal pipe, gauge no.																
16	1.0	1.5	1.5						1.5	2.0	2.0					
14	1.0	1.0	1.0	1.5	2.0				1.0	1.0	1.5	2.0	2.5			
12	1.0	1.0	1.0	1.0	1.5	1.5	2.0		1.0	1.0	1.5	1.5	2.0	2.5	2.5	
10			1.0	1.0	1.0	1.0	1.5	1.5			1.0	1.5	1.5	2.0	2.0	2.5
8					1.0	1.0	1.0	1.0					1.0	1.5	1.5	2.0

TABLE 9-9 Recommended Minimum Depth of Cover for Pipe, ft

Flexible Pavement																
Kind of Pipe	Pipe Cover															
	Nominal Diameter of Pipe, In								Nominal Diameter of Pipe, In							
	12	18	24	30	36	42	48	60	12	18	24	30	36	42	48	60
	Wheel Load 45,000 lb								Wheel Load 60,000 lb							
Clay culvert pipe	3.0	3.5	3.5	4.0	4.0				3.5	4.0	4.5	4.5	5.0			
Concrete sewer pipe (extra-strength)	3.0	3.5	3.5						3.5	4.0	4.5					
Reinforced-concrete culvert pipe																
Class I																
Class II	3.0	3.5	3.5	4.0	4.0	4.5	4.5		3.5	4.0	4.5					
Class III	2.5	3.0	3.0	3.5	3.5	3.5	4.0	4.0	3.0	3.5	4.0	4.0	4.5			
Class IV	2.0	2.5	2.5	2.5	2.5	2.5	3.0	3.0	2.5	3.0	3.0	3.0	3.5	3.5	3.5	4.0
Class V	1.5	1.5	2.0	2.0	2.0	2.0	2.5	2.5	2.0	2.0	2.5	2.5	2.5	3.0	3.0	3.0
Corrugated metal pipe, gauge no.																
16	2.0	2.5	3.0						2.5	3.0	3.5					
14	1.5	2.0	2.5	3.0	3.0				2.0	2.5	3.0	3.5	4.0			
12	1.5	2.0	2.0	2.5	2.5	3.0	3.5		1.5	2.0	2.5	3.0	3.5	4.0	4.0	
10			1.5	2.0	2.0	2.5	3.0	3.5			2.0	2.5	3.0	3.5	3.5	4.0
8					1.5	2.0	2.5	3.0					2.5	3.0	3.0	3.5

	Wheel Load 75,000 lb								Wheel Load 100,000 lb							
Reinforced-concrete culvert pipe																
Class I																
Class II																
Class III	3.0	4.0	4.5	4.5	5.0				3.5	4.5	5.0	5.0	5.5			
Class IV	3.0	3.0	3.5	3.5	4.0	4.0	4.5	5.0	3.5	3.5	4.0	4.0	4.5	5.0	5.0	5.5
Class V	2.5	2.5	3.0	3.0	3.0	3.5	4.0	4.0			3.5	3.5	3.5	4.0	4.5	4.5
Corrugated metal pipe, gauge no.																
16	3.0	3.5	4.0						3.5	4.0	4.5					
14	2.5	3.0	3.5	4.0	4.5				3.0	3.5	4.0	4.5	5.0			
12	2.5	3.0	3.0	3.5	4.0	4.5	5.0		3.0	3.0	3.5	4.0	4.5	5.0	5.5	
10			2.5	3.0	3.5	4.0	4.5	5.0			3.0	3.5	4.0	4.5	5.0	5.5
8					3.0	3.5	4.0	4.5					3.5	4.0	4.5	5.0
Cover depths measured from top of flexible pavement or unsurfaced areas to top of pipe. Cover for pipe in areas not used by aircraft shall be in accordance with cover requirements for 15,000-lb wheel loads.																
Rigid Pavement																
Pipe placed under rigid pavements shall have a minimum cover, measured from the bottom of the slab, of 1.0 ft.																

Note: The recommended minimum depth of cover for pipe does not provide protection against freezing conditions in seasonal freezing areas.

Source: Federal Aviation Administration [2].

TABLE 9-9 Recommended Minimum Depth of Cover for Pipe, ft (*Continued*)

	<i>n</i>
Pipe	
Clay and concrete	
Good alignment, smooth joints, smooth transitions	0.013
Less favorable flow conditions	0.015
Corrugated metal	
100% of periphery smoothly lined	0.013
Paved invert, 50% of periphery paved	0.018
Paved invert, 25% of periphery paved	0.021
Unpaved, bituminous-coated or noncoated	0.024
Open channels	
Paved	0.015–0.020
Unpaved	
Bare earth, shallow flow	0.020–0.025
Bare earth, depth of flow over 1 ft	0.015–0.020
Turf, shallow flow	0.06–0.08
Turf, depth of flow over 1 ft	0.04–0.06

Source: Federal Aviation Administration [2].

TABLE 9-10 Coefficients of Roughness *n*

As a guide for the design of storm drains, the coefficient of roughness *n* for various types of pipes and open channels is listed in Table 9-10.

Subsurface Drainage

The functions of subsurface drainage are to (1) remove water from a base course, (2) remove water from the subgrade beneath a pavement, and (3) intercept, collect, and remove water flowing from springs or pervious strata.

Base drainage is normally required (1) where frost action occurs in the subgrade beneath a pavement, (2) where the groundwater is expected to rise to the level of the base course, and (3) where the pavement is subject to frequent inundation and the subgrade is highly impervious.

Subgrade drainage is desirable at locations where the water may rise beneath the pavement to less than 1 ft below the base course.

Intercepting drainage is highly desirable where it is known that subsurface waters from adjacent areas are seeping toward the airport pavements.

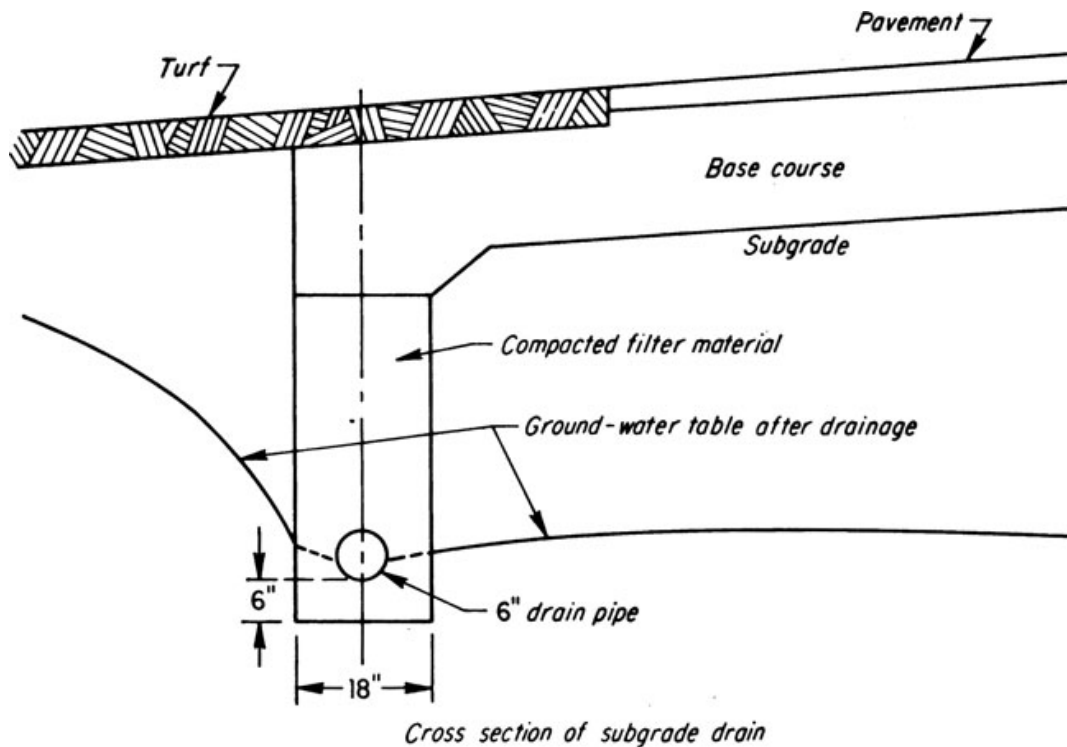


FIGURE 9-15 Subgrade subdrainage details (*Corps of Engineers*).

Methods for Draining Subsurface Water

Base courses are usually drained by installing subsurface drains adjacent to and parallel to the edges of the pavement. The pervious material in the trench should extend to the bottom of the base course, as shown in Fig. 9-15. The center of the drainpipe should be placed a minimum of 1 ft below the bottom of the base course.

Subgrades are drained by pipes installed along the edges of pavement and in some instances, where the groundwater is extremely high, underneath the pavements. The center of the subsurface drain should be placed no less than 1 ft below the level of the groundwater. When subgrade drains are installed along the edges of the pavement, they may also serve for draining the base course.

Intercepting drainage can be accomplished by means of open ditches well beyond the pavement areas. If this is not practical, then subdrains can be used.

Types of Pipe

The following types of pipe have been used for subdrainage:

1. Perforated metal, concrete, or vitrified clay pipe. The joints are sealed. The perforations normally extend over about one-third of the circumference of the pipe. The perforated area is usually placed adjacent to the soil.
2. Bell-and-spigot pipes are laid with the joints open. Vitrified clay, cast iron, and plain concrete are used in the manufacture of bell-and-spigot pipes.

3. Porous concrete pipe collects water by seepage through the concrete wall of the pipe. This type of pipe is laid with the joints sealed.
4. Skip pipe manufactured of both vitrified clay and cast iron is a special type of bell-and-spigot pipe with slots at the bells.
5. Farm tile is made of clay or concrete with the ends separated slightly to permit the entrance of water. This type of pipe is rarely used on airport projects.

Pipe Sizes and Slopes

Experience has shown that a 6-in-diameter drain is adequate, unless extreme groundwater conditions are encountered. If desired, the flow may be estimated by means of the available theories for soil drainage [7]. These theories require knowledge of the effective porosity and coefficient of permeability of the soil which is being drained, as well as the head on the pipe and the distance which the water must flow to reach the drain. Rarely is theory relied on to compute pipe sizes.

The recommended minimum slope for subdrains is 0.15 ft in 100 ft. A minimum thickness of 6 in of filter material should surround the drain. The gradation of the filter material is discussed in succeeding paragraphs.

Utility Holes and Risers

For cleaning and inspection, utility holes and risers are often installed along the drains. The Corps of Engineers recommends that utility holes be placed at intervals of not more than 1000 ft, with one riser approximately midway between the holes [7]. The function of the riser is to be able to insert a hose for flushing the system. The function of a utility hole is to permit inspection of the pipes.

Gradation of Filter Material

The term *filter material* applies to the granular material which is used as backfill in the trenches where subdrains are placed. To permit free water to reach the drain, the filter material must be many times more pervious than the protected soil. Yet if the filter is too pervious, the particles of soil to be drained will move into the filter material and clog it.

On the basis of some general studies conducted by K. Terzaghi, the Corps of Engineers has developed an empirical design for filter material which has been substantiated by tests [10]. The criteria for selecting the gradation of the filter material are as follows:

1. To prevent clogging of a perforated pipe with filter material, the following requirement must be satisfied:

$$\frac{\text{85\% size of filter material}^*}{\text{Diameter of perforation}} > 1$$

2. To prevent the movement of particles from the protected soil into the filter material, the following conditions must be satisfied:

$$\frac{\text{15\% size of filter material}}{\text{85\% size of protected soil}} \leq 5$$

and

$$\frac{\text{50\% size of filter material}}{\text{50\% size of protected soil}} \leq 25$$

3. To permit free water to reach the pipe, the following condition must be fulfilled:

$$\frac{\text{15\% size of filter material}}{\text{15\% size of protected soil}} \geq 5$$

A typical example of design is shown in Fig. 9-16. Concrete sand has proved to be a satisfactory filter material for the majority of fine soils which are drainable. A single gradation of filter material is preferred for simplicity of construction.

Filter materials tend to segregate as they are placed in trenches. To minimize this tendency, the material should not have a coefficient of uniformity greater than 20. For the same reason, filter materials should not be skip-graded. Filter materials should always be placed in a moist state. The presence of moisture tends to reduce segregation.

Drainability of Soils

Certain types of soils, such as gravelly sands, sand, and sandy loams, are usually self-draining and require very little, if any, subsurface drainage. Subsurface drainage can be effective for draining clay loams, sandy clay loams, and certain silty loams. The amount of sand in these soils largely determines how drainable they are. For soils containing a high percentage of silt and clay, subsurface drainage becomes very problematic.

*This means that 85 percent (by weight) is finer than the specified size.

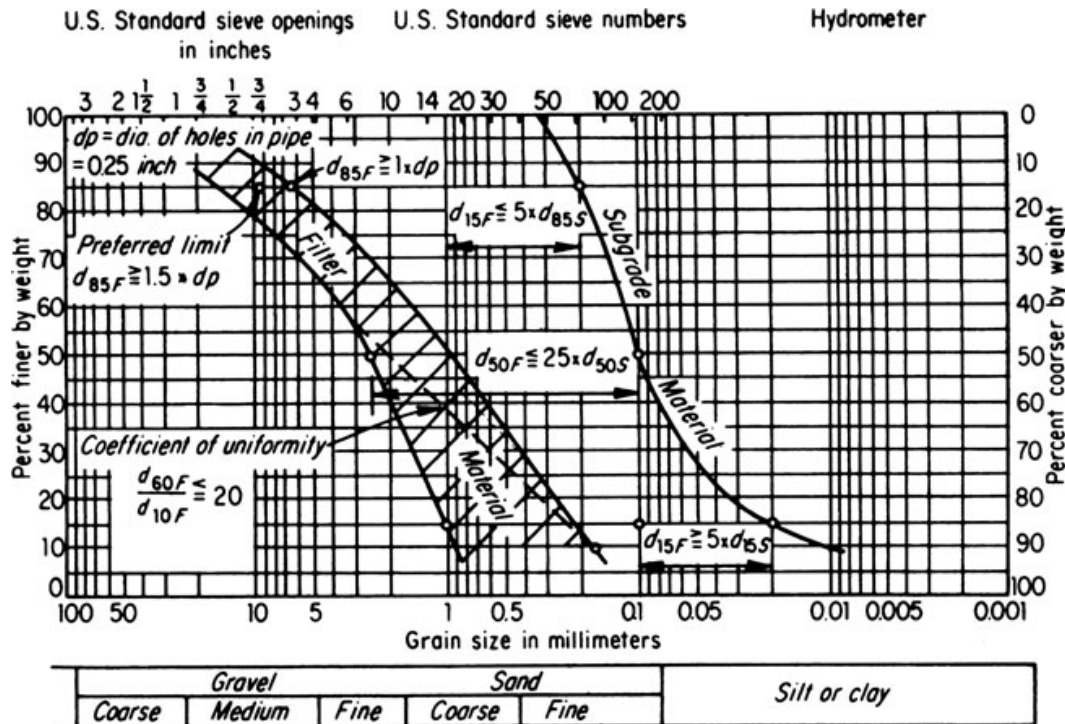


FIGURE 9-16 Design example for filter materials (*Corps of Engineers*).

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**SCHOOL OF MECHANICAL ENGINEERING
DEPARTMENT OF AERONAUTICAL ENGINEERING**

UNIT – V – Air Traffic Control – SAE1605

CHAPTER 3

Air Traffic Management

Introduction

In order that the airport planner and designer may be aware of the importance of the rules and technologies that define the aviation operations within the airspace, a very brief summary of what constitutes air traffic control, increasingly being known as air traffic management, how it is managed and operated, and the principal aids to air navigation, is presented in this chapter.

An appreciation of air traffic management and its current and future operating and technological characteristics will focus attention on the fact that any extensive reorientation of runways on existing airports or the construction of entirely new airports requires consultation with the organizations in charge of operating surrounding airspace and very often an airspace study. This is particularly true in large metropolitan areas where several airports are present and the existing airspace must be shared by several airports. In addition, the design of local airspace procedures include procedures for the departure and arrival of aircraft to airport runways requires a fundamental knowledge in current and future air traffic control technologies and policies. Conflicts in air traffic procedures can seriously affect the efficiency of any single airport or a system of airports in a region. The planning of airports must include provisions for facilities located at airports that support the air traffic management system.

As enhanced air traffic management technologies and strategic plans continue to be implemented, consideration of local air traffic procedures has become increasingly relevant and important for even the smallest of airports.

As of 2008, the air traffic management system was just beginning a complete system transformation. As such it is imperative of the airport planner to have an understanding of both the fundamentals of air traffic management and the constant enhancements to the system. This chapter is intended to only briefly introduce the system to the airport planner.

A Brief History of Air Traffic Management

The first attempt to set up rules for air traffic control was made by the International Commission for Air Navigation (ICAN), which was under the direction of the League of Nations. The procedures which the commission promulgated in July of 1922 were adopted by 14 countries. Although the United States was not a member of the League of Nations, and therefore did not officially adopt the rules, many of the procedures established by ICAN were used in the promulgation of air traffic procedures in the United States as well as in most regions of the world.

Construction and operation of the airways system in the United States prior to 1926 were controlled by the military and by the Post Office Department. The formal entry of the federal government into the regulation of air traffic came with the passage of the Air Commerce Act of 1926 (Public Law 64-254). This act directed the Bureau of Air Commerce to establish, maintain, and operate lighted civil airways. At the present time the Federal Aviation Administration maintains and operates the airways system of the United States.

The establishment of the International Civil Aviation Organization (ICAO) in 1944 helped to standardize recommended air traffic control procedures internationally. Today, air traffic control in each country is operated either by its federal government or by private corporations under governmental supervision and regulations. Examples of international air traffic control organizations include the Federal Aviation Administration in the United States, National Air Traffic Services Ltd. (NATS) serving the United Kingdom, NAV Canada in Canada, and Air Services Australia serving the Australian continent. In addition, Eurocontrol, an intergovernmental organization comprising 38 member states within the European Union, coordinates, standardizes, and assists in managing air traffic in the airspace over the European continent.

The primary mission of the Federal Aviation Administration, as well as its international counterparts, is to provide for safe and efficient movement of aircraft throughout the airspace system. The primary function of the air traffic management system is to prevent collisions between aircraft. As such, the FAA office of air traffic management is made up of and responsible for a series of hierarchical control facilities, ground and satellite based navigational aides and aircraft routing procedures, as well as a defined system of air routes and airspace classifications. While much of the current air traffic system is in many ways based on the original development of air traffic control in the early twentieth century, it should be noted that air traffic control policies are constantly changing as the most modern technologies are implemented to better manage increasing air traffic volumes.

The Organizational Hierarchy of Air Traffic Management in the United States

In general, aircraft operate in what is known as the National Airspace System (NAS). The NAS is defined by a series of air routes, airspace classifications, and navigational aids. Aircraft operate within the NAS under varying levels of air traffic control, based primarily on the weather conditions and the type and amount of flight activity within the area. In areas with very low volumes of flight activity during excellent visibility conditions, aircraft may operate in the complete absence of air traffic control, whereas in the busiest airspace or when visibility is limited, aircraft may be under full “positive” control, only being able to change speed, course, or altitude by direct orders from an air traffic controller.

The NAS is operated and managed by a hierarchical organization of air traffic control facilities. The specific purpose of the air traffic control service is to prevent collisions between aircraft and on the maneuvering area between aircraft and obstructions, to expedite and maintain an orderly flow of air traffic [3].

The Air Traffic Control System Command Center

In the United States, air traffic control is managed on a macro level at the air traffic control system command center (ATCSCC) in Herndon, Virginia. In 2007, ATCSCC monitored an average of 25,000 flights per day, with an average of 6000 flights airborne during peak periods. In addition, ATCSCC manages flights planned 6 to 12 h in the future, with the purpose of planning for limiting congestion within the nation’s airspace. In doing so, ATCSCC has the authority to implement ground delay programs by dictating certain aircraft to remain at their airports of departure to prevent further congestion in points of the airspace or at airports suffering from delays due to weather or heavy traffic volumes.

Air Route Traffic Control Centers

Air route traffic control centers (ARTCCs) have the responsibility of controlling the movement of en route aircraft along the airways and jet routes, and in other parts of the airspace. Each of the 21 air traffic control centers within the United States has control of a defined geographical area which may be greater than 100,000 mi² in size. At the boundary point, which marks the limits of the control area of the center, control of aircraft may be transferred to an adjacent center or an approach control facility, or radar service may be terminated and VFR aircraft are free to contact the next center. Air traffic control centers are normally not located at airports. Air traffic control centers can also provide approach control service to nontowered airports and to nonterminal radar approach control airports.

Each ARTCC geographical area is divided into sectors. The configuration of each sector is based on equalizing the workload of the controllers. Control of aircraft is passed from one sector to another. The geographical area is sectorized not only in the horizontal but also in the vertical plane. Thus there can be a high-altitude sector above one or more low-altitude sectors. Each sector is manned by one or more controllers, depending on the volume and complexity of traffic. The average number of aircraft that each sector can handle depends on the number of people assigned to the sector, the complexity of traffic, and the degree of automation provided.

Each sector is normally provided with one or more air route surveillance radar (ARSR) units which cover the entire sector and allow for monitoring of separation between aircraft in the sector. In addition, each sector has information on the identification of the aircraft, destination, flight plan route, estimated speed, and flight altitude, which is posted on pieces of paper called flight progress strips, and are superimposed on the radarscope adjacent to the blips which identify the position and identity of aircraft. The strips are continuously updated as the need arises.

At present, communication between the pilot and controller is by voice. Therefore each ARTCC is assigned a number of VHF and UHF radio communication frequencies. The controller in turn assigns a specific frequency to the pilot. However, modernization of air traffic control is planned to include further proliferation of digital communications, known as controller pilot data link communications (CPDLC) between controllers and pilots.

Terminal Approach Control Facilities

The terminal approach control facility (TRACON) monitors the air traffic in the airspace surrounding airports with moderate to high density traffic. It has jurisdiction in the control and separation of air traffic from the boundary area of the air traffic control tower at an airport to a distance of up to 50 mi from the airport and to an altitude ranging up to 17,000 ft. This is commonly referred to as the terminal area. Where there are several airports in an urban area, one facility may control traffic to all of these airports. In essence the facility receives aircraft from the ARTCC and guides them to one of several airports. In providing this guidance, it performs the important function of metering and sequencing aircraft to provide uniform and orderly flow to the airports.

The organizational structure of an approach control facility is very similar to the ARTCC. Like the ARTCC, the geographic area of the facility is divided into sectors to equalize the workload of the controllers. The approach control facility transfers control of an arriving aircraft to the airport control tower when it is lined up with the runway about 5 mi from the airport. Likewise, control of departing aircraft is transferred to the approach control facility by the airport control tower.

Airport Traffic Control Tower

The airport traffic control tower (ATCT) is the facility which supervises, directs, and monitors the arrival and departure traffic at the airport and in the immediate airspace within 5 mi from the airport. The tower is responsible for issuing clearances to all departing aircraft, providing pilots with information on wind, temperature, barometric pressure, and operating conditions at the airport, and for the control of all aircraft on the ground except in the maneuvering area immediately adjacent to the aircraft parking positions called the ramp area. In the United States in 2007, there were more than 550 air traffic control towers. While most towers are operated by the FAA, as of January 2007, 233 were operated by the private sector under the FAA's contract tower program. The number of operating contract towers, as they are known, has increased tremendously since the inception of the program in 1982. Figure 3-1 provides an illustration of the new ATCT, overshadowing the previously active ATCT, at the Hartsfield-Jackson Atlanta International Airport, Atlanta, Georgia.



FIGURE 3-1 The new ATCT dwarfing the old tower at Hartsfield-Jackson Atlanta International Airport (*ATCmonitor.com*).

Flight Service Stations

While not providing specific control, flight service stations (FSS) are the element of the air traffic management system that provides information and other noncontrol communications to aircraft operating in the system. Their principal functions are to accept and close flight plans, brief pilots about their routes of flight, and to provide important information, in the form of notices to airmen (NOTAMs) before flight and in flight, on such items as severe weather, the status of navigational aids, airport runway closures, and changes in published approach and departure procedures. A secondary function is to relay traffic control messages between aircraft and the appropriate control facility on the ground.

Flight service has gone through a number of changes since the early 1990s. In the 1990s the FAA consolidated more than 180 flight service stations into approximately 60 automated flight service stations (AFSS) which allow many functions, particularly with respect to disseminating weather and other NOTAMs and the filing of flight plans to be performed electronically by voicemail or computer.

In 2005, the FAA awarded a contract to operate the AFSS system to the Lockheed Martin Corporation, representing another step in the privatization of major components of the nation's air traffic control system. While the privatization of the AFSS system has caused some controversy within the aviation industry, there has been relatively little impact of this or any other FAA privatization efforts on airport planning and design.

Air Traffic Management Rules

Air traffic rules are traditionally applied based on prevailing meteorological conditions. Visual meteorological conditions (VMC) are applied when there is sufficient visibility for pilots of aircraft to be able to navigate by referencing locations on the ground, as well as to be able to see and avoid other aircraft in the area. Around airports, VMC is defined as at least 3 statute miles visibility and cloud "ceilings" (defined as at least $\frac{5}{8}$ of the sky covered by clouds) of at least 1000 ft above the ground (AGL). Conversely, instrument meteorological conditions (IMC) exist when visibilities are less than 3 statute miles and cloud ceilings are less than 1000 ft above the ground.

At its most basic level, aircraft operating in VMC tend to fly under visual flight rules (VFR). VFR flight rules depend on aircraft operators to visually maintain adequate separation from terrain, clouds, and other aircraft. Under VFR, aircraft navigation is based on visual reference to locations on the ground, including visual identification and approaches to airports.

While flying under VFR conditions, pilots may request from air traffic control to be under "flight following." Under flight following, air traffic control operators provide assistance to pilots by supervising

course and altitude changes, as well as actively notifying pilots of nearby aircraft. Pilots flying under VFR conditions are required to fly under flight following in the busiest of airspace.

Aircraft flying in IMC or at altitudes over 18,000 ft above sea level (AMSL) fly under instrument flight rules (IFR). Aircraft flying under IFR navigate using ground-based and satellite-based navigation aides and are fully controlled along planned routes by air traffic control personnel. Often times, flights operating under IFR will fly defined departure and approach procedures to and from airports which depend on flying precise courses and altitudes to and from waypoints as defined by ground- and satellite-based navigation systems. These published instrument procedures provide for aircraft to safely and efficiently depart from and arrive to airport runways while avoiding collisions with terrain and other aircraft during poor visibility conditions. In many ways, IFR rules, routes, and departure and approach procedures have significant influence on the planning, design, and operation of airports.

Airspace Classifications and Airways

In the United States, domestic airspace is defined into six classes, plus areas with special operating restrictions, and a designated series routes between airports and waypoints. Aircraft are subject to different levels of air traffic control depending on which airspace classification they are currently operating in, the type of defined route they are on, and whether they are flying under VFR or IFR flight rules.

Classes of airspace in the United States are identified alphabetically, as Class A, B, C, D, E, or G airspace, as illustrated in Fig. 3-2.

Class A airspace, also known as positive control airspace, is the airspace between 18,000 ft above mean sea level (AMSL) (known as FL 180) and 60,000 ft (FL 600) AMSL over the 48 contiguous United States and Alaska, extending out to 12 nm off the coast of the United States.

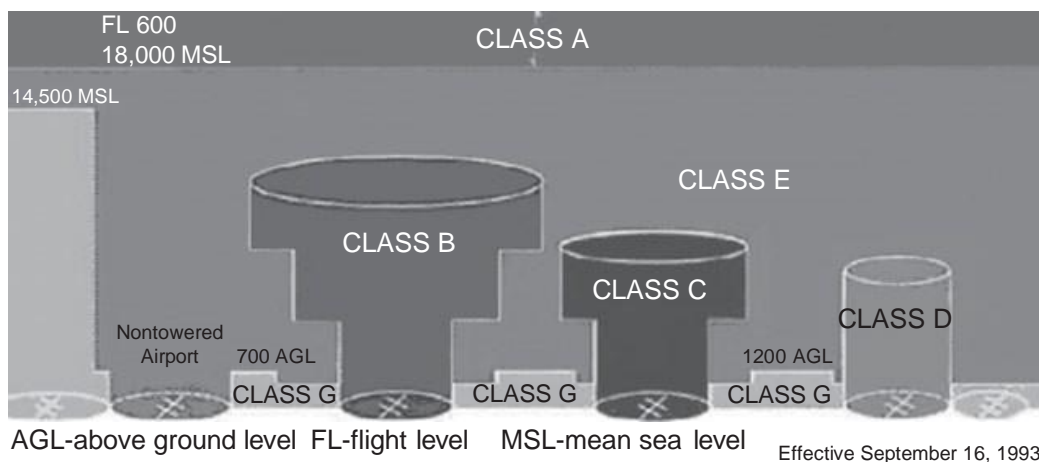


FIGURE 3-2 Illustration of airspace classes.

Since aircraft flying in Class A airspace are generally fast moving commercial airline or general aviation aircraft, all aircraft operating in Class A airspace operate under IFR.

Class B airspace are defined areas within a 30 nm radius around the busiest airports, including areas of multiple large airports, in the United States. Class B airspace surrounds 36 of the busiest commercial service airports in the United States. Class B airspace is typically shaped in the form of what is known in the industry as an “inverted wedding cake.” Nearest the busiest airports within the radius of Class B airspace, Class B airspace extends from the surface of the busiest airports in the area to generally 10,000 ft MSL. Farther away from the airport, Class B may begin at some altitude above the surface and extend to 10,000 ft MSL. The purpose of Class B airspace is to provide an area of positive air traffic control to coordinate the many high-speed aircraft transitioning from high altitudes to landing at the busiest airports, and vice versa, with local lower altitude traffic within the area, while providing airspace at lower altitudes further away from the airport to be used with lower levels of control for smaller and slower general aviation aircraft in the region. Aircraft operating within Class B airspace are under positive air traffic control, and as such must either be flying under IFR rules or, with permission from air traffic control, under VFR rules with flight following. An example depiction of Class B airspace is illustrated in Fig. 3-3. This illustration is a portion of an airspace sectional chart, provided by the U.S. Department of Defense and the Federal Aviation Administration as one standard for identifying classes of airspace, airports, navigational aids, and air routes in the NAS.

United States Airspace Class B Areas, centered around the following civil airports:

- PHX Phoenix Sky Harbor International
- LAX Los Angeles International
- SAN San Diego International Lindbergh Field
- SFO San Francisco International
- DEN Denver International
- MIA Miami International
- MCO Orlando International
- TPA Tampa International
- HNL Honolulu International
- ORD Chicago O'Hare International
- CVG Cincinnati/Northern Kentucky International
- MSY Louis Armstrong New Orleans International
- BWI Baltimore/Washington International

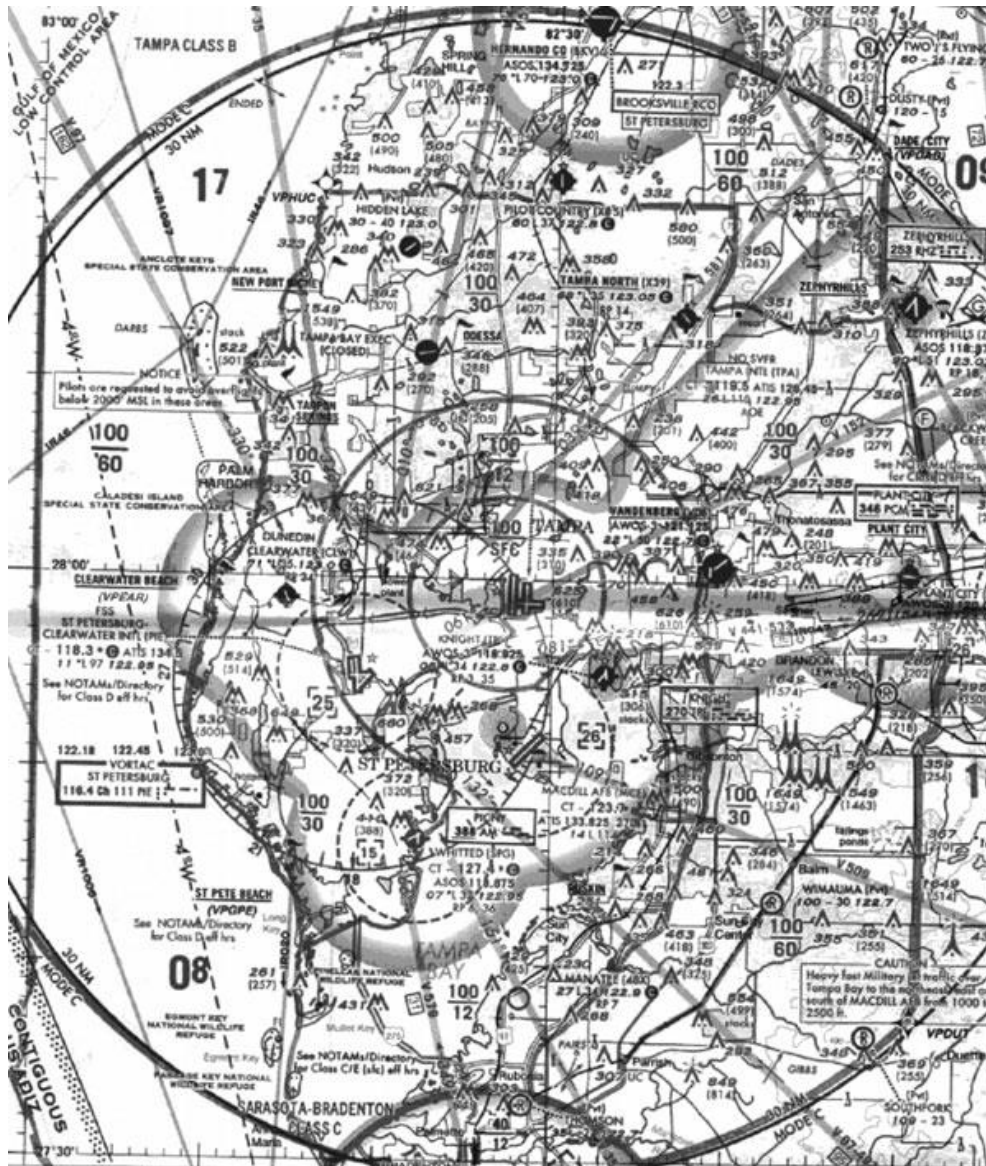


FIGURE 3-3 Class B airspace around Tampa International Airport, Tampa, Florida.

- BOS General E. L. Logan International (Boston)
- DTW Detroit Metropolitan Wayne County
- MSP Minneapolis-St. Paul International
- MCI Kansas City International
- STL Lambert-St. Louis International
- LAS Las Vegas McCarran International
- EWR Newark Liberty International
- JFK John F. Kennedy International
- LGA New York LaGuardia
- CLT Charlotte/Douglas International
- CLE Cleveland-Hopkins International

- CVG Cincinnati/Northern Kentucky International Airport
- PHL Philadelphia International
- PIT Pittsburgh International
- MEM Memphis International
- DAL Dallas Love Field
- DFW Dallas-Fort Worth International
- HOU Houston William P. Hobby
- IAH George Bush Intercontinental (Houston)
- SLC Salt Lake City International
- DCA Ronald Reagan Washington National
- IAD Washington Dulles International
- SEA Seattle-Tacoma International

Class C airspace is found around airports without as much operating volume as those around Class B airspace, but is busy enough to warrant some active level of air traffic control within 10 mi of the airport. VFR traffic operating within Class C airspace must adhere to strict cloud separation requirements and have at least 3 mi of visibility so that they may sufficiently be able to see and avoid other traffic. In addition, all traffic operating within Class C airspace must have established radio communication with air traffic control. The shape of Class C airspace is also in the form of an upside down wedding cake, extending from the surface to typically 4000 ft AGL around the inner 5-nm radius around the airport, and from 1000–2000 ft to 4000 ft AGL from 5 to 10 nm from the airport. Figure 3-4 provides an illustration of Class C airspace surrounding the Daytona Beach International Airport, depicted by a two concentric rings of radii 5 and 10 mi around the airport.

Class D airspace is found within a 5-mi radius of an airport with an operating air traffic control tower, extending from the surface to typically 2500 ft AGL. The purpose of Class D airspace is to provide an area of air traffic control authority to controllers in the airport's control tower, who are responsible for the safe separation of arriving and departing aircraft to and from the airport. Aircraft operating under VFR flight rules are allowed to operate within Class D airspace as long as they establish communication with the air traffic controllers in the tower. When an airport's control tower is in operation, the airport is said to be a "controlled" airport. When the airport's tower is not operational, the airport is considered "uncontrolled" and Class D airspace is no longer active. Airports without a control tower are considered "uncontrolled airports," as well. Figure 3-5 illustrates Class D airspace surrounding the Southwest Georgia Regional Airport in Albany, Georgia, depicted by a dashed 5-mi radius circle around the

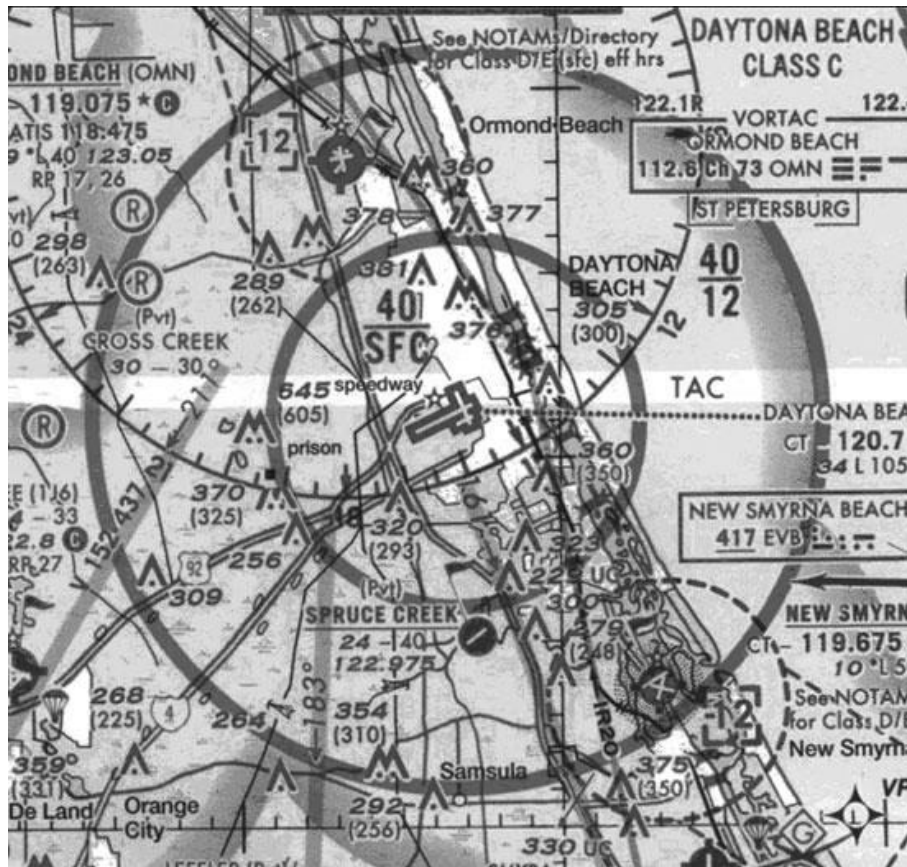


FIGURE 3-4 Class C airspace around Daytona Beach International Airport.

airport. The outer shaded ring depicts Class E airspace beginning at 700 ft above ground level.

Class E airspace is found in several locations with the purpose of providing areas of at least “passive” control for airplanes flying in areas of low altitude but moderate traffic activity, on defined air routes,



FIGURE 3-5 Class D airspace around Southwest Georgia Regional Airport, Albany, Georgia.

as well as those flying on instrument-based approach and departure procedures to or from an airport. Specifically, Class E airspace exists in most parts of the United States, from the surface, 700 ft AGL, or 1200 ft AGL to 14,500 ft AGL, and 3 nm surrounding the nation's airways.

Any airspace that does not fall within Class A, B, C, D, or E airspace is considered Class G, or uncontrolled airspace. This airspace is found only at very low altitudes (typically less than 700 or 1200 ft AGL) or in rural areas of low volume air traffic. Within Class G airspace, aircraft may move freely as long as there is sufficient visibility (1 mi during day hours, 3 mi during night hours, 5 mi day or night when above 10,000 ft AMSL) to see and avoid other air traffic.

Within the National Airspace System are a number of special use airspace classifications. Some of these define permanent location of special use or restricted activity, others define locations where flight operations are restricted for security or other reasons.

Prohibited areas are defined within the NAS as areas prohibited to any civil aviation activity. These areas are typically defined around highly sensitive locations, such as the White House in Washington, D.C.

Restricted areas are defined within the NAS as areas where regular, but not constant, sensitive operations occur, precluding the safe passage of civil aircraft. These areas, such as around the Kennedy Space Center on the east coast of Florida, will periodically restrict civilian access when sensitive activities are occurring.

Military operations areas (MOAs) are defined as areas with periodic military aviation or other activity. These areas may be entered only by permission from air traffic control, which coordinates with the military for civilian use.

Temporary flight restrictions (TFRs) are defined as areas that temporarily restrict or prohibit most civil aviation operations for reasons of national security. TFRs are implemented with little advance notice for a variety of reasons, ranging from protecting nuclear power facilities, to national sporting events, to the travels of the President of the United States. Oftentimes, the activation of a TFR will have serious impacts on the accessibility of an airport to the aviation system.

Figure 3-6 provides an illustration of multiple classes of airspace within the same region, including Class E airspace under Palatka-Larkin Airport and restricted areas within a military operations area south of the airport. Restricted use airspace presents challenges to airport planners seeking maximum efficiency of air traffic to and from the airfield.

Airways

Aircraft flying from one point to another have traditionally followed designated routes. In the United States these are referred to as victor airways and jet routes. These routes have evolved over time as discussed below.



FIGURE 3-6 MOAs and restricted areas south of Kay Larkin Airport, Palatka, Florida.

Colored Airways

The earliest airways, created in the 1920s were initially given a color designation on aeronautical charts and described by their color. The trunk lines east and west were green, trunk lines north and south were amber, secondary lines east and west were red, and secondaries north and south were blue. Each of these colored airways was then given a number, such as green 3, red 4, etc. The numbering for the airways began at the Canadian border and the Pacific Coast, then progressed to the south and east. These airways were then assigned an altitude level, which for green and red was at odd-thousand feet eastbound and at even-thousand feet westbound. On the amber and blue airways northbound, odd-thousand-foot levels were assigned, and southbound even-thousand-foot levels were assigned. These airways were delineated on the ground by low-frequency medium-frequency (LF/MF) four course radio ranges. The colored airways were phased out as aircraft became equipped to use the victor airways in the late 1940s.

Victor Airways

Following the development of the LF/MF four course radio ranges the routes now known as the victor airways were established. The victor airways are delineated on the ground by very high frequency omnirange radio equipment (VORs). Each VOR station has a discrete radio frequency to which a pilot could tune a navigational radio and thus be able to maintain a course from one VOR to the next. The numbering system for these airways is even numbers east and west, odd numbers north and south. The advantages of the victor airways were that the VORs were relatively free of static and it is much easier for a pilot to determine air position relative to a VOR station than with the LF/MF four course radio range. Victor airways are designated on aeronautical charts as V-1, V-2, etc. The airway includes the airspace within parallel lines 4 mi each side of the centerline of the airway. If two VORs delineating an airway are more than 120 mi apart, the airspace included in the airway is as indicated for jet routes.

Jet Routes

With the introduction of commercial jet aircraft in 1958, the altitudes at which these aircraft flew increased significantly. At higher altitudes the number of ground stations (VORs) required to delineate a specific route is smaller than at lower altitudes because the signal is transmitted on a line of sight. Therefore there was no need to clutter the high altitude routes with all the ground stations required for low altitude flying. All the routes in the continental United States could be placed on one chart. These were established what are known as jet routes. Although in one sense these routes are airways, they are not referred to as such. Today both victor airways and jet routes exist. Thus the jet routes are delineated by the same aids to navigation on the ground (VORs) as are victor airways but fewer stations are used. Victor airways extend from 1200 ft above the terrain to, but not including, 18,000 ft AMSL. Jet routes extend from 18,000 ft to 45,000 ft AMSL. Above 45,000 ft there are no designated routes and aircraft are handled on an individual basis. The numbering system for the jet routes is the same as for the victor airways. Jet routes are designated on aeronautical charts as J-1, J-2, etc.

Area Navigation

For many years all aircraft were required to fly on designated routes, airways, or jet routes. That is, all aircraft had to fly from one VOR to the next VOR since the VORs delineate the airways and jet routes. This required the funneling of all traffic on the designated routes that resulted in congestion on certain routes. Also the designated routes were often not the shortest distance between two points, resulting in additional fuel consumption, flight time, and cost. Furthermore, if the

designated route penetrates into an area of thunderstorms, aircraft have to be vectored around the storm by controllers on the ground. This imposed an extra workload on the controllers that is compensated for by the use of the severe weather avoidance program (SWAP). Despite these inefficiencies, the vast majority of transient aviation still fly along the victor airways and jet routes. This trend, however, has begun to change dramatically since the beginning of the twenty-first century, with the proliferation of GPS based navigation systems, under what is known as RNAV.

Area navigation, RNAV, is a method of aircraft navigation that permits aircraft operation on any desired course within the coverage of station-referenced navigational signals or within the limits of a self-contained system capability. Area navigation routes are direct routes, based upon the area navigation capability of aircraft, between waypoints defined in terms of latitude and longitude coordinates, degree and distance fixes, or offsets from established routes and airways.

RNAV is possible due to the proliferation of onboard aircraft technologies that take advantage of the global positioning system (GPS). GPS is based on 24 satellites located approximately 12,000 mi about the earth in a geostatic orbit. Technology that references an aircraft's position in relation to these satellites allows an aircraft to navigate by referencing its position to a detailed database that identifies airports, waypoints, terrain, and man-made infrastructure. Enhancements to the accuracy of the GPS system, with technologies such as the wide area augmentation system (WAAS) have made it possible for the air traffic control system to approve defined approaches to airport runways with far greater accuracy than with traditional radio-frequency-based systems.

Area navigation provides a more flexible routing capability that allows for better utilization of the airspace. The greater utilization reduces delays in the airspace and results in more economical operation of the aircraft. For example, routes parallel to the designated routes from one VOR to another can be established without requiring additional aids to navigation on the ground. Another example is the establishment of a more direct route from one point to another by establishing waypoints that provide for a shorter trip. Routing around a thunderstorm without continuous radar guidance from the ground is another example.

Area navigation is not limited to the horizontal plane but can also be utilized in the vertical plane, termed VNAV. It can also include a time reference capability. A properly equipped aircraft could arrive at a specified point in space, called a fix, with no need for ground vectoring or directions and could additionally be at that point at a specified altitude and time. This is a four-dimensional capability giving latitude, longitude, altitude, and time (4D RNAV). Thus area navigation

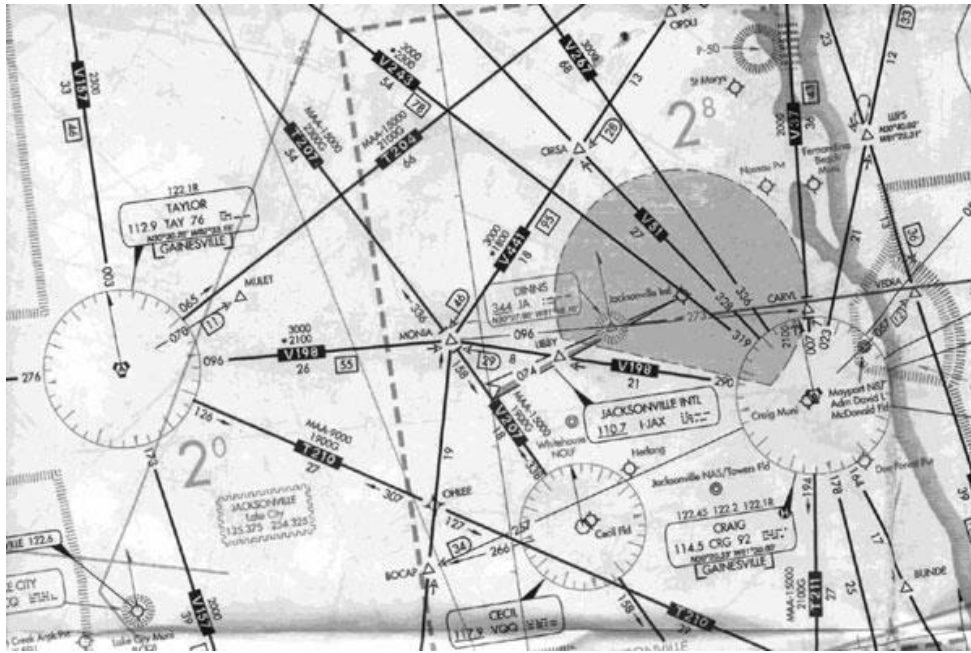


FIGURE 3-7 Airways, navigational aids, and airports as depicted on an IFR en route low altitude navigational chart.

has the potential of increasing airspace capacity, enhancing safety, and reducing the workload of the pilot and the air traffic controller.

As part of RNAV enhancements, the FAA began establishing T-routes as alternative routes to the victor airways for those aircraft equipped with GPS systems. These T-routes were established to provide aircraft with more direct routing, often around congested traffic areas such as Class B airspace and areas where victor airways intersect, often in the vicinity of a VOR station.

Figure 3-7 illustrates a portion of airspace as depicted in an IFR en route low altitude chart, published by the FAA's National Aeronautical Charting Office. This figure depicts both victor airways (identified by V followed by a route number) and T-routes (identified by a T followed by a route number), as well as the locations of airports, classes of airspace (such as the shaded area around Jacksonville International Airport), VOR stations (such as Taylor, Cecil, and Craig), and other navigational facilities. It is recommended that airport planners become familiar with understanding the information provided on this and other aeronautical charts.

Air Traffic Separation Rules

Air traffic rules governing the minimum separation of aircraft in the vertical, horizontal or longitudinal, and lateral directions are established in each country by the appropriate government authority. The current rules described in this text are those that are prescribed by the

FAA for use in the United States. The separation rules are prescribed for IFR operations and these rules apply whether or not IMC conditions prevail. Minimum separations are a function of aircraft type, aircraft speed, availability of radar facilities, navigational aids, and other factors such as the severity of wake vortices [3].

Vertical Separation in the Airspace

The minimum vertical separation of aircraft outside of the terminal area from the ground up to and including 41,000 ft AMSL is 1000 ft. In 2005, vertical separation minimums above 29,000 ft AMSL were reduced from 2000 to 1000 ft under the reduced vertical separation minima (RVSM) program. Implementation of this program allowed for additional jet routes thereby increasing the capacity within the NAS. Within a terminal area a vertical separation of 500 ft is maintained between aircraft, except that a 1000-ft vertical separation is maintained below a heavy aircraft.

Assigned Flight Altitudes

To formalize the separation of air traffic in the airspace, air traffic control assigns flight altitudes to aircraft based on their direction, or more precisely magnetic heading, of flight, and whether or not they are flying under VFR versus IFR rules.

Aircraft flying under IFR are typically assigned altitudes of odd-thousand feet (i.e., 3000 ft, 5000 ft, etc.) AMSL while on an easterly heading (magnetic compass heading of 0° to 179°) and even-thousand feet (i.e., 4000 ft, 6000 ft, etc.) while on a westerly heading (magnetic compass heading of 180° to 359°). Between 29,000 ft AMSL (FL 290) and 41,000 ft AMSL (FL 410), aircraft are assigned a flight level of either FL 290, FL 330, FL 370, or FL 410 when traveling on an easterly heading, and either FL 310, FL 350, FL 390 when traveling on a westerly heading. If an aircraft is RVSM certified, it may be assigned an RVSM altitude of FL 300, 320, 340, etc. between FL 290 and FL 410.

Aircraft flying under VFR above 3000 ft AMSL are typically assigned altitudes of odd-thousand feet plus 500 ft (i.e., 3500 ft, 5500 ft, etc.) while on an easterly heading, and even-thousand feet plus 500 feet (i.e., 4500 ft, 6500 ft, etc.) while on a westerly heading (magnetic compass heading of 180° to 359°). Above 29,000 ft (FL 290), VFR traffic is assigned every-other even or odd thousand (FL 290, FL 330, etc. if traveling on an easterly heading, and FL 320, FL 360, etc. if traveling on a westerly heading). It should be noted that above FL 180, all traffic is required to be on an IFR flight plan.

Longitudinal Separation in the Airspace

The minimum longitudinal separation depends on a number of factors. Among the most important are aircraft size, aircraft speed, and the availability of radar for the control of air traffic. For the purposes

of maintaining aircraft separations aircraft are classified by the FAA as heavy, large, or small based upon their maximum gross takeoff weight (MGTOW). Heavy aircraft are classified as those aircraft which have a MGTOW of 300,000 lb or more. Large aircraft are as those aircraft which have a MGTOW of in excess of 12,500 lb but less than 300,000 lb. Small aircraft are as those aircraft which have a MGTOW of 12,500 lb or less. Aircraft size is related to wake turbulence. Heavy aircraft create trailing wake vortices which are a hazard to lighter aircraft following them.

The minimum longitudinal separations en route are expressed in terms of time or distance as follows:

1. For en route aircraft following a preceding en route aircraft, if the lead aircraft maintains a speed at least 44 kn faster than the trail aircraft, 5 mi between aircraft using distance measuring equipment (DME) or area navigation (RNAV) and 3 minutes between all other aircraft
2. For en route aircraft following a preceding en route aircraft, if the lead aircraft maintains a speed at least 22 kn faster than the trail aircraft, 10 mi between aircraft using DME or RNAV and 5 min for all other aircraft
3. For en route aircraft following a preceding en route aircraft, if both aircraft are at the same speed, 20 mi between aircraft using DME or RNAV and 10 min for all other aircraft
4. When an aircraft is climbing or descending through the altitude of another aircraft, 10 mi for aircraft using DME or RNAV if the descending aircraft is leading or the climbing aircraft is following and 5 min for all other aircraft
5. Between aircraft in which one aircraft is using DME or RNAV and the other is not, 30 mi

Minimum longitudinal separations over the oceans is normally 10 minutes for supersonic flights and 15 minutes for subsonic flights but in some locations it can be slightly more or less than this value [3].

When the aircraft mix is such that wake turbulence is not a factor and radar coverage is available, the minimum longitudinal separation for two aircraft traveling in the same direction and at the same altitude is 5 nm, except that when the aircraft are in the terminal environment within 40 nm of the radar antenna the separation can be reduced to 3 nm. For this reason the minimum spacing in the terminal area is 3 nm because the airport is almost always within 40 nm of a radar antenna. Under certain specified conditions a separation between aircraft on final approach within 10 nm of the landing runway may be reduced to 2.5 nm [3].

If wake turbulence is a factor, the minimum separation in the terminal area between a small or large aircraft and a preceding heavy

Lead Aircraft Type	VFR* Trall Aircraft Type			IFR (Wake Vortex) Trall Aircraft Type		
	Heavy	Large	Small	Heavy	Large	Small
Heavy	2.7	3.6	4.5	4.0	5.0	6.0
Large	1.9	1.9	2.7	3.0	3.0	4.0
Small	1.9	1.9	1.9	3.0	3.0	3.0

*These are shown to appropriately represent these operations and are not regulatory in nature.

Source: Federal Aviation Administration [18].

TABLE 3-1 Horizontal Separation in Landing for Arrival-Arrival Spacing of Aircraft on Same Runway Approaches in VFR and IFR Conditions (nautical miles)

aircraft is 5 nm. The spacing between two heavy aircraft following each other is 4 nm. The spacing between a heavy aircraft and a preceding large aircraft is 3 nm.

For landing aircraft when wake turbulence is a factor, the longitudinal separation is increased between a small aircraft and a preceding large aircraft to 4 mi and between a small aircraft and a preceding heavy aircraft to 6 mi.

The IFR separation rules for consecutive arrivals on the same runway which are used when wake vortices are a factor are shown in Table 3-1. The VFR and IFR separation rules for consecutive departures from the same runway are expressed in terms of time and these are shown in Table 3-2.

Lateral Separation in the Airspace

The minimum en route lateral separation below 18,000 ft MSL is 8 nm, and at and above 18,000 ft MSL the minimum en route lateral separation

Lead Aircraft Type	VFR* Trall Aircraft Type			IFR Trall Aircraft Type		
	Heavy	Large	Small	Heavy	Large	Small
Heavy	90	120	120	120	120	120
Large	60	60	50	60	60	60
Small	50	45	35	60	60	60

*These are shown to appropriately represent these operations and are not regulatory in nature.

Source: Federal Aviation Administration [18].

TABLE 3-2 Separation for Same Runway Consecutive Departures in VFR and IFR Conditions (seconds)

is 20 nm. Over the oceans the separation varies from 60 to 120 nm depending on location [3].

Navigational Aids

Aids to navigation, known as NAVAIDS, can be broadly classified into two groups, ground-based systems and satellite-based systems. Each system is complimented by systems installed in the cockpit.

Ground-Based Systems

Nondirectional Beacon

The oldest active ground-based navigational aid is the nondirectional beacon (NDB). The NDB emits radio frequency signals on frequencies between 400 and 1020 Hz modulation. NDBs are typically mounted on a pole approximately 35 ft tall. They may be located on or off airport property, at least 100 ft clear of metal buildings, power lines, or metal fences. While the NDB is quickly being phased out in the United States, it is still a very common piece of navigational equipment in other parts of the world, particularly in developing nations. Figure 3-8 provides an illustration of an NDB.

Aircraft navigate using the NDB by referencing an automatic direction finder (ADF) located on the aircraft's panel. The ADF simply points toward the location of the NDB. Figure 3-9 illustrates an ADF system.

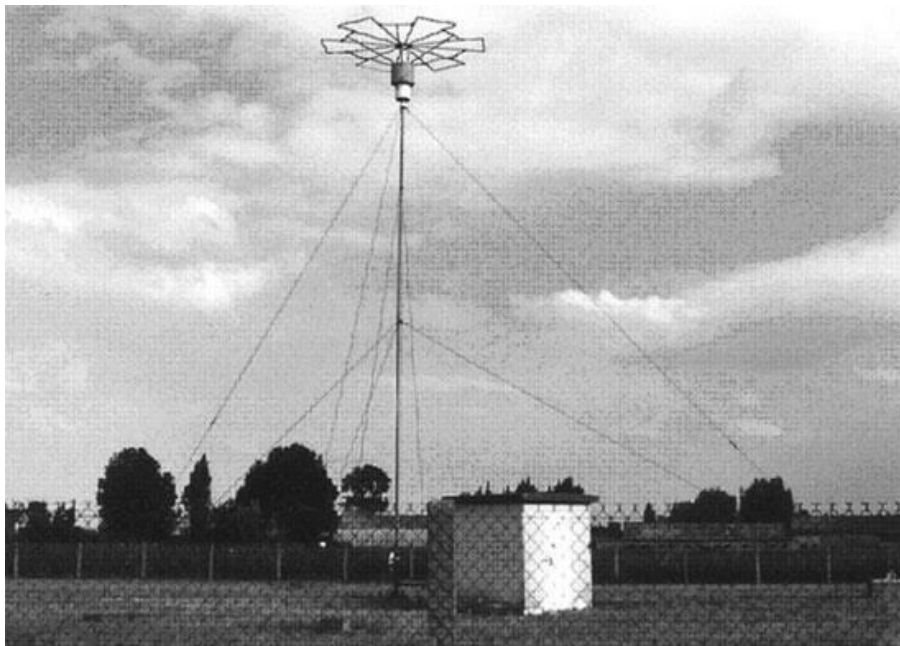


FIGURE 3-8 Nondirectional beacon.



FIGURE 3-9 Automatic direction finder.

Very High Frequency Omnidirectional Radio

The advances in radio and electronics during and after World War II led to the installation of the very high frequency omnirange (VOR) radio stations. These stations are located on the ground and send out radio signals in all directions. Each signal can be considered as a course or a route, referred to as a radial that can be followed by an aircraft. In terms of 1° intervals, there are 360 courses or routes that are radiated from a VOR station, from 0° pointing toward magnetic north increasing to 359° in a clockwise direction. The VOR transmitter station is a small square building topped with what appears to be a white derby hat. It broadcasts on a frequency just above that of FM radio stations. The very high frequencies it uses are virtually free of static. The system of VOR stations establish the network of airways and jet routes and are also essential to area navigation. The range of a VOR station varies but is usually less than 200 nm. A typical VOR beacon is illustrated in Fig. 3-10.

Aircraft equipped with a VOR receiver in the cockpit have a dial for tuning in the desired VOR frequency. A pilot can select the VOR radial or route he wishes to follow to the VOR station. In the cockpit there is also an omnibearing selector (OBS) which indicates the heading of the aircraft relative to the direction of the desired radial and whether the aircraft is to the right or left of the radial. An illustration of an OBS is provided in Fig. 3-11.

Distance Measuring Equipment

Distance measuring equipment (DME) has traditionally been installed at VOR stations in the United States. The DME shows the pilot the



FIGURE 3-10 VOR beacon on the airfield at Ronald Reagan Washington National Airport.

slant distance between the aircraft and a particular VOR station. Since it is the air distance in nautical miles that is measured, the receiving equipment in an aircraft flying at 35,000 ft directly over the DME station would read 5.8 nm.

An en route air navigation aid which best suited the tactical needs of the military was developed by the Navy in the early 1950s. This aid is known as TACAN, which stands for tactical air navigation. This aid combines azimuth and distance measuring into one unit instead of two and is operated in the ultra-high-frequency band. As a compromise between civilian and military requirements, the FAA replaced the DME

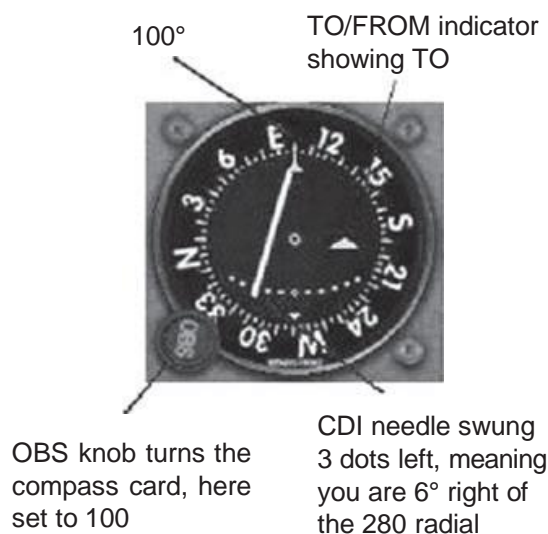


FIGURE 3-11 Omnibearing selector.

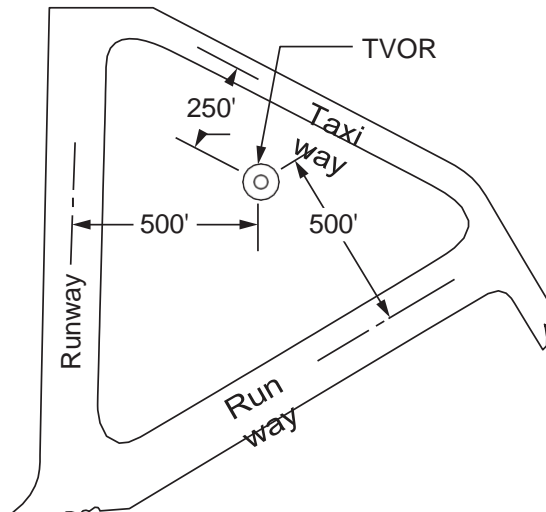


FIGURE 3-12 TVOR location standards.

portion of its VOR facilities with the distance measuring components of TACAN. These stations are known as VORTAC stations. If a station has full TACAN equipment, both azimuth and distance measuring equipment, and also VOR, it is designated as VORTAC.

NDB and VOR systems are often located on airport airfields. The location of these systems on airport, known as TVORs, are significant to airport planners and designers, as the location of other facilities, such as large buildings, particularly constructed of metal, may adversely affect the performance of the navaid.

As illustrated in Fig. 3-12, TVORs should be located at least 500 ft from any runways and 250 ft from any taxiways. Any structures or trees should be located at least 1000 ft from the TVOR antenna. There should also be a clearance angle of at least 2.5° for any structures and 2.0° for any trees beyond 1000 ft, as illustrated in Fig. 3-13.

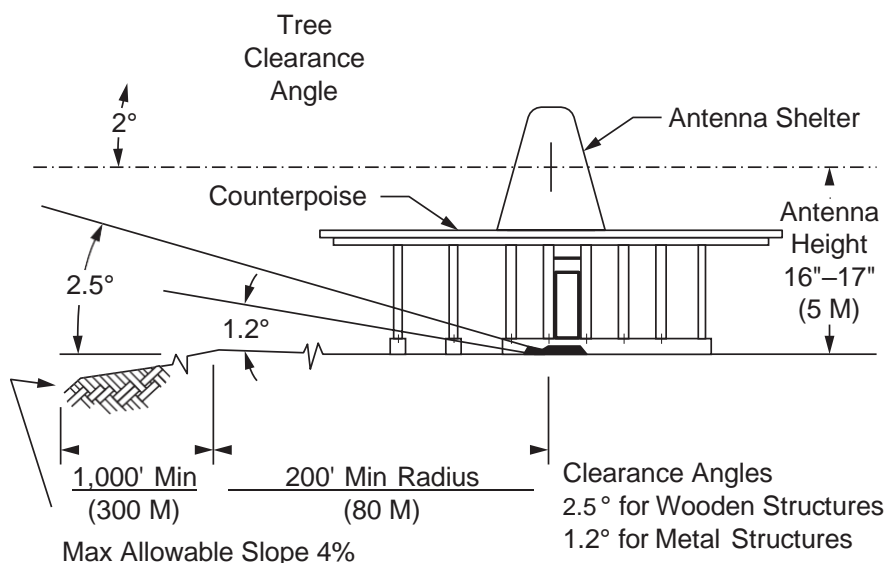


FIGURE 3-13 TVOR clearance requirements.

Air Route Surveillance Radar

A long-range radar for tracking en route aircraft has been established throughout the continental United States and in other parts of the world. While in the United States there is complete radar coverage in the 48 contiguous states, this is not the case elsewhere in the world. These radars have a range of about 250 nm. Strictly speaking radar is not an aid to navigation. Its principal function is to provide air traffic controllers with a visual display of the position of each aircraft so they can monitor their spacings and intervene when necessary. However, it can be and is used by air traffic controllers to guide aircraft whenever this is necessary. For this reason it has been included as an aid to navigation.

The VOR and NDB, often combined with radar-based surveillance from air traffic control, have traditionally been used in both en route navigation and for navigation on approach to landing at an airport. Navigation on approach to an airport using these ground-based systems is performed by following predetermined, published, approach procedures. These procedures are often updated and published by the FAA in the form of approach charts. Figure 3-14 provides an example of an approach chart depicting an approach procedure using an NDB as an aid to navigation, while Fig. 3-15 illustrates a similar approach using a VOR as the primary aid to navigation. It is strongly recommended that the airport planner understand the information provided in these charts. Approaches based on NDB and VOR nav aids are considered “nonprecision” approaches, as they provide lateral navigation assistance but not vertical navigation. That is, these instruments may be referenced to determine which direction to fly when approaching an airport, but do not provide instrument-based guidance in determining the appropriate altitude or descent rate on approach.

Instrument Landing System

Until the recent proliferation of published navigation procedures which rely on the satellite based GPS system, the instrument landing system (ILS) was the only ground-based system certified to provide both lateral and vertical guidance to aircraft on approach to an airport, and as of 2008 is the only navigational aid certified by the FAA to provide “precision” navigation for aircraft, and is still the most widely used method of approach navigation at the world’s larger airports.

An ILS system consists of two radio transmitters located on the airport. One radio beam is called the localizer and the other the glide slope. The localizer indicates to pilots whether they are left or right of the correct alignment for approach to the runway. The glide slope indicates the correct angle of descent to the runway. Glide slopes are in the order of from 2°–3° to 7.5°.

In order to further help pilots on their ILS approach, up to three low-power fan markers called ILS markers are usually installed so

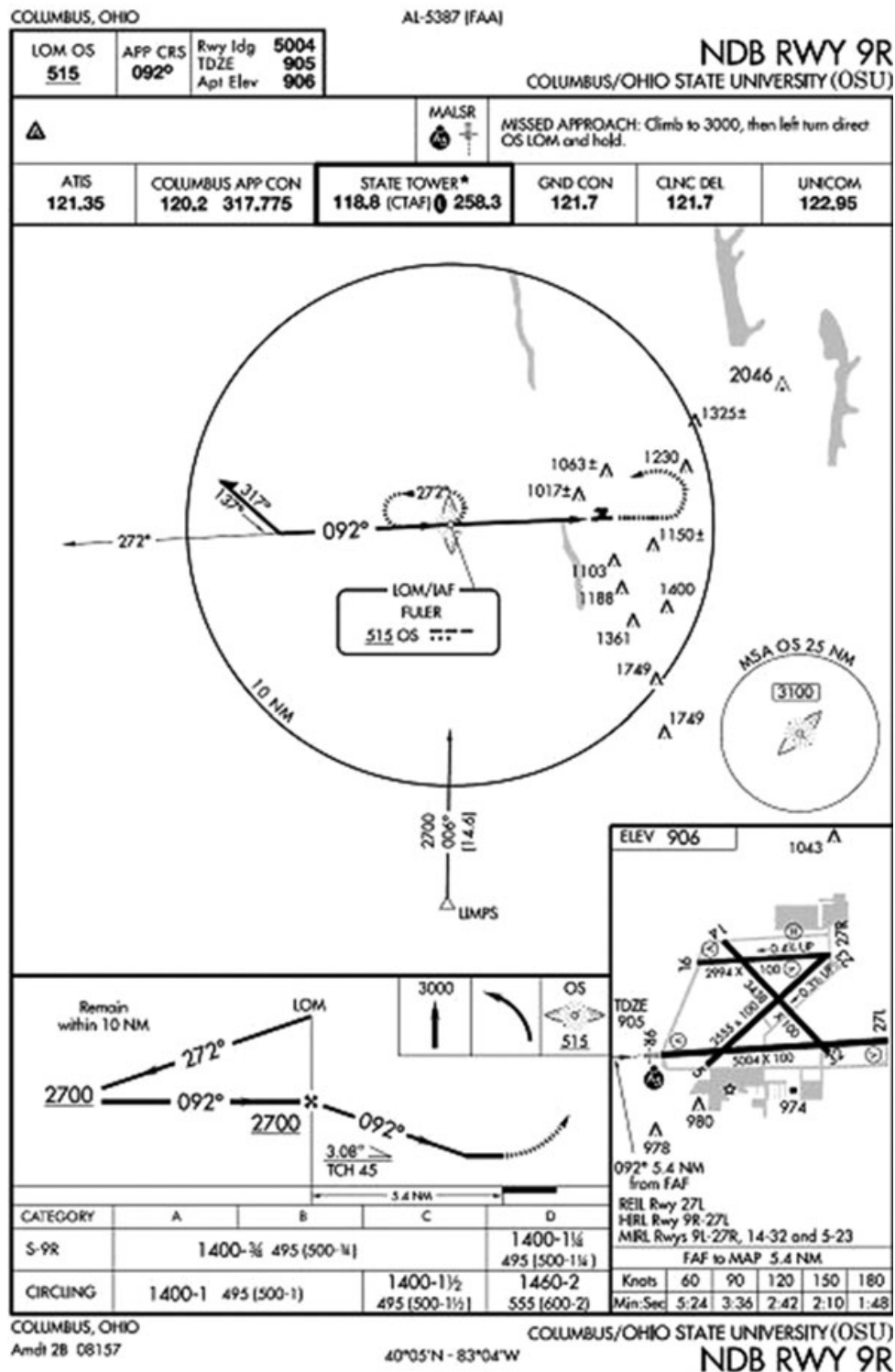
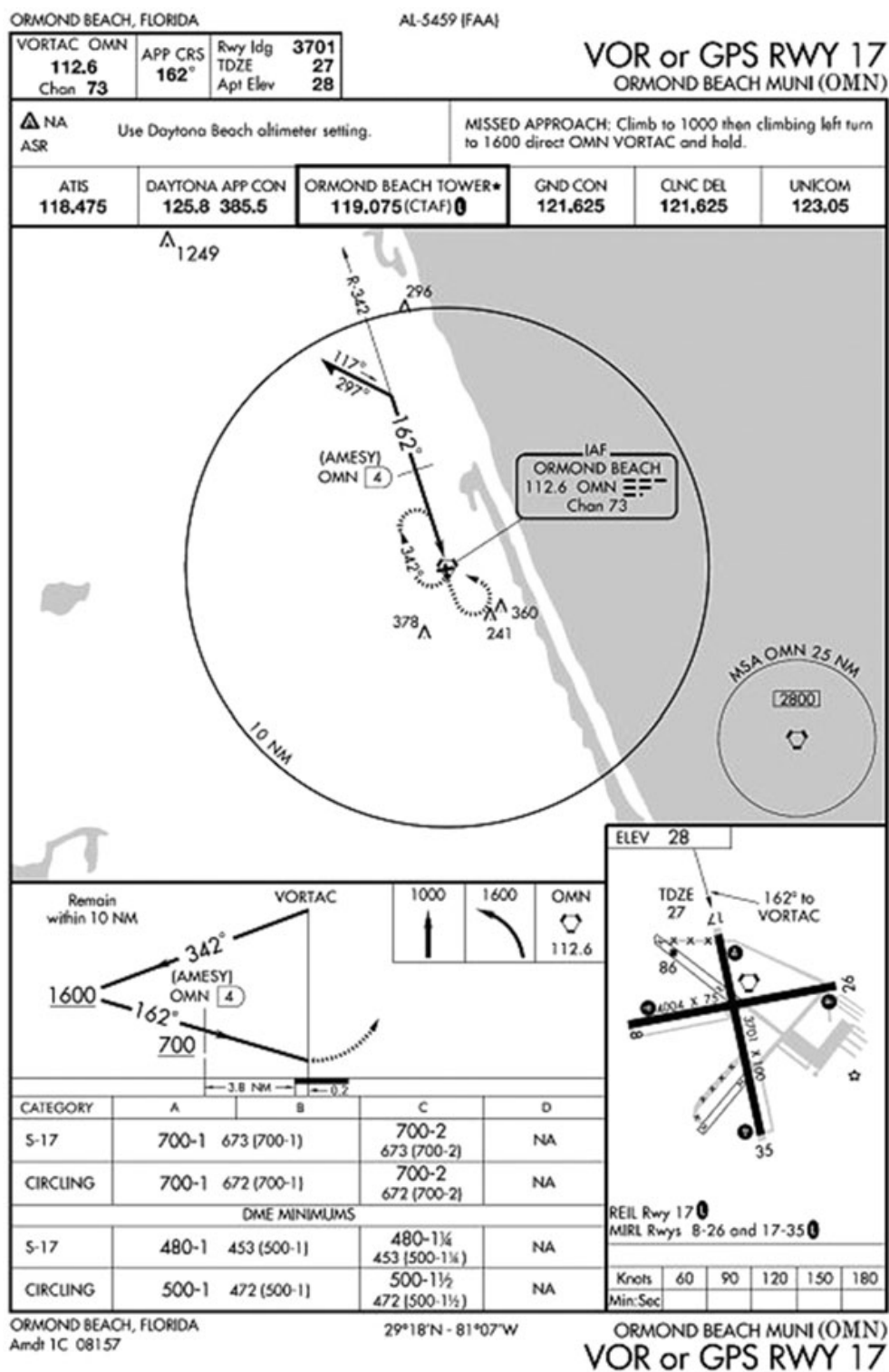


FIGURE 3-14 NDB approach.

that they may know just how far along the approach to the runway they have progressed. The first is called the outer marker (LOM) and is located about 3.5 to 5 mi from the end of the runway. The middle marker (MM) is located about 3000 ft from the end of the runway. On

**FIGURE 3-15** VOR approach.

some ILS systems, an additional marker called the inner marker (IM) is located 1000 ft from the end of the runway. When the plane passes over a marker, a light goes on in the cockpit and a tone sounds. The configuration of the ILS system is shown in Fig. 3-16.

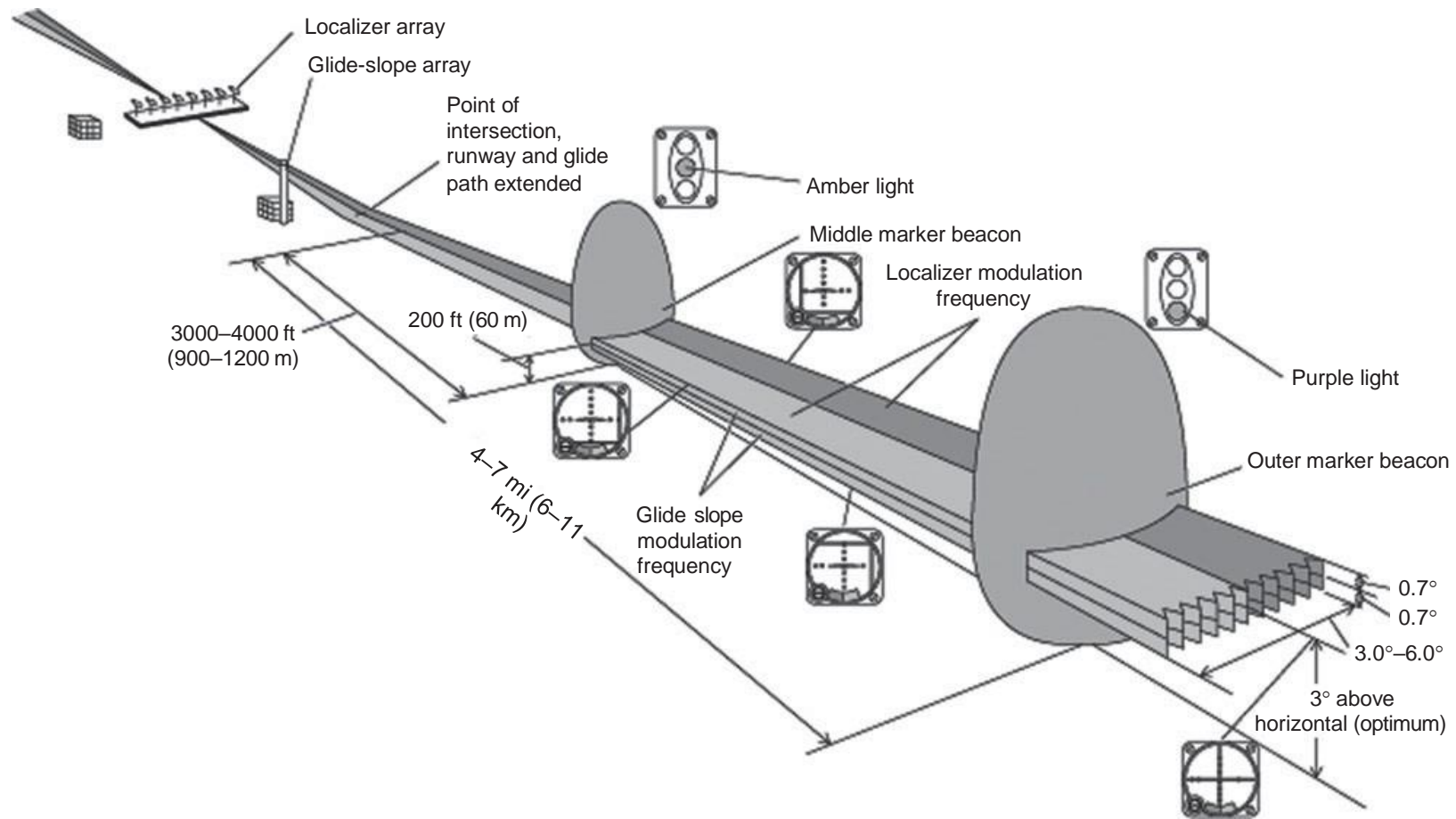


FIGURE 3-16 ILS system configuration.

The localizer consists of an antenna, which is located on the extension of the runway centerline approximately 1000 ft from the far end of the runway, and a localizer transmitter building located about 300 ft to one side of the runway at the same distance from the end of the runway as the antenna. The glide slope facility is placed 750 to 1250 ft down the runway from the threshold and is located to one side of the runway centerline at a distance which can vary from 400 to 650 ft. The functioning of the localizer and the glide slope facility is affected by the close proximity of moving objects such as vehicular and aircraft traffic. During inclement weather the use of the ILS critical areas inhibit aircraft and vehicles from entering into areas that would impede an aircraft inside of the outer marker from receiving a clear signal. Stationary objects nearby can also cause a deterioration of signals. Abrupt changes of slope in proximity of the antennas are not permitted or the signal will not be transmitted properly. Another limitation of the ILS is that the glide slope beam is not reliable below a height of about 200 ft above the runway.

As with VOR and NDB systems, the localizer and glide slope components of the ILS are highly sensitive to their proximity to surrounding objects that may interfere with their radio signals. As such, there are specific restrictions to construction in the immediate vicinity of these systems.

ILS systems may also be accompanied by runway visual range (RVR) equipment, which provide a measurement of lateral visibility to a pilot. RVR systems determine the distance a pilot should be able to see down the runway, given current atmospheric conditions and existing lighting systems. Depending on the type of RVR system installed, pilots can safely approach to land on a runway using ILS navigation in varying levels of cloud ceiling levels and horizontal visibility. Table 3-3 provides the ceiling and visibility levels for ILS systems equipped with RVR. Figure 3-17 illustrates a published approach using an ILS.

The most critical point of approach to landing comes when the aircraft breaks through the overcast and the pilot must change from instrument to visual conditions. Sometimes, only a few seconds are

ILS Category	Cloud Ceiling	Visibility (RVR)
I	200 ft	1,800–2,400 ft
II	100 ft	1,200 ft
IIIa (auto land)	0–100 ft	700 ft
IIIb (auto rollout)	0–50 ft	150 ft
IIIc (auto taxi)	0 ft	0 ft

TABLE 3-3 ILS Capabilities

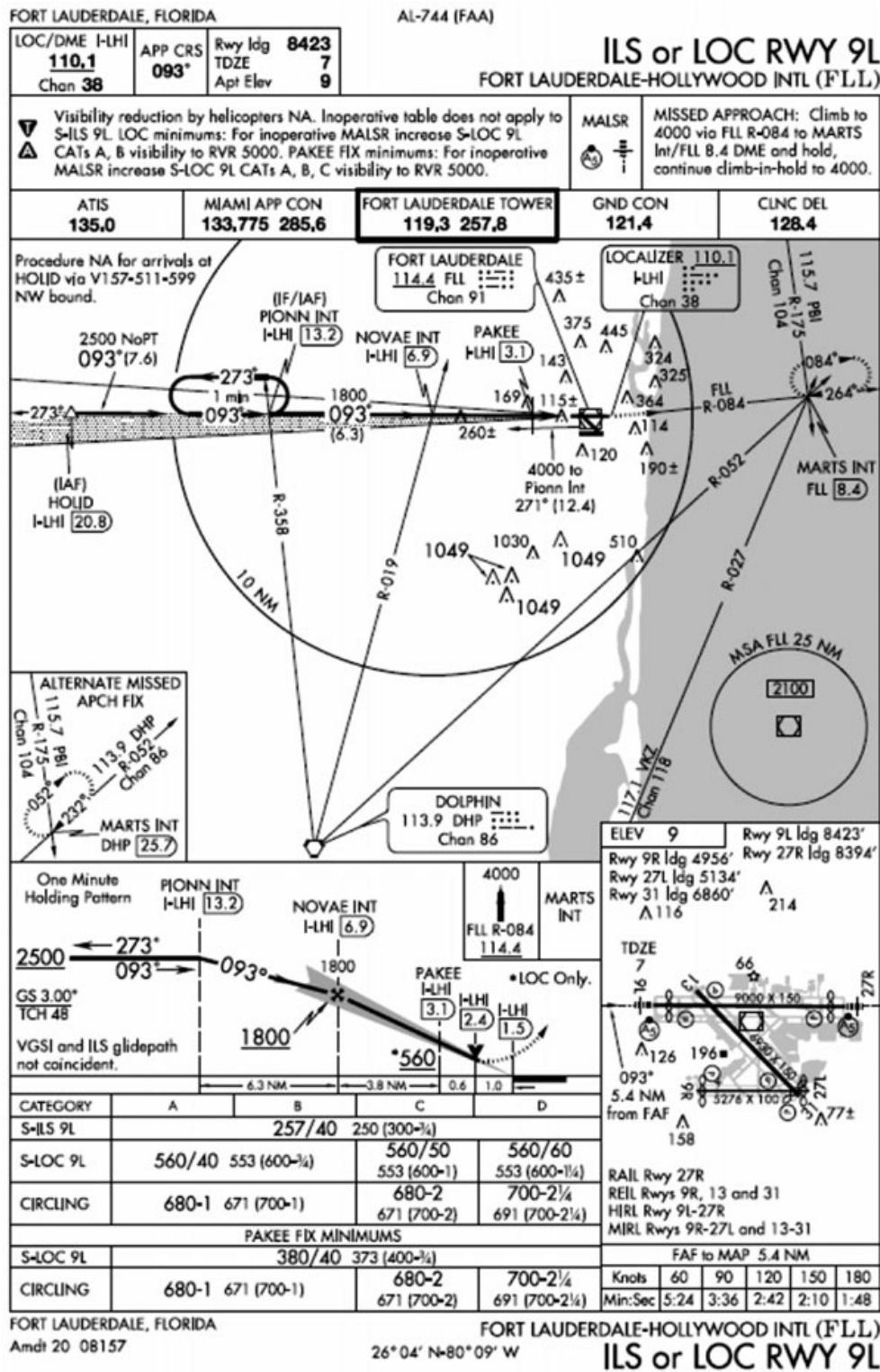


FIGURE 3-17 Published ILS approach procedure.

available for the pilot to make the transition and complete the landing. To aid in making this transition, lights are installed on the approach to and on the runways. These are generally termed approach lighting systems (ALS). More details concerning these systems are contained in the chapter on signing, marking, and lighting.

Airport Surface Detection Equipment

In use in many of the busier airports within the United States and elsewhere, specially designed radar called airport surface detection equipment (ASDE), often referred to as *ground radar*, has been developed to aid the controller in regulating traffic on the airport. The system gives the air traffic controller in the control tower a pictorial display of the runways, taxiways, and terminal area with radar indicating the position of aircraft and other vehicles moving on the surface of the airport.

ASDE technologies are most commonly available in two forms. The airport movement area safety system (AMASS) integrates third generation (ASDE-3) ground-based radar systems with audio and visual warning systems to prevent runway incursions on the airfield. ASDE-X, a system that integrates ASDE technology with transponder systems to identify the aircraft operating on the aircraft, has been employed at less busy commercial service airports throughout the United States. These systems are monitored by air traffic management personnel in the airport's ATCT.

Satellite-Based Systems: Global Positioning System

Perhaps the greatest impact on air traffic management since the beginning of the twenty-first century has been the development, acceptance, and proliferation of navigational procedures which rely on the global positioning system.

The global positioning system (GPS) is a satellite-based radio positioning and navigation system. The system is designed to provide highly accurate position and velocity information on a continuous global basis to an unlimited number of properly equipped users. The system is unaffected by weather and provides a common worldwide grid reference system. The GPS concept is predicated upon accurate and continuous knowledge of the spatial position of each satellite in the system with respect to time and distance from a transmitting satellite to the user. The GPS system consists of 24 satellites in near-circular orbit about the earth. GPS receivers onboard aircraft automatically select the appropriate signals from typically four satellites which are in view of the receiver and translate these signals into a three-dimensional position, and when the receiver is in motion, velocity.

GPS was developed by the United States military to aid in reconnaissance and strategic operations. Under military operations, signals transmitted from the orbiting GPS satellites were encrypted to reduce the accuracy of positioning for nonauthorized users. This policy became known as "selective availability." However, on May 1, 2000, President Bill Clinton ordered the removal of selective availability, allowing the civilian world to take advantage of GPS positioning accuracy on the order of 1 to 3 m. This level of accuracy has been

deemed sufficient to allow aircraft to navigate using properly equipped GPS receivers for both en route navigation and approaches to airports, in both VFR and IFR conditions.

In the first decade of the twenty-first century, the proliferation of GPS-based air navigation systems has been dramatic, to the point where the use of traditional ground-based navigation aids such as the NDB and VOR is becoming obsolete. GPS navigation systems have become available as both in-panel fixed navigation systems, and portable units, and have become widely used in many areas of society outside of aviation.

As Fig. 3-18 illustrates, GPS systems, particularly enhanced with comprehensive databases of area terrain, landmarks, and airport infrastructure, provide pilots with “virtually visual conditions” and the ability to navigate from origin to destination without any reliance on traditional ground-based analog navigational aids.

GPS-based RNAV approaches have been refined with improving technology, known as the wide area augmentation system (WAAS), and training to allow aircraft to approach airports using very precise navigation procedures. These approaches, known as RNP (required navigation performance), have allowed aircraft to navigate around such obstacles as mountainous terrain and security sensitive areas, resulting in a more efficient use of airport runways. Juneau, Alaska and Washington, D.C. are airports that have benefited from these enhancements in air traffic control. An example RNP approach is illustrated in Fig. 3-19.



FIGURE 3-18 Aircraft GPS-based navigation equipment (*Cirrus Aircraft Inc.*).

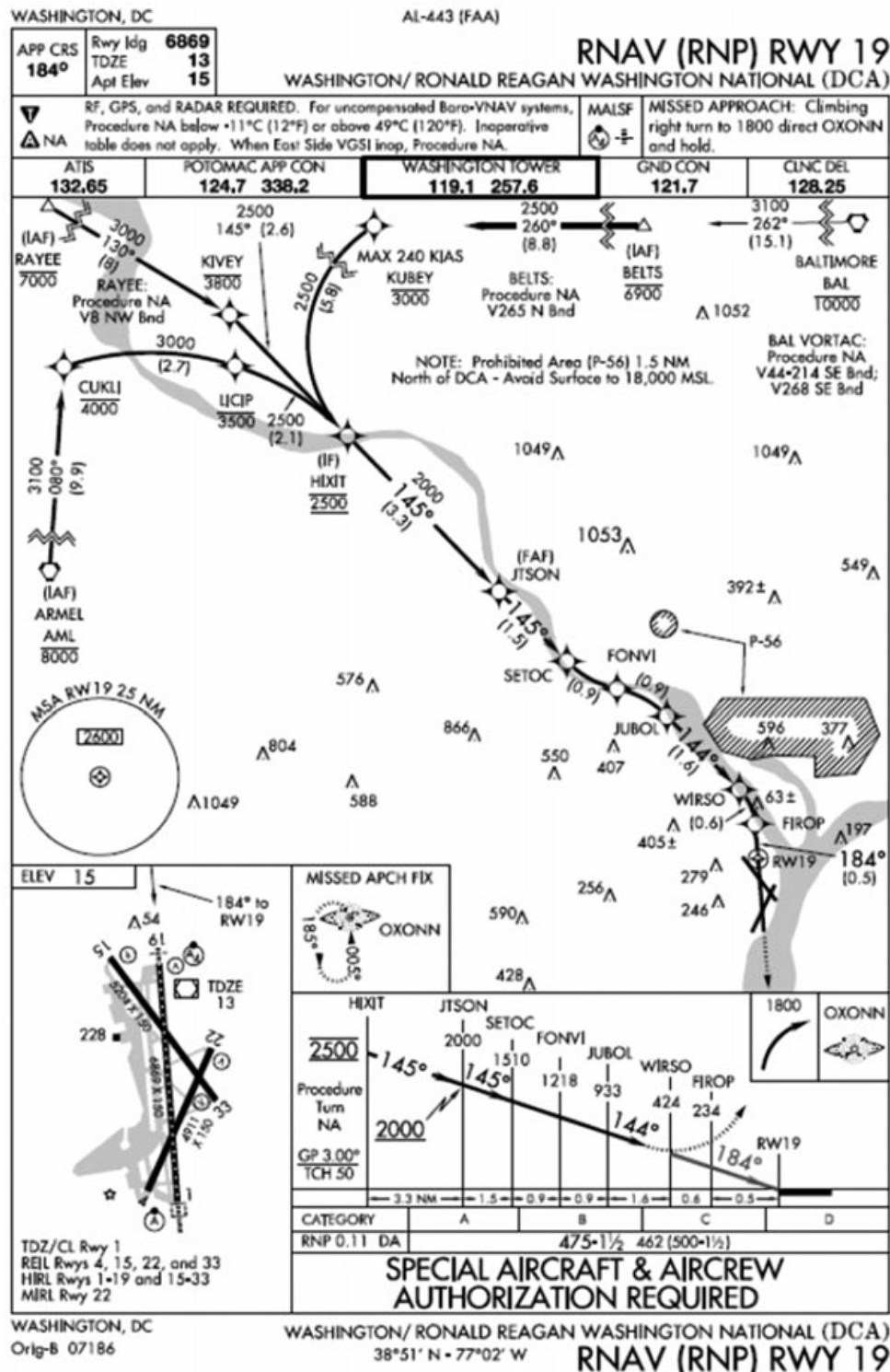


FIGURE 3-19 RNP approach into Ronald Reagan Washington National Airport, Washington, D.C.

ADS-B

Further enhancements to the air traffic management system include the use of advanced digital data-link systems, known as automated dependent surveillance. ADS-address (ADS-A) systems send digitally transmitted information between specific aircraft and ADS-broadcast

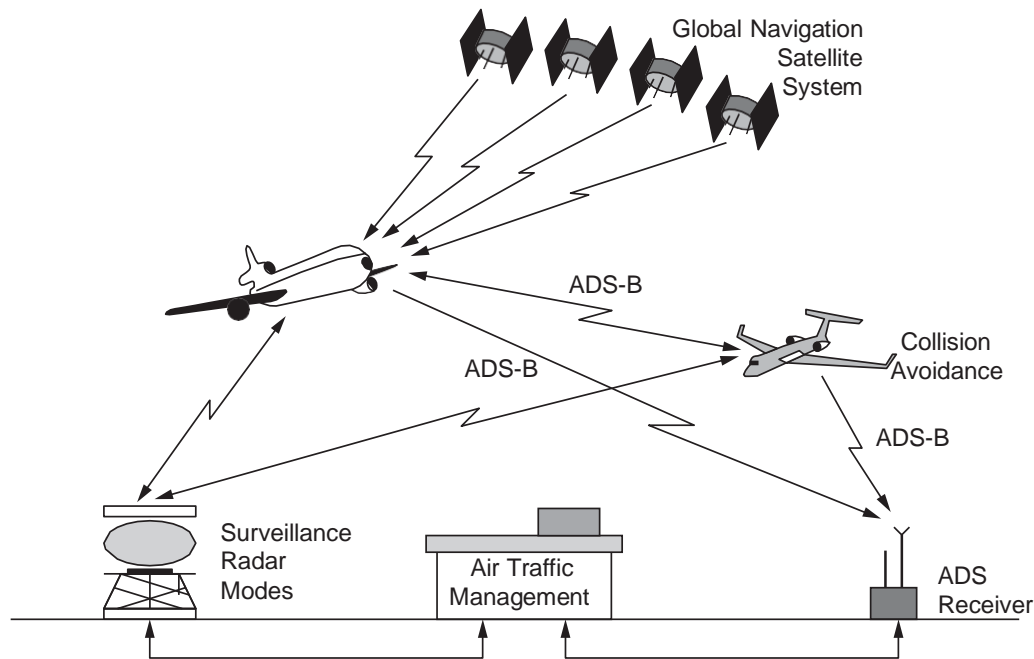


FIGURE 3-20 Rendering of ADS-B system (*Atcmonitor.com*).

(ADS-B) systems broadcast information to all equipped aircraft and air traffic management facilities, identifying their locations to other traffic in the system, providing the added ability to safely avoid collisions even in poor visibility conditions. Originally tested in Alaska between 2000 and 2003, ADS-B is quickly becoming a standard component of air traffic navigation systems in the United States. A rendering of the ADS-B system is Fig. 3-20.

The Modernization of Air Traffic Management

Despite the proliferation of GPS-based navigation since the beginning of the twenty-first century, the principal aids for the control of air traffic by air traffic management personnel are still voice communication and radar. Air traffic controllers monitor the spacing between aircraft on the radarscope and instruct pilots by means of voice communication.

Radar returns appear on the radarscope as small blips. These are reflections from the aircraft body. Primary radar requires the installation of rotating antennas on the ground and the range of the primary radar is a function of its frequency. Secondary radar consists of a radar receiver and transmitter on the ground that transmits a coded signal to an aircraft if that aircraft has a transponder. A transponder is an airborne receiver and transmitter which receives the signal from the ground and responds by returning a coded reply to the interrogator on the ground. The coded reply normally contains information on aircraft identity and altitude.

Information from primary and secondary radar returns are provided to air traffic controllers via an alphanumeric display on their radar “scopes,” as illustrated in Fig. 3-21. The first line shows the



identity of the aircraft, the second line its altitude and ground speed, and the third line gives the beacon code transponder number and the aircraft track number. To be able to have this information presented on the radarscope, the aircraft must carry a mode-C or mode-S transponder that has the capability of altitude reporting along with aircraft identity. All commercial airline aircraft carry a transponder, which satisfies the requirement for reporting altitude. Further, all aircraft flying in Class A, B, or C airspace are required to have an operating transponder onboard.

NextGen

For more than 50 years air traffic control systems have gone through a number of incremental technological enhancements, such as enhanced radar capabilities, automated flight service systems, and ground-based navigation systems. Despite these upgrades, it has been widely recognized that the traditional radar and analog-based communication system will not be sufficient to accommodate the increasing demands on the system in the twenty-first century.

As part of the Vision 100 Century of Aviation Reauthorization Act of 2003, the U.S. federal government called for a complete transformation of the national airspace system and a modernization of its air traffic control facilities. This modernization has come to be known as the “next generation air traffic system” or *NextGen*. Through the act, Congress directed the formation of a “Joint Planning and Development Office” (JPDO) to facilitate the mammoth task of converting the current system to a fully automated, digital, satellite-based air traffic management system. The JPDO comprises representatives from the FAA, NASA, The U.S. Departments of Transportation, Defense, Homeland Security, Commerce, and the White House Office of Science and Technology Policy, and directed to develop a next generation air traffic system that is technologically advanced and fully integrates the interests of all who use the nation’s aviation system.

NextGen will focus on making the satellite-based GPS system and digital data communications the backbone of air traffic management. Integrated into NextGen are GPS, WAAS, and ADS-B technology to allow for digital surveillance of air traffic between both ground-based air traffic management facilities as well as among aircraft themselves. In addition to ADS-B technology, NextGen features the following capabilities, as described by the FAA:

SWIM

System wide information management (SWIM) provides the infrastructure and services to deliver network-enabled information access across the NextGen air transportation operations. As an early opportunity investment, SWIM will provide high-quality, timely data to many users and applications—extending beyond the previous focus

on unique, point-to-point interfaces for application-to-application data exchange. By reducing the number and types of interfaces and systems, SWIM will reduce redundancy of information and better facilitate multiagency information-sharing. SWIM will also enable new modes of decision making, as information is more easily accessed by all stakeholders affected by operational decisions.

NextGen Data Communications

NextGen data communications will provide for two-way digital communications between air traffic controllers and pilots for air traffic control clearances, instructions, and other advisories. In addition, digital communications will provide broadcast text-based and graphical advisory information such as weather reports and notices to airmen without relying on voice communications.

NextGen Enabled Weather

The NextGen network enabled weather (NNEW) will serve as the core of the NextGen weather support services and provide a common weather picture across the national airspace system. These services will, in turn, be integrated into other key components of NextGen required to enable better air transportation decision making. It is anticipated that tens of thousands of global weather observations and sensor reports from ground-, airborne-, and space-based sources would fuse into a single national weather information system, updated as needed in real time.

NextGen is due to be a phased transformation of the NAS through 2025 at an estimated cost of \$20 to \$25 billion. It should be noted the early stages of NextGen development have been very volatile with regard to the selection of technology platforms on which to base the future air traffic management system, and it should be expected that further developments in technology will result in variations to current system plans. As such, it is imperative that the airport planner keep up with current progress. The JPDO and the FAA frequently update their Internet sites with NextGen system progress.

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CHAPTER 15

Heliports

20

Introduction

A rotorcraft is a rotary winged aircraft that can lift vertically and sustain forward flight by power-driven rotor blades turning on a vertical axis. The helicopter is a vehicle which essentially can take off from and land in a nearly vertical direction. This is known as vertical takeoff and landing (VTOL). Since the helicopter is by far the most advanced and utilized of the vertical takeoff aircraft, the emphasis on this chapter is on ground facilities for helicopters and other rotary wing type aircraft, referred to as heliports.

Heliports

A heliport is defined as an identifiable area on land, water, or structures, including buildings or facilities, used or intended to be used for the landing and takeoff of helicopters or other rotary wing type aircraft [10]. A helideck is a heliport located on a floating or an off-shore structure. A heli-stop is defined as an area developed and used for helicopter landings and takeoffs to drop-off or pickup passengers or cargo. A helipad is defined as a paved or other surface used for parking helicopters at a heliport.

The Nature of Helicopter Transportation

Helicopters are used for a variety of aviation activities including aerial observation, sightseeing, agricultural application, law enforcement, fire fighting, emergency medical services, transporting personnel and supplies to offshore oil rigs, traffic and news reporting, corporate and business transportation, personal transportation, and heavy lifting. Helicopters are also extensively used in military operations throughout the world. Transportation by helicopter can generally be classified into two general categories, namely, private operations and commercial operations. Private operations are of the same nature as general aviation and commercial operations are similar to scheduled air carrier activity.

Most of the helicopters used in private operations have a capacity of 1 to 5 persons and have maximum gross weights between 3000 and 6000 lb. Helicopters in commercial operations are of greater capacity,

typically between 10 and 50 passengers, and have maximum gross weights between 10,000 and 50,000 lb. Primarily because of the difference in size, heliport facilities for private operations are normally much smaller than those for commercial operations.

Private operations include construction, forest and police patrol, crop dusting, advertising, emergency medical service and rescue, and transport to off-shore oil well locations. Commercial operations may be classified into two types. One is transportation in large metropolitan areas between several airports in the region and between airports and the business center in the region. The second is intercity transportation between cities not necessarily in the same metropolitan area.

Experience to date shows that the helicopter has achieved its greatest success in the first type of service, the operating service area being within a radius of about 50 mi. The continued development of the second type of service will depend on the ability of helicopters to compete favorably with fixed-wing aircraft with respect to speed and economics over stage lengths up to about 300 mi.

Despite the growth of transport by helicopter, it only accounts for a small percentage of the total number of persons traveling by air. Helicopter operating costs have gradually been reduced but they are still considerably higher than those for fixed-wing aircraft.

Characteristics of Helicopters

A helicopter is a powered aircraft which gains its lift from the rotary motion of airfoil surfaces. The distinctive characteristic of a helicopter is its ability to hover through application of power to the rotating airfoils. The practical consequences of this characteristic are a much greater range of flight speeds and flight attitudes than is the case with conventional aircraft and the ability to land on and takeoff from comparatively small areas. When on the ground, helicopters have the ability of taxiing under their own power. Helicopters in private operation typically have cruise speeds between 90 and 130 kn, ranges between 300 and 400 nm, and passenger capacities between 2 and 10. Helicopters in the transport category typically have cruise speeds from 100 to 150 kn, ranges between 300 and 700 nm, and passenger capacities between 10 and 50.

While helicopters can ascend vertically from the ground, prolonged vertical ascents severely restrict load-carrying capacities. The usual procedure is to employ vertical ascent only to initiate the takeoff. As with other aircraft, takeoffs are usually made into the wind. The initial vertical rise for takeoff is aided by a ground cushion built up by the pressure of the air directed against the ground by the revolving rotors. After a few feet of vertical ascent, horizontal acceleration is begun until climb-out speed is reached. Prior to reaching climb-out speed, the helicopter can be flown in a horizontal path or in a slightly ascending path. Climb-out and descent speeds vary from 30 to 60 kn. Just before touchdown, the helicopter hovers momentarily 5 to 10 ft above the landing pad.

From a safety standpoint, the operation of single-engine helicopters requires that emergency landing areas be available along the entire flight path. In case of engine failure, safe landings using autorotation can be made if space is available. Autorotation is the continuation of rotor rotation in flight after cessation of power. Sufficient height must be reached by the helicopter in order to utilize the principle of autorotation. Twin-engine helicopters, on the other hand, are designed to permit continuation of flight and even a moderate rate of climb in the event once the engine fails. For these types of helicopters, it is not necessary from a safety standpoint to have space available for emergency landing along the entire route. Virtually all small helicopters in private operation are single engine. Most of the helicopters, with the exception of a few models, are also single engine. Twin-engine helicopters are not, however, designed to hover with only one engine in operation. On takeoff, therefore, in the event an engine fails before the helicopter has reached a one-engine-out flight speed, a landing must be made. To take care of this eventuality, sufficient space must be provided ahead of the landing area for an emergency landing. The single-engine helicopter also needs this area, but in addition requires space for emergency landings all along the flight path. If helicopters are designed to hover with one engine out, space ahead of the landing area is not required.

Because it is not economically practical for a helicopter to ascend and descend vertically, unobstructed approach-departure paths leading to the heliport are required. To protect the approach-departure path, obstructions are not permitted to extend above a prescribed inclined plane, an approach surface, beginning at the heliport and extending to specified distance from the heliport. The obstruction clearance requirements specified by the FAA are discussed later and ICAO has adopted similar recommendations.

Helicopters can be either single-rotor or tandem rotor and powered by one or two engines. The landing gear can consist of pontoons for landing on water, skids, or wheels equipped with rubber tires. When wheels are used, the landing gear normally consists of two main wheels and a single nose or tail wheel, or four wheels. Figure 15-1 shows several small utility helicopters at the Port Authority of New York and New Jersey heliport on the East River in New York City. The principal dimensions of representative helicopters used for private and commercial operations are shown in Fig. 15-2 and tabulated in Table 15-1.

Factors Related to Heliport Site Selection

The selection of a heliport site in an urban area requires the consideration of many factors, the most important of which are:

1. The best locations to serve potential traffic
2. The provision of minimum obstructions in the approach and departure areas



FIGURE 15-1 Port Authority of New York and New Jersey heliport (*Bell Helicopter Textron, Inc.*).

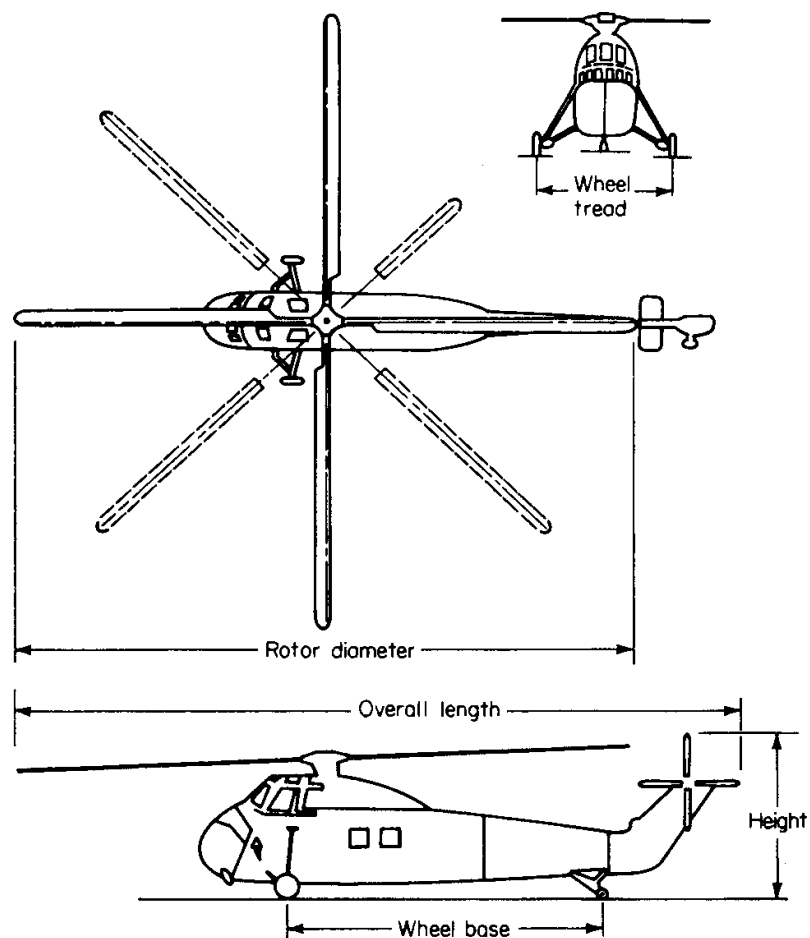


FIGURE 15-2 Dimensional definitions for helicopters.

Aircraft	Rotor Diameter, ft	Overall Length, ft	Height, ft	Wheelbase, ft	Wheel Tread, ft	Gross Weight, lb	Maximum Passengers
Aerosp 330J	49.5	59.8	16.9	13.2	7.9	16,315	19
Aerosp 332L	51.2	61.4	16.2	17.2	9.8	18,410	24
Bell-212	48.0	47.3	13.0	7.6	8.3	11,200	14
Bell-214ST	52.0	62.2	13.2	8.1	8.3	17,500	18
B-Vertol 107II	50.0	83.3	16.9	24.9	12.9	20,000	25
B-Vertol 234	60.0	99.0	18.7	25.8	10.5	48,500	44
B-Vertol 360	83.7	49.7	19.4	32.7	11.4	36,160	30
Sikorsky S-61N	62.0	73.0	18.9	23.5	14.0	20,500	28
Sikorsky S-64	72.0	88.5	25.4	24.4	19.8	42,000	45
Sikorsky S-76B	44.0	52.5	14.5	16.4	8.0	11,400	12
Westland 30300	42.5	52.1	16.3	17.8	9.3	16,000	19

Sources: International Civil Aviation Organization and Federal Aviation Administration.

TABLE 15-1 Dimensions of Typical Commercial Helicopters

3. The provision of minimum disturbance from noise and desirable location with respect to adjacent land use
4. The provision of adequate access to surface transportation and parking
5. The cost to acquire and develop
6. The provision of two approach paths separated by at least 90° and oriented with respect to prevailing winds
7. The avoidance of traffic conflicts between helicopters and other air traffic
8. The consideration of turbulence and visibility restrictions presented by nearby buildings
9. The provision of emergency landing areas along the entire route for single-engine helicopters

Final selection of a heliport site will usually require a compromise among these various factors. The most severe problems can be expected in large, highly developed metropolitan areas. In large urban areas heliports should be planned on a regional basis. The first step is to prepare an estimate of the demand for helicopter services and the origins and destinations of this demand. The second step is to select a heliport site or sites which can reasonably satisfy the demand and yet meet the requirements cited above.

The principal market for commercial helicopter transportation has been in large urban areas between one or more airports and the central business district. Therefore, it is essential that the downtown heliport be centrally located near the hotel area and the business district. Likewise adequate provision for helicopters should be made at airports. In extremely large urban complexes, there may be outlying smaller centers, and secondary heliports are needed so that the benefits of air transportation can extend to these centers. A heliport must have good access to streets, highways, and public transit facilities so that passengers using buses, personal vehicles, or mass transit can easily reach the facility.

Noise

The noise caused by helicopter operations within or adjacent to built-up urban areas is and will continue to be an extremely important factor in planning for helicopter transport, as it has been with fixed-wing aircraft. Manufacturers are aware of this problem and continue to study ways in which noise can be minimized.

A heliport should be located so that the noise generated by helicopters will not cause excessive disturbance to surrounding developments. The noise factor is most critical underneath the flight path on takeoff and landing. The amount of sound that can be tolerated by the average person is dependent upon a number of factors, including the

overall noise level, its frequency, and its duration, the type of development (residential, industrial, etc.) surrounding the source of the noise, and the ambient sound level in the area. A greater amount of noise can be tolerated in industrial areas than in residential areas. Docks and other waterfront sites offer some of the best possibilities for heliport location in large, congested urban centers. Approach and noise problems can usually be overcome by making the use of water areas for heliport location. The downtown heliport in New York City is an example of such a facility.

Noise generated by small two- and three-seat helicopters can be tolerated in business and industrial areas, but the noise generated by large multiengine helicopters powered by turbine engines can exceed tolerable levels even in business and industrial areas. It is well to check with the manufacturers concerning the latest information on the levels of noise generated by the several transport type helicopters.

To minimize the noise, it is desirable to orient the landing pad so that landings and takeoffs are made over areas where noise would be least objectionable. Considerably more latitude can be exercised in this respect for helicopters than with fixed-wing aircraft.

Protection of Approach and Departure Paths

Zoning is necessary both to control the location of heliport sites for maximum benefit to the community and to provide safety in helicopter operations by protection of the surrounding airspace. The dimensions of the approach-departure paths for various types of heliport operations are discussed below.

Turbulence and Visibility

Another factor which must be considered in the selection of a site for a heliport is the effect of turbulence over roof surfaces and down-drafts near buildings. This factor is of particular importance for rooftop heliports. If there is doubt in the planner's mind, the site should be flight-checked with a helicopter.

Poor visibility can be an important factor to consider for sites on tall buildings, that is, those of 100 ft or more in height. The cloud deck seldom reaches the ground, but at higher levels the heliport might find itself enveloped in fog when the ground is clear.

Physical Characteristics of a Heliport

A heliport is defined as a facility which is intended to be used for the landing and takeoff of helicopters, and may include space for helicopter parking, buildings, servicing facilities, and vehicular parking. The *final approach and takeoff (FATO) area* is a defined area over which the final phase of the approach maneuver to a hover or

a landing is completed and from which the takeoff maneuver is commenced. The *touchdown and liftoff (TLOF) area* is a hard surfaced load bearing area typically located within the final approach and takeoff area on which a helicopter may touch down or lift off. Functionally, the terminal area requirements for the parking, servicing, and fueling of helicopters and the processing of passengers and ground vehicles are no different from the requirements for fixed-wing aircraft.

Heliports are usually classified according to use as follows:

Military heliport: Facilities operated by one of the branches of the armed services. The design criteria are specified by the branch of the service and usually prohibit nonmilitary uses.

Federal heliport: Facilities operated by a nonmilitary agency or department of the federal government. They are used to carry out the functions appropriate to the agency.

Private-use heliport: Facilities which are restricted in use by the owner. These may be publicly owned but their use is restricted, as in police or fire department use.

Public-use heliport: Facilities which are open to the general public and do not require the prior permission of the owner to land. The extent of the facilities available may limit operations to helicopters of specified sizes or weights.

Commercial service heliport: Public use and public owned facilities which are designed for the use of helicopters in commercial passenger or cargo service which enplane 2500 passengers annually and receive scheduled passenger service with helicopters.

Personal-use heliport: Facilities which are used exclusively by the owner.

The principal components of a heliport are the final approach and touchdown area, the touchdown and liftoff area, and, for large heliports, taxiways, helicopter parking areas, and the terminal building area. The relationships between these components are shown in Fig. 15-3.

Final Approach and Takeoff Area

The final approach and takeoff area (FATO) is a surface from which the helicopter can land or take off. The FAA allows the FATO to be any shape as long as it is enclosed by a square of the dimensions indicated in Table 15-2. ICAO specifies that the FATO is a circle. Its size depends primarily on the overall length of the largest helicopter to be accommodated by the heliport. Because of the dust that can be created by the rotor of a helicopter, it is necessary to prepare the surface of the landing and takeoff area so that it will be free of dust

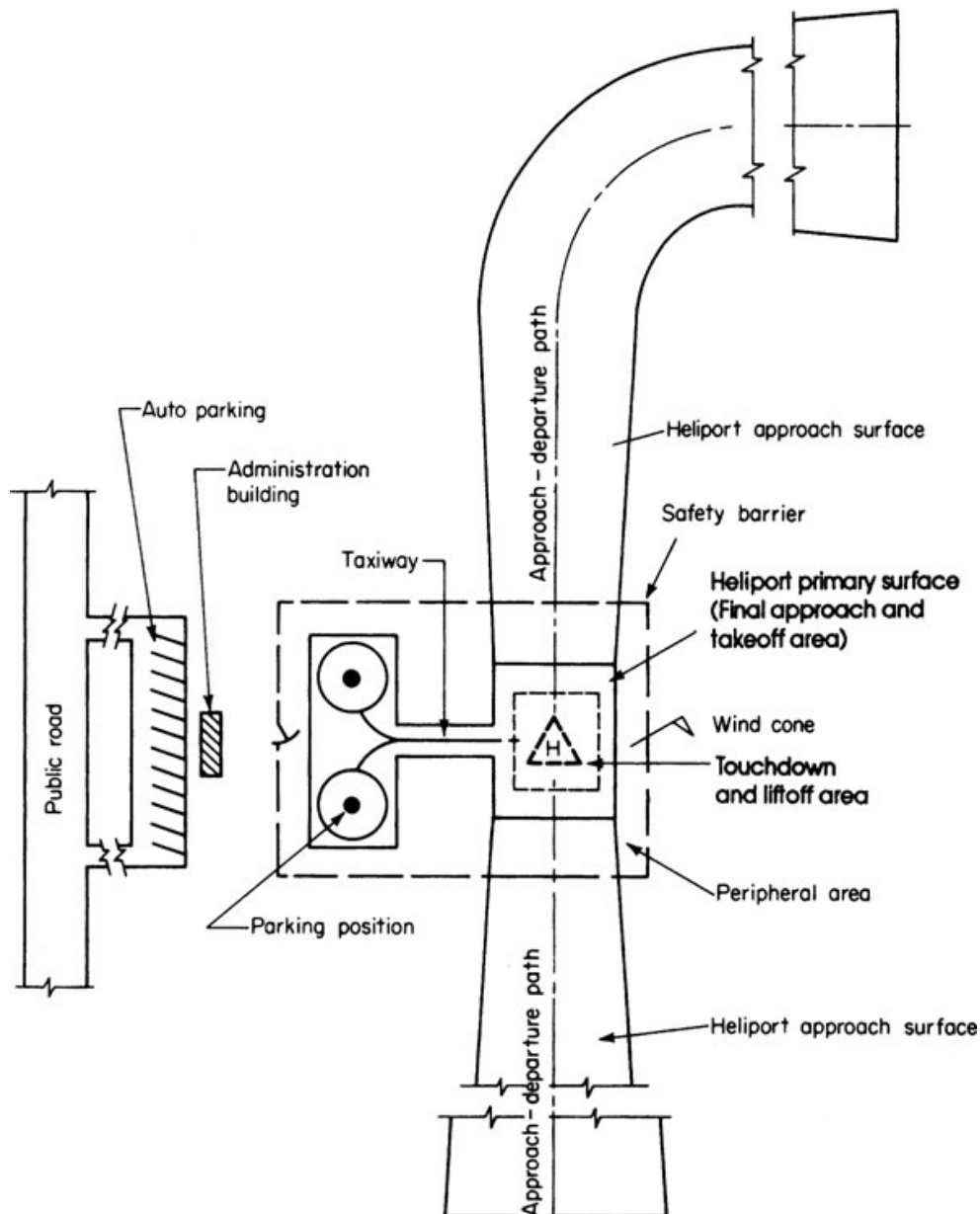


FIGURE 15-3 Typical heliport layout (*Federal Aviation Administration*).

(e.g., turf or pavement). It is recommended that paving or stabilizing the soil of the takeoff and landing area to improve the load carrying ability of the surface, minimize the erosive effects of rotor downwash, and to facilitate surface runoff due to rain or snow.

The recommended dimensions of the final approach and takeoff area are given in Table 15-2. For precision instrument operations the final approach and takeoff area is 300 ft wide by 1225 ft long and incorporates a *final approach reference area* (FARA) which is an obstruction-free area 150 ft by 150 ft located at the far end of the final approach and takeoff area.

	Private	FAA Public Utility	Commercial Transport	ICAO	
				Land	Water
Final approach and takeoff area					
Length	$1.5L^b$	$1.5L^b$	200 ft ^b	$1.5D^h$	$1.5D + 10\%^i$
Width	$1.5L^b$	$1.5L^b$	$2R^d$	$1.5D^h$	$1.5D + 10\%^i$
Clearance	$1/3R^c$	$1/3R^c$	30 ft	$0.25D^j$	$0.25D^j$
Touchdown and liftoff area					
Length and width	$1.5U1.0L1.0L$		^e	$1.5U$	
Parking area					
Clearance ^f	$1/3R^g$	30 ft	30 ft	^l	
Minimum width		$1.5U1.5U$		^l	
One way taxiway route width					
Hover operations		$R + 60 \text{ ft}$ $R + 60$ ft	$2R$		
Ground operations		$R + 40 \text{ ft}$ $R + 40$ ft	7.5–20 m	^j	
Parallel taxiway route width					
Hover operations		$R + 90 \text{ ft}$ $R + 90$ ft			
Ground operations		$R + 70 \text{ ft}$ $R + 70$ ft			
Taxiway pavement width		$2T2T$			
Air transit route				$7R^k$	

^a*L* is overall length of design helicopter; *R* is rotor diameter of design helicopter; *U* is maximum of undercarriage length or width of design helicopter; *D* is the overall length or width of the design helicopter, whichever is greater; *T* is wheel tread of design helicopter.

^bMay need to be adjusted for elevation; see AC 150/5390-2B.

^cMinimum of 10 ft for private; minimum of 20 ft for public.

^d100 ft for public owned.

^ePosition on major axis of FATO with its center at least 50 ft from end or edge of FATO.

^fCannot lie under approach or climb path.

^gMinimum of 20 ft.

^hThe overall length or width of the design helicopter, whichever is greater; for class 2 or class 3 helicopters on water heliports this is $2D$; the FATO described is circular with this diameter.

ⁱFor VFR; for IFR the clearance should be at least 45 m on each side of the centerline and 60 m beyond the ends.

^jDepending upon the main gear span; separation between parallel taxiways should be 60 m on the side and 90 m on the ends of the final approach and takeoff area.

^kFor daytime operations; $10D$ for nighttime operations

^lThe same as for an aircraft parking area; see Chap. 6.

Sources: Federal Aviation Administration [10] and International Civil Aviation Organization [3].

TABLE 15-2 Geometric Design Standards for Heliports^a

Touchdown and Liftoff Area

Within the final approach and takeoff area, an area is designated for the normal everyday landing of helicopters. The TLOF area is usually defined by a solid border painted on the pavement surface. The recommended dimensions of the touchdown and liftoff area are given in Table 15-2.

Peripheral Area

A peripheral or clearance area surrounding the final approach and takeoff area is recommended as an obstruction free safety zone. The area should be kept free of objects hazardous to the operation of helicopters. The clearance from the edges of the final approach and takeoff area required for this area is also given in Table 15-2.

Effect of Wind

Although helicopters can maneuver in much higher crosswinds than fixed-wing aircraft, the takeoff and landing area should preferably be oriented as nearly as possible to permit operation into the wind. At the present time it appears that the crosswind characteristics of the helicopter will be such that for a majority of cases a rectangular takeoff and landing area need be oriented in one direction only.

Terminal Area

At heliports where the volume of traffic is relatively small, the loading and unloading of passengers can be accomplished within the final approach and takeoff area. As traffic increases, it becomes necessary to provide additional space for the parking of helicopters and passenger processing. This is usually accomplished on a helipad which is an area adjacent to the terminal building for processing passengers. This area provides for one or more parking spaces for helicopters and is similar in nature and function to the gate or ramp area provided on the apron adjacent to airport terminal facilities. Clearances between adjacent helicopter parking positions are provided so that a separation is provided between the rotor planes of helicopters as shown in Table 15-2.

The helipad is connected to the final approach and takeoff area by taxiways and taxilanes. Helicopters may traverse taxiways in a hover or ground mode on single or parallel taxiway routes. The width of these taxiway routes and the taxilanes adjacent to the helipad are given in Table 15-2. The taxilane widths are the same as the taxiway routes in a ground mode of operation.

Approach and Takeoff Climb Path

There is a requirement that heliports have at least one approach and takeoff climb path which is free of obstructions that should be established on the basis of the direction of prevailing winds and the access route that has the fewest obstacles in the flight path. As conditions permit, additional approach and takeoff climb paths should be established to facilitate operations at times when winds are from other directions. At private use heliports, it is recommended that these

paths flare out in the horizontal plane from the final approach and takeoff area at the rate of 1:20 and slope upward at the rate of 8:1. These paths terminate when the design helicopter attains a safe en route altitude. These paths may curve to avoid objects or noise sensitive areas when necessary. The FAA is in the process of developing design criteria for curved visual approaches and recommends that the FAR Part 77 specifications be applied until these criteria are developed. For a commercial service airport it is recommended that at least one instrument approach and takeoff climb path be established. For public use heliports the dimensions of the final approach and climb paths correspond to those specified under FAR Part 77 as discussed below.

Obstruction Clearance Requirements

Imaginary obstruction clearance surfaces are established for each class of heliport in FAR Part 77 [10]. For heliports, the principal surfaces are the approach and departure surfaces, the transitional surfaces, and the heliport protection zone. The *heliport protection zone* is the area on the ground below the approach surface from the edge of the final approach and takeoff area to the point where the approach surface is 35 ft above the elevation of the final approach and takeoff area. The horizontal surface required for airports is not necessary for heliports. The approach surface requirements for visual, nonprecision instrument and precision instrument operations specified by the FAA are given in Table 15-3. Similar requirements are specified

	Types of Approach		
	Visual	Nonprecision	Precision
Length	4,000	10,000	25,000*
Inner width	†	500	1,000
Outer width	500	5,000	6,000
Slope	8:1	20:1	34:1‡
Transitional			
Inner width	†	†	600
Outer width	250	600	1,500
Slope	2:1	4:1	7:1

*Begins 1225 ft from the far end of the final approach and takeoff area.

†Width of final approach and takeoff area.

‡For a 3° glide slope; 22.7:1 for a 4.5° glide slope; 17:1 for a 6° glide slope; glide slope can be increased in 0.1° increments with corresponding corrections to approach slope.

Source: Federal Aviation Administration [10].

TABLE 15-3 FAR Part 77 Approach Surface Dimensions for Heliports, ft

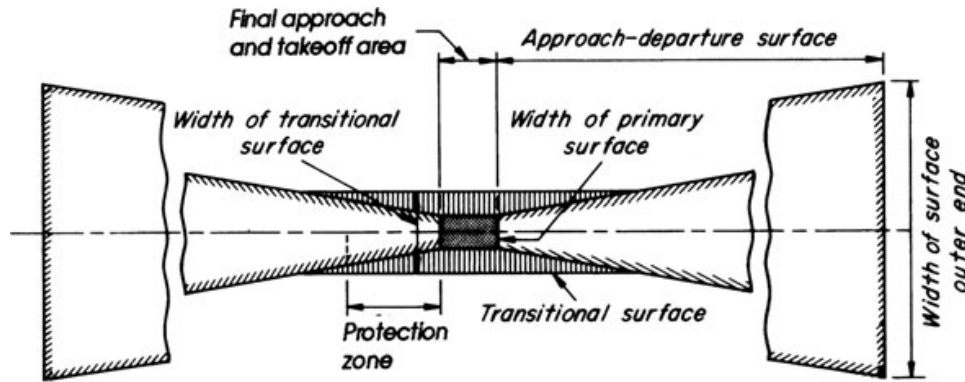


FIGURE 15-4 Obstruction clearance requirements for heliports (*Federal Aviation Administration* [10]).

by ICAO [3]. Precision instrument operations by commercial helicopters are very limited and thus visual or nonprecision instrument operation criteria will suffice for most heliports. The various FAR Part 77 surfaces are shown in Fig. 15-4. It should be noted that curved-path approaches and departures are currently not permitted under IFR conditions but with the implementation of microwave landing systems (MLS) this is subject to change. The FAA has developed specifications for the MLS critical areas and siting requirements which should be consulted if the installation of an MLS is contemplated. The specifications for IFR are a function of the nature of the navigational aids and references [10] should be consulted prior to establishing landing and takeoff paths.

The various dimensions specified by the FAA for a commercial service heliport are illustrated in Example Problem 15-1.

Example Problem 15-1 Let us design the layout of a commercial service heliport for operations with a Boeing-Vertol 234 design helicopter. Let us assume that the helipad or parking apron adjacent to the passenger terminal building will require space for four helicopters. Let us assume that the heliport elevation is 800 ft above mean sea level. A ground taxiway route is to be provided from the touchdown and liftoff area to the helipad.

From Table 15-1 the design helicopter has a rotor diameter D of 60 ft, an overall length L of 99 ft, a height of 18.7 ft, a wheelbase of 25.8 ft, a wheel tread of 10.5 ft, a maximum gross weight of 48,500 lb, and a maximum capacity 44 passengers.

For a commercial service heliport, from Table 15-2 the final approach and takeoff area is required to have a minimum length of 200 ft and a minimum width of 2 times the rotor diameter or $2(60)$ or 120 ft. An object-free clearance width of at least 30 ft from the edges of the final approach and takeoff area is also required.

The length and width of the liftoff and touchdown area is equal to the overall length of the design helicopter or 99 ft. Let us use 100 ft for these dimensions. To provide for the length of the helicopter and the minimum object-free area distance from the final approach and takeoff area, the minimum length of the taxiway leading to and from the touchdown and liftoff area is $100 + 30$ or 130 ft.

Parking positions must have a minimum width of 1.5 times the undercarriage length or undercarriage width whichever is greater. The greater of these two dimensions is the wheelbase and therefore, the required minimum width of a parking position is 1.5 (25.8) or 39 ft. Let us provide 40 ft. However, since the rotor diameter of the design helicopter is 60 ft, the minimum width of the parking position must be 60 ft of which 40 ft will be paved. There must also be a clearance between the edges of adjacent parking positions 30 ft. This results in the minimum distance between the centerlines of adjacent parking positions being 90 ft.

A paved ground taxiway will be provided and this must have a minimum width of 2 times the wheel tread or $2(10.5)$ or 21 ft. The ground taxiway route is also required to have a safety area width of the rotor diameter plus 40 ft or $60 + 40$ or 100 ft. Taxiways or taxilanes in the vicinity of the terminal building must also have this safety area to provide clearances between the building and parked helicopters. Therefore, the minimum distance from the centerline of the taxiway or taxilanes is 50 ft.

The layout of the heliport with the corresponding dimensions is shown in Fig. 15-5.

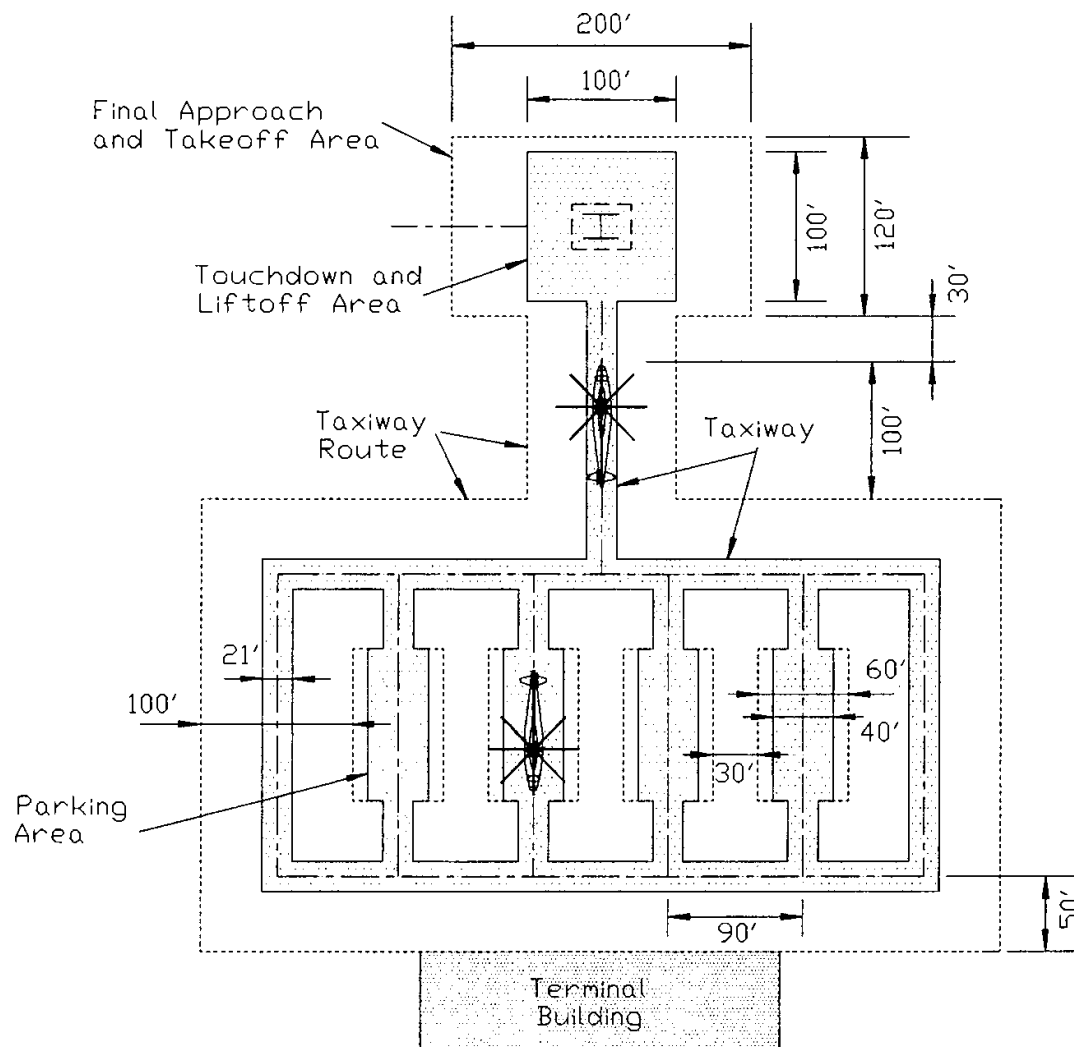


FIGURE 15-5 Commercial service heliport layout for Example Problem 15-1.

Marking of Heliports

The primary purpose for marking heliports is to identify the area clearly as a facility for the use of helicopters. The requirements for marking heliports are specified by the FAA and the ICAO. Essentially these requirements consists of painting an equilateral square with an "H" in the center of the touchdown and liftoff area. For hospital heliports a white cross is also inscribed within the square along with the letter "H" as shown in Fig. 15-6 for the heliport located at the Alexian Brothers Medical Center in Dade County, Florida. Marking delineating the edges of the final approach and takeoff area should be broken white lines whereas on the touchdown and liftoff area the edges should be delineated by continuous white lines. Taxiway centerlines are delineated by solid yellow lines and taxi route centerline and apron edge markings should be solid yellow lines. Taxiway edges should be marked by a double solid yellow line. A painted yellow line is also recommended to define the centerline of parking positions and when these positions vary in the amount of clearance provided a number enclosed by a circle should be painted on the entrance to the parking position to indicate the largest helicopter that can be accommodated.

Lighting of Heliports

For operation during hours of darkness various types of lights are suggested [10]. The amount of lighting depends on the character and

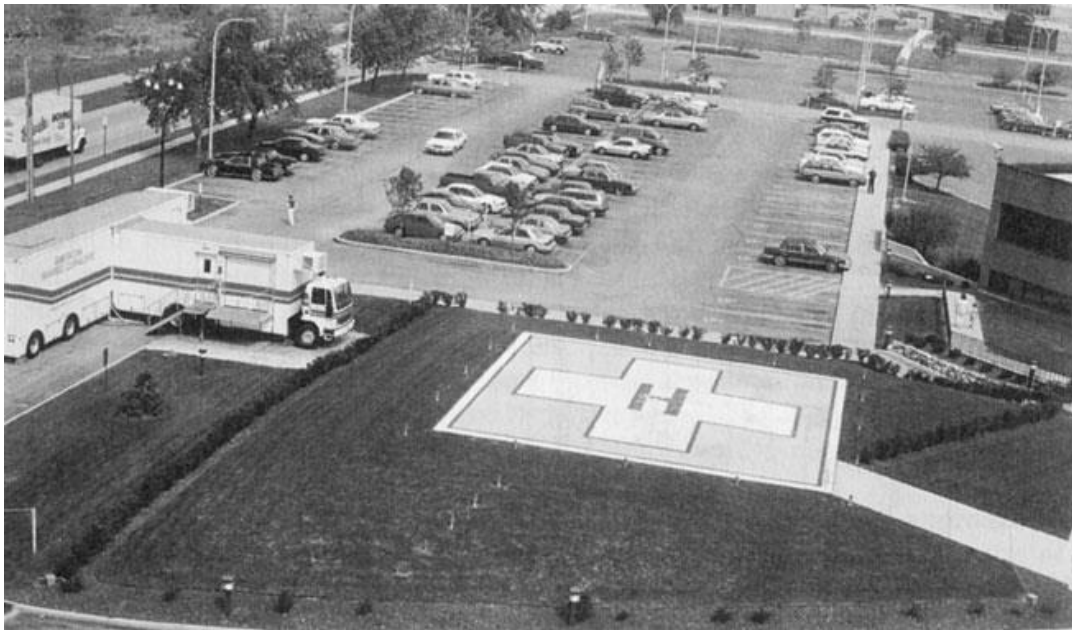


FIGURE 15-6 Heliport located at Alexian Brothers Medical Center in Dade County, Florida (*Howard Needles Tammen & Bergendoff and Alexian Brothers Medical Center*).

volume of operations. More lighting is required for scheduled air carrier operations than for private heliports with occasional use.

The minimum recommendations for private use and public use heliports consist of lighting of the perimeter of either the final approach and takeoff area or the touchdown and liftoff area with lights with yellow lenses uniformly spaced at 25-ft intervals. At public use heliports green lights are used to define the taxiway and taxilane centerlines. For commercial service heliports the touchdown and liftoff area is delineated by yellow lights located 10 ft from the outside edge. At such heliports in-pavement green lights are recommended for taxiway and taxilane centerlines. Blue retroreflective markers are also used at these airports to identify taxiway entrance and exit points and to define taxiway edges. Perimeter lighting defining the touchdown and liftoff area of a commercial service heliport is shown in Fig. 15-7. All objects that penetrate the obstruction clearance surfaces should be lighted with red colored lights.

Other useful lighting aids are the landing direction lights, visual glide path indicators, and heliport identification beacons. The landing direction lights are miniature approach lights since they extend only 75 ft. The color is yellow. Visual glide path indicators are also recommended for visual operations at commercial service heliports. The lowest on-course signal should provide a 1° clearance over any object in the approach path within 10° horizontally on either side of the approach path centerline. The optimal location is on the extended runway centerline of the approach path such that it will bring the

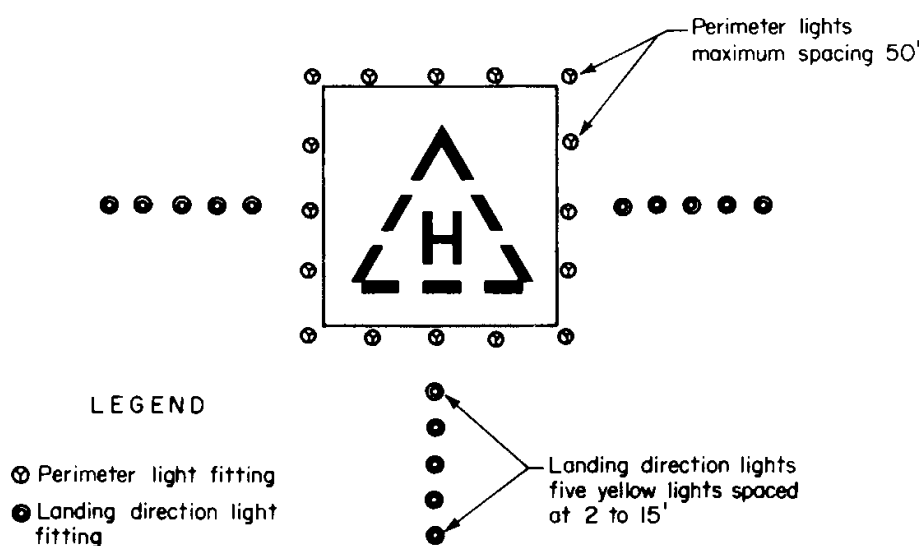


FIGURE 15-7 Heliport lighting configuration (*Federal Aviation Administration [10]*).

helicopter to between a 3- and 8-ft hover distance over the touchdown and liftoff area. Heliport identification beacons, alternate flashing white-green-yellow lights should be located with one quarter of a mile from a commercial service heliport.

Elevated Heliports

When ground level sites are not available or are unsuitable, an elevated site may be practical. Elevated heliports may be located on piers or other structures over water, as well as on buildings. The dimensions of the touchdown and liftoff area are the same as for heliports on the ground, but the final approach and takeoff area can be smaller and there is no need for peripheral areas. When planning a rooftop heliport, a thorough study should be made of the air currents caused by the presence of adjacent buildings. Roof areas make it possible to locate the heliport closer to the center of business activities in a city, provided the facility is environmentally acceptable. Another advantage is that the land cost is partially absorbed by the tenancy of the lower floors of the building. However, it should be realized that of the operations of any size, space on the floors below the takeoff and landing area may have to be devoted to uses such as lobby, freight, and baggage handling. The possible disadvantage of height with respect to visibility was mentioned earlier. A heliport 100 ft or more above the ground would require a higher cloud base than a ground heliport to provide the same operating safety. A downtown commercial heliport would require, in addition to lobby space, car parking facilities relatively close by.

Where heliports are built on elevated structures, the strength of the floor should be greater than the strength of the landing gear of the helicopter. The loads imposed by helicopters and recommendations concerning the structural design of elevated structures are discussed in the next section.

Structural Design of Heliports

Helicopters using facilities on land are usually supported on tubular skids or wheels equipped with rubber tires. Helicopters equipped with conventional landing gear wheels are normally supported by two main wheels and one tail or nose wheel. For larger helicopters each main landing gear consists of two wheels. Each main gear typically supports 40 to 45 percent of the weight of the helicopter and the tail or nose wheel supports the remainder of the weight, approximately 10 to 20 percent. If the helicopter is supported by tubular skids, 50 percent of the weight is supported by each skid.

The strength requirements for the touchdown and liftoff area are determined by considering the static load, dynamic load, and down-wash load of the helicopter. Both the static load and the dynamic load are applied through the landing gear contact area whereas the down-wash load is applied over a contact area defined by the diameter of the rotors. The FAA recommends that for design purposes the touchdown and liftoff area should be capable of supporting 150 percent of the maximum takeoff weight of the design helicopter.

Heliports at Airports

A large number of helicopters will operate into airports to serve traffic from the downtown area and surrounding communities. Accordingly, provisions should be made at an airport for the landing and takeoff of helicopters. The takeoff and landing area should be located to

1. Provide maximum separation from fixed-wing aircraft traffic patterns so as to avoid creating a conflict in takeoff and landing operations.
2. Be as close as possible to passenger check-in areas for fixed-wing aircraft to avoid long walking distances for passengers.
3. Avoid as much as possible the mixing of taxiing fixed-wing aircraft and helicopters, since helicopters taxi at relatively low speeds.

It is recommended that if simultaneous same direction diverging helicopter and fixed-wing aircraft operations are to be conducted under visual flight rule conditions, a runway centerline to a final approach and takeoff area and a touchdown and liftoff area centerline separation of 700 ft be provided. A 2500-ft minimum separation is required for radar departures under instrument flight rule conditions [10]. Helicopter parking apron areas should meet the same runway clearance standards as those required for fixed-wing aircraft parking [5].

Alternative locations for heliports at an airport are the roof of the terminal building, the apron adjacent to the terminal building used by fixed-wing aircraft, and the area adjacent to the terminal building separate from the fixed-wing aircraft apron. There are advantages and disadvantages to all three locations. Normally, a ground-level site is preferred. The most convenient and least expensive method for accomplishing this is to reserve a part of the fixed-wing aircraft apron for the takeoff and landing of helicopters. If this is not convenient, a special pad for helicopter operations on the aircraft side of the terminal building should be provided.

Figure 15-8 shows the heliport at Miami International Airport.

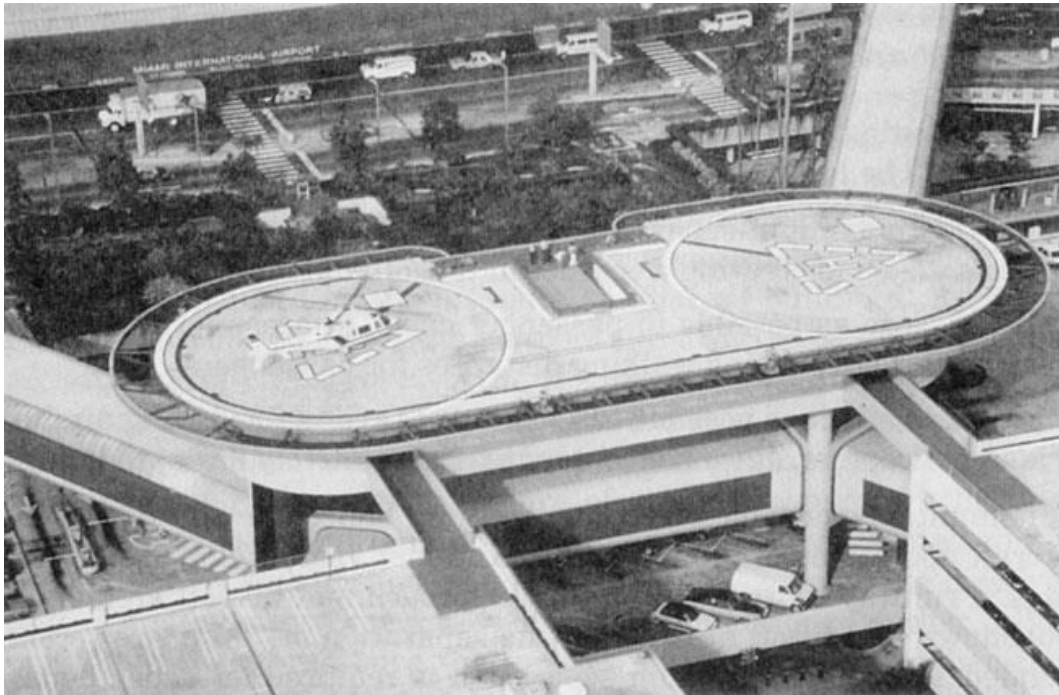


FIGURE 15-8 Heliport at Miami International Airport (*Howard Needles Tammen and Bergendoff & Miami International Airport*).

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